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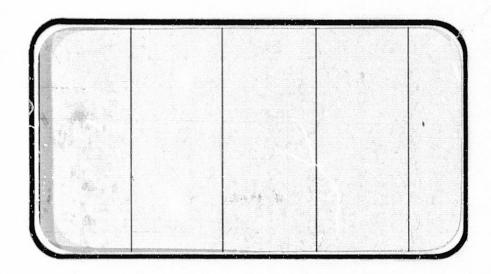
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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

NASA CR-

141823



(NASA-CR-141823) RESULTS OF AN INVESTIGATION OF THE 0.003-SCALE SPACE SHUTTLE EXTERNAL TANK MSFC MODEL 460 IN THE NASA/MSFC 14 X 14-INCH TRISONIC WIND TUNNEL TO DETERMINE STATIC PRESSURE DISTRIBUTIONS

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER HOUSTON, TEXAS

DATA MANagement services



DMS-DR-2165 NASA CR-141,823

VOLUME 1 OF 5

RESULTS OF AN INVESTIGATION OF THE 0.003-SCALE

SPACE SHUTTLE EXTERNAL TANK MSFC MODEL 460

IN THE NASA/MSFC 14 X 14-INCH TRISONIC WIND TUNNEL

TO DETERMINE STATIC PRESSURE DISTRIBUTIONS DURING

REENTRY (TA2F)

by

P. E. Ramsey, MSFC G. W. Winkler, NSI

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services
Chry ler Corporation Space Division
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center National Aeronautics and Space Administration Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number:

MSFC TWT 596

NASA Series Number:

TA2F

Model Number:

460

Test Dates:

July 20-23, 1974

Occupancy Hours:

104

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

RESULTS OF AN INVESTIGATION OF AN 0.003-SCALE

SPACE SHUTTLE EXTERNAL TANK MSFC MODEL 460 IN THE

NASA/MSFC 14 x 14-INCH TRISONIC WIND TUNNEL TO

DETERMINE STATIC PRESSURE DISTRIBUTIONS DURING REENTRY

(TA2F)

by

P. E. Ramsey, MSFC, and G. W. Winkler, NSI

ABSTRACT

Objective of the test was to obtain static pressure distributions for the ET at reentry conditions. Basic configuration of the model was the MCR 0200 ET modified to include a rectangular crossbar at the aft ET/orbiter attach point. Mach numbers were 1.96, 3.48, and 4.96. Reynolds number per foot at these Mach numbers were 6.95 million, 6.42 million, and 4.95 million, respectively. Angle of attack range was -8 to 100 degrees and roll angle was 0 to 315 degrees. Occupancy hours were 104.

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- A) CP vs. X/LB
- B) CP vs. THETA
- C) DCNM/D(X/LB) vs. X/LB
- D) DCYM/D(X/LB) vs. X/LB
- E) CNM vs. ALPHA CLMM CYM CYNM

NOMENCLATURE

Symbol	Plot <u>Symbol</u>	Definition	<u>U</u> nits
à		speed of sound	m/sec, ft/sec
Ab		base area; cross-sectional area of the cylindrical ET	in.2
^b ref	BREF	reference span; diameter of the cylindrical section of the model	in.
ET		external tank	
FA		axial force (AF), positive in the negative direction of x_{m}	7Ь
FN		normal force (NF), positive in the negative direction of $z_{\tilde{m}}$	1b
Fγ		side force (SF), positive in the positive direction of y_{m}	1b
ъВ	LBODY	length of the ET	in.
^l ref	LREF	reference length; diameter of the cylindrical section of the model	in.
М	MACH	Mach number; V/a	
MRP	MRP	moment reference point located in the x_m , y_m , z_m axes by XMRP, YMRP, and ZMRP (See Data Reduction section)	
M _X		rolling moment (RM); a moment about the x_m axis (a positive rolling moment tends to rotate the positive y_m axis toward the positive z_m axis)	in16
My		pitching moment (PM); a moment about the y_m axis (a positive pitching moment tends to rotate the positive z_m axis toward the positive x_m axis)	in1b
ċ g		center of gravity	

NOMENCLATURE (Continued)

Symbol	Plot Symbol	<u>Definition</u>	<u>Units</u>
M _Z		yawing moment (YM); a moment about the z_m axis (a positive yawing moment tends to rotate the positive x_m axis toward the positive y_m axis)	inlb
p_{∞}	þ	pressure, freestream	psi
p _o q∞	PO Q(PSI)	stagnation pressure free stream dynamic pressure	psi psi
S _{ref}	SREF	reference area; cross-sectional area of the cylindrical section of the model	in. ²
RN/L	RN/L	unit Reynolds number	per m, per ft
SRB		solid rocket booster	
V		velocity	m/sec, ft/sec
x _m ,y _m ,z _m		missile axis system (see Data Reduction section)	
Х		distance from nose of ET model in the negative $\boldsymbol{x}_{\boldsymbol{m}}$ direction	in.
x_T, y_T, z_T		model stations; (see figure 2a)	in.
XCb\rB	XCP/L	longitudinal position of the center of pressure, expressed as a fraction of the ET length, measured from the ET nose	
		$\frac{x_{CP}}{x_B} = \frac{x_{MRP}}{x_B} - \frac{c_{m_m}}{c_{N_{fin}}} \frac{x_{ref}}{x_B}$	
č		aerodynamic chord	m,ft
COEFFICIE	NTS		
$c_{A_{m}}$	CA	axial force coefficient; FA/q Sref	

NOMENCLATURE (Continued)

Symbol	Plot Symbol	<u>Definition</u>	Units
$c_{A_{b_{m}}}$	CAB	base axial force coefficient; $(p_{\infty} - p_b) A_B/q S_{ref}$	
c _{Af}	CAF	forebody axial force coefficient; c_{A_m} - c_{A_b}	
•			
c _{lm}	CBL	rolling moment coefficient; M _X /q S _{ref} b _{ref}	
C _{mm}	CLMM	pitching moment coefficient; My/q Sref ^l ref	
c_{N_m}	CNM	normal force coefficient; F _N /q S _{ref}	
c _{nm}	CYNM	yawing moment coefficient; M _Z /q S _{ref} b _{ref}	
$c_{\mathbf{p}}$	СР	pressure coefficient; $(p-p_{\infty})/q$	
$c_{Y_{\overline{m}}}$	СҮМ	side force coefficient; Fy/q S _{ref}	
C _{Nm}	DCN/DX	local normal force coefficient; aC _N /a(X/D)	
Cy_m	DCY/DX	local side force coefficient; aCy/a(X/D)	
SYMBOLS			
α	ALPHA	angle of attack	deg.
β	BETA	angle of sideslip	deg.
ф	PHI	angle of roll	deg.
ψ	PSI	angle of yaw	deg.

NOMENCLATURE (Concluded)

C	Plot	D_022.2	
Symbol	Symbol	<u>Definition</u>	<u>Units</u>
θ	THETA	circumferential location	deg.
ρ		mass density	kg/m³, slugs/ft³
ref		reference conditions	
&		free stream conditions	
Ь		base	
c		cavity	
t		total conditions	
В		model body	
T		external tank	
m		missile axis system	
1		local	
S		static conditions	
	MOUNT	1.0 indicates tail mounted (T₁)2.0 indicates side mounted (T₂)	

INTRODUCTION

After the solid rocket boosters and the external tank separate from the orbiter, the ET will reenter the earth's atmosphere at high supersonic or even hypersonic Mach numbers. This test is the second of two tests conducted in the NASA-MSFC 14-inch Trisonic Wind Tunnel to obtain force and pressure data on the 324-inch diameter ET at typical reentry angles of attack.

Model (MSFC No. 460) configuration is a 0.003-scale representation of the ET with fuel lines and forward and aft SRB and orbiter attach hardware. Also included is the ET/orbiter rectangular crossbar attach structure.

Pressure taps (192 total) were used to obtain data for evaluating the load distribution on the ET. Further evaluation of the ET aerodynamic characteristics can be made by comparing data from this test with data from TWT 583 (reference 4).

Pressure data were taken at three Mach numbers: 1.96, 3.48, and 4.96. Angle of attack range was -8 to 100 degrees, which was obtained by using two ET model mountings. Range -8 to 30 degrees used a tail-mounted model (T_1) for each of eight roll positions, 0 to 315 degrees. This model had attach structure and protuberances. For the range of 51 to 100 degrees, a side-mounted model (T_2) at 0° roll position was used.

MODEL DESCRIPTION

The model is a 0.003-scale of the MCR 0200 space shutils FT configuration modified to include a crossbar at the aft orbiter/ET attach points. General arrangement of the model is shown in figure 2a. The model is designated MSFC #460, and it consists of two ET models (one tail-mounted and one side-mounted); protuberances simulating fuel lines, attachment hardware, etc.; and model adapters which allowed the tanks to be supported in the tunnel on RI stings #1 and #3. The models were built by NASA to conform to the configuration specified by Rockwell International drawing VL78-000041B (Reference Drawing 6) and Martin-Marietta memo SA-A-74-9 (Reference Report 2).

Both ET models were made of stainless steel and contained 192 pressure orifices each. From these orifices, stainless steel and annealed 0.032-inch OD tubing was routed out the base (or the side) of the model. Four feet of 0.050-inch OD tubing was brazed onto each of the 0.032-inch tubes as close to the exit cavity as possible.

When placed in the tunnel test section, the tubing bundle from the model was secured along the sting and routed down the sector through the tunnel floor. At this point, Tygon tubing was used to connect the steel tubing to quick disconnects, which were connected to the scanivalves. Installation photographs for the tail mounted (T_1) and side mounted (T_2) models are in figures 3a and 3b, respectively.

Model stations are sometimes used to describe locations of various components of the model. When used, these stations will be given in

MODEL DESCRIPTION (Concluded)

inches model scale and the zero reference points will be same as in Rock-well International drawing VL72-000088"D" (Reference Drawing 2). Zero reference points are shown in figure 2a.

CONFIGURATIONS INVESTIGATED

Two ET configurations investigated are defined as follows:

T₁--MCR 0200 tail-mounted, modified to include crossbar configuration with protuberances.

T2--MCR 0200 side-mounted, "clean" configuration (without protuberances).

Each of the configurations consists of the following model components:

T1--T12 AT5 AT6 AT7 AT8 AT9 PT1 PT2 PT3 FL1 FL2 FR6

T2--T12

Brief descriptions of each component are below. Refer to table III for dimensional data.

T₁₂ Baseline 324-inch diameter external oxygen-hydrogen tank

AT5 Forward orbiter/ET attach structure

AT₆ Left rear orbiter/ET attach structure

AT₇ Right rear orbiter/ET attach structure

AT8 Forward SRB/ET attach structure

ATg Aft SRB/ET attach structure

PT₁ LOX vent line fairing

PT₂ LOX feed line

PT₃ LH₂ feed line

FL₁ LOX feed line

FL₂ LH₂ feed line

FR6 Aft ET/orbiter crossbar

TEST FACILITY DESCRIPTION

The Marshall Space Flight Center 14" x 14" Trisonic Wind Tunnel is an intermittent blowdown tunnel which operates by high pressure air flowing from storage to either vacuum or atmospheric conditions. A Mach number range from .2 to 5.85 is covered by utilizing two interchangeable test sections. The transonic section permits testing at Mach 0.20 through 2.50, and the supersonic section permits testing at Mach 2.74 through 5.85. Mach numbers between .2 and .9 are obtained by using a controllable diffuser. The range from .95 to 1.3 is achieved through the use of plenum suction and perforated walls. Mach numbers of 1.44, 1.93 and 2.50 are produced by interchangeable sets of fixed contour nozzle blocks. Above Mach 2.50 a set of fixed contour nozzle blocks is tilted and translated automatically to produce any desired Mach number in .25 increments.

Air is supplied to a 6000 cubic foot storage tank at approximately -40°F dew point and 500 psi. The compressor is a three-stage reciprocating unit driven by a 1500 hp motor.

The tunnel flow is established and controlled with a servo actuated gate valve. The controlled air flows through the valve diffuser into the stilling chamber and heat exchanger where the air temperature can be controlled from ambient to approximately 180°F. The air then passes through the test section which contains the nozzle blocks and test region.

Downstream of the test section is a hydraulically controlled pitch sector that provides a total angle of attack range of 20° ($\pm 10^{\circ}$). Sting offsets are available for obtaining various maximum angles of attack up to 95°.

TEST FACILITY DESCRIPTION (Concluded)

The diffuser section has movable floor and ceiling panels which are the primary means of controlling the subsonic Mach numbers and permit more efficient running supersonically. The sector assembly and supersonic diffuser telescope into the subsonic diffuser to allow easy access to the model and test section.

Tunnel flow is exhausted through an acoustically damped tower to atmosphere or into the vacuum field of 42,000 cubic feet. The vacuum tanks are evacuated by vacuum pumps driven by a motor of 500 hp.

Data are recorded by a solid-state digital data acquisition system.

The digital data are transferred to punched cards during the run to be reduced later by a computer to proper coefficient form.

TEST PROCEDURE

First part of the test was conducted using a side-mounted, "clean" configuration (T₂ without protuberances). Since it was a "clean" configuration, the roll angle was considered to always be 0 degrees. Angle of attack range was from 51 to 100 degrees in increments of 3 degrees. Data were obtained at Mach numbers of 1.96, 3.48, and 4.96.

Second part of the test consisted of using a tail-mounted model with attach hardware, fuel lines, and electrical tunnel. Angle of attack range was from -8 to 30 degrees in increments of 4 degrees. Data were obtained at eight roll positions, 0 to 315 degrees in 45-degree increments. All orifices and tubing were checked for leakage at the beginning of the test and after each roll position change. A leak check after rolling the model insured that correct measurements were being received from the orifices. Response time for the scanivalve function was within the one-second intervals allowed each scanivalve.

List of average test conditions is in table I. Dataset run number collation summary is in table II.

DATA REDUCTION

BIGGE

Location of each pressure orifice and the numbering system are presented in tables IV and V. Also special identification of blocked or inoperative pressure orifices is made for both tail-mounted and side-mounted models in these tables. Locations of these orifices are shown in figure 2b.

Sting deflections were 'asured outside the tunnel by using check weights. Sting deflections were assured outside the tunnel by using check weights. Sting deflections were used curve for the pressure test (TWT 596) was found to be the same, within allowable accuracy, as that of the force test (TWT 583). The same ET configuration and only slightly different support hardware were used in both force and pressure tests. Increments of α due to sting bending in the force test were added to the nominal α 's for the pressure test. This gave reasonably accurate values of angle of attack, accuracy comparable to force test, when the pressure model was tested at the same Mach number and tunnel total pressure as the force model.

Pressure data were reduced to coefficient form and are tabulated along with wind tunnel parameters, configuration, and run number in the appendix. Plots are presented for both longitudinal and circumferential pressure distributions (Cp vs X/R_B and Cp vs θ). These plots are shown for each Mach number, angle of attack, and roll position at which tests were conducted. In addition, the pressure coefficients were integrated to obtain the following missile axis force and mument coefficients:

DATA REDUCTION (Concluded)

$c_{N_{m}}$	=	F _N /q S _{ref}	normal force coefficient
Cym	=	Fy/q S _{ref}	side force coefficient
c _{mm}	=	My/q Sref tref	pitching moment coefficient
$c_{n_{m}}$	=	M _Z /q S _{ref} b _{ref}	yawing moment coefficient
c _{Nm}	=	aCM/a(X/D)	local normal force coefficient
Cyń	=	acy/a(X/D)	local side force coefficient

Force and moment coefficients obtained from the integration of pressures are for comparison with the results from the force test.

Model reference dimensions used in the data reduction are presented in table VI. The axis system diagram is presented in figure 1. The missile axis system (x_m, y_m, z_m) is a non-rolling body axis system that is frequently used in wind tunnel tests and studies of missile flight dynamics. It is a system of axes that rotates with a missile or wind tunnel model through angles of sideslip and angles of attack but never through angles of roll; i.e., it never rotates about the missile or model longitudinal axis. The orientation of the missile axis coefficients is defined in figure 1. The missile axis system is identical with the body axis system at zero roll angle.

Moment reference point (MRP) for the 0.003-scale model is taken to be at the dry weight center of gravity of the ET. For the full-scale ET, the center of gravity is located at $X_T = 1395.4$ inches. Thus, the MRP for the 0.003-scale ET model is 3.259 inches from the model nose, on the centerline (figure 2a).

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- 2. SA-A-74-9, "Space Shuttle External Tank Entry Force and Moment Wind Tunnel Test Requirements," Michna, D. J., Michoud Operations, Martin Marietta Corporation, February 1974.
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- 2. VL72-000088 "D", 8-3-73; Shuttle Configuration Control, MCR 0200 Baseline Rev. III, Dated 7-2-73; Rockwell International.
- 3. VL78-000031 "A", 6-29-73; Thermal Protection-External Tank MCR 0200 Baseline Dated 4-11-73; Rockwell International.
- 4. VL77-000051 "A", 9-10-73; SRB Single PT.-Fwd Thrust Fitting (MCR 0190 Rev. 3 Baseline 8-13-73); Rockwell International.
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- 6. VL78-000041 "B", 5-30-73; External Tank Configuration Control MCR 0200 Revision I Dated 5-16-73; Rockwell International.

TEST:	MSFC TWT 596		1	DATE: Aug 1974
		TEST CON	DITIONS	
MACH NUMBER	REYNOLDS NUMBER (per foot)	DYNAMIC PRESSURE (pounds/sq.inch)	S TAGNATION TEMPERATURE (degrees Fahrenheit)	STAGNATION PRESSURE (pounds/sq.inch)
1.96	6.95 x 70 ⁶	10,2	104	28.0
3.48	6.42 x 10 ⁶	6.9	143	60.0
4.96	4.95 x 10 ⁶	3.1	143	90.0
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TABLE II.

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TABLE II. (Continued)

CONFIGURATION	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE) 2 154 153 2 154 153 1 15 1 152 1 175 176 1 176 1 176 1 176 1 176 1 176 1 176 1 176 1 176 1 176 1
CONFIGURATION	3.48 151 179 175 175 175 175 176 186 186
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PROTUBERANCES > -8 0 (35 -4 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	
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MSFC - Form 163-1 (Rev. May 1973)

TABLE II. (Continued)

DATA SET/RUN NUMBER COLLATIN		77.1.1
ASET CONFIGURATION OF SCHOL PARAMETERS/VALUES NO. MACH 1816 3 67 17 033	COLLATION SUMMARY	
FEER CONFIGURATION	MACH NUMBERS (OR ALTERNATE INDEPENDENT	VARIABLE)
037 T.CTAIL-MOUNTED 16 0 20 Tail 190 3 67 039 E.T. WITH 20 20 3 69 040 A A A 3 70 040 A A A A 3 70 042 A <t< th=""><th>9.1</th><th></th></t<>	9.1	
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040 28 V V 3 70 042 -8 0 225 2 2 042 -4 0 225 2 2 043 4 4 V 2 2 044 6 6 7 2 2 045 6 7 7 2 2 045 6 6 7 7 2 048 7 7 7 2 2 049 7 7 7 3 9 050 7 7 7 4 7 7 7 3 9 050 7 7 7 7 7 7 4 7 7 7 4 7 7 7 7 4 7 7 7 4 7 7 7 7 4 7 7 7 7 7 4 7 7 7 7 7 7 7 7 7 7	6	
042 -8 0 225 2 042 -4 -4 2 2 044 -4 -4 2 2 045 -4 -6 2 2 045 -6 -7 2 2 047 -6 -7 2 2 048 -7 -7 -7 -7 -7 050 -7 -7 -7 -4 0 -7 -7 054 -7 -4 0 -7	0	
042 4 2 0443 6 2 0444 6 7 0445 8 7 2 0446 12 2 2 0447 16 2 2 048 20 2 2 049 24 7 7 050 270 3 96 051 -4 0 270 3 96 053 4 4 4 4 4 3 99 7 13 13 13 37 4	255 126	
044 4 4 2 0446 8 V 2 045 8 V 2 046 12 20 2 047 16 2 2 048 20 2 2 048 2 2 2 050 28 V V 2 051 -8 0 270 3 96 052 4 V V 3 98 054 4 V V 3 99 13 19 25 31 37 4	254 127	TES
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050 051 052 053 054 V V V 3 99	257 124	
0.51 0.52 0.53 0.54 \ 4 \ 7 \ 7 \ 3 99 0.54 \ 31 37 4	256 /25	
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TABLE II. (Continued)

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	DATE: July 1974
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. (Continued)	DATA SET/RIN NIIMRED COI : ATION CHAMBON
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TABLE II. (Continued)

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TABLE II. (Concluded)

CONFIGURATION SCHOL PARAMETERS/VALLES NO. MACH HUMSERS LOR ALTERNATE INDEPENDENT VARIATION	EST: MSFC	TWT 596	\neg	0	ATA SI	ET/RU	NON N	BER (COLL	ATION	ATA SET/RUN NUMBER COLLATION SUMMARY	ARY	긔	DAIE: DAIX	_	3	-
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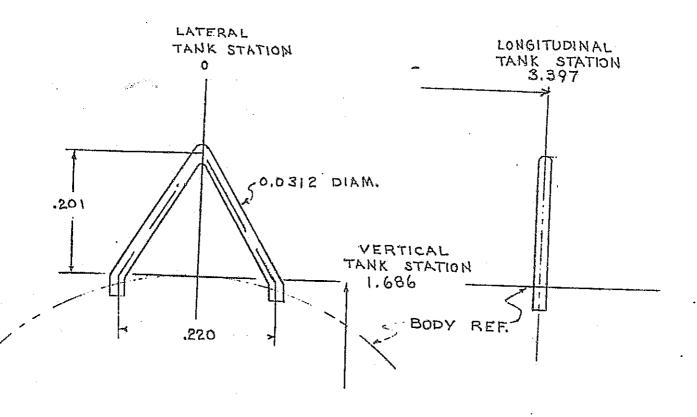
TABLE III. MODEL DIMENSIONAL DATA

MODEL COMPONENT: EXTERNAL TANK	- T ₁₂	
GENERAL DESCRIPTION: EXTERNAL O	XYGEN - HYDROGEN TANK	WITH OGIVE NOSE AND
SEMI-ELIPTICAL TAIL: BEGINNING	AT MODEL TANK STATION	0.927 AND ENDING AT STATION
6.522		
MODEL SCALE: 0-003		
REFERENCE DRAWING: VL78-00004	<u>1B</u>	
	THEOR	ETICAL
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Length	1865 in.	5.595 in.
Max. Width	324 in.	0.972 in.
Fineness Ratio	5.756 in.	5.756 in.
Max. Cross-Sectional	572,555_ft ²	0.742 in. ²
Base	572.555_ft ²	0.742 in. ²
WL OF TANK CENTERLINE	400 in.	1.200 in.

TABLE III. (Continued)

MODEL COMPONENT:	ATTACH STRUCTURE - AT5	_
GENERAL DESCRIPTION:	FORWARD ORBITER/ET ATTACH STRUCTURE	
(2 MEMBERS)		
***************************************		_
		_
MODEL SCALE: 0.003		
REFERENCE DRAWING: VL	72-000088D	_

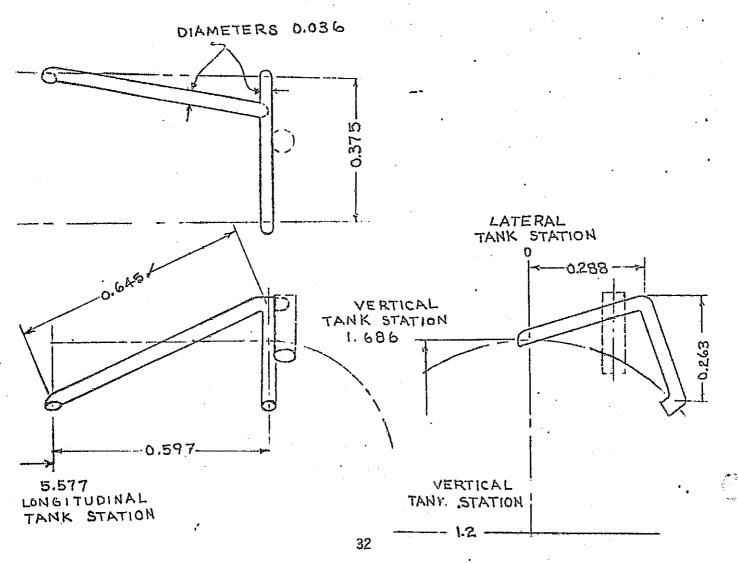
ALL DIMENSIONS IN INCHES MODEL SCALE



MODEL COMPONENT:	ATTACH STRUC	TURE - AT6	-t	, .		
GENERAL DESCRIPTION:	LEFT REAR OR	BITER/ET AT	TTACH STR	UCTURE (2	MEMBERS)	
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•				•		
MODEL SCALE: 0.003						
REFERENCE DRAWING: VL	.78-000050	4				
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			NK STA			

MODEL COMPONENT: ATTACH STRUCTURE - AT7	t and the second
GENERAL DESCRIPTION: RIGHT REAR ORBITER/E	T ATTACH STRUCTURE (3 MEMBERS)
MODEL SCALE: 0.003	
REFERENCE DRAWING: VL78-000050	

ALL DIMENSIONS IN INCHES MODEL SCALE



MODEL COMPONENT:	ATTACH STRU	CTURE - A	r _e	• • • • • • • • • • • • • • • • • • • •	•		
GENERAL DESCRIPTION	I: FORWARD SRI	B/ET ATTAC	CH STRUCTUR	<u>e (et p</u>	ORTION	TESTED (ONLY)
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					1.	. 1	
MODEL SCALE: 0.0	03				***************************************		,
REFERENCE DRAWING:	VL77-000051A		vana.	i		نيق ورب استخدادات	
•			ALL DIMENS	SIONS I	N INCHES	MODEL.	SCALE
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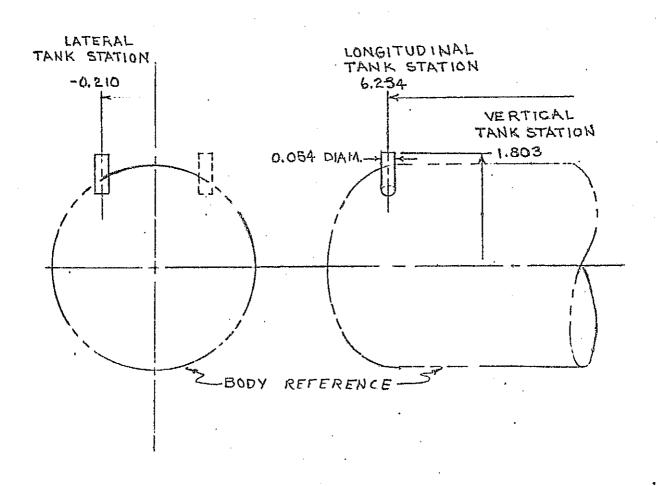
GENERAL DESCRIPTION: AFT SRB/ET ATTACH STRUCTURE (3 MEMBERS) (ET PORTION TESTED ONLY)
ONLY)
MODEL SCALE: 0.003 '
REFERENCE DRAWING: VL72-000106
O.018 O.018 O.018 O.018 O.018 O.0480 LATERAL TANK STATION
0.15 VERTICAL TANK STATION 0.474 LATERAL LONGITUDINAL

MODEL COMPONENT: LOX VENT LINE FAT	RING - PT.	
GENERAL DESCRIPTION: VENT LINE AL	ONG UPPER RIGHT S	IDE OF ET OGIVE NOSF
BEGINNING AT MODEL STATIONS X _T = 0	.927. Y = 0, AND	7 _T = 1.2; TERMINATING AT
$X_{T} = 2.841, Y_{T} = 0.162, Z_{T} = 1.658$		
		to the second se
MODEL SCALE: 0.003		
REFERENCE DRAWING: VL78-000031A		
	-	
		RETICAL
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Length:	638 in.	. <u>1.914 in.</u>
Max. Width	17.7 in.	0.053 in.
Max. Depth	9.3 in.	0.028 in.
Radial Position	19 1/2°	<u>19 1/2°</u>

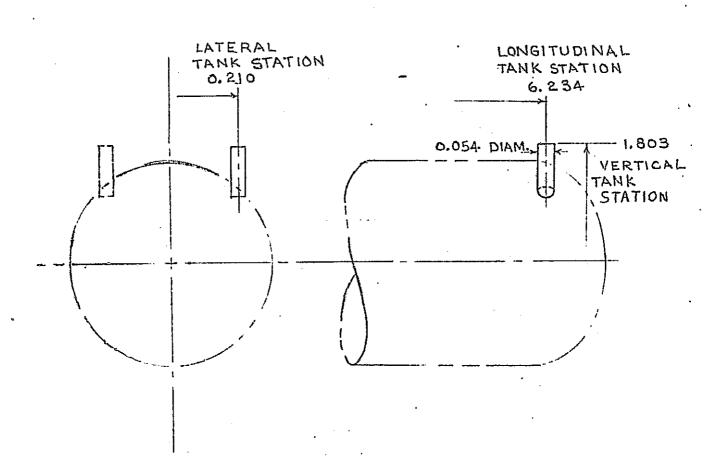
MODEL COMPONENT: LOX FEED L	INE - PT,	
GENERAL DESCRIPTION: LONGITUD BEGINNING AT MODEL STATIONS X	·	
AT $X_T = 6.116$, $-Y_T = 0.194$, AN	$1D Z_{T} = 1.645$	
MODEL SCALE: 0.003 REFERENCE DRAWING: VL78-0000)31A	
	THEOR	RETICAL
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Length	1092 in.	3,275 in.
Max. Width	30.7 in.	0.092 in.
Max. Height	28 in.	0.084 in.
m 41.7 Maritian	22 1 /20	23 1/2°

MODEL COMPONENT: LH2 FEED LINE - PT3								
GENERAL DESCRIPTION: LONGITUDINAL FUEL LINE ALONG UPPER LEET SIDE OF ET								
BEGINNING AT MODEL STATIONS $X_{\uparrow} = 2.84$	1, Y _T = 0.275, AN	ID 7 = 1.601						
TERMINATING AT STATIONS X _T = 6.116. Y	_ = 0.275, AND Z.	_ ≈ 1.601						
MODEL SCALE: 0.003								
REFERENCE DRAWING: VL78-000031A								
REFERENCE DRAWING. VE70-VOGVOIG								
	THEOR	ETICAL						
DIMENSIONS:	FULL-SCALE	MODEL SCALE						
Length	1092 in.	3.275 in.						
Max. Width	25.7 in.	<u>0.077 in.</u>						
Max. Depth	14.7 in.	0.044 in.						
Radial Position	-33°	33°						

MODEL COMPONENT: LO	X FEED LI	NE - FL	,					·
GENERAL DESCRIPTION:	18-INCH	<u>DI AMETER</u>	VERTICAL	FUEL.	LINE A	T AFT	END OF E	T ON
RIGHT			·			- 	<u> </u>	
						,	• 1	
		<u> </u>				*		****
MODEL SCALE: 0.003					, -			
REFERENCE DRAWING:	VL 78-0000)50						
_			Δ(1	n ntme	NSTONS	TN IN	CHES MOD	EL SCAL



SEMERAL DESCRIPTION:	18-INCH	DIAMETER	VERTICAL	FUEL	LINE A	<u>r aft</u>	END OF	ET
ON LEFT .		•	•					
		<u> </u>			,	•		
ODEL SCALE: 0.003		•						
EFERENCE DRAWING:	VL78-000	050		······	<u>, , , , , , , , , , , , , , , , , , , </u>			



MODEL COMPONENT: ATTACH STRUCTURE - FR₆

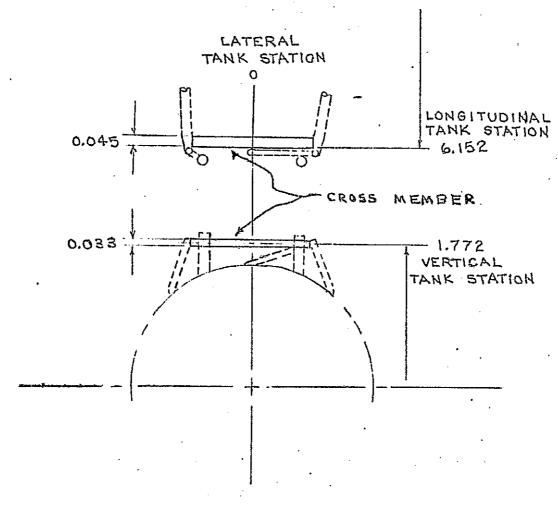
GENERAL DESCRIPTION: AFT ET/ORBITER CROSS MEMBER (CROSS SECTION 11 IN. x 15 IN.)

LOCATED AT ET-STATION 205D.5

MODEL SCALE: 0.003

REFERENCE DRAWING: FIGURE 3, MARTIN MARIETTA MEMO SA-A-74-9

ALL DIMENSIONS IN INCHES MODEL SCALE



TABULATED DATA PRINT-OUT FORMAT AND CORRELATION BETWEEN TUBE NUMBER, ORIFICE NUMBER, AND ORIFICE LOCATION ON MODEL TABLE IV.

inoperable orifice

ر با المستور ((المستور (المناسلة)

SIDE-MOUNTED ET (au_2 CONFIGURATION)

あるがら 0.954 142 154 7 60/ 177 W اله و 611 165 ري 0 9 181 N o Ø 3 2 ന 61 M N 1 ø N 5,160 5260 104 176 107 130 18 (V) 180 4 3 8 4 2 و ທ 5 4 ~ 5 5 ĐO ū Ö 4,985 0.892 140 175 3 Q 106 129 152 179 190 67 0 4 9 4 m ā 8 9/ ~ Do 0.860 4.810 174 128 105 162 178 681 60 24 о О (م وج 47 59 2 5 0-O (v) 0 N 0.735 0//: 38 401 150 127 3 <u>(1)</u> (r) ານ ໝ , 00 173 4 161 ρ_0 Ø ∞ 6 0-90 ڼ 0.610 3.4.0 149 ※/87 103 126 137 09/ 172 ※ 45 73 φ 3 0 5 4 1 1 × 0.578 2.900 0 */86 ro Ø S S N 3 171 ω ω 々 64 5 06 .0 17 0 5 S 7 4 4 *0,322 1.800 124 135 147 158 170 101 b N 'n 9 78 ۍ ص **(^)** u) <u>৩</u> иì ē, ₫ 0,216 1,205 40 184 134 157 3 000 691 4 45 00 وي 77 ល * ហ <u>~</u>3 4 ď 0.905 0.162 123 W 156 M 99/ 4 94 2 M 0 a 6.3 4 0 M) ന 4 M 00 00 œ 0,605 0.108 44 132 155 167 2 S N 0 Φ S 3 V ی N O_Q N ~ 8 4 ហ s ø Ů. 0,055 0.305 143 90 α_{i} 47 O <u>-</u> 7 4 ē. **(**)_ 6 H Ø ∢ Ω Ш 1 Σ Δ L 1 Q Ů 2 7211 1572 315 2472 292 202= 672 0 135 225 270 378 346 45 8 AND THE PARTY OF T 24 王 Q REPUBLICATION LACATION

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NUSBER,	* inope	2680	 -	0	01	2.1	25	37	48	3	9	11	83	94	90/	117		127	140	152	163	175	179	061	\ \ \ \ \ \
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TABULATED AND ORIFI			0.108	0,605	2	2	53	X	54	40	Si	70	63	7.5	86	86	601		24	132	144	155	167	X	182
TABLE IV.			0,055 6		,		X	X	28	X		2	X	74	X	9.7			071		14.3	X	166	X	X
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TABLE V. TABULATED DATA PRINT-OUT FORMAT AND CORRELATION BETWEEN TUBE NUMBER, ORIFICE NUMBER, AND ORIFICE LOCATION ON MODEL

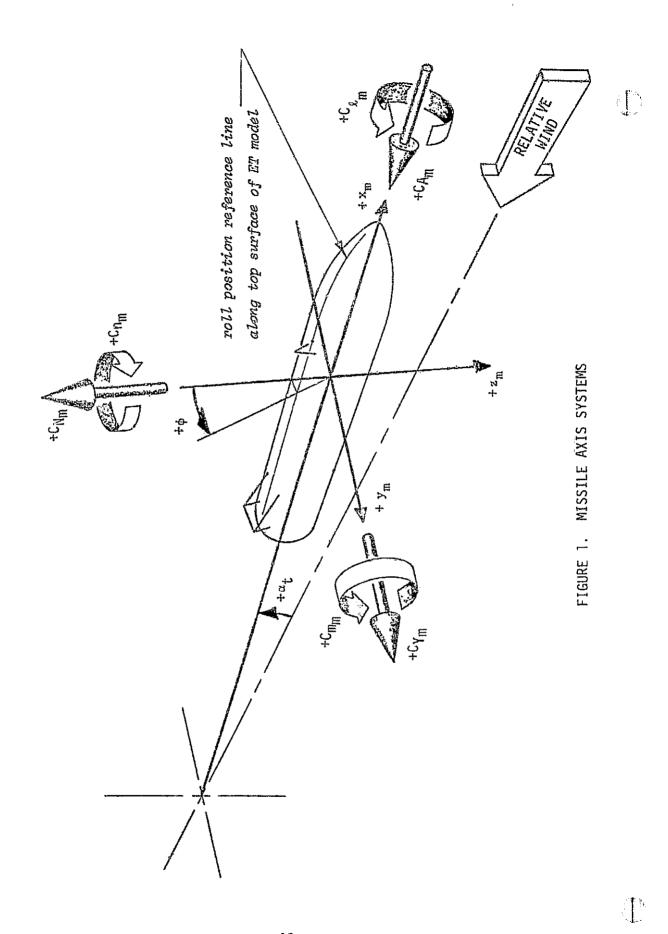
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rable o;	5250	5.160	"	: 7	200	1 0	1 6 8 a	0 0	1,3		1 8	, O		118	180	4	153	497	176	08/	161
, inoperable	268.0	4 985	9	0/	- 0	וות	2 6	ο σ	0,9	7	6.0		70/	211	129	140	152	163	175	179	\$ 96
	0.860	4.810	0	0	0	40	36	77	500	70		93	105	116	100 E	139	151	162	174	178	183
TIOM)	0.735	4.110	00	00	6	X	LC iY		500	69	~ ~	26	100	11.5	127	/38	150	191	173	X	821
(7 ₁ CONFIGURATIOW)	0.610	3.410	7	7	00	X	460	4.5	57	88	80	ō	103	114	126	137	601	1,60	172	X	187
ET (1 ₁ C	0.518	2.900	٥	9	71	X	83	44	56	67	79	06	102	113	125	136	148	/29	121	X	981
TAIL-MOUNTED	0,322	1.800	to	5	2/	X	32	43	※55	99	78	68	/0/	112	124	135	*/47	158	1 70	$\langle \rangle$	185
TAIL-	0216	1,205	#	4	15	X	18	42	54	65	77	88	001	111	123	134	* 901	157	691	M	134
	0.162	0.905	W	(Y)	14	X	30	4-1	53	64	76	87	66	071	122	133	145	156	168	X	183
	0.108	0.605	Ŋ	7	13	\bigvee	29	40	5 5	63	75	56	9.8	601	121	132	144	155	167	X	182
	0,055	0.305	/	,	\bigvee	\bigvee	20	\bigvee	5.	\bigvee	74	\bigvee	126	X	120	\bigvee	143	\bigvee	166	X	X
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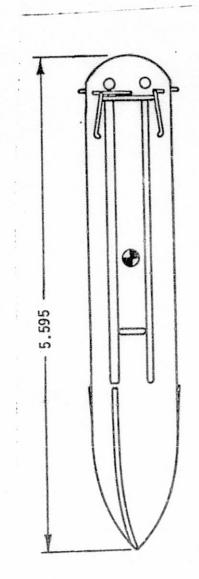
TABLE V. TABULATED DATA PRINT-OUT FORWAT AND CORRELATION BETWEEN TUBE NUMBER, ORIFICE NUMBER, AND ORIFICE LOCATION ON MODEL (CONCLUDED)

	ſ	7	la	T	1	T-		<u> </u>	T-		-	_			——·	·			,						
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able ori		0.923	5.160	11		22	26	ίλ. «	49	13	7.2	4	84	25	1		2/8/	130	14:1	153	164	1	- -	_	19i
* inoperable c	.	0.872	4,985	0/	01	21	25	37	84	09	17		83	94	90/		111	129	140	152	163	-	┤-	+	06/
		2000	4.810	6	6	20	24	36	47	59	102	*	מא	93	105	1:	9//	128	139	151	162	174	-	╁	/57
OM)	0720		4/10	Ø	ထ	61	\bigvee	35	46	58.	69	T	ō	92	104	115	57	121	/38	150	191	173	X	_	00/
CONFIGURATION)	0610	7 7 7 7	1,470	7	1	8/	X	34	45	57	89	0		5	/03	1,4	137	+	+	149	160	172		187	-
(1	0.518	7,900	3	٥	o	17	X	33	44	56	67	20		2/	102	113	10 11	一	+	748	159	171		786	\dashv
NTED ET	0,322	1,800	L	0	5	9/		32	43	55	٥٥	78	000	70	10/	112	124	-	╁	/#/	158	170	X	85.	$\frac{1}{2}$
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•	0.162	0.905	M	2	3	<u>+</u>	(.30	1 4 1	00	6.4	76	24			110	122	/33	14.5	7 1		\ \(\cdot\)	$\left\langle \right\rangle$	687	1
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	0,055	0.305		,	$\left \right\rangle$	$\langle \rangle$	۵	2	1	;		#	X	9.7		1	120	X	143	X	13	-K $-$	\uparrow		
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		· ·	AMARAN S	0	#	777	45	÷19	90	7211	7 2.17		[1572]	8	1,000		225	2472	270	2924	1		_Ł_		
									-													-1. "	<u></u>	_1	

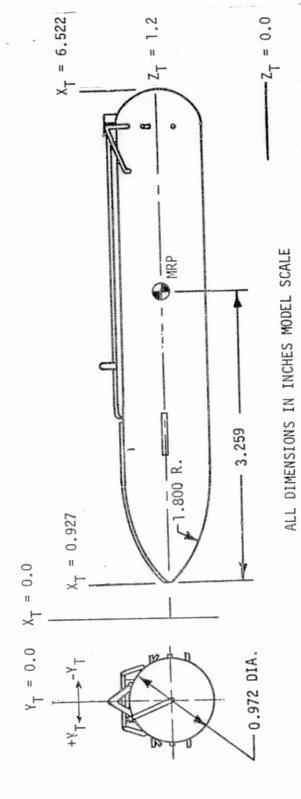
Table VI.
0.003-SCALE 324-INCH ET REFERENCE DIMENSIONS

DIMENSION	FULL SCALE	MODEL SCALE
Reference Area, S _{ref} (cross-sectional area of ET)	572.555 FT ²	0.742 IN. ²
Reference Length, L _{ref} (ET diameter)	324 IN.	0.972 IN.
Reference Span, b _{ref} (ET diameter)	324 IN.	0.972 IN.
Moment Reference Point, MRP (dry weight c.g.) XMRP (from nose) YMRP ZMRP (model centerline)	1086.4 IN. O 400 IN.	3.259 IN. O 1.2 IN.
Base Area, A _b (cross-sectional area of ET)	572.555 FT ²	0.742 IN. ²

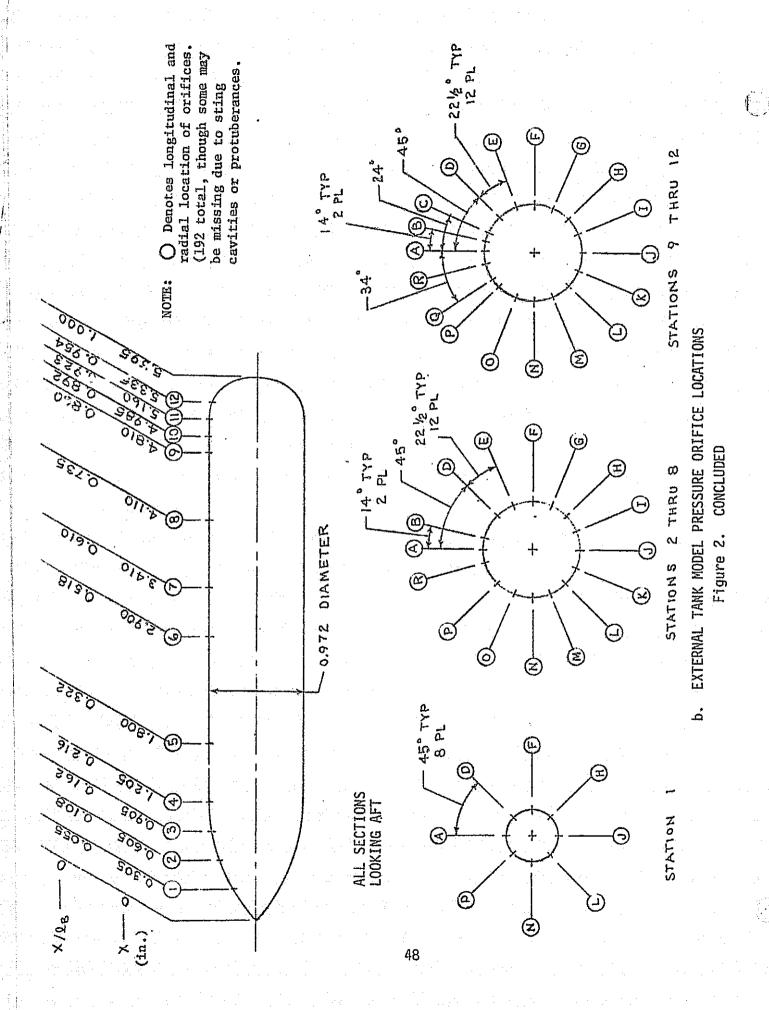


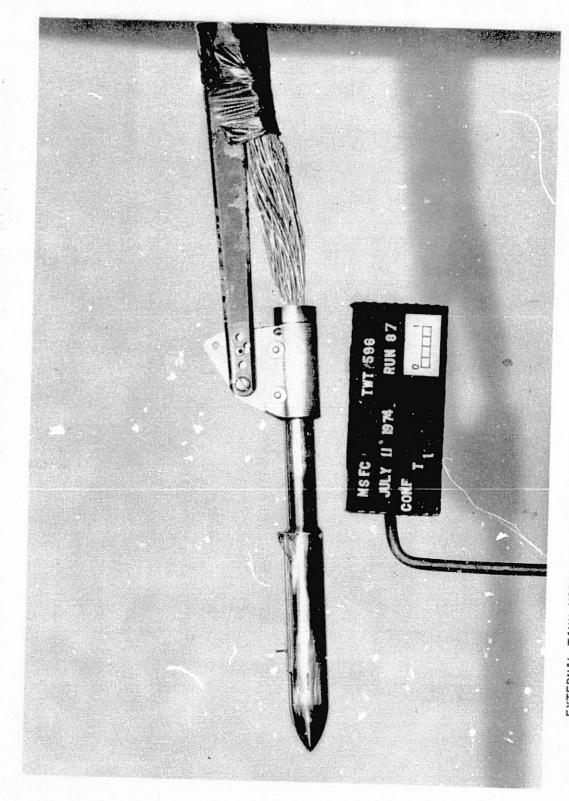


(1)

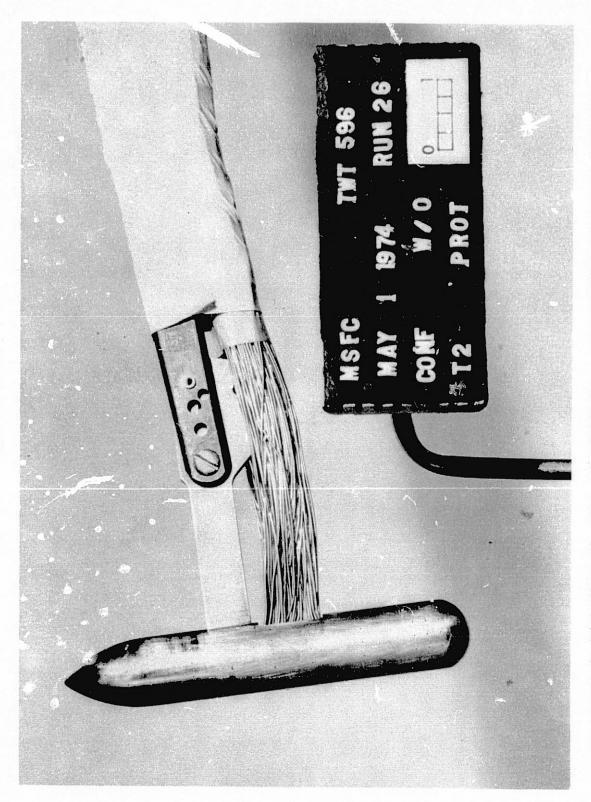


a. GENERAL ARRANGEMENT OF MSFC MODEL NO. 460, CONFIGURATION T1 EXTERNAL TANK WITH PROTUBERANCES Figure 2. MODEL SKETCHES





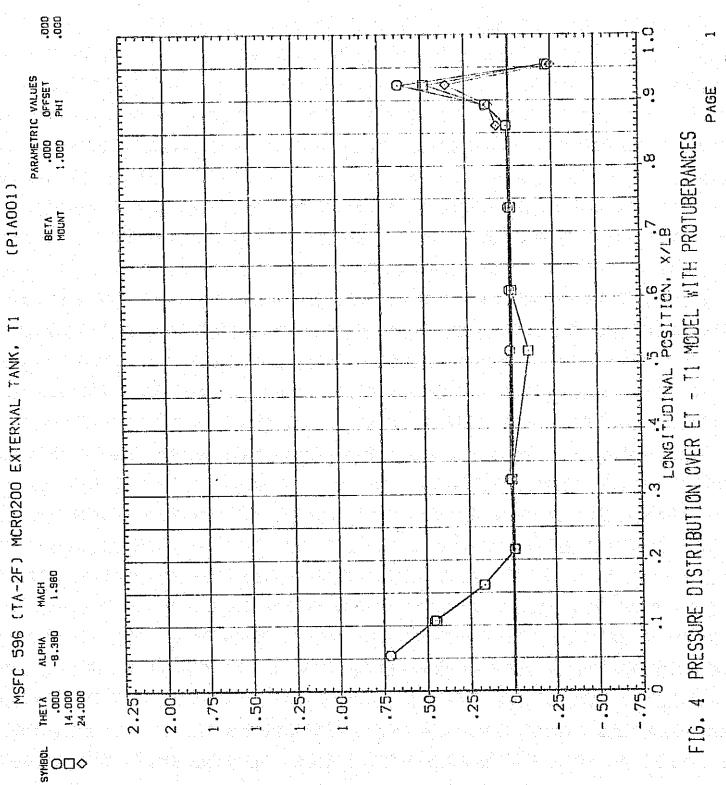
EXTERNAL TANK MODEL NO. 460, CONFIGURATION T1 TAIL-MOUNTED WITH PROTUBERANCES Figure 3. MODEL PHOTOGRAPHS



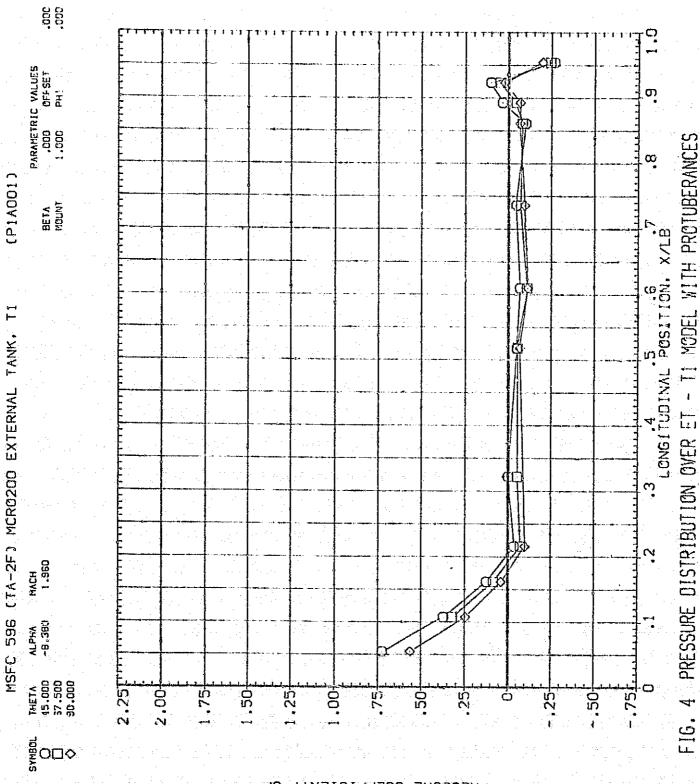
EXTERNAL TANK MODEL NO. 460, CONFIGURATION T₂ SIDE-MOUNTED WITHOUT PROTUBERANCES Figure 2. CONCLUDED b.

DATA FIGURES

VOLUME 1--Pages 1-720 VOLUME 2--Pages 721-1200 VOLUME 3--Pages 1201-2000 VOLUME 4--Pages 2001-2740



PRESSURE COEFFICIENT, CP.



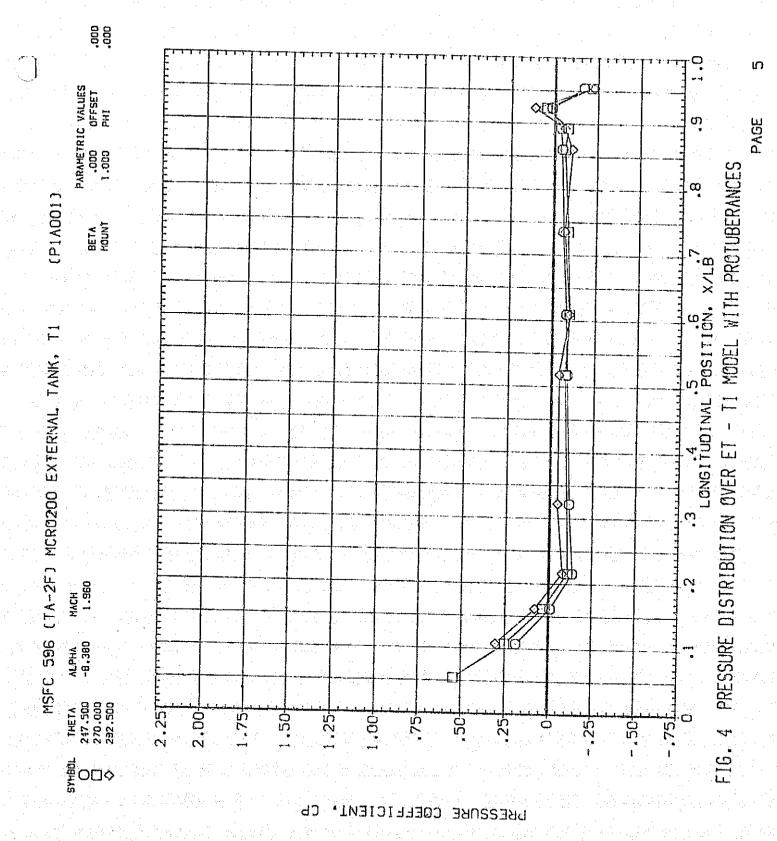
PRESSURE COEFFICIENT, CP

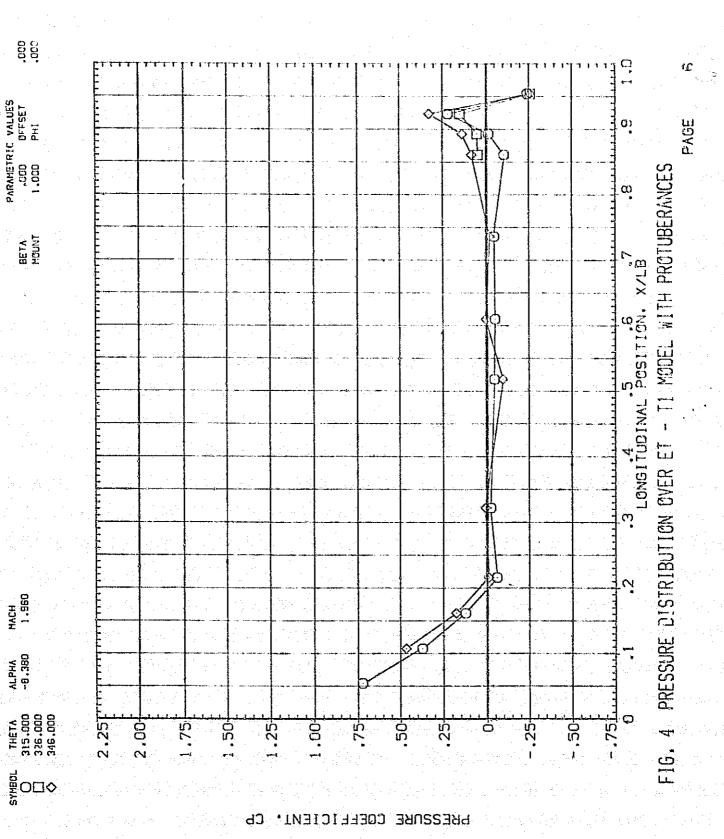
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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PRESSURE COEFFICIENT



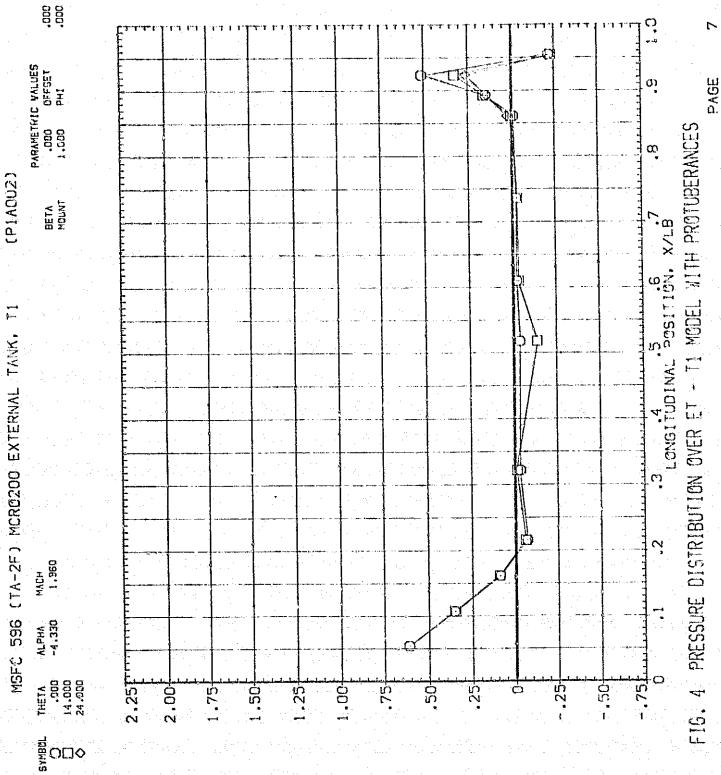


(P1A001)

MSFC 596 CTA-2F) MCR0200 EXTERNAL TANK, T1

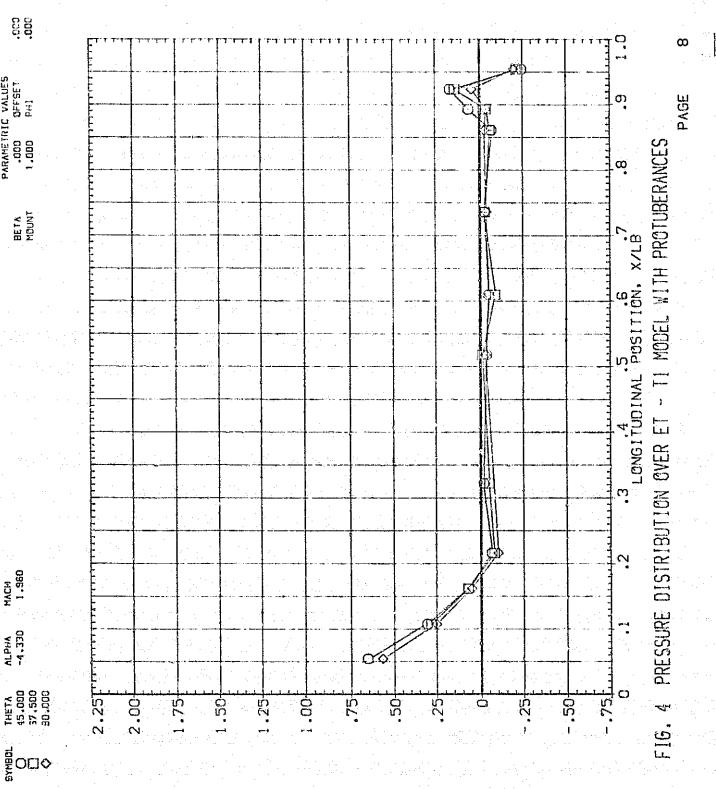
MACH 1.960

ALPHA -8.380



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(P1A002)

MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

MACH 1.960

ALPHA -4.330

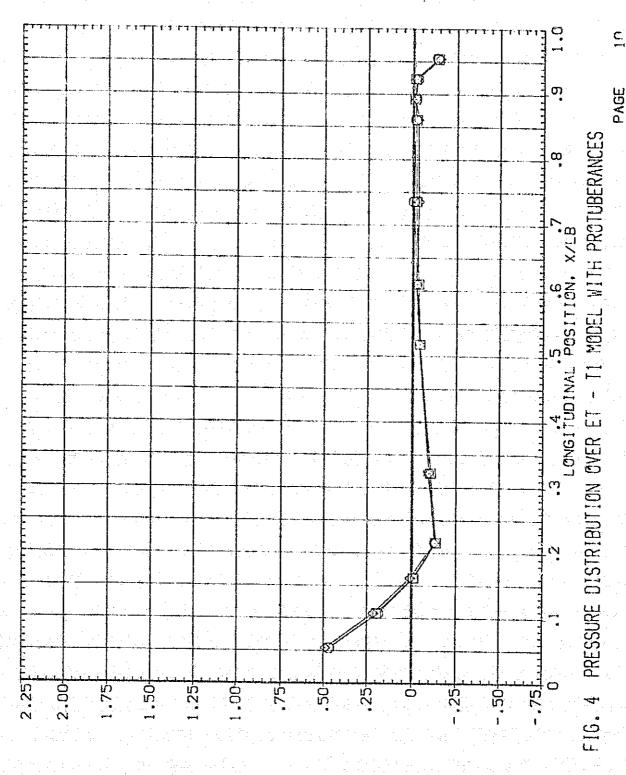
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK,

МАСН 1.960

ALPHA -4,330

THETA 130,000 232,500 225,000

SY-HO□◇

PRESSURE COEFFICIENT, CP

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BETA

(P1A002)

MSFC 596 (TA-2F) MCR0200 EXTERMAL TANK, TI

MACH 1.960

ALPHA -4.330

THETA 247.500 270.000 232.500

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2.00

1.75

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COEFFICIENT, **b**BE220BE

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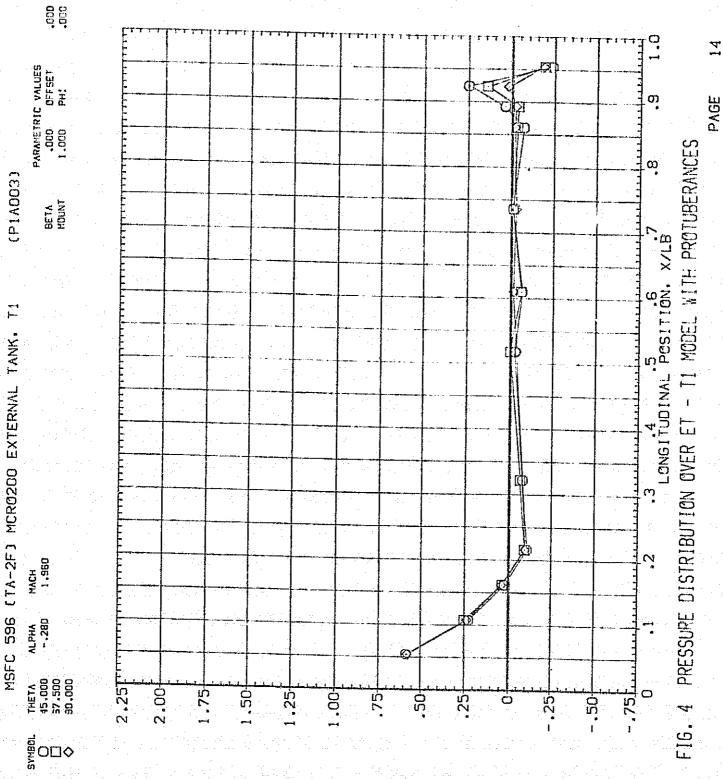
CP1A0023

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(TA-2F) MCR0200 EXTERWAL TANK.

MSFC 596

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PRESSURE COEFFICIENT, CP

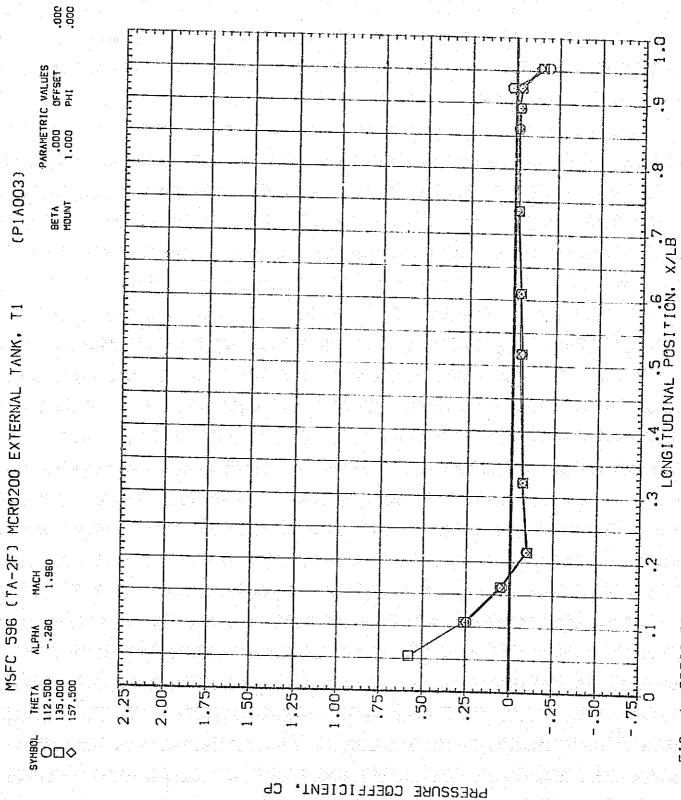
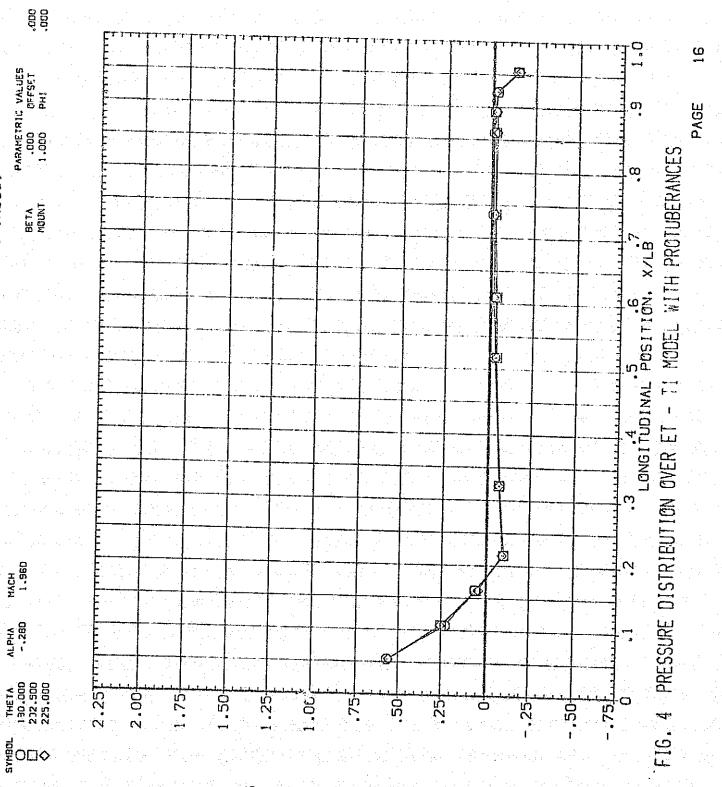


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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PARAMETRIC VALUES .000 OFFSET 1.000 PHI

BETA MOUNT

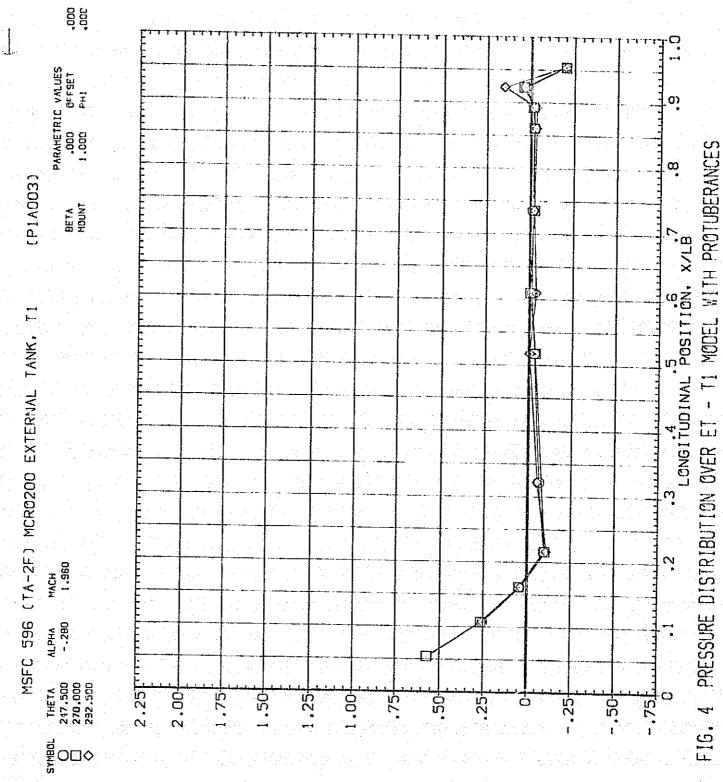
(P1A003)

MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

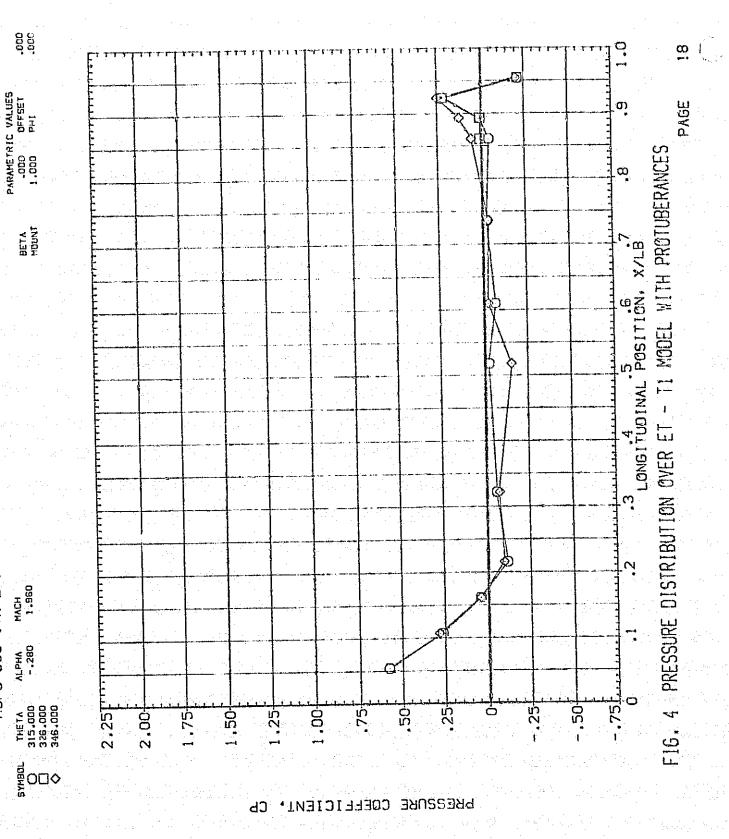
мAСН 1.960

ALPHA -.280

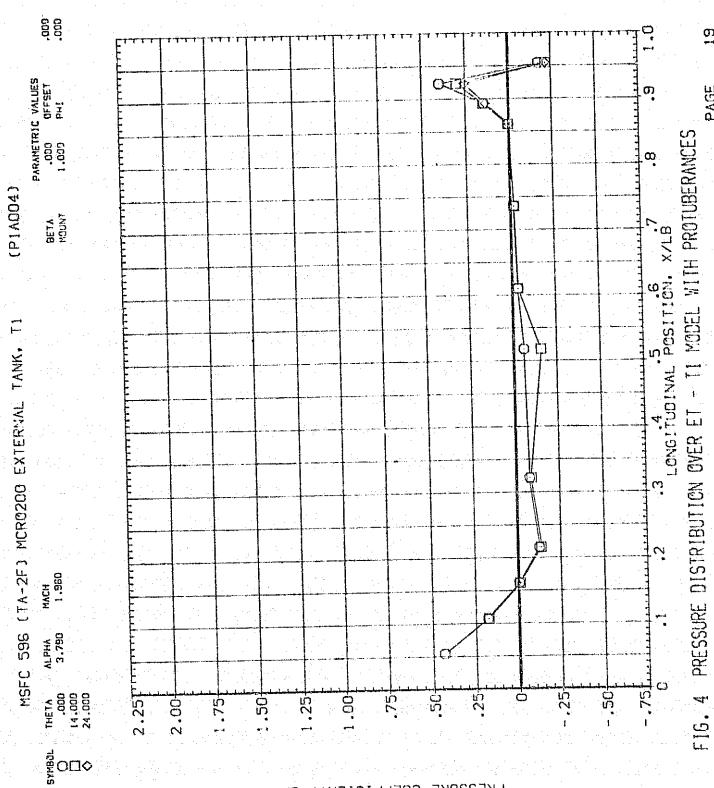
COEFFICIENT, PRESSURE



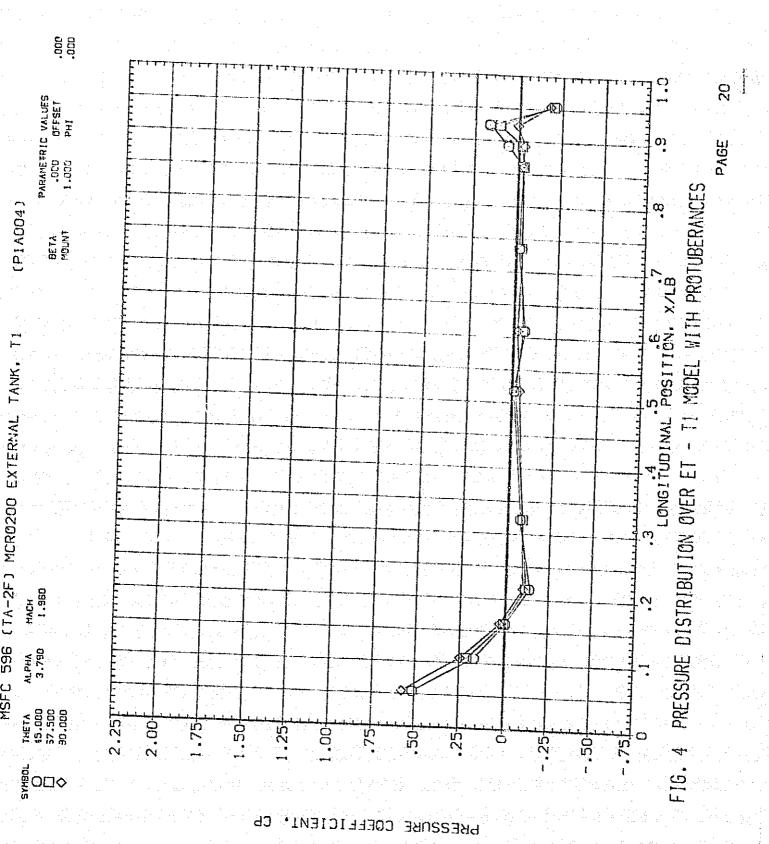
PRESSURE COEFFICIENT, CP

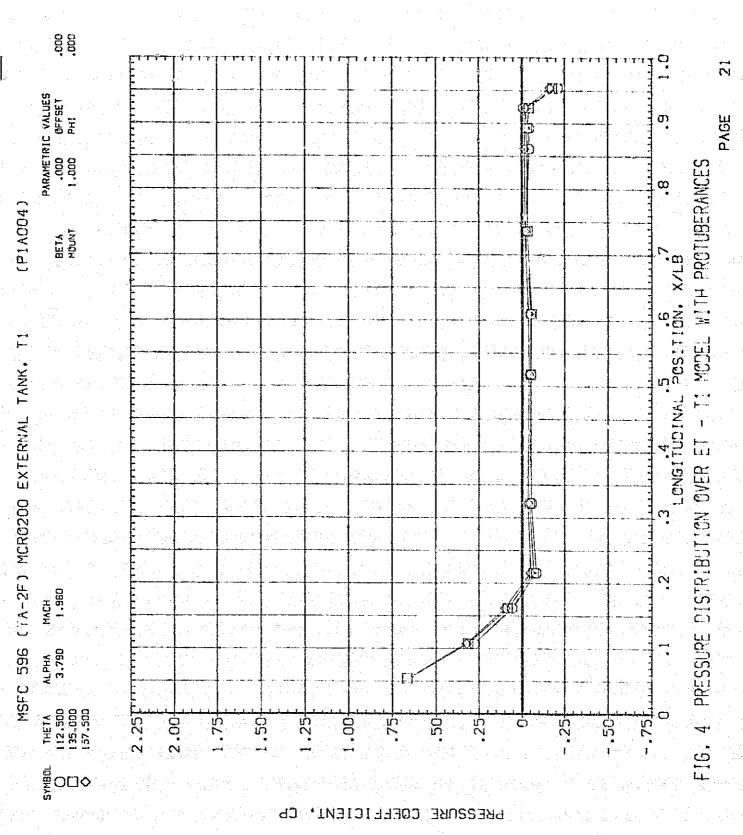


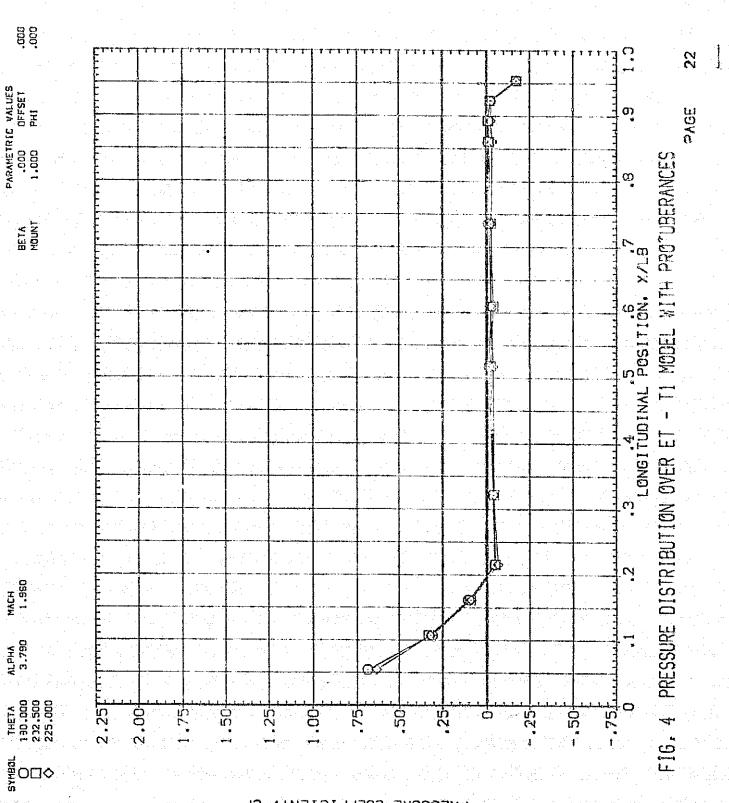
46,1



COEFFICIENT **BRESSURE**





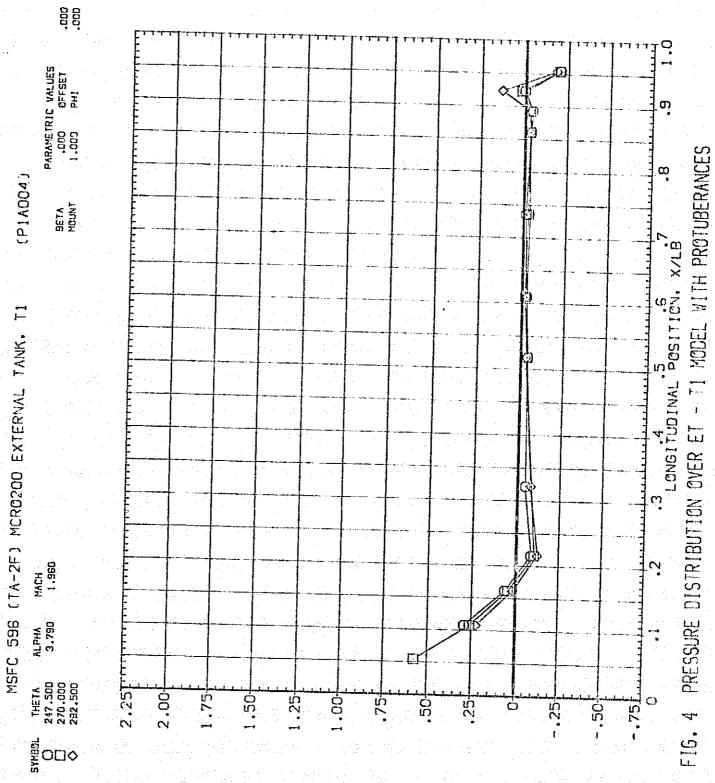


(P1A004)

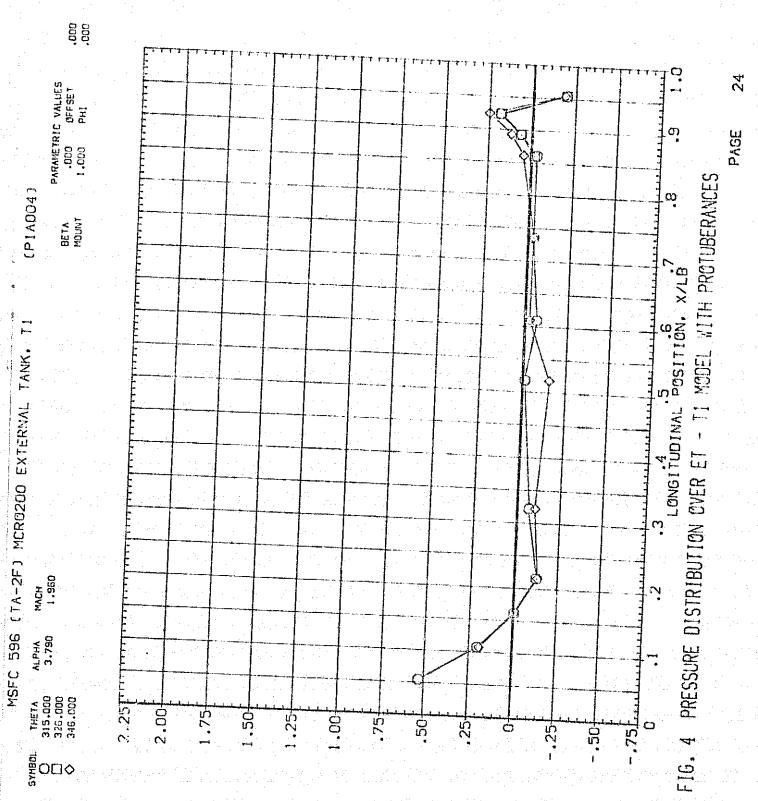
MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI

MACH 1,950

ALРНА 3.790



RESSURE COEFFICIENT, CP

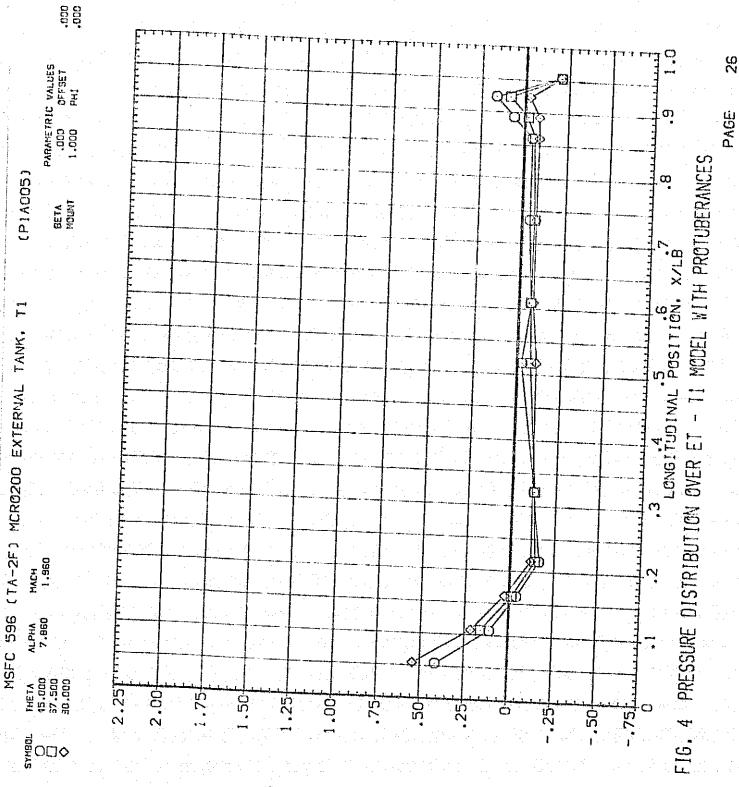


PRESSURE COEFFICIENT, CP

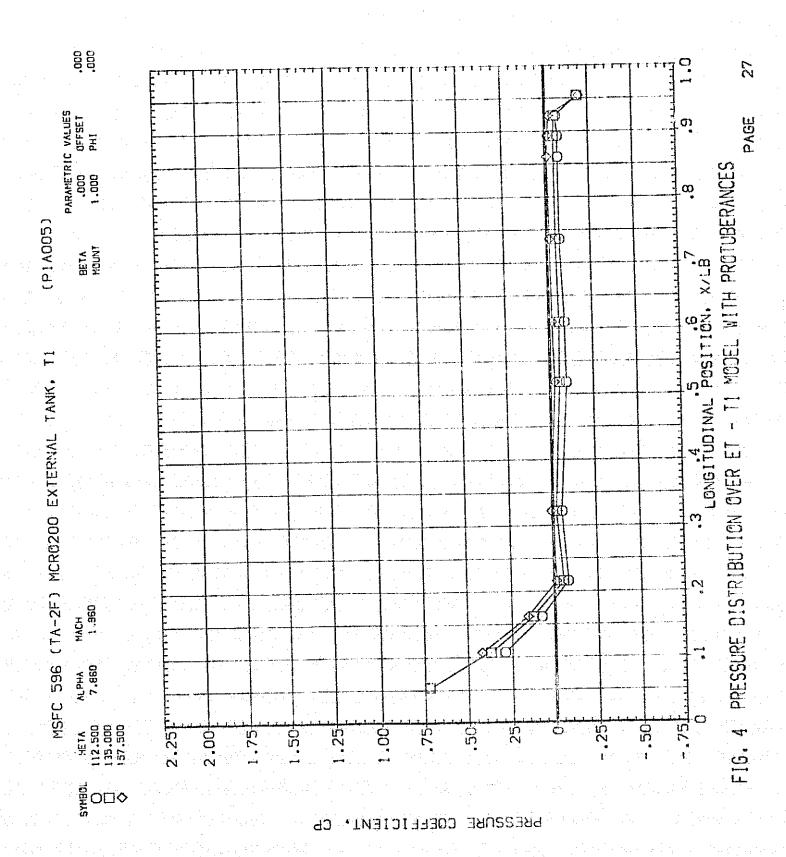
BRESSURE

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

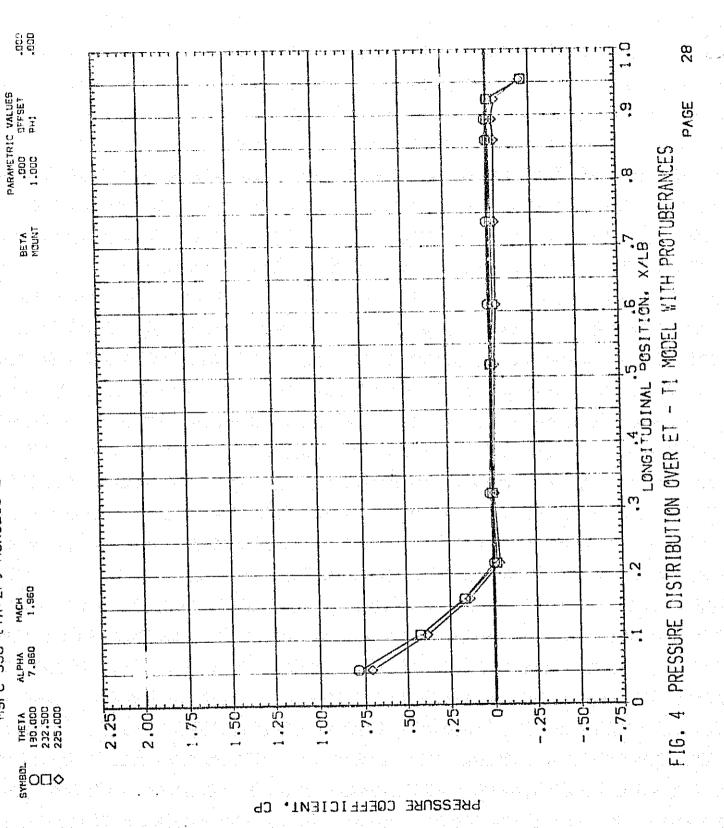
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PRESSURF COEFFICIENT, CP



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(P1A005)

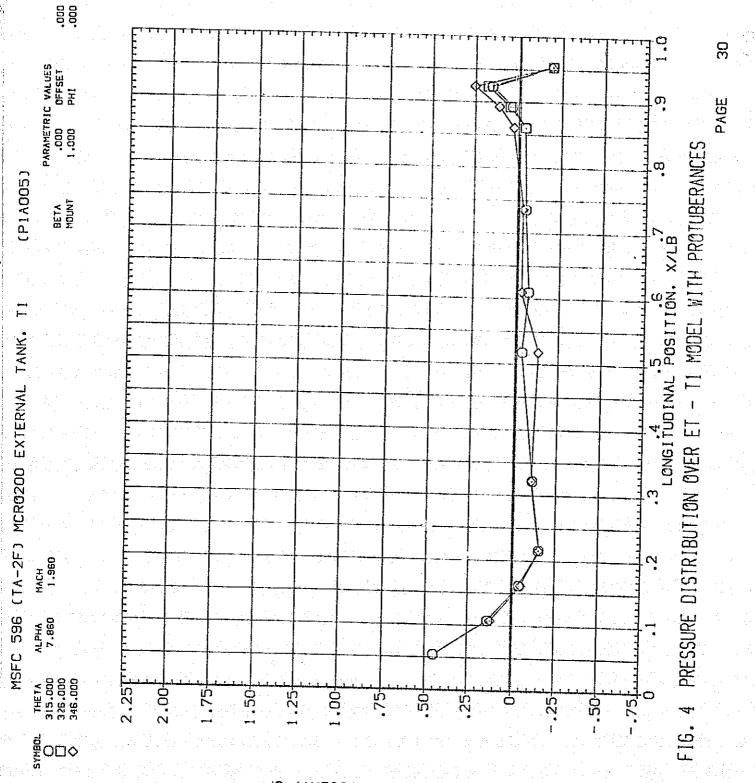
MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK,

MACH 1,960

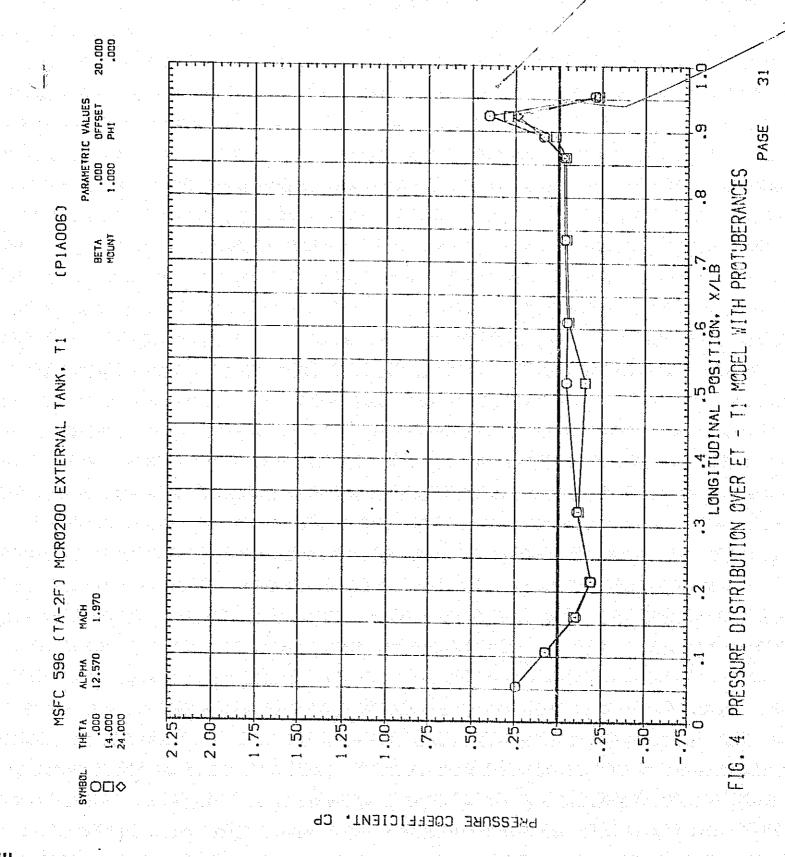
COEFFICIENT

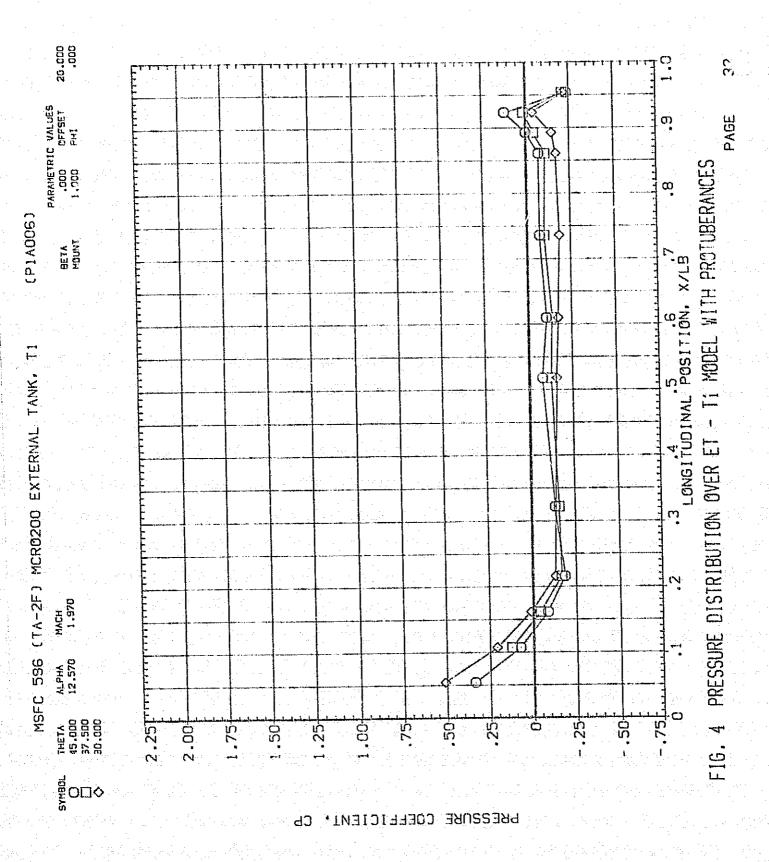
BRESSURE

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH FROTUBERANCES



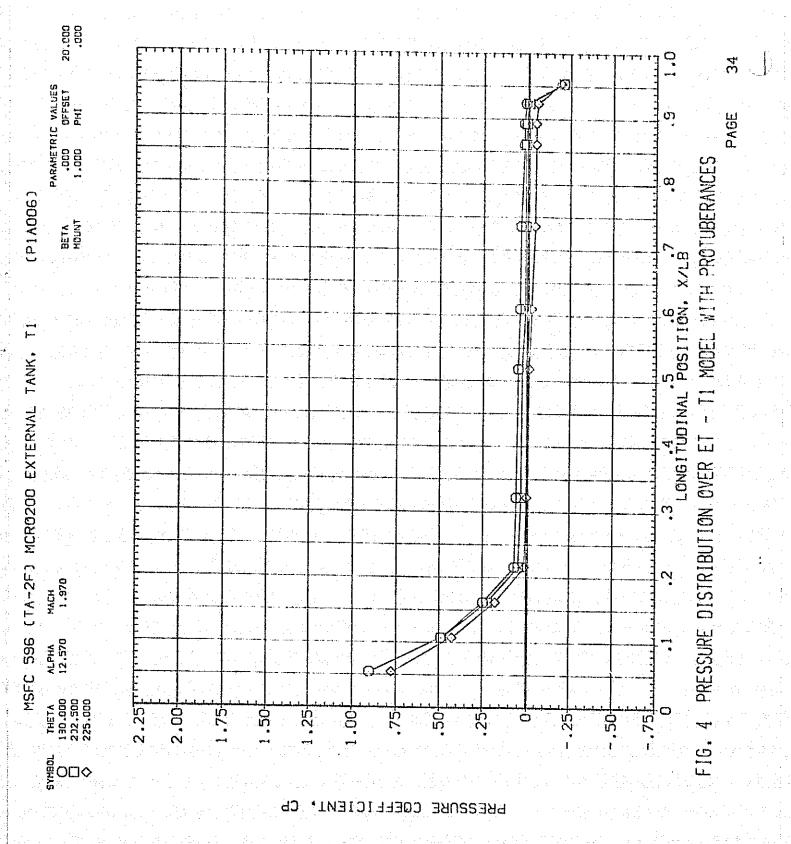
PRESSURE COEFFICIENT, CP





PRESSURE COEFFICIENT.

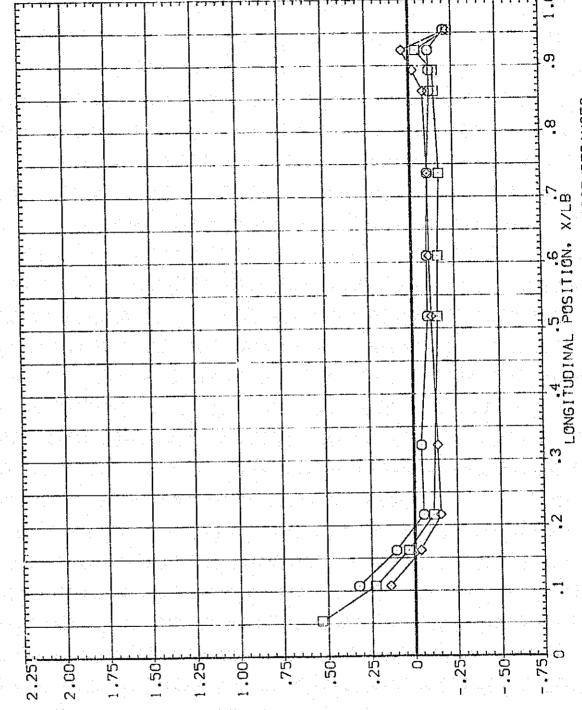
33



PAGE

20.000 PARAMETRIC VALUES .000 OFFSET 1.000 PHI (P1A006) BETA MDUNT MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK. MACH 1.970 ALPHA 12.570 THETA 247.500 270.000 232.500 Sweet Co Co

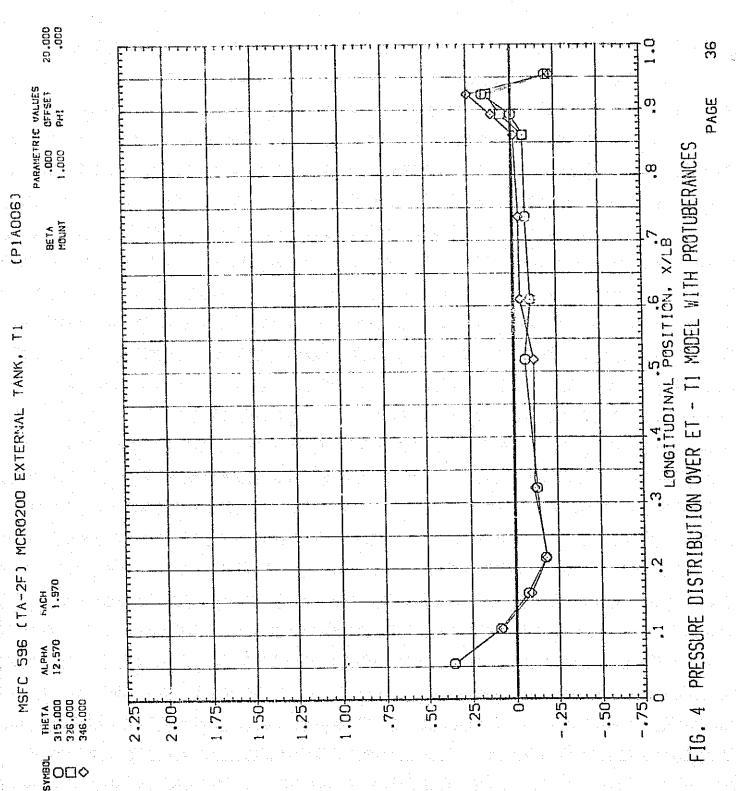




PRESSURE COEFFICIENT.

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



PRESSURE COEFFICIENT, CP

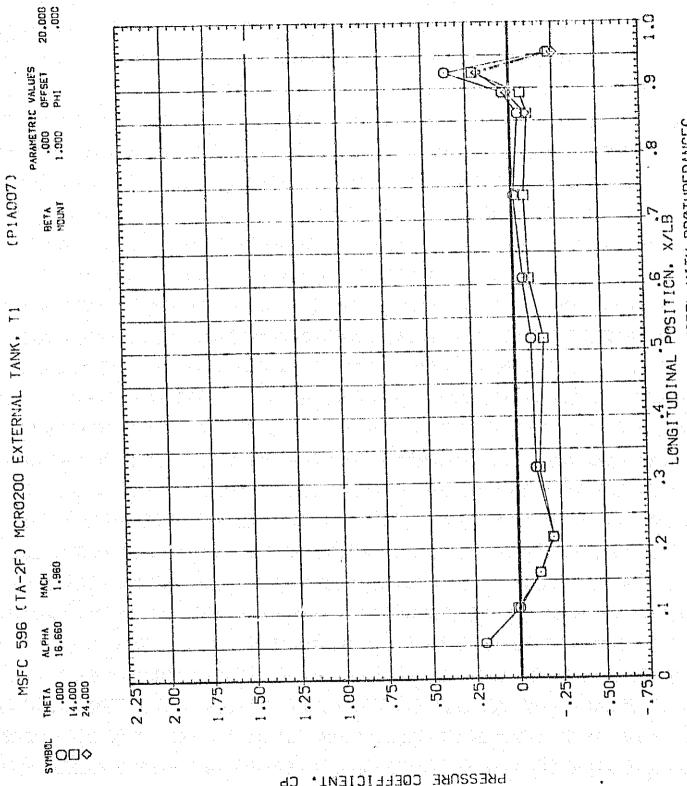
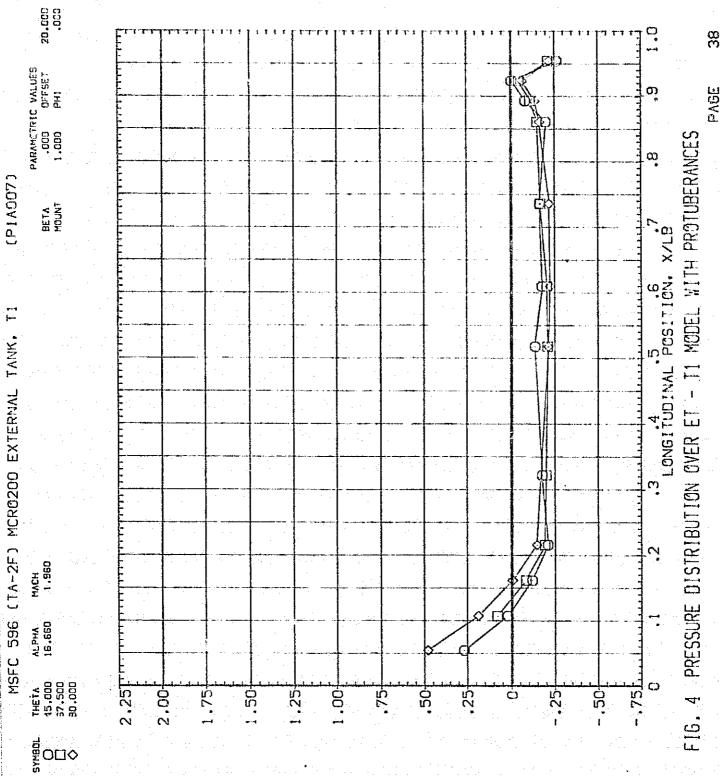


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



PRESSURE COEFFICIENT, CP

(P1A007)

THETA 112.500 135.000 157.500

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2.25

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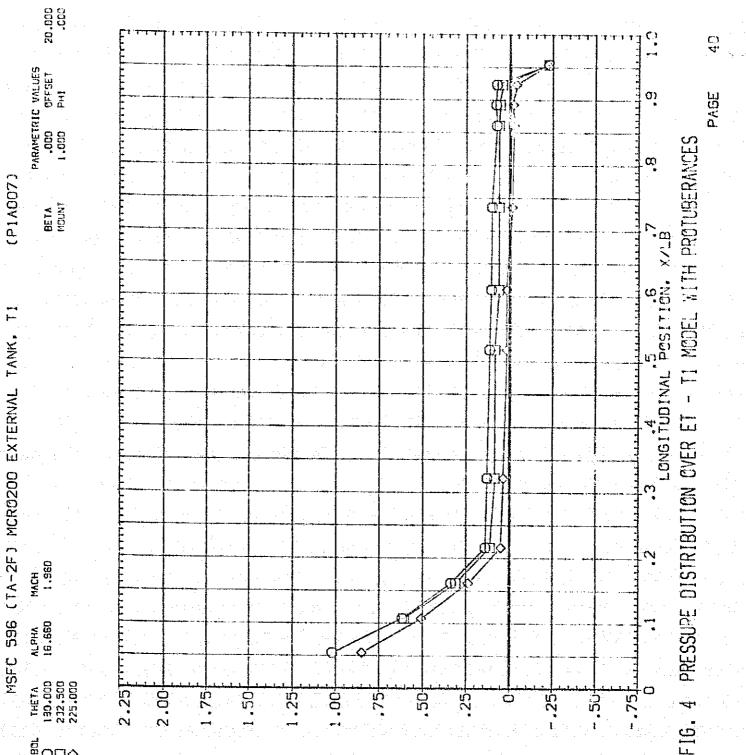
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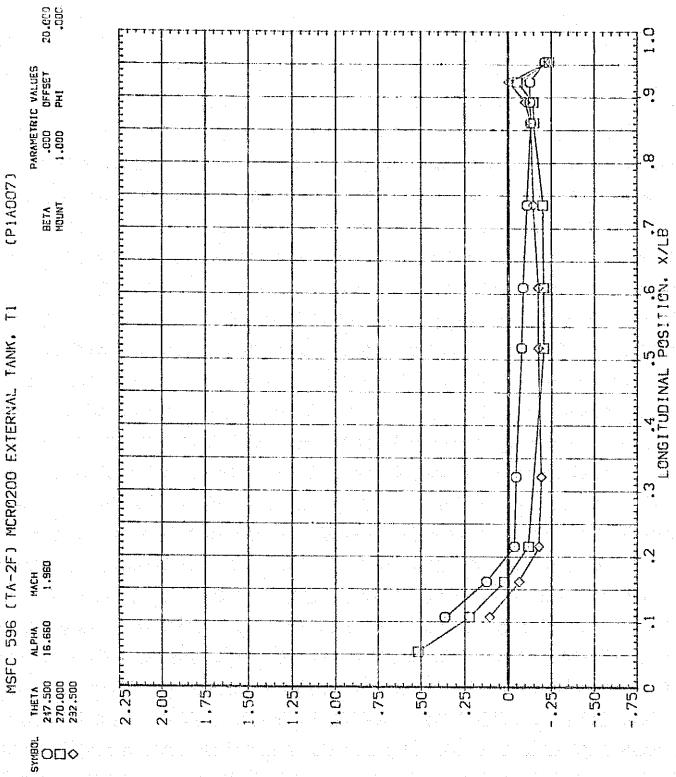
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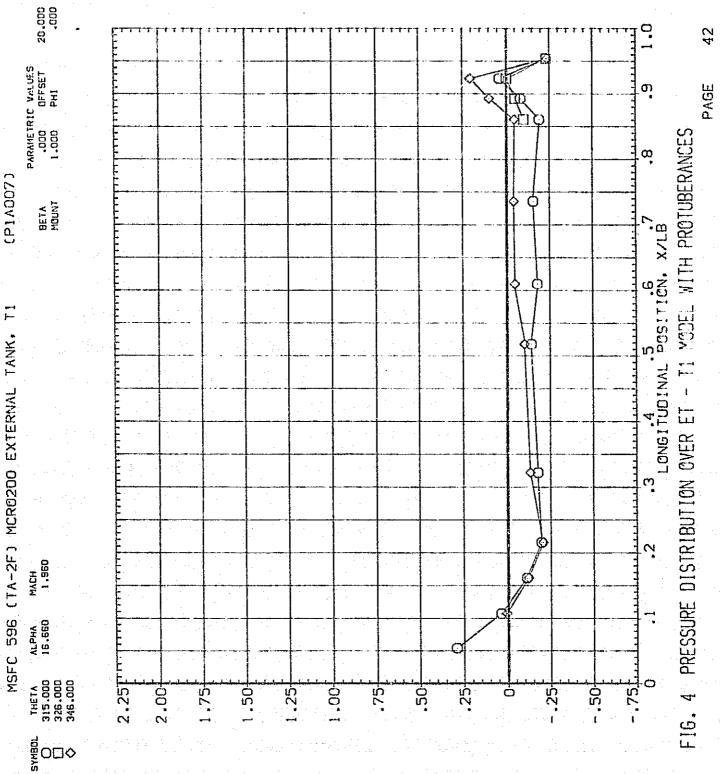
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



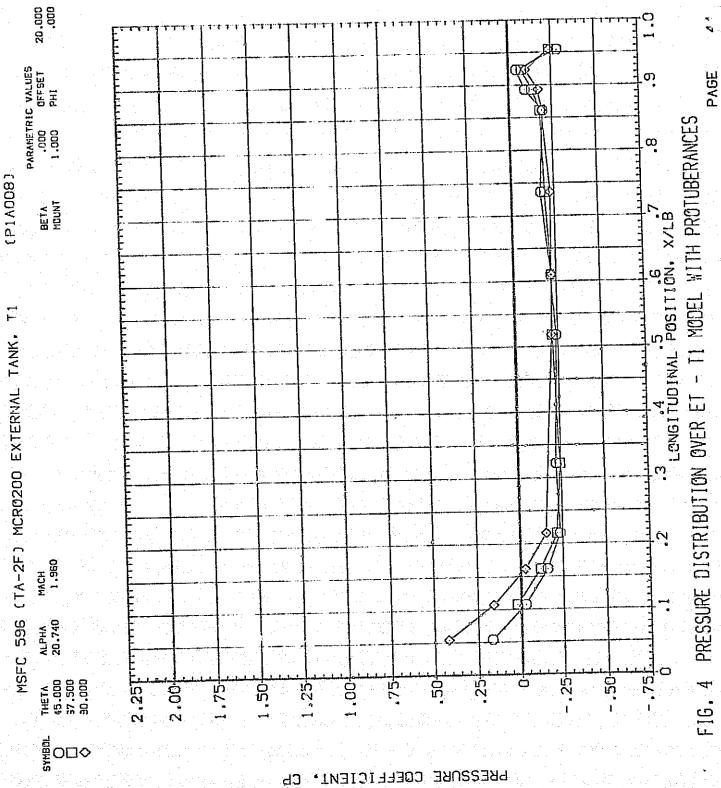
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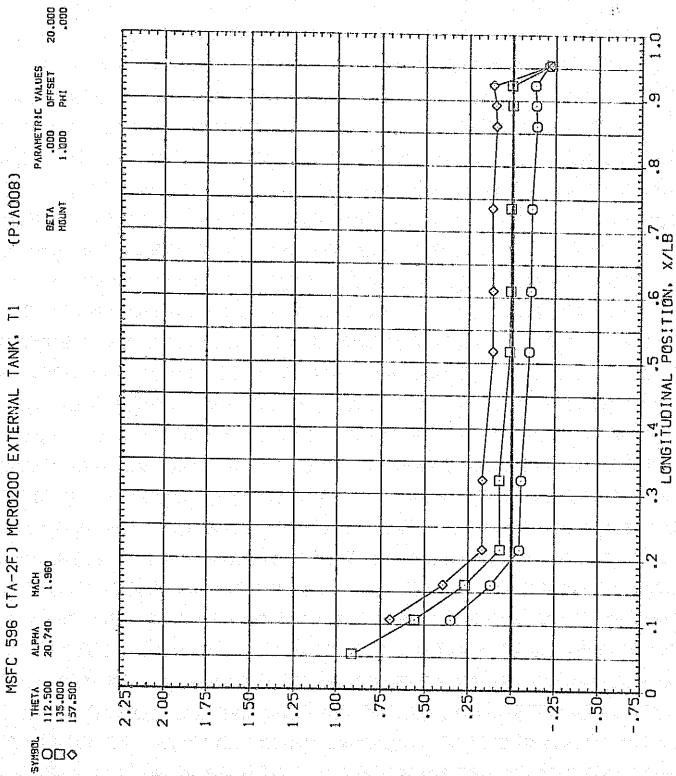
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

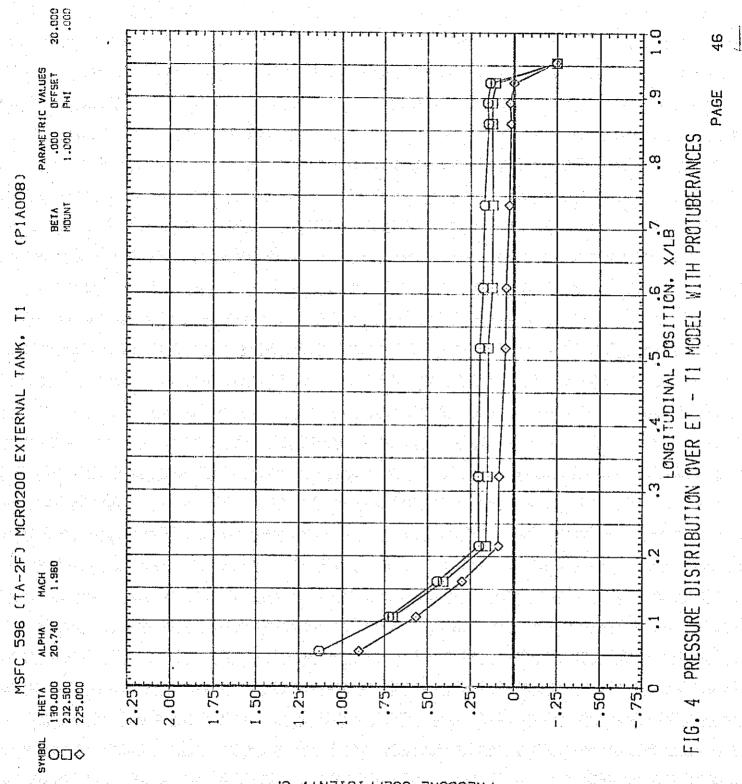




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PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES FIG. 4

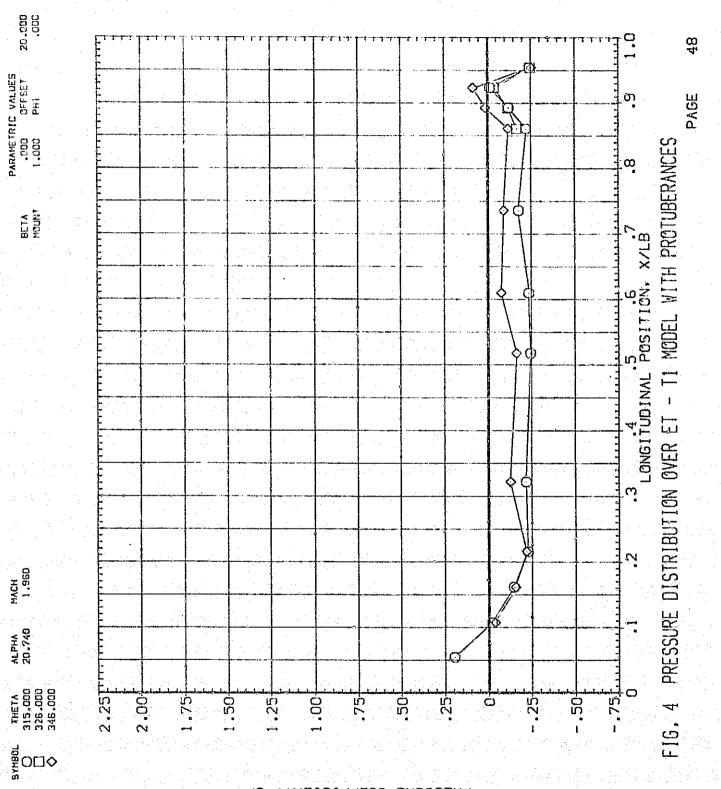


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

PRESSURE COEFFICIENT. CF

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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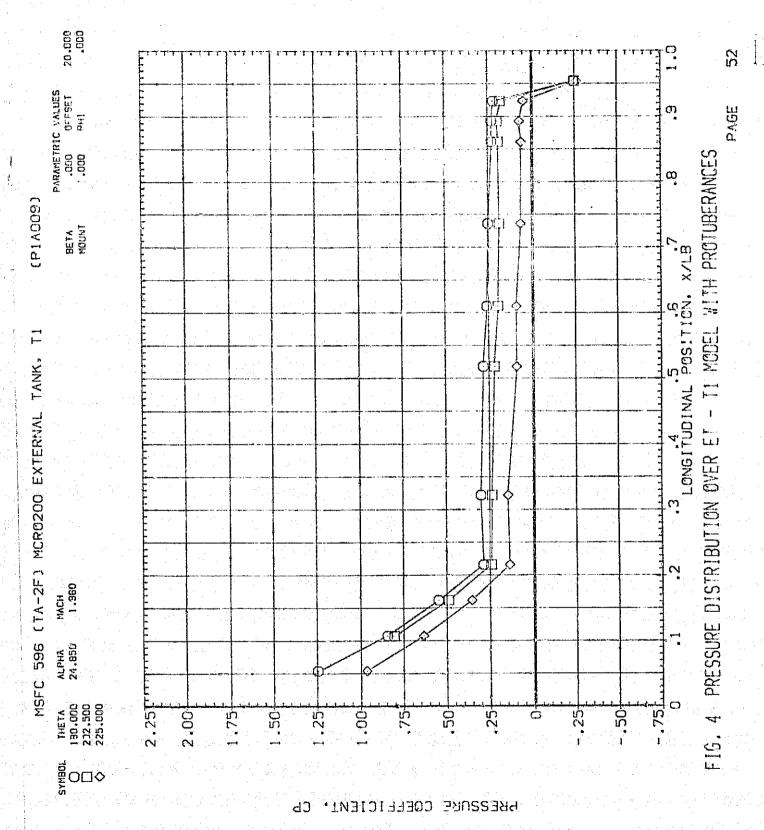
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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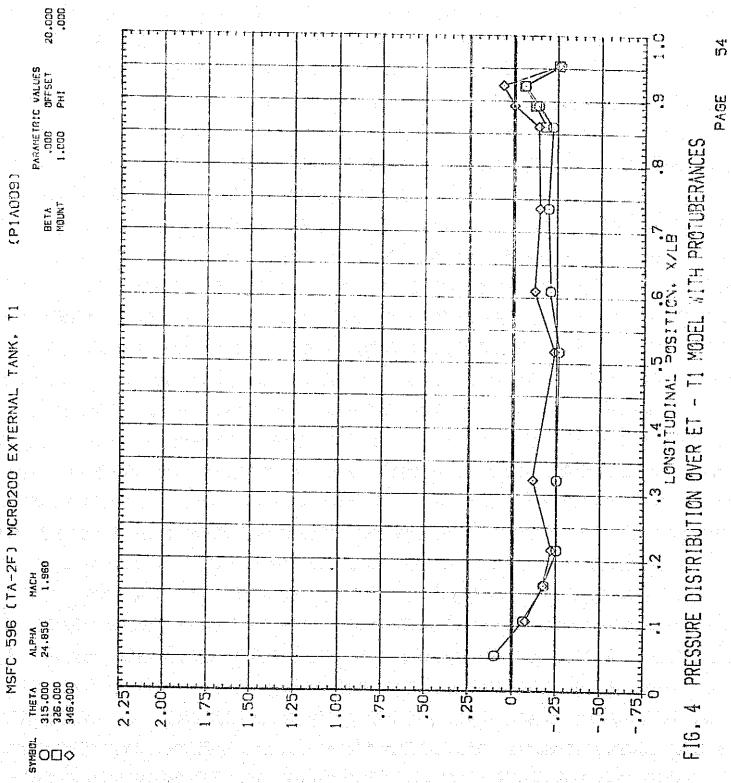
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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

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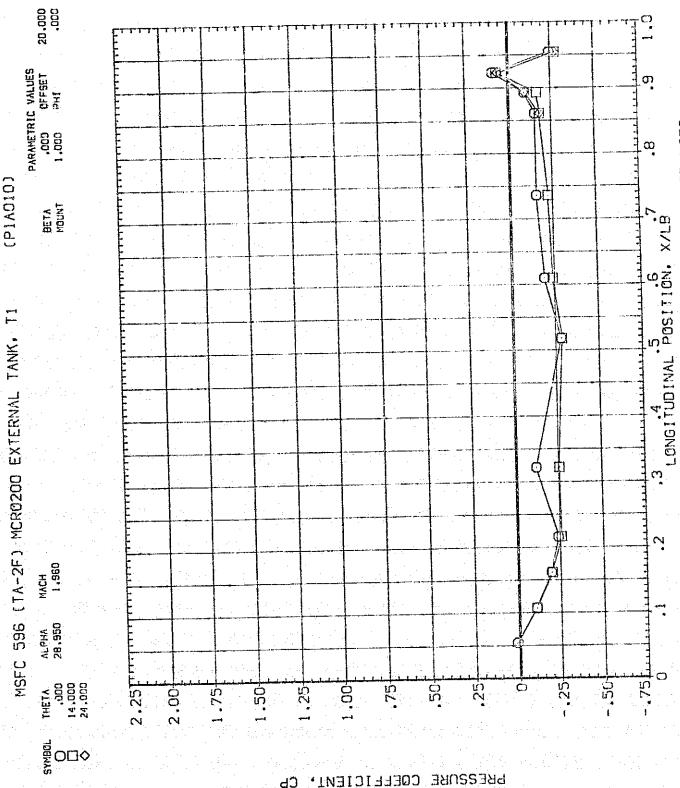
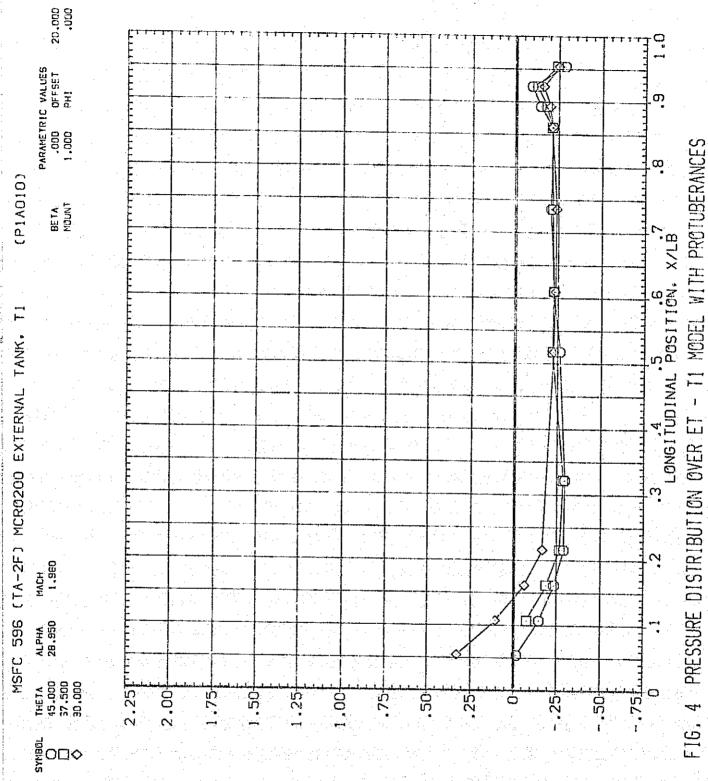


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



PRESSURE COEFFICIENT, CP

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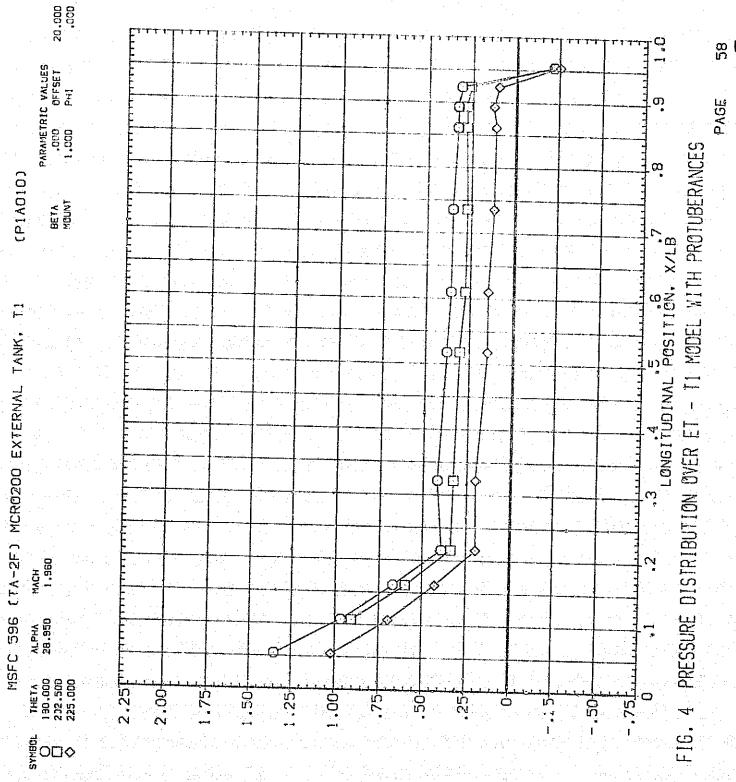
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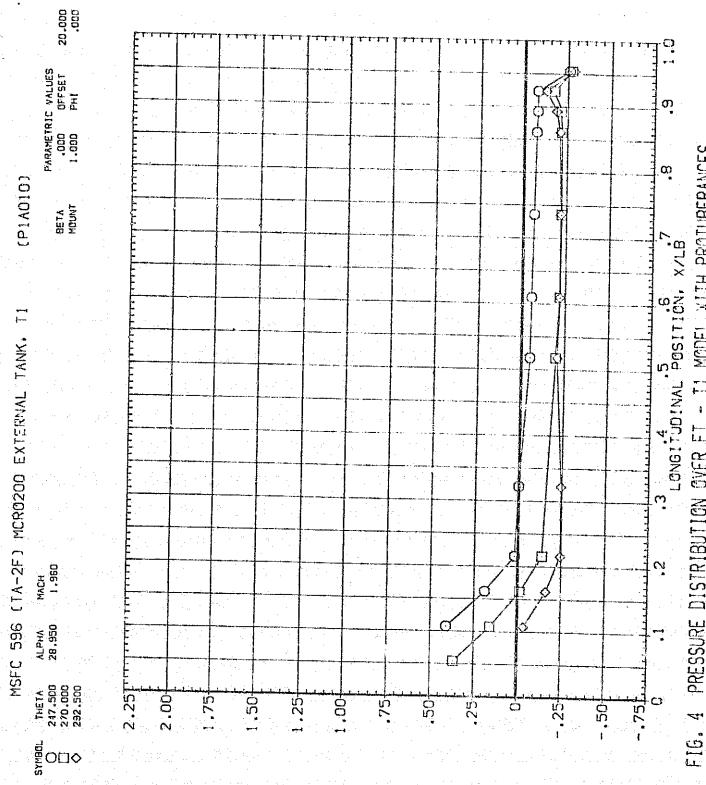
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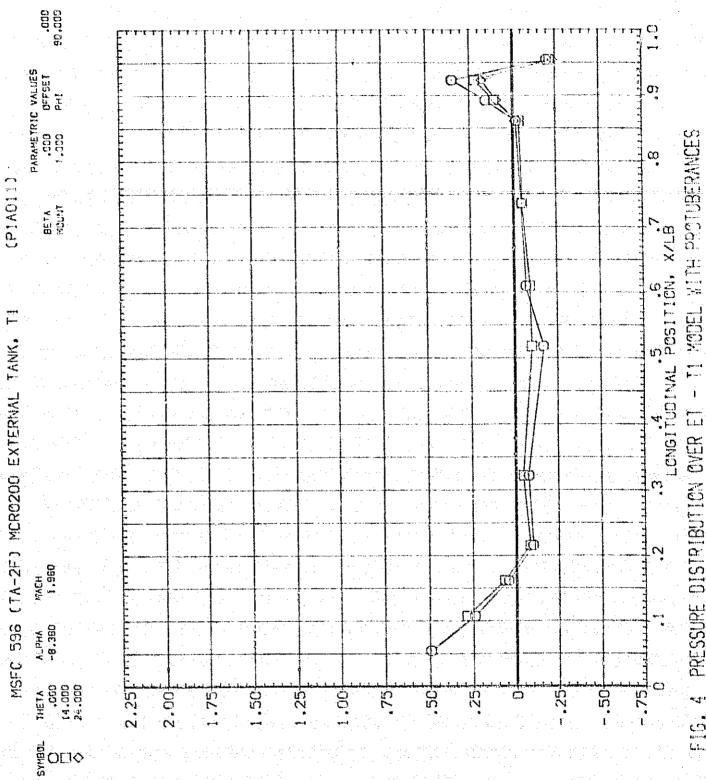
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BRESSURE COEFFICIENT

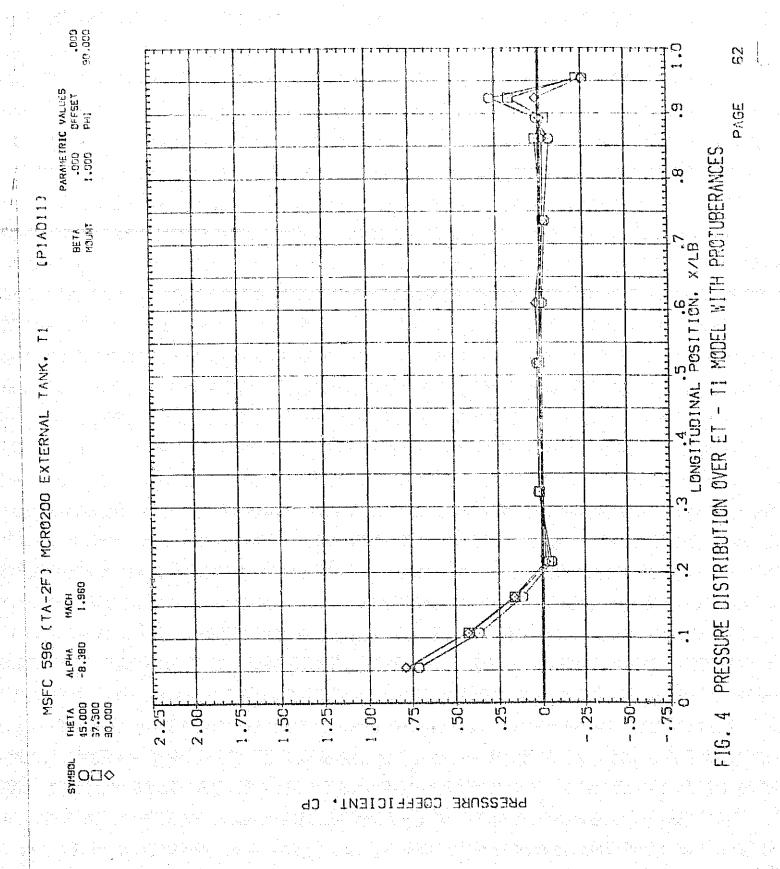
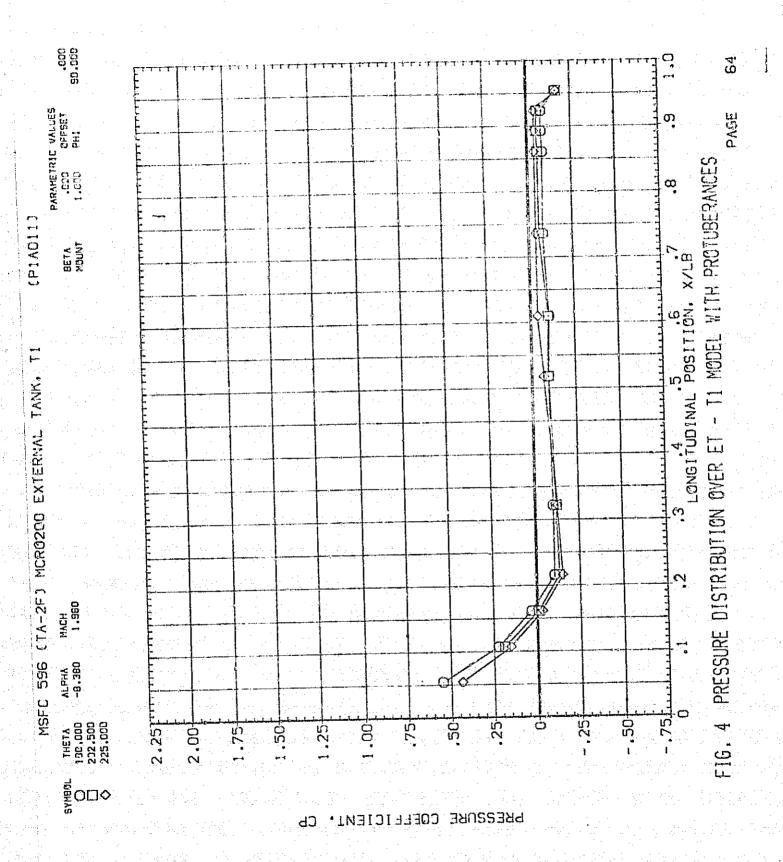
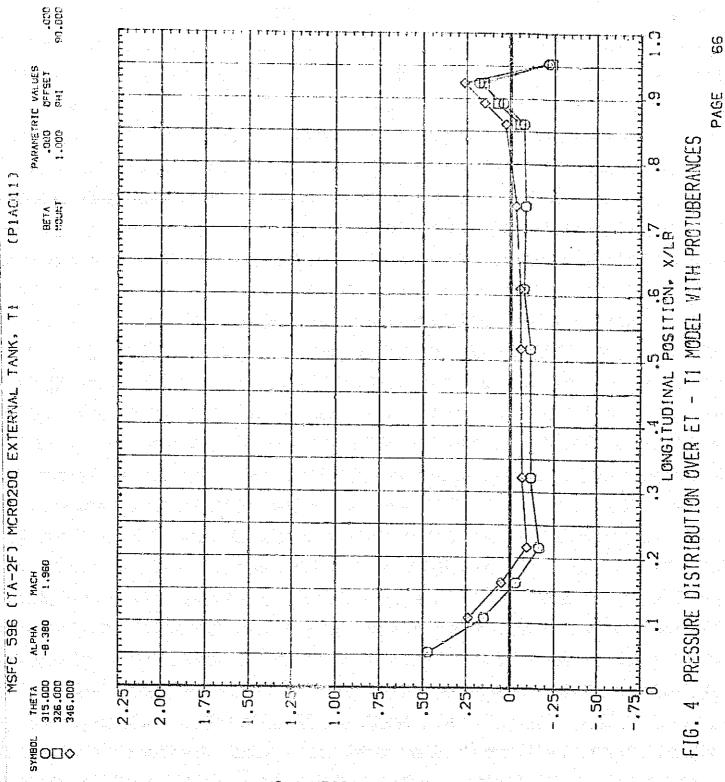


FIG. 4 PRESSURE DISTRIBUTION GVER ET - TI MODEL WITH PROTUBERANCES



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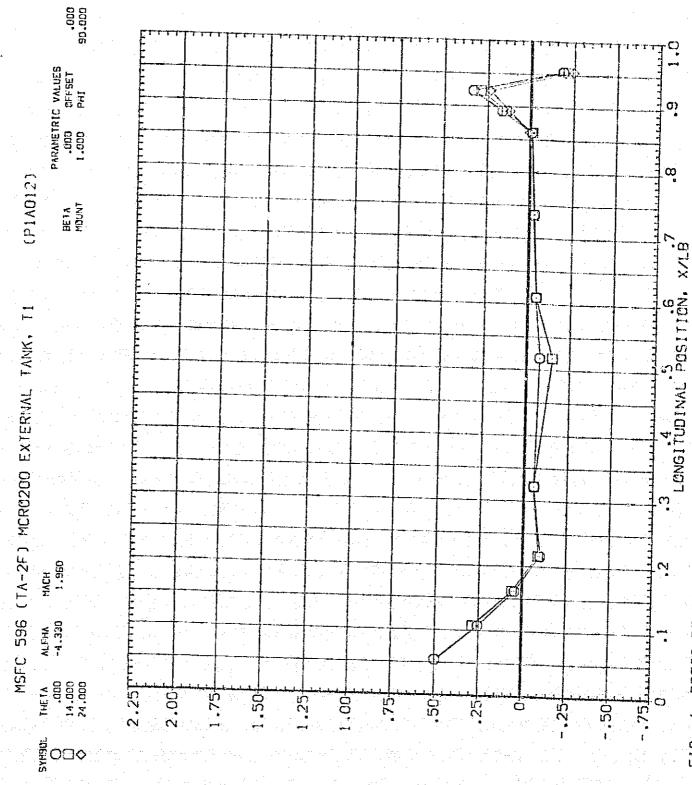
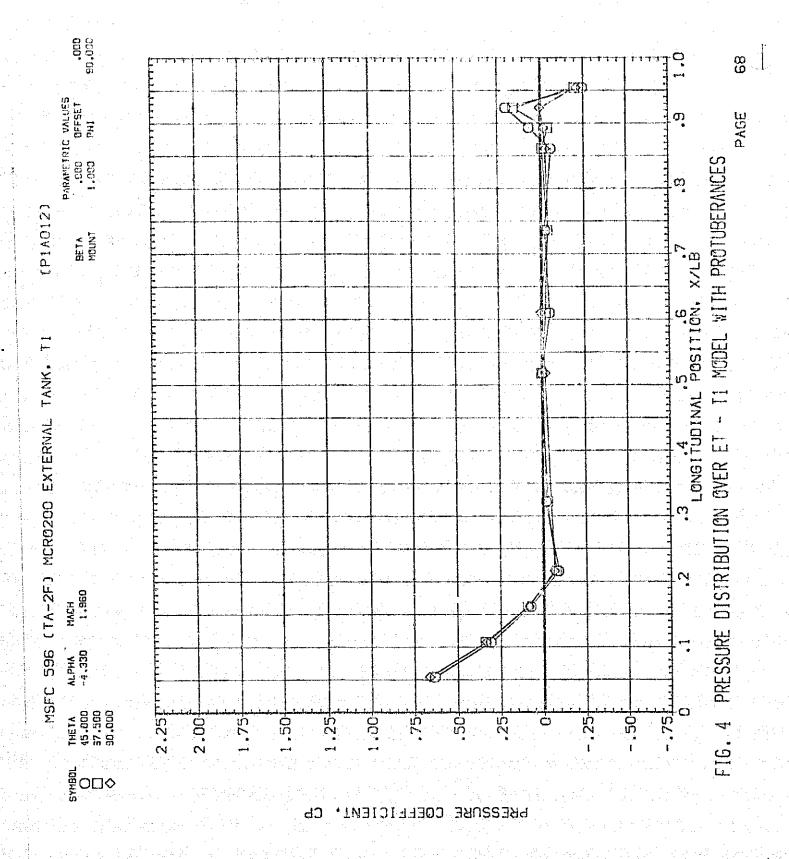
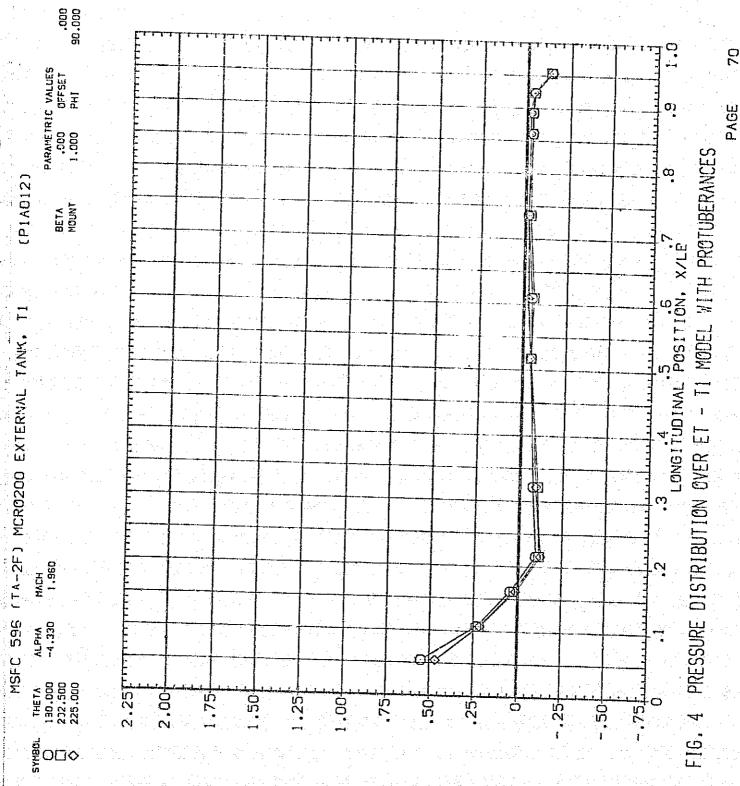


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

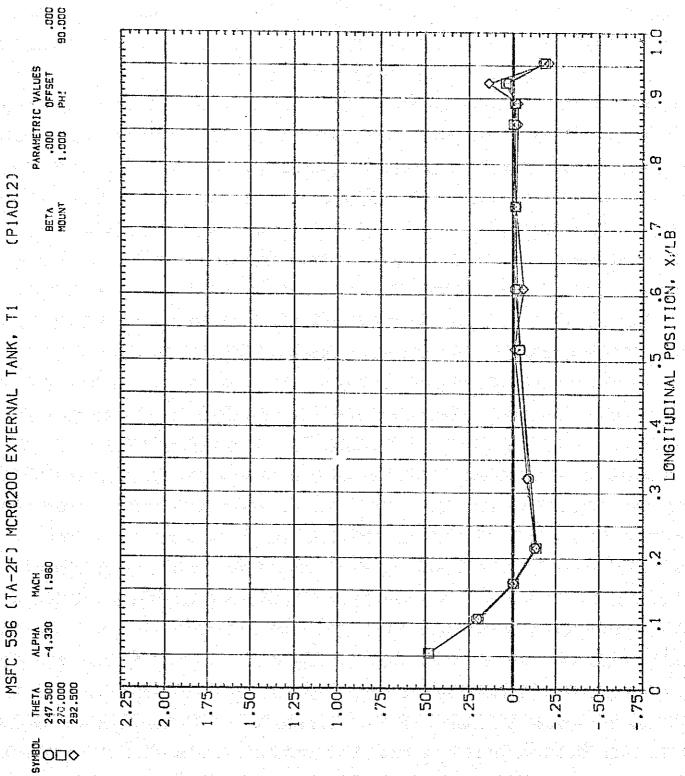
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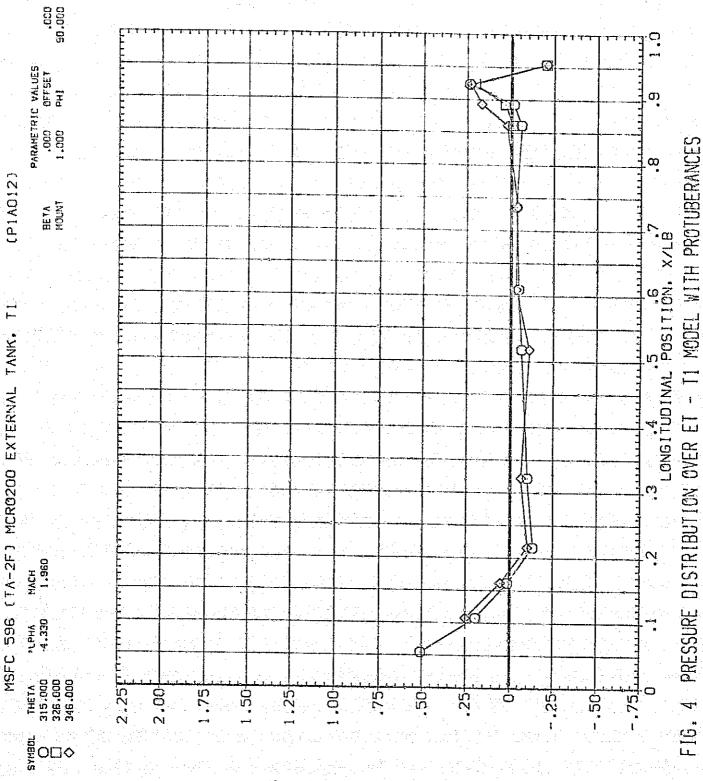


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

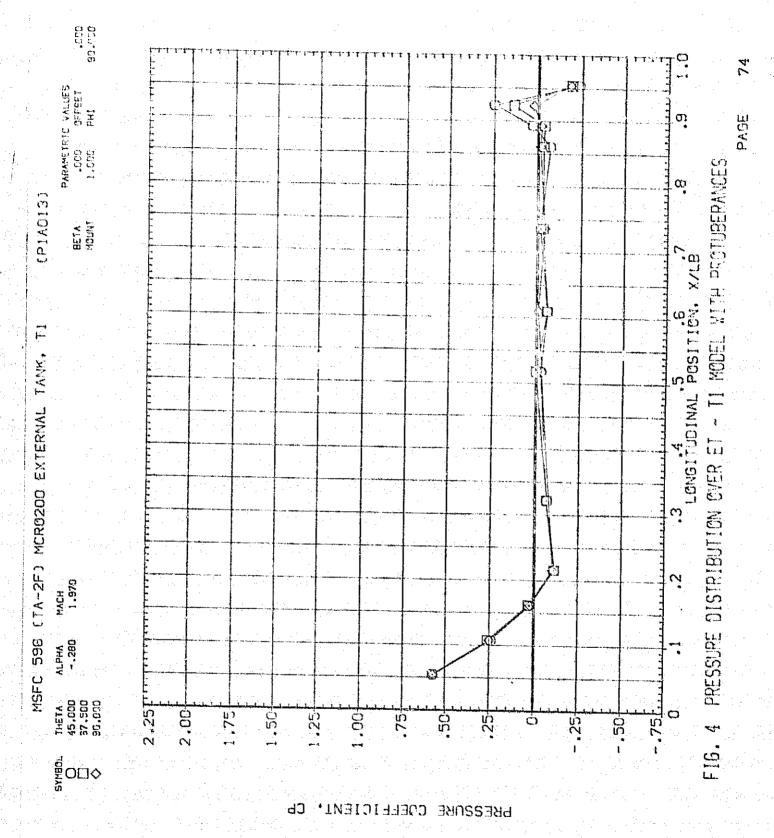


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PRESSURE COEFFICIENT, CP

PRESSURE COEFFICIENT,

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



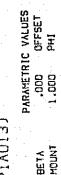
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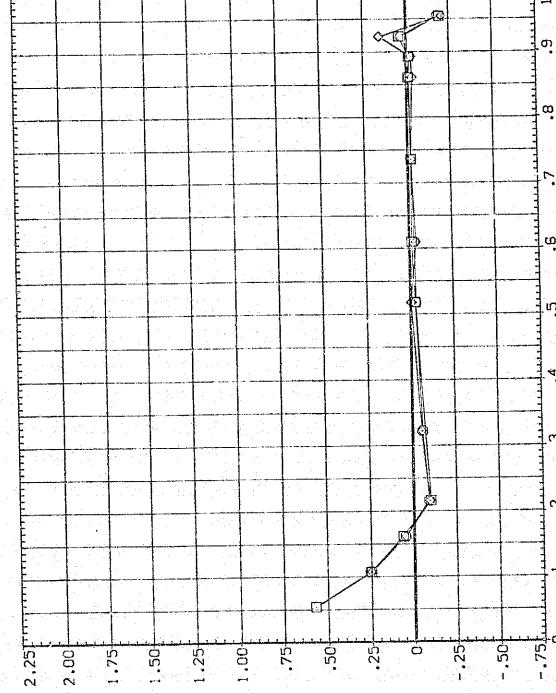
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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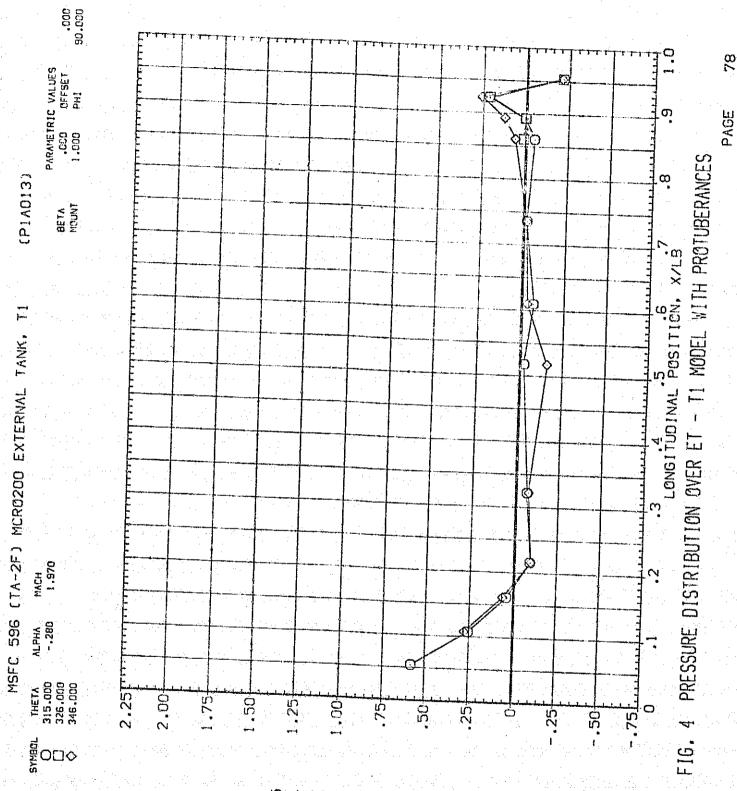


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES LGNGITUDINAL POSITION, X/LB

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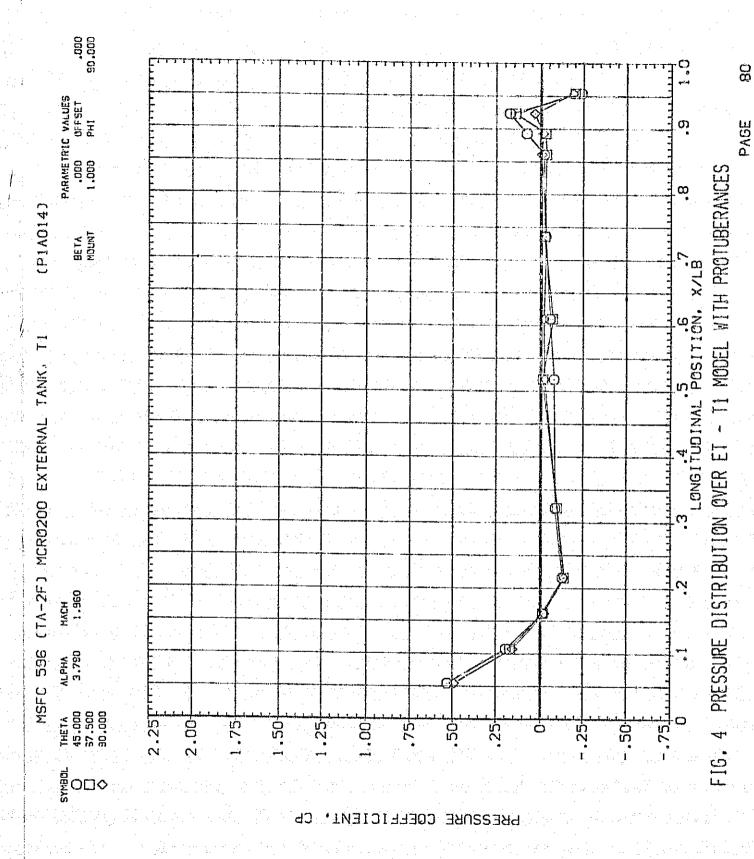


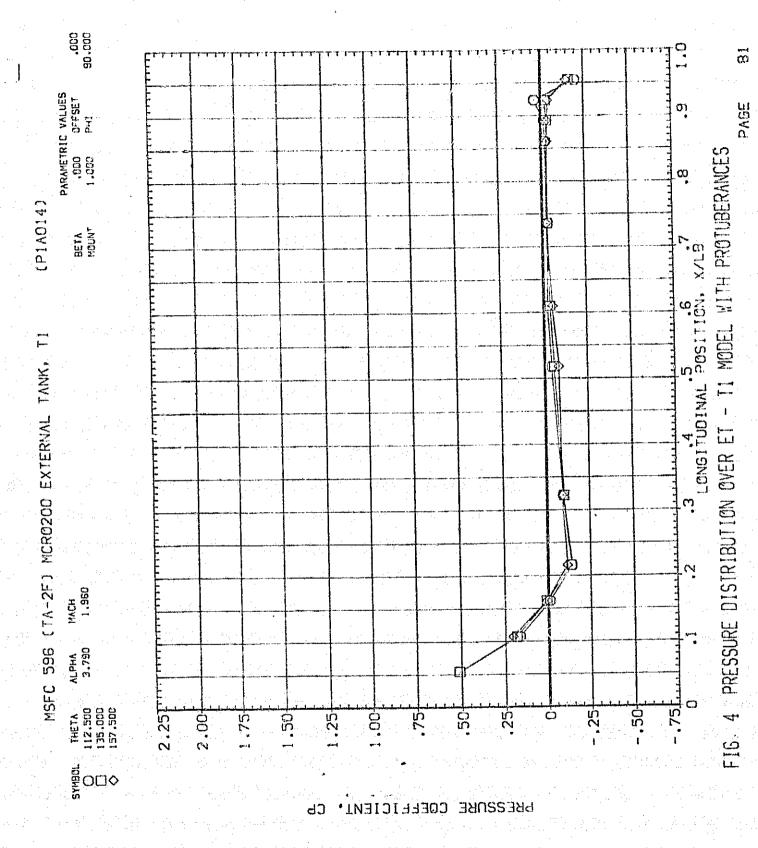
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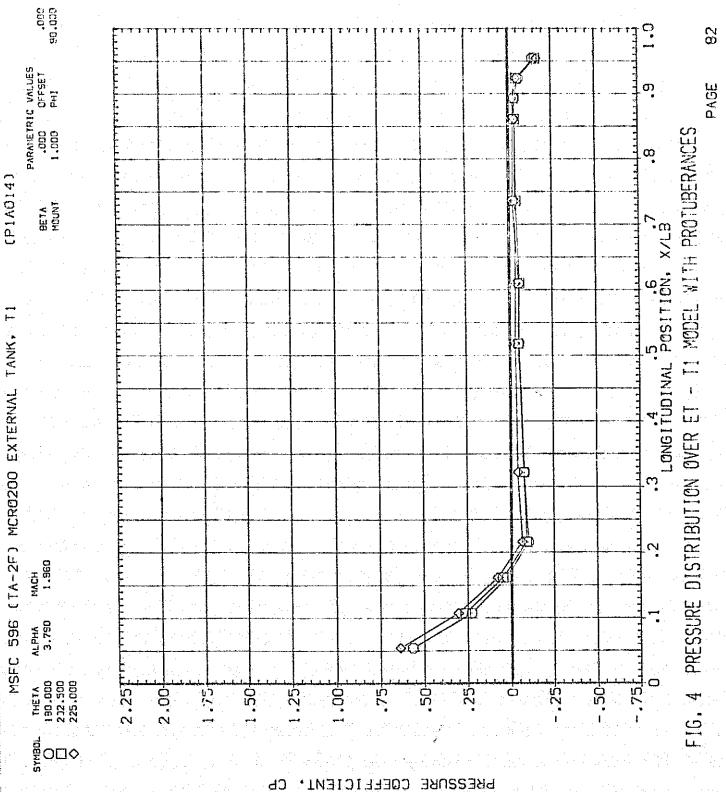
000.06 PARAMETRIC VALUES .000 OFFSET 1.000 PHI 0 ດ FIG. 4 PRESSURE DISTRIBUTION OVER ET - 71 MODEL WITH PROTUBERANCES ω (P1A014) BETA LONGITUDINAL POSITION, X/LB MSEC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI 4 ø Ø1 MACH 1.960 3.795 THETA .000 14.000 24.000 2.00 75 1.25 1.50 00 75 -.75 50 25 -, 25 . 50 Ó

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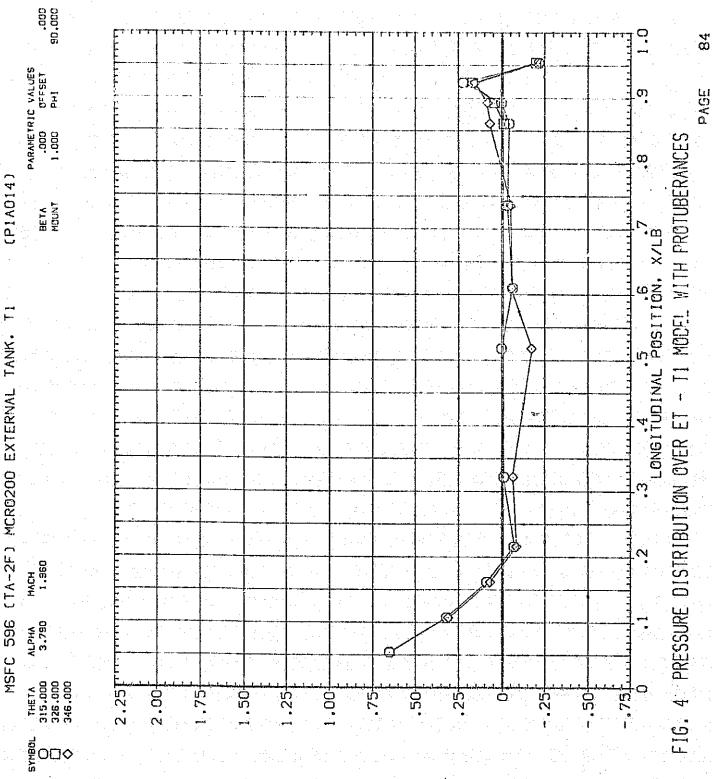




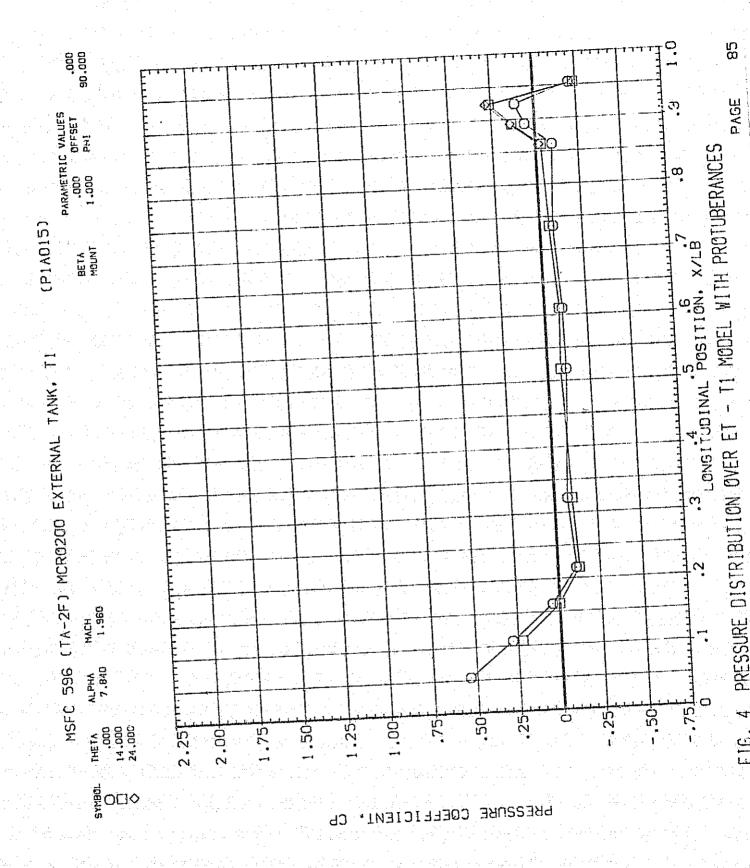


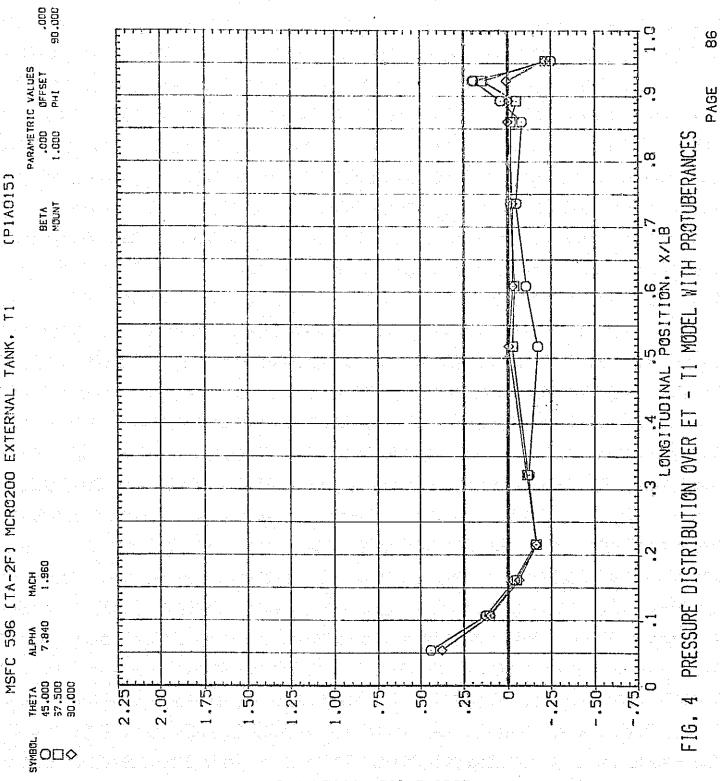
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PRESSURE DISTRIBUTION OVER ET - TI MONEL WITH PROTUBERANCES FIG. 4

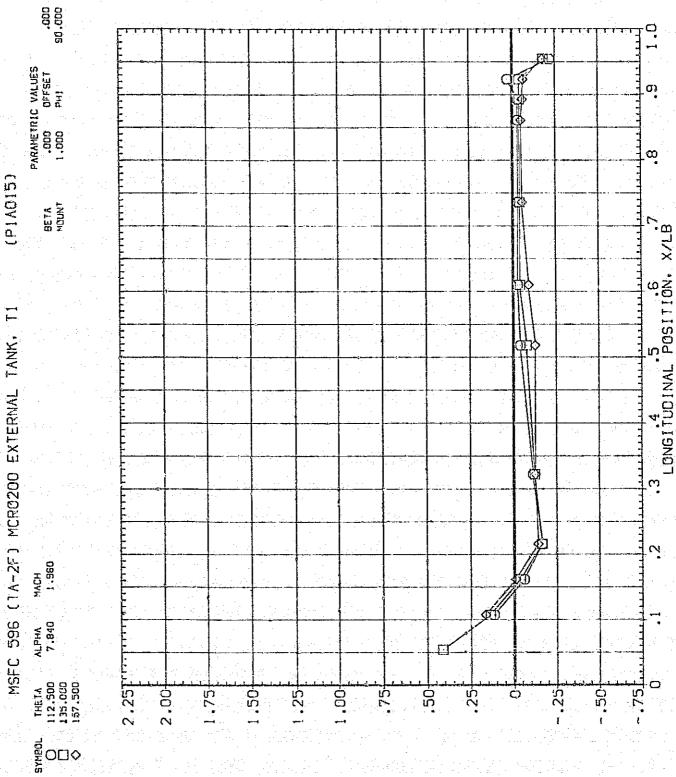


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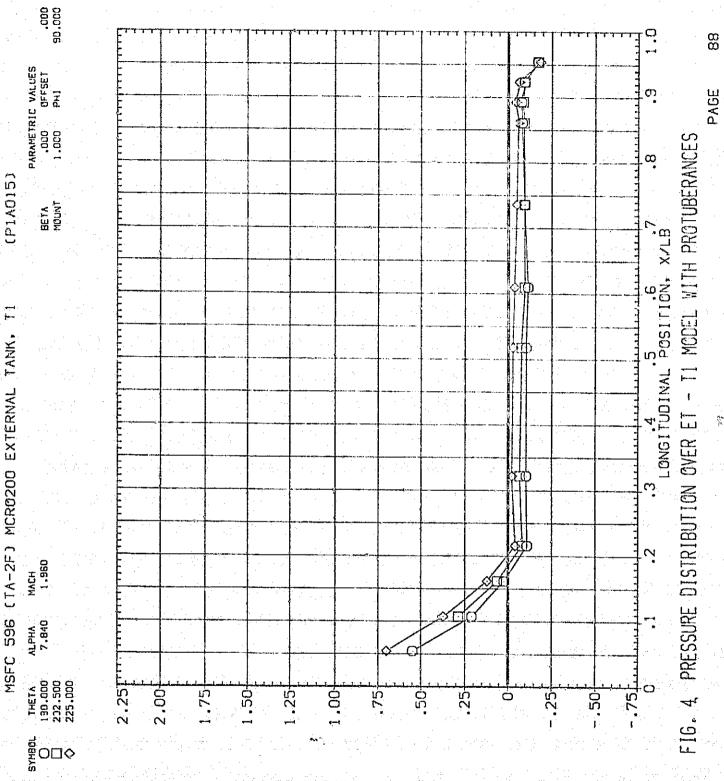
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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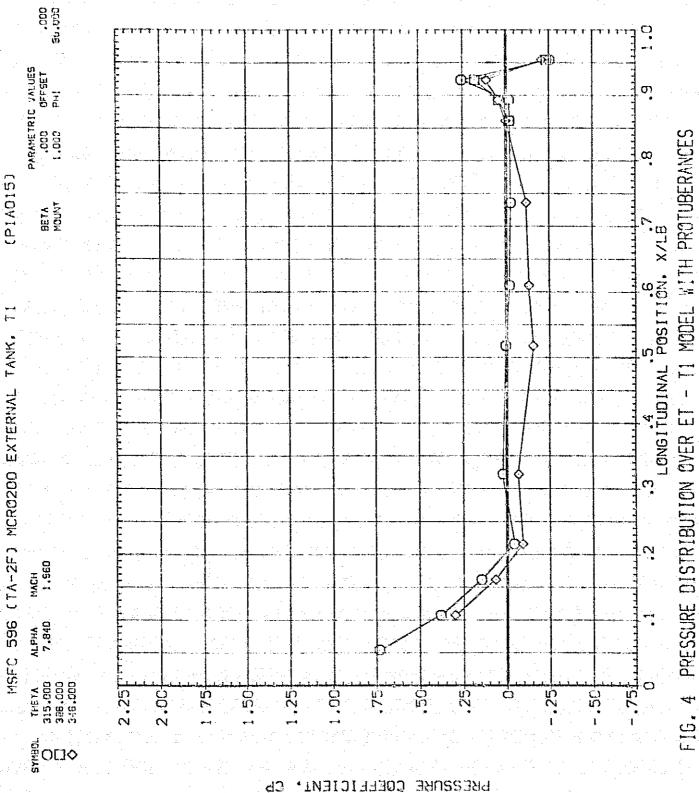
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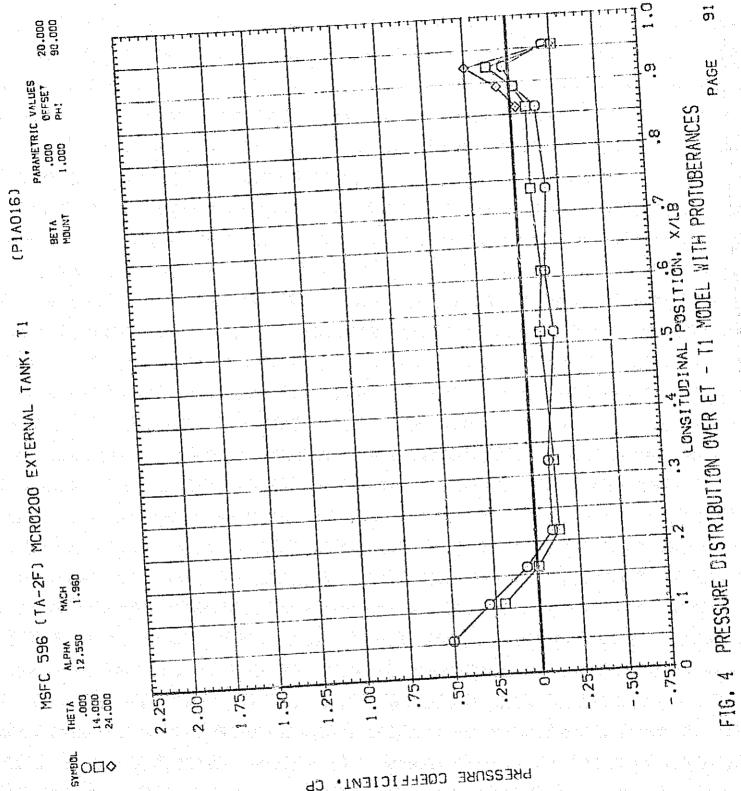
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

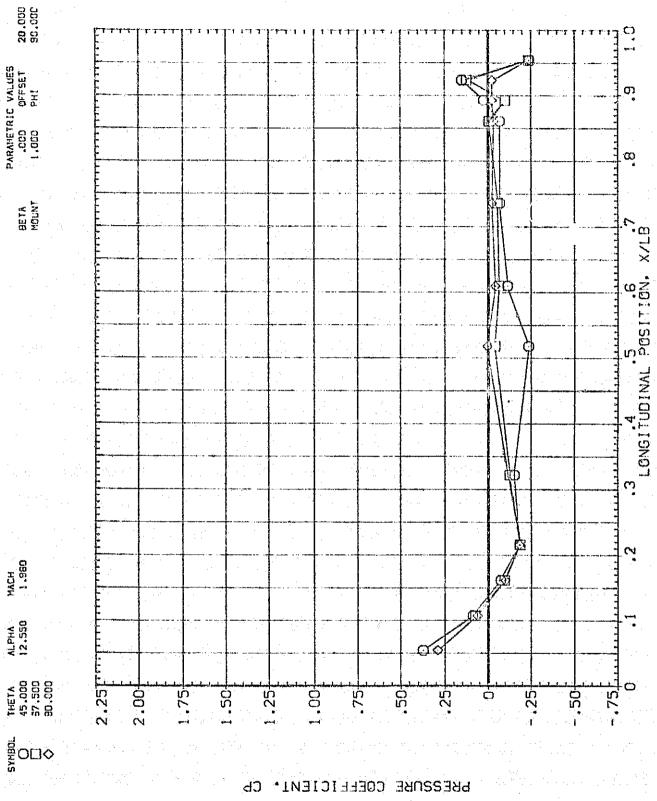
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(TA-2F) MCR0200 EXTERNAL TANK.

MSFC 596

- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

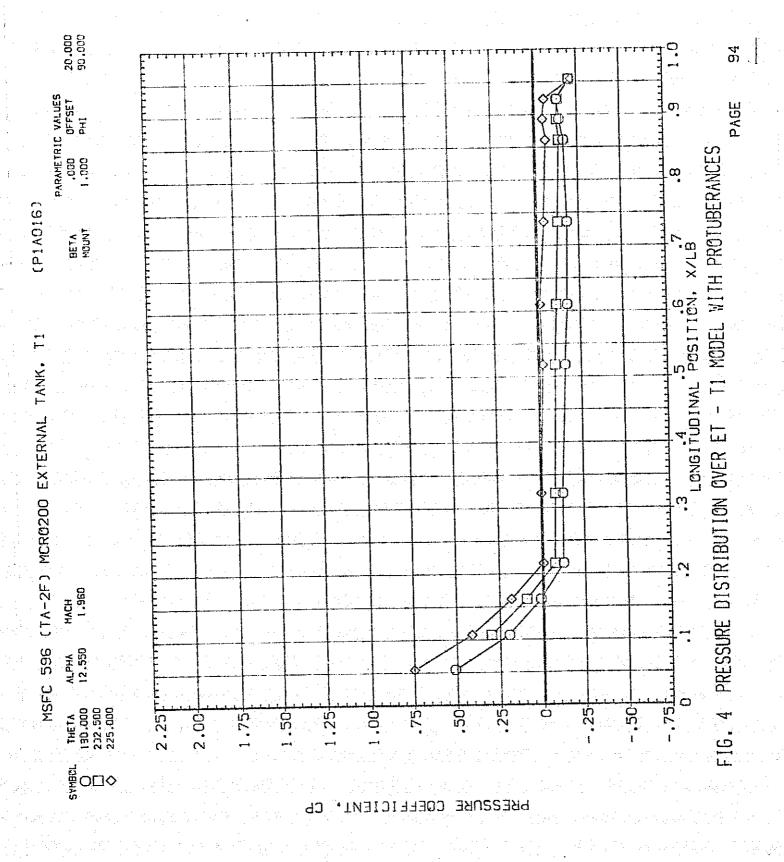
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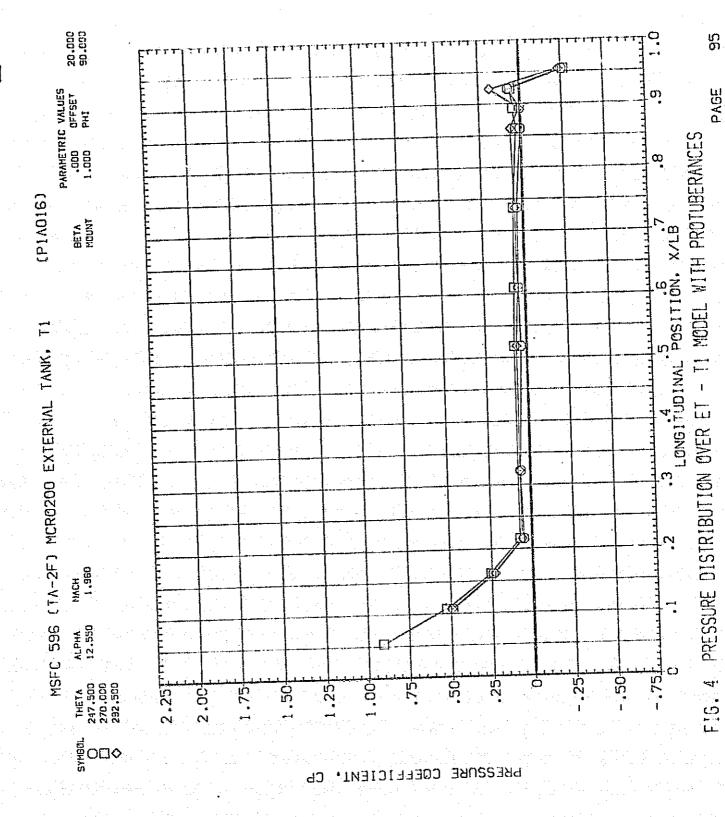
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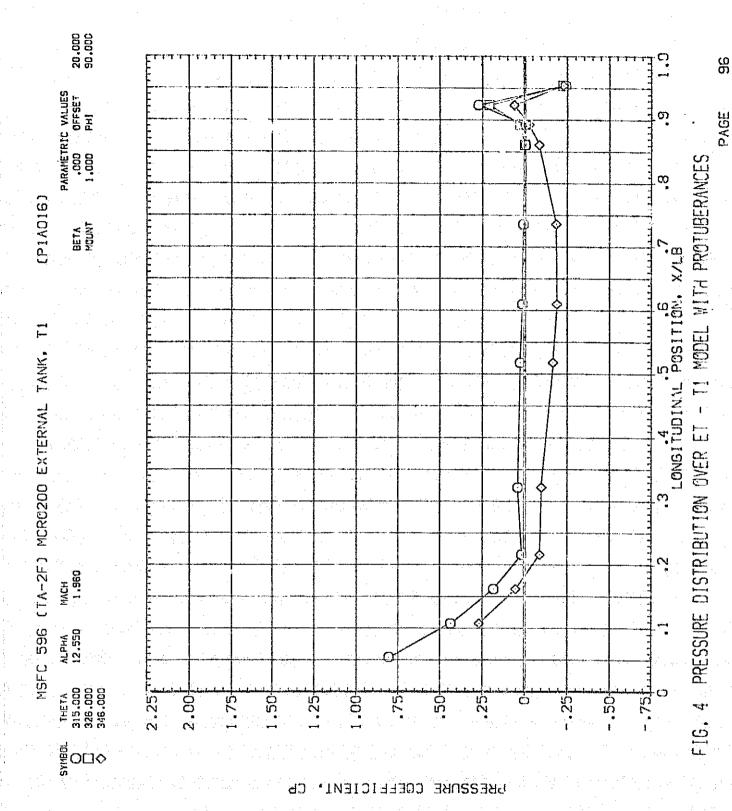
MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK,

FIG. 4 PRESSURE DISTRIBUTION OVER ET

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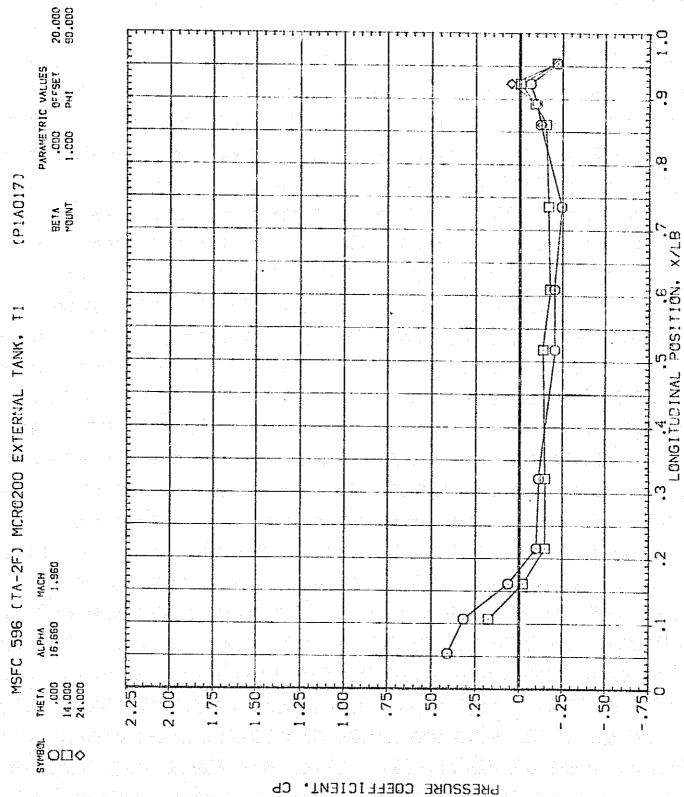
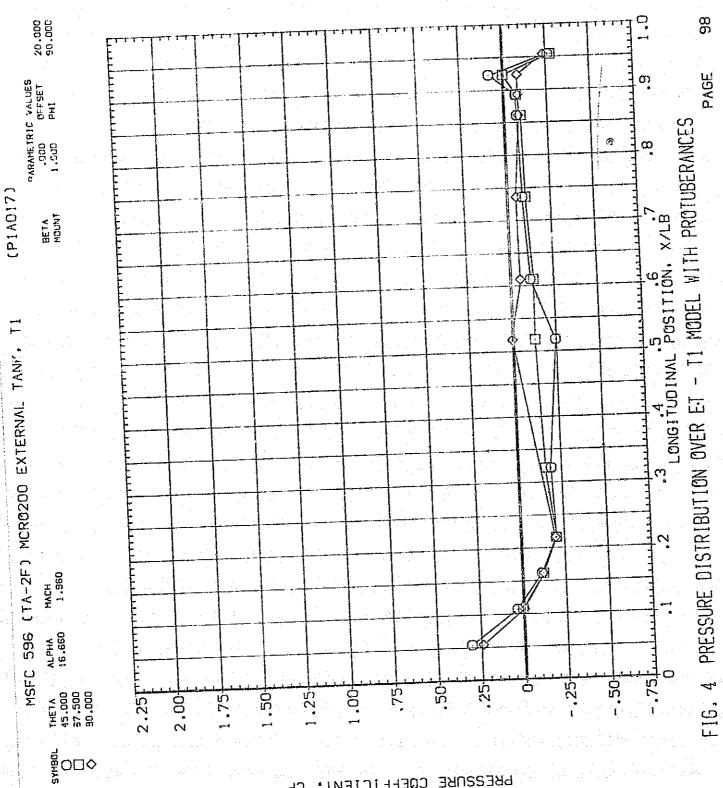
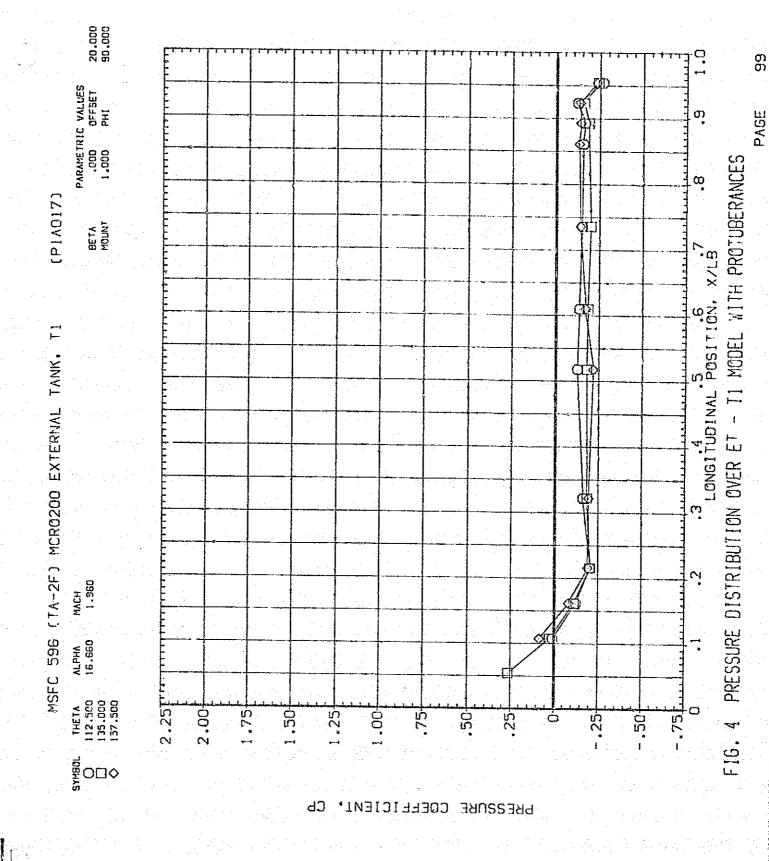


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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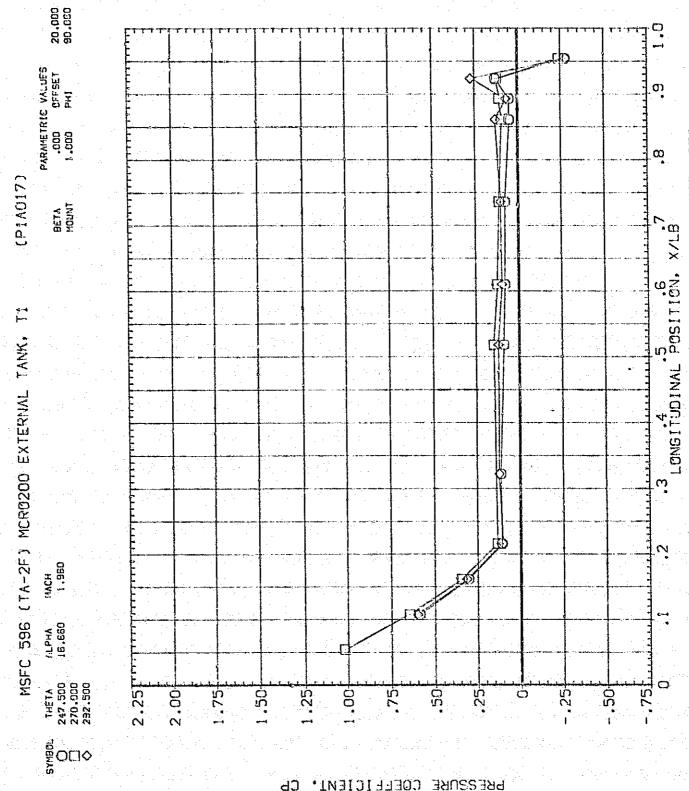
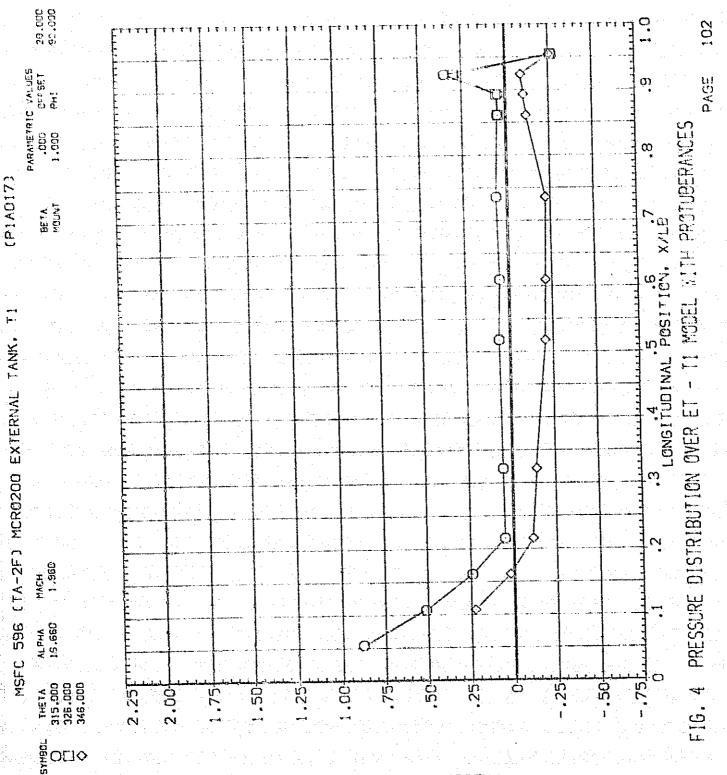
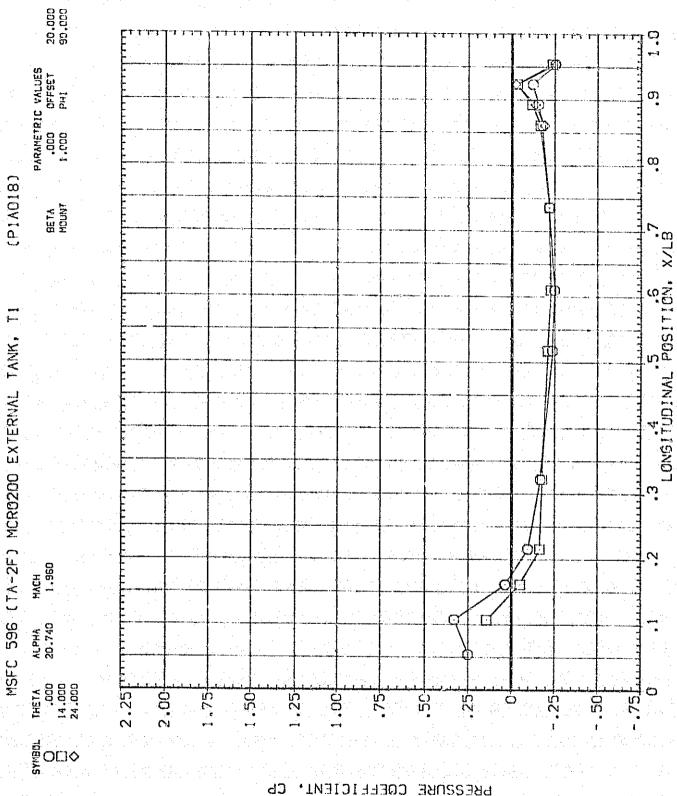


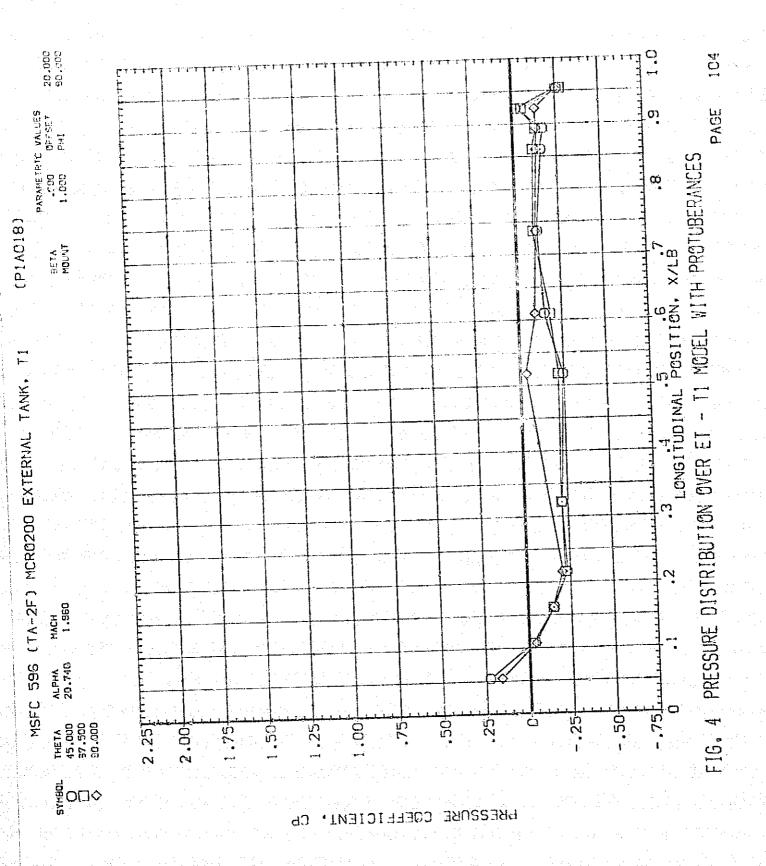
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES FIG. 4



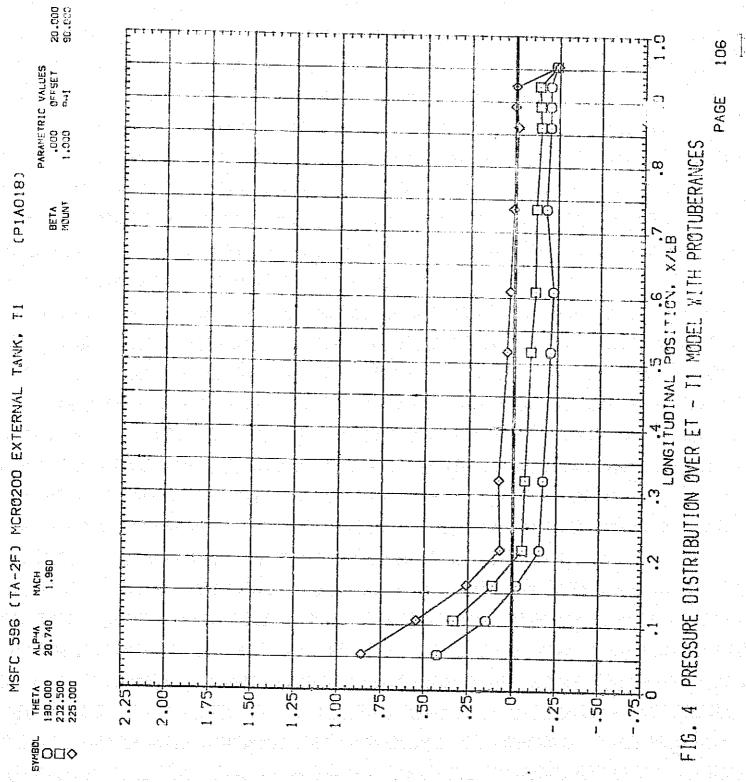
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WITH PROTUBERANCES

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PRESSURE COEFFICIENT, CP

- TI MODEL WITH PROTUBERANCES LONGITUDINAL POSITION, X/LB PRESSURE DISTRIBUTION OVER ET F16, 4

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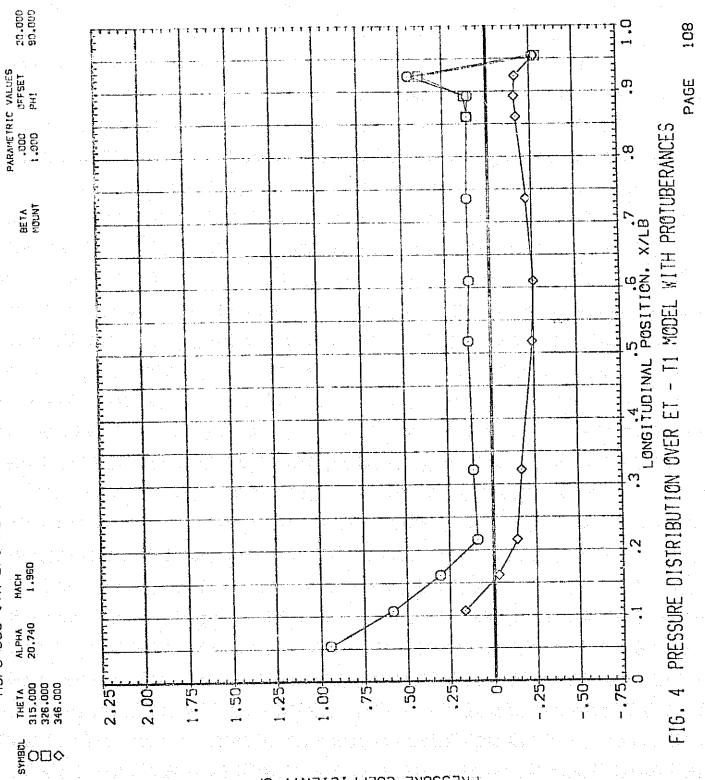
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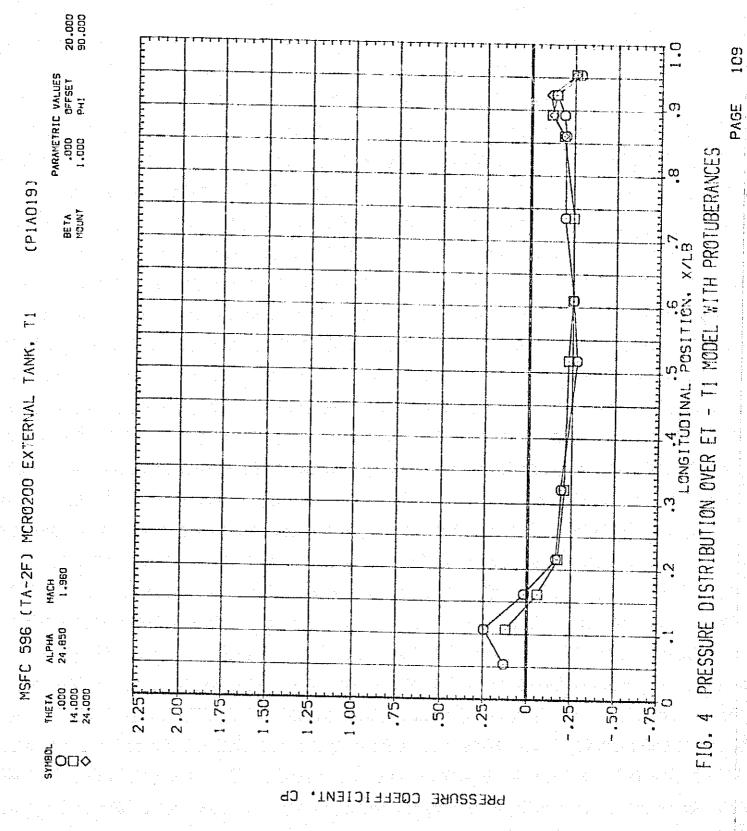
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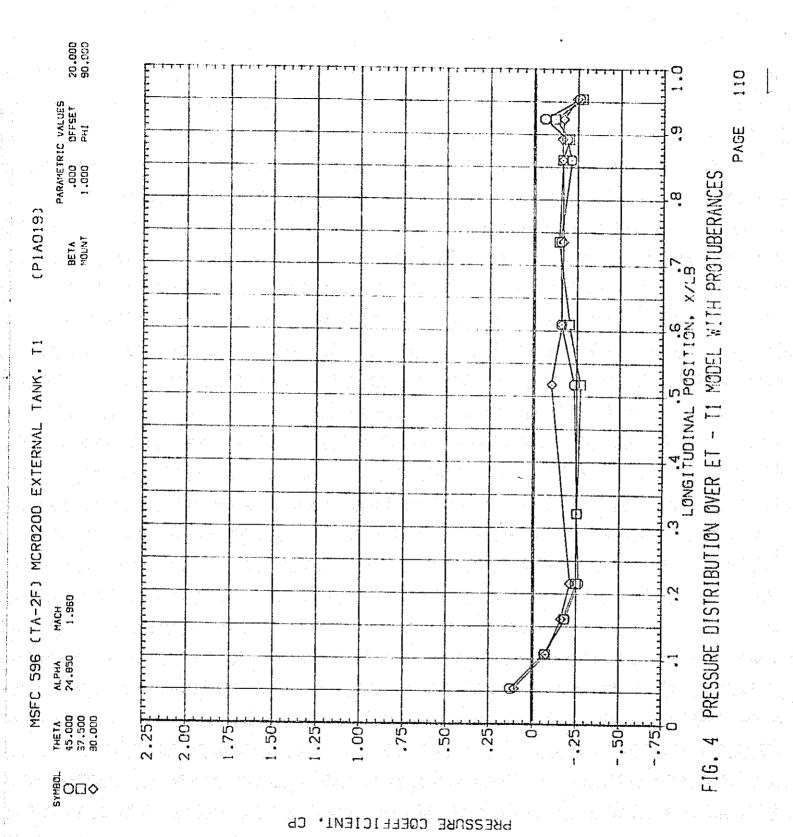
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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

PRESSURE COEFFICIENT, CP



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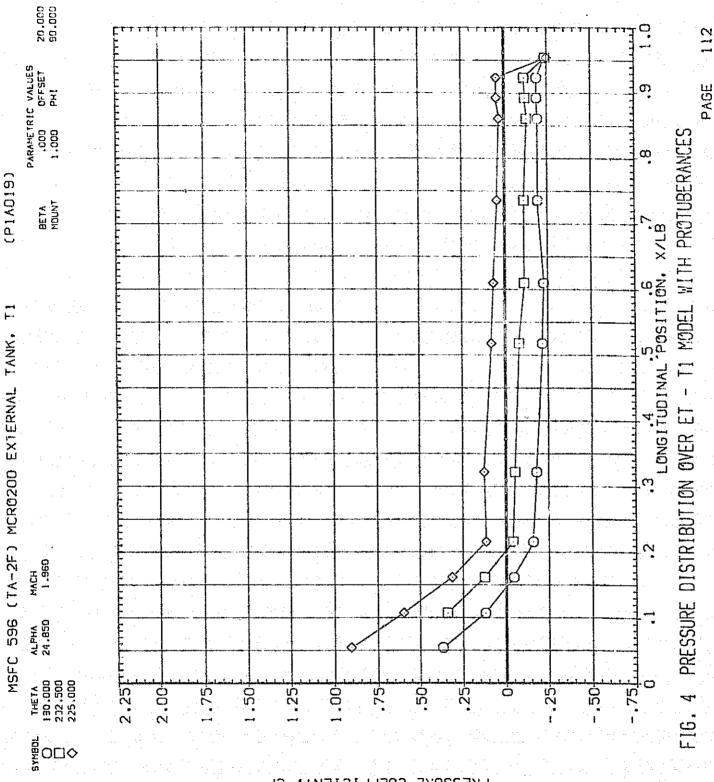
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TI MODEL WITH PROTUBERANCES LONGITUDINAL POSITION, X/LB FIG. 4 PRESSURE DISTRIBUTION OVER ET -

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PRESSURE COEFFICIENT, CP

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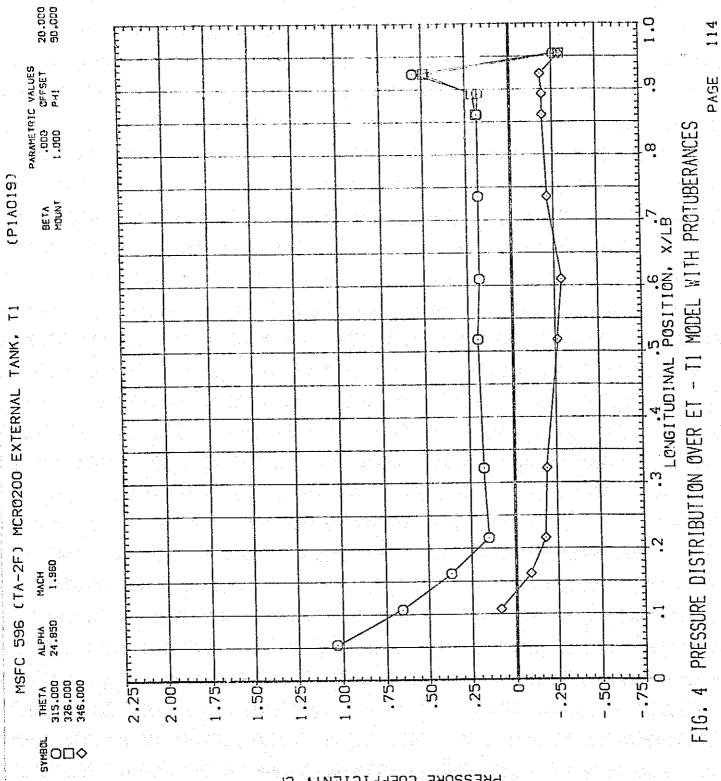
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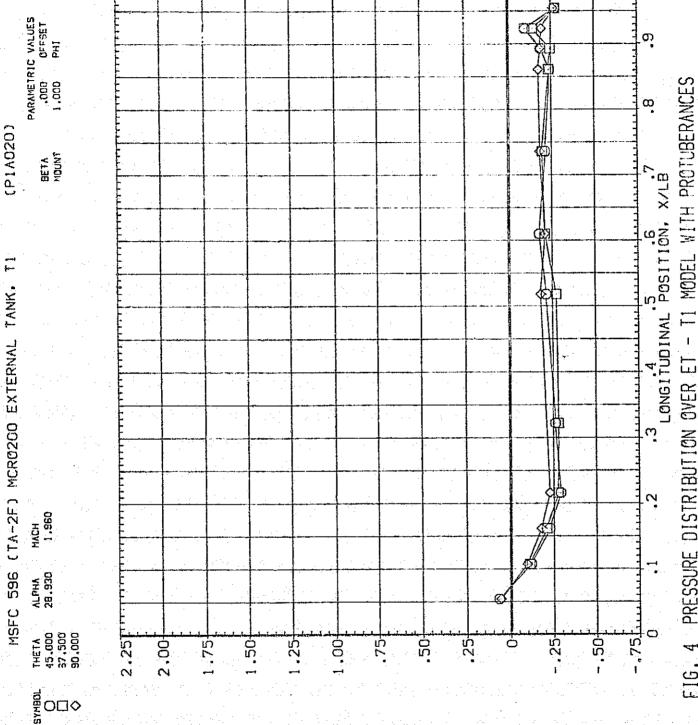
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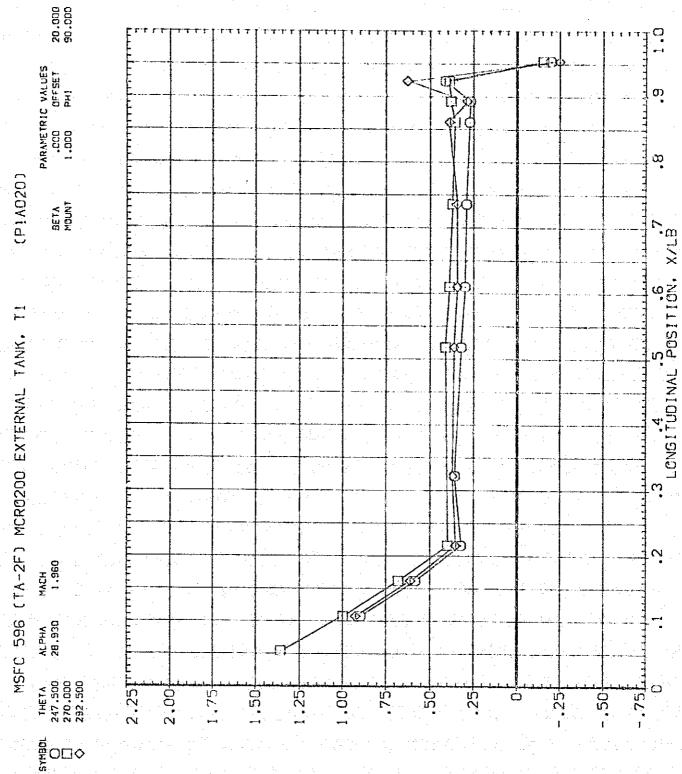
PRESSURE DISTRIBUTION OVER ET

FIG. 4

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

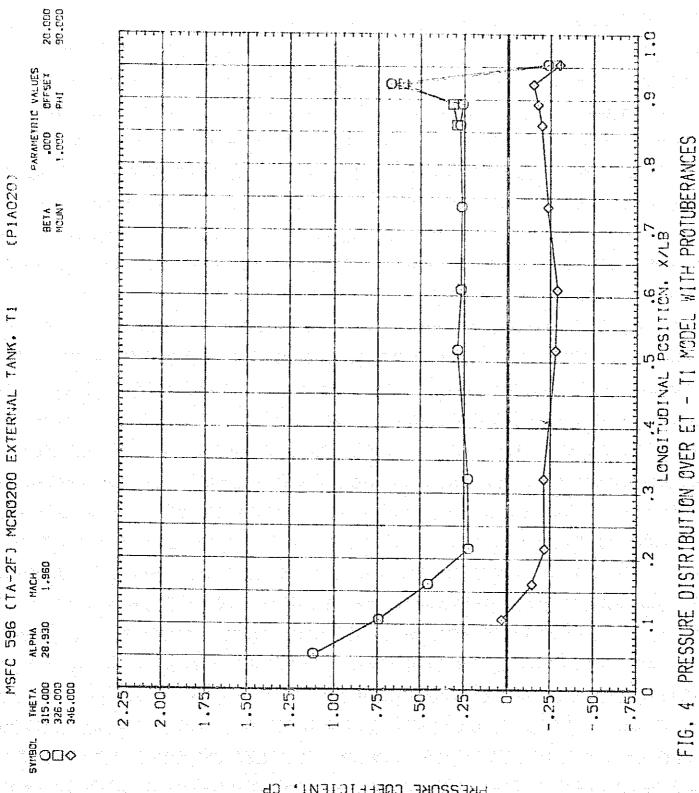
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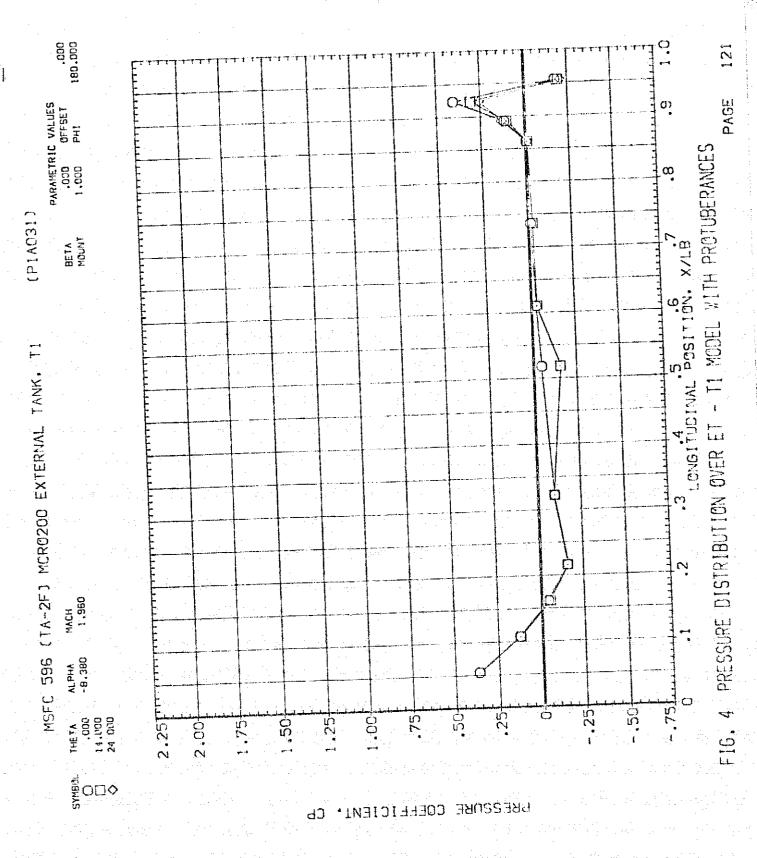


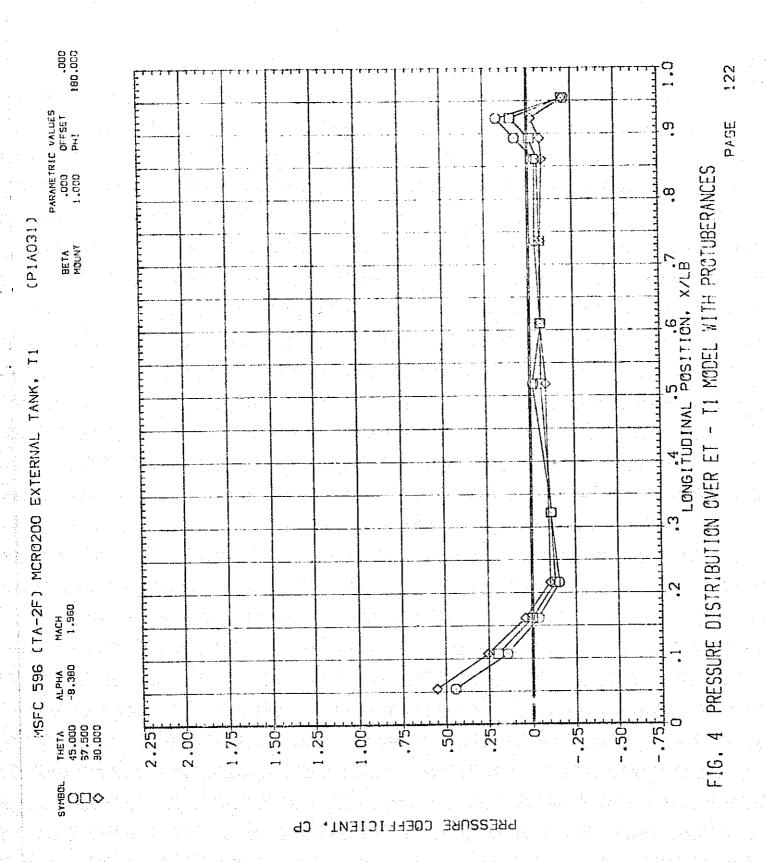
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WITH PROTUBERANCES -PRESSURE DISTRIBUTION OVER F16. 4

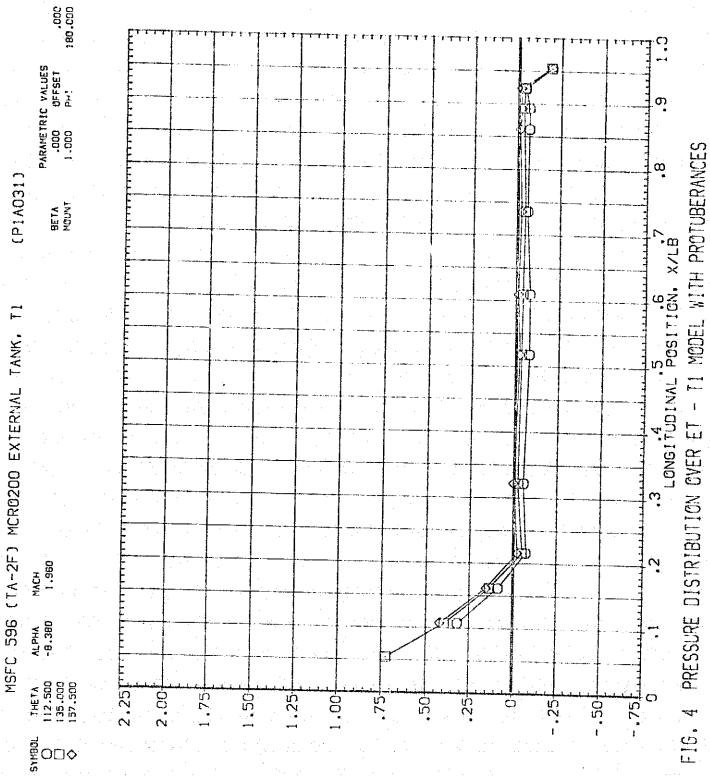


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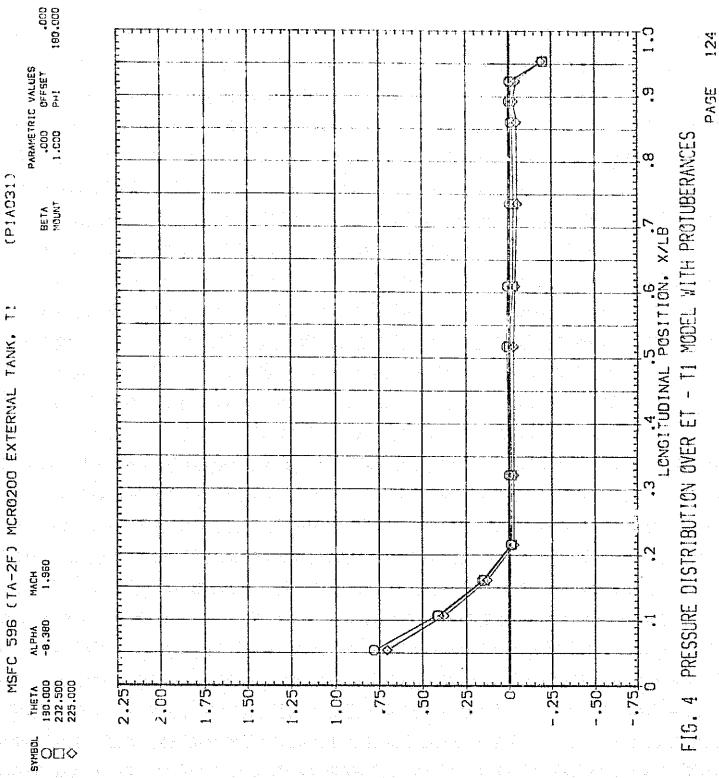




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PRESSURE COEFFICIENT, CP

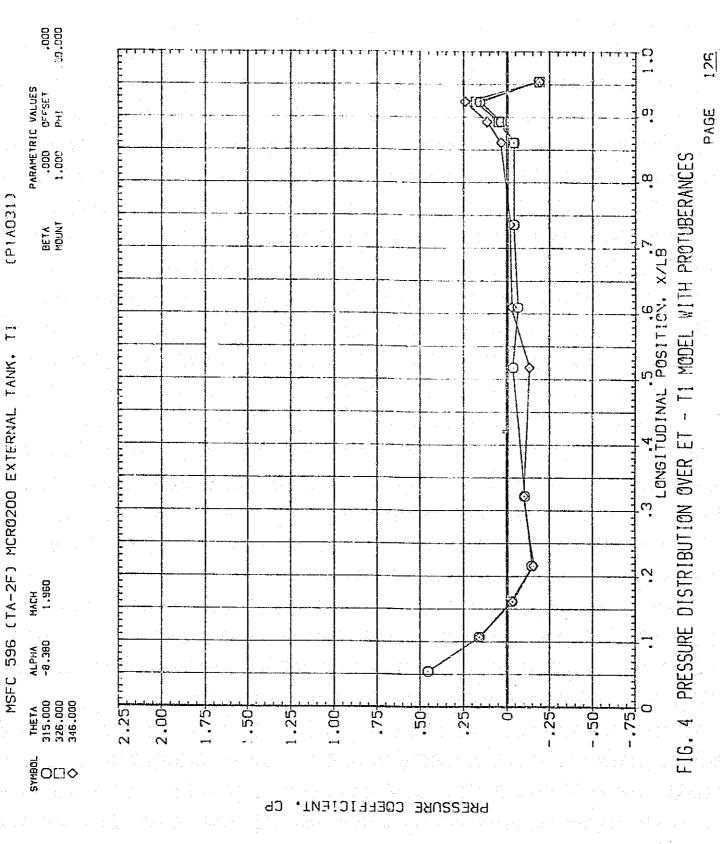
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PRESSURE DISTRIBUTION OVER

F16, 4

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PRESSURE COEFFICIENT.



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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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Lengitudinal Positien, X/LB

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COEFFICIENT PRESSURE

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(P1A032)

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МАСН 1.960

ALPHA -4,330

THETA .000 14.000 24.000

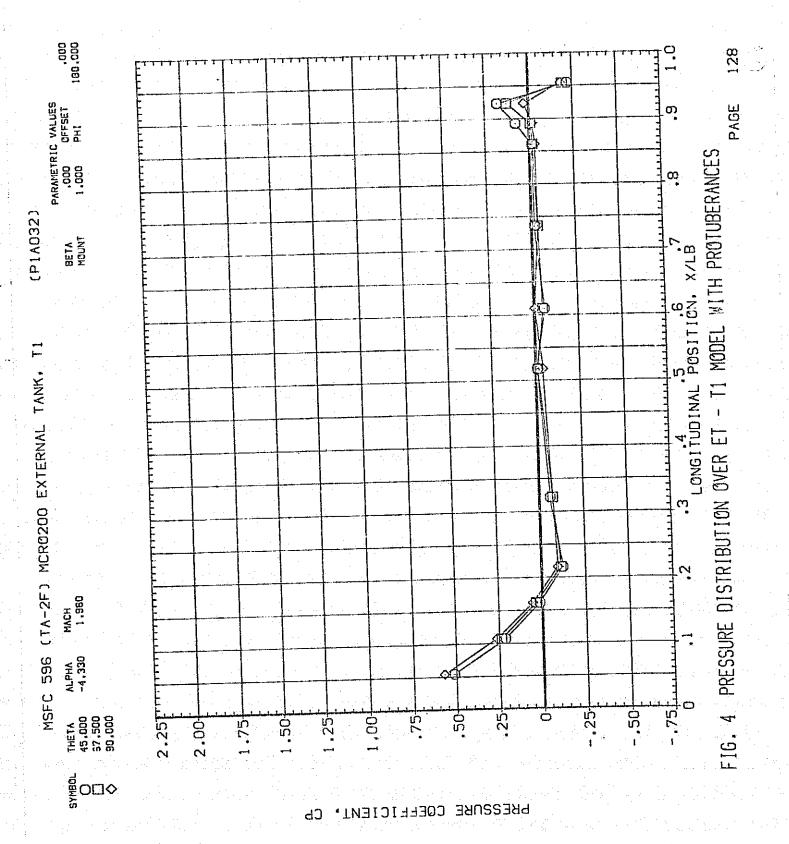
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- TI MODEL WITH PROTUBERANCES

FIG. 4 PRESSURE DISTRIBUTION OVER ET

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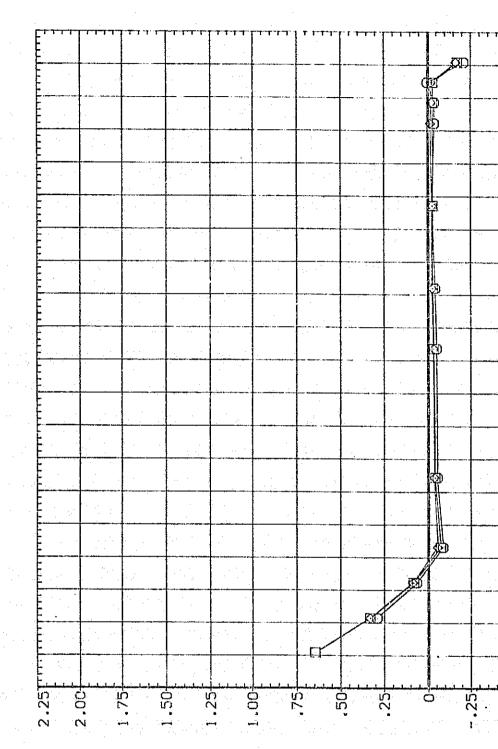
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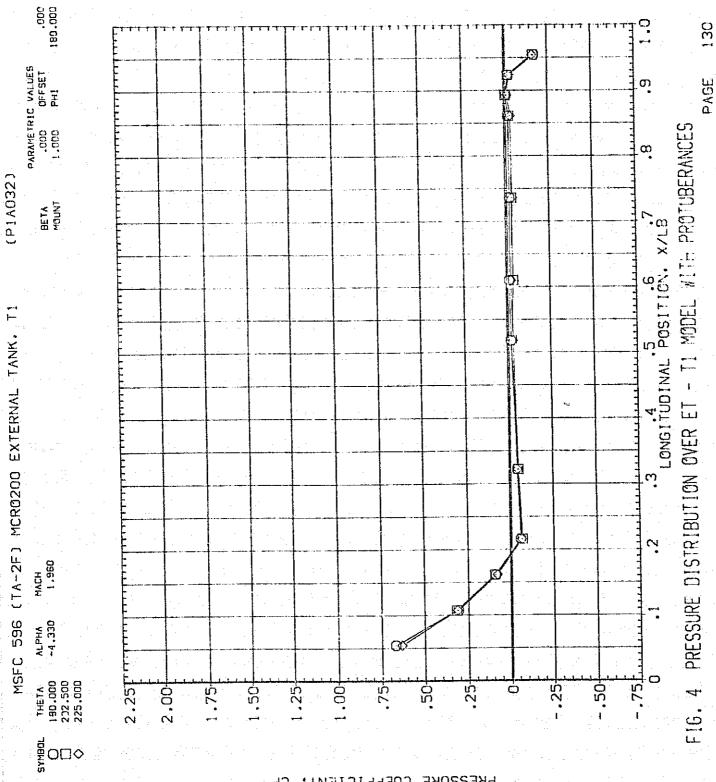
LONGITUDINAL POSITION. X/LB

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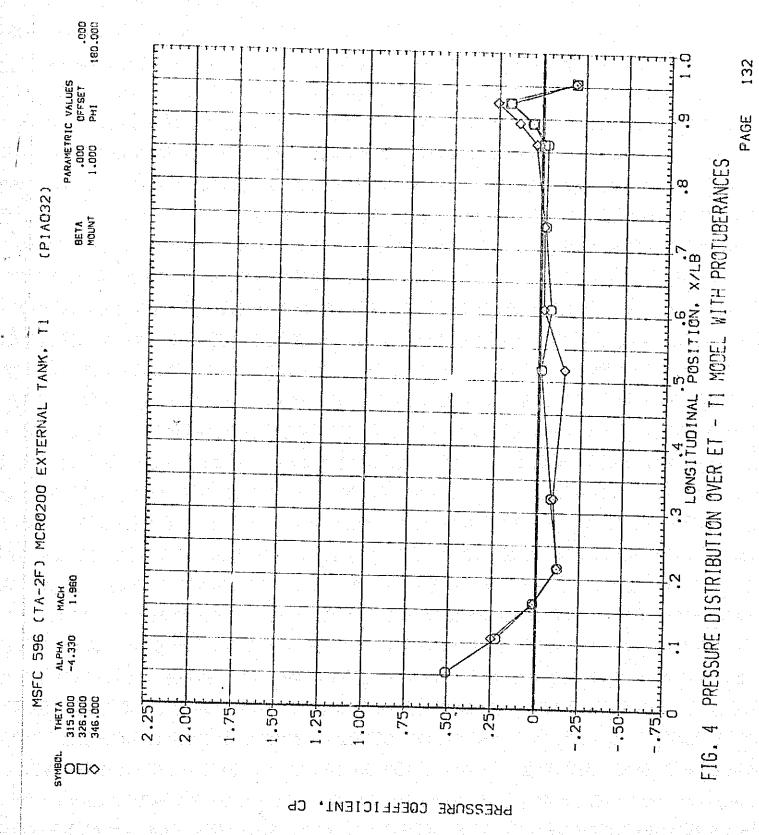
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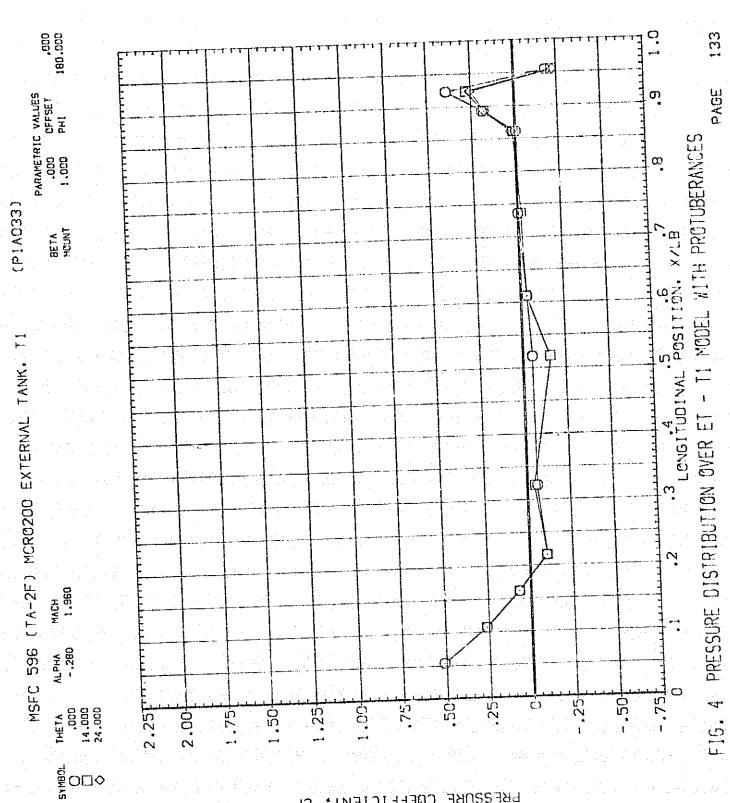


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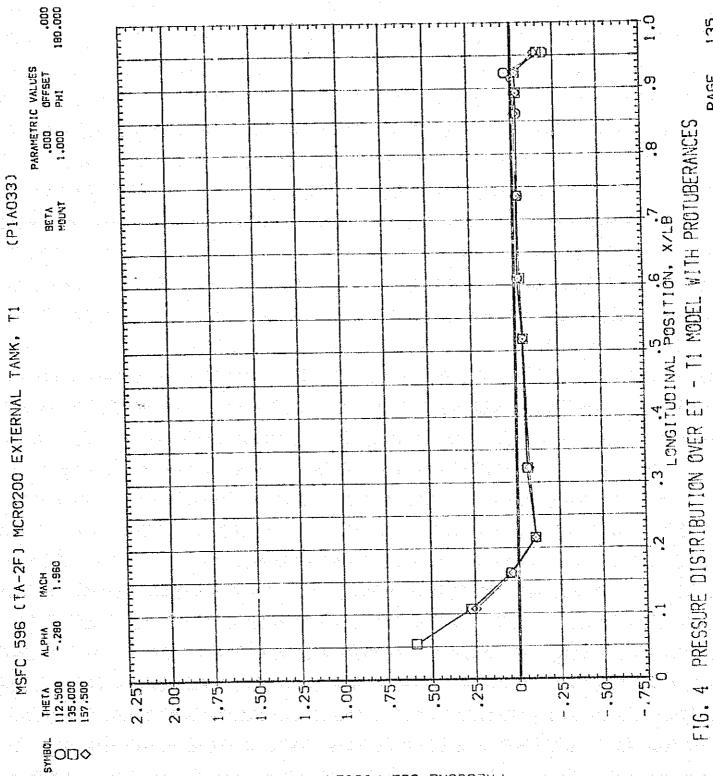




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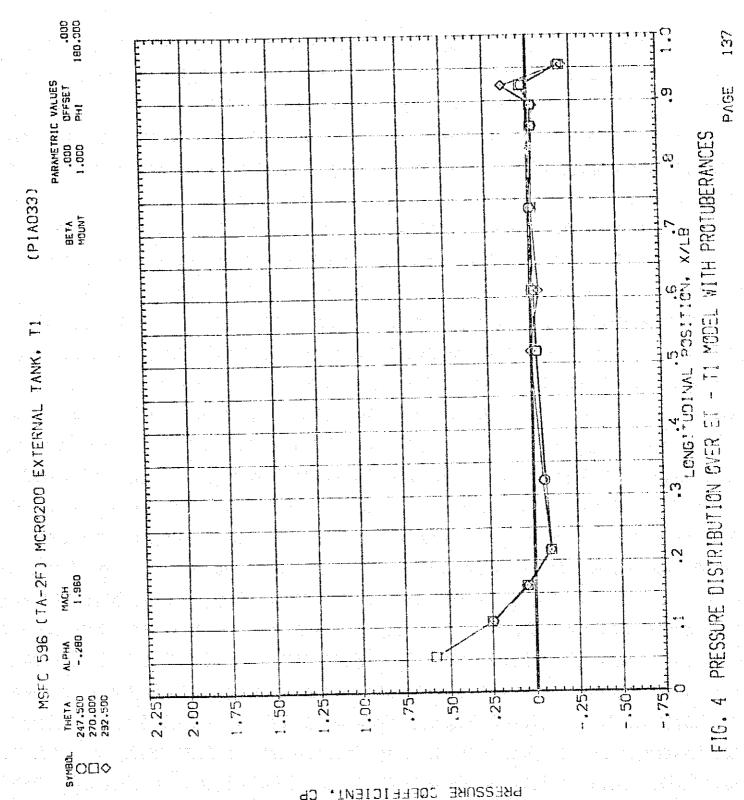
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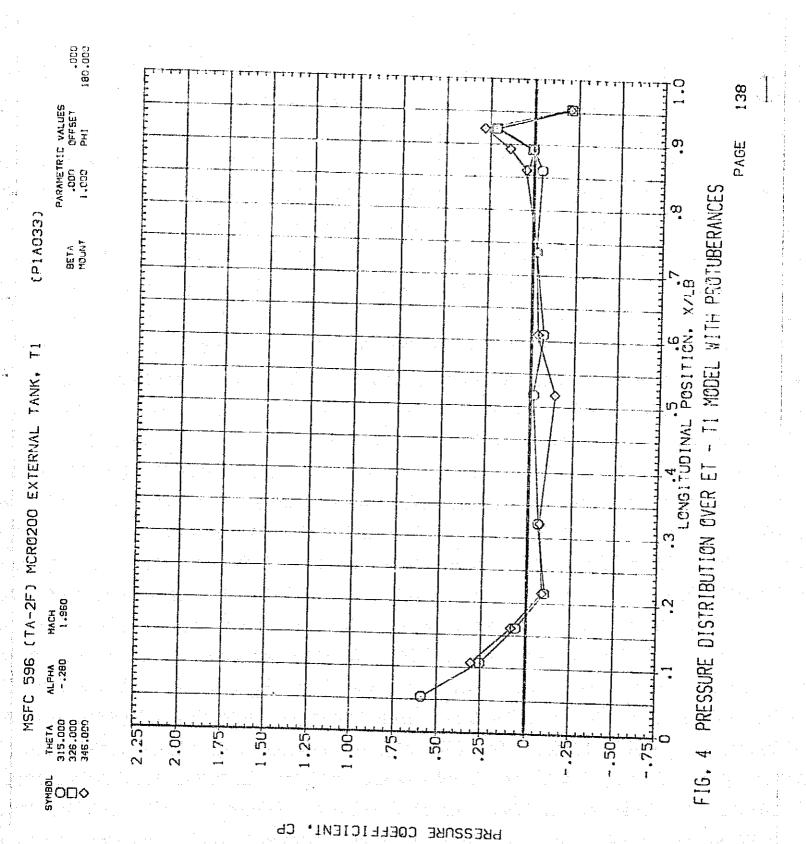
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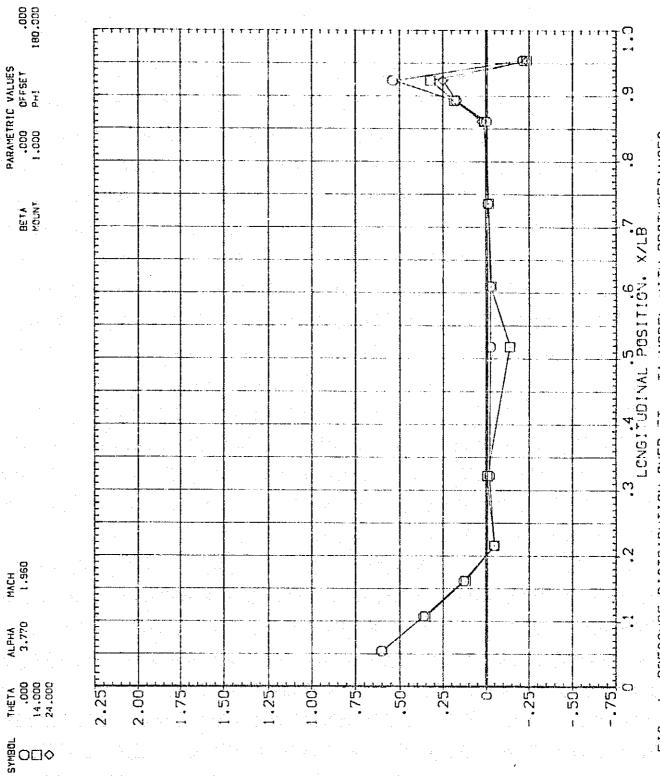
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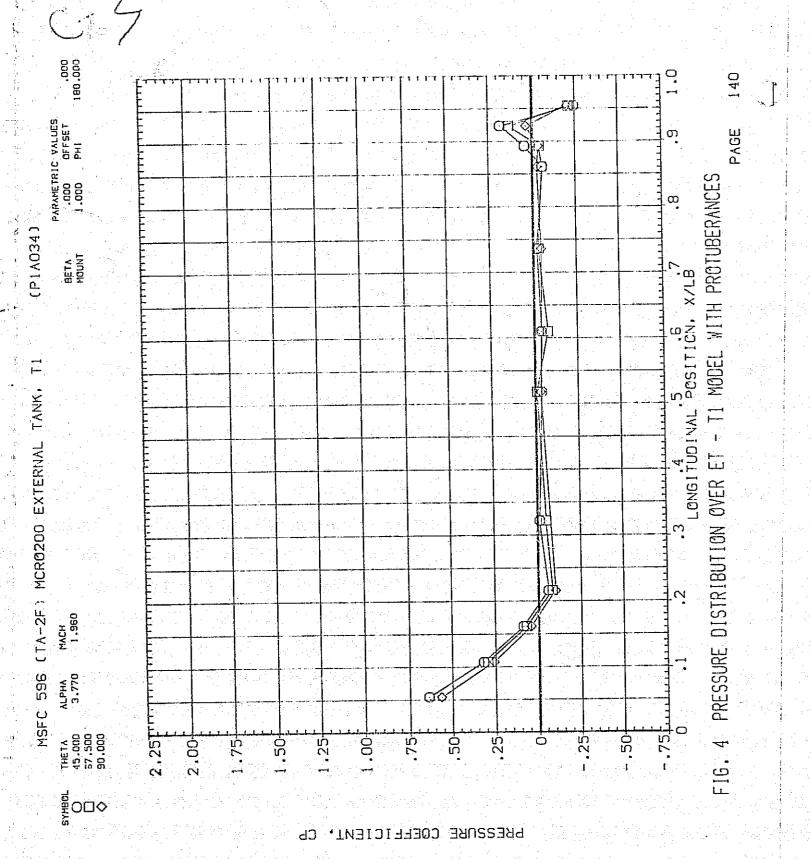
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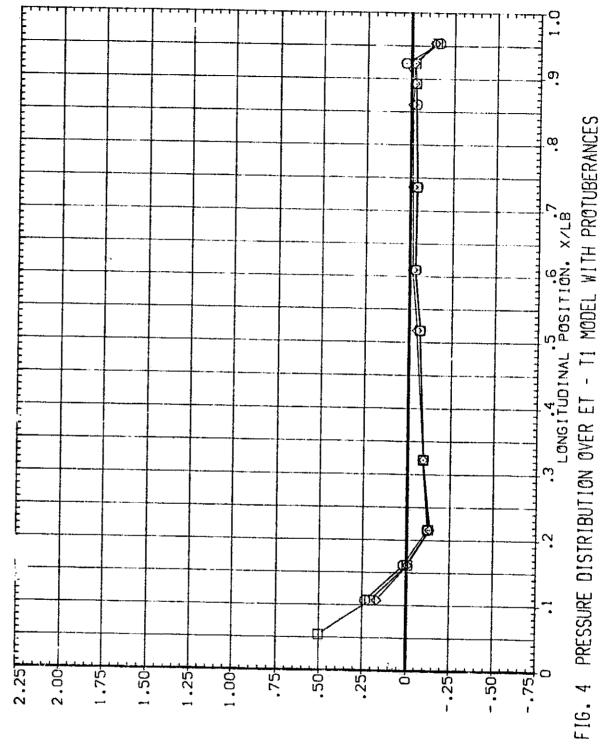


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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI

- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET





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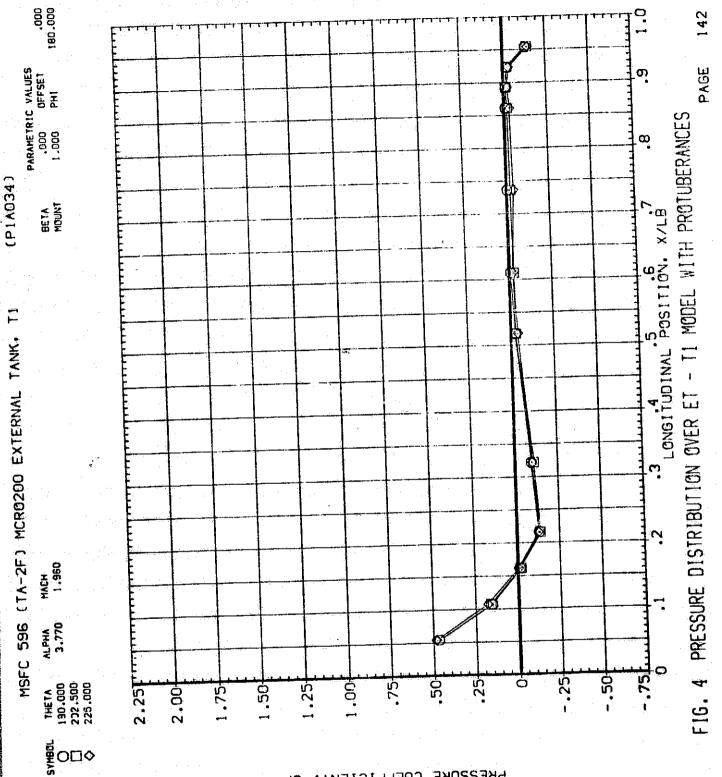
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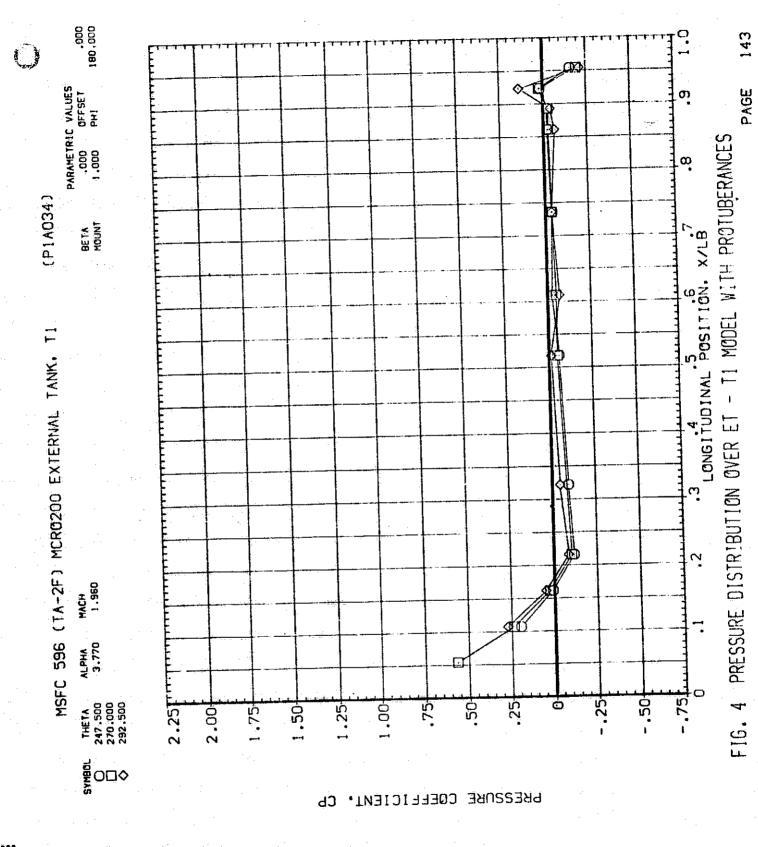
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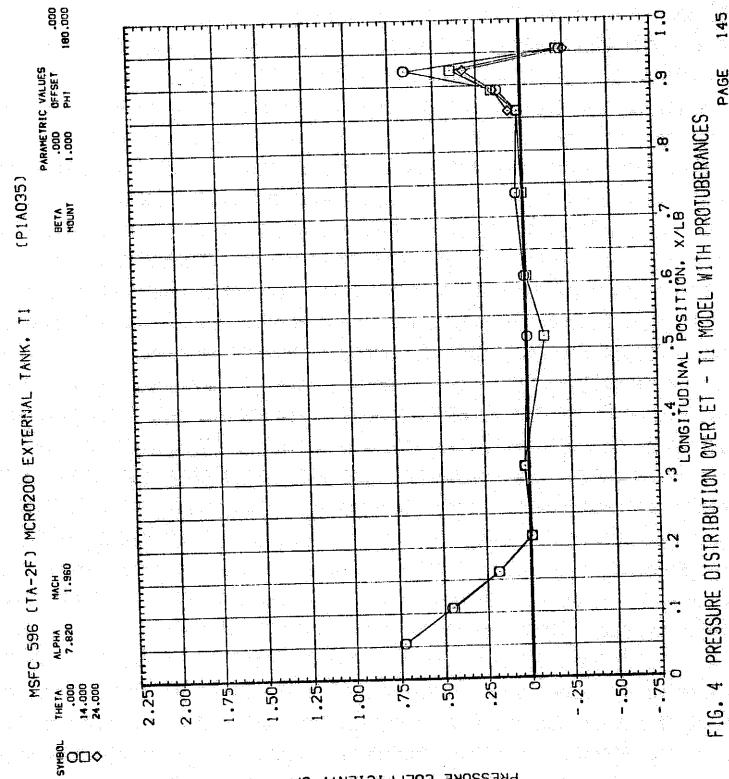
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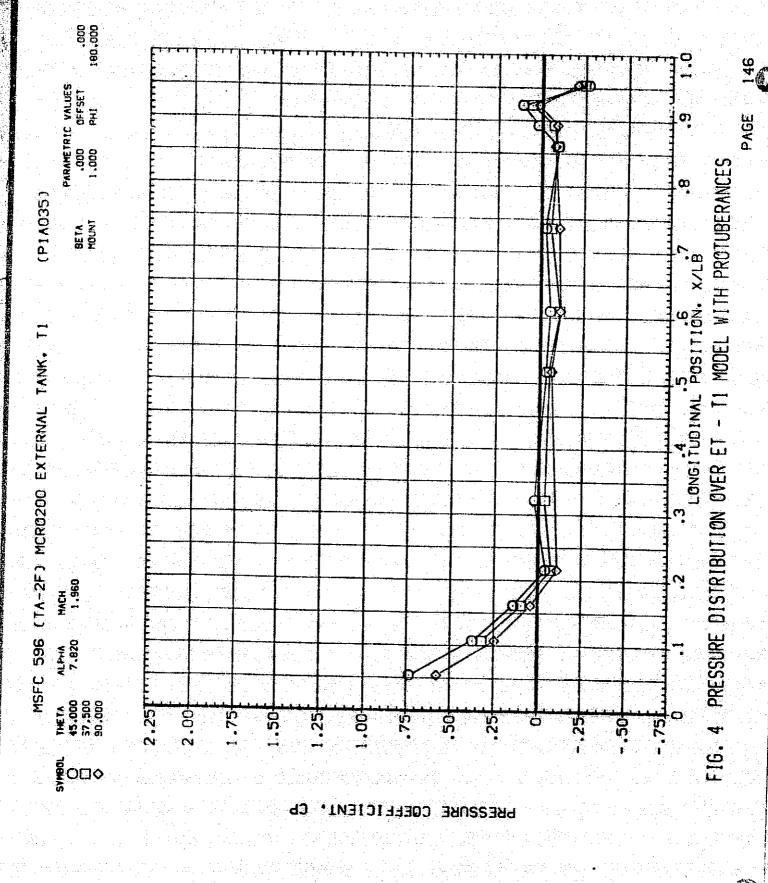
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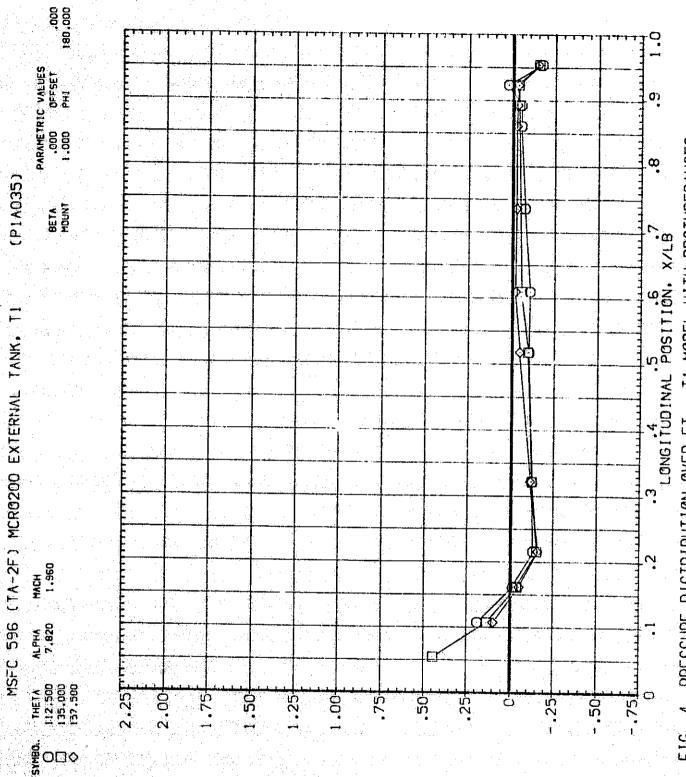
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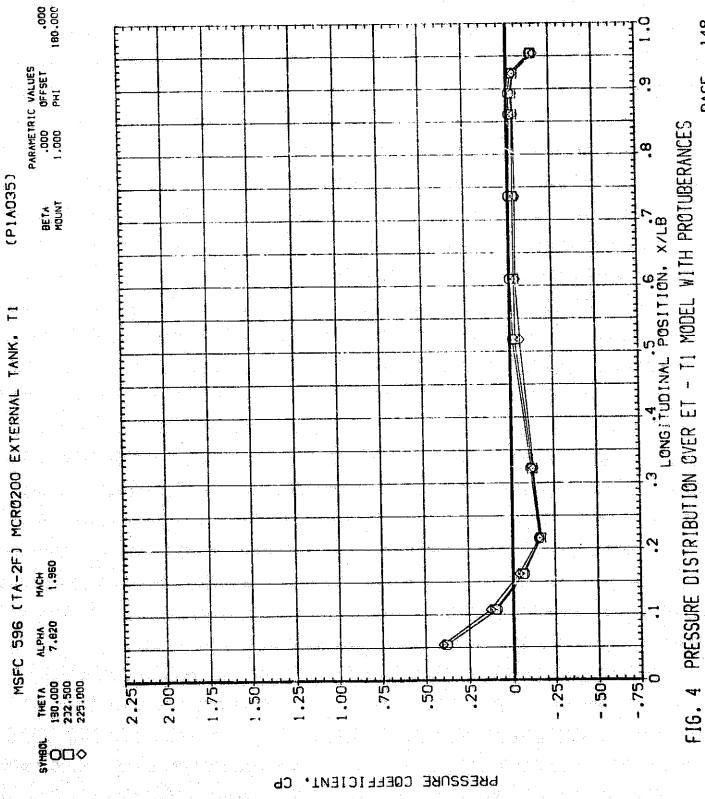


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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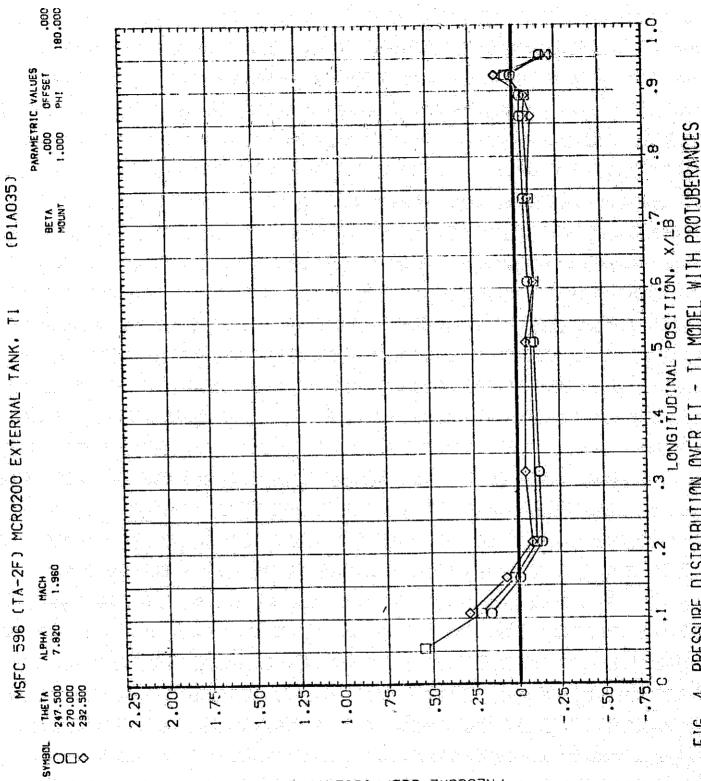


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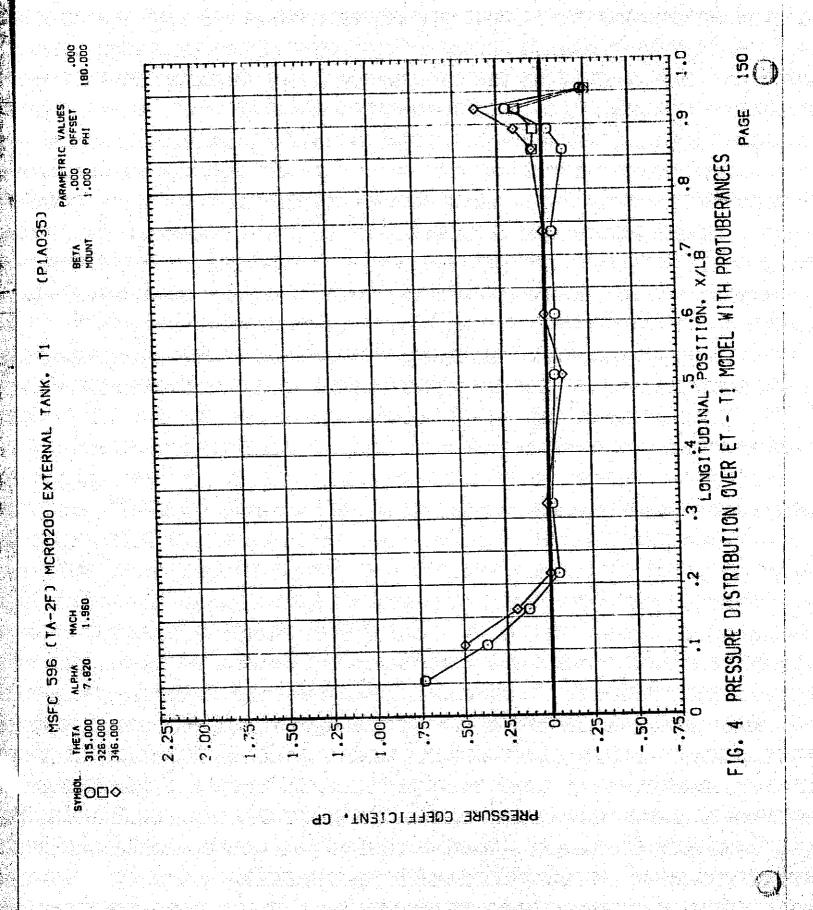


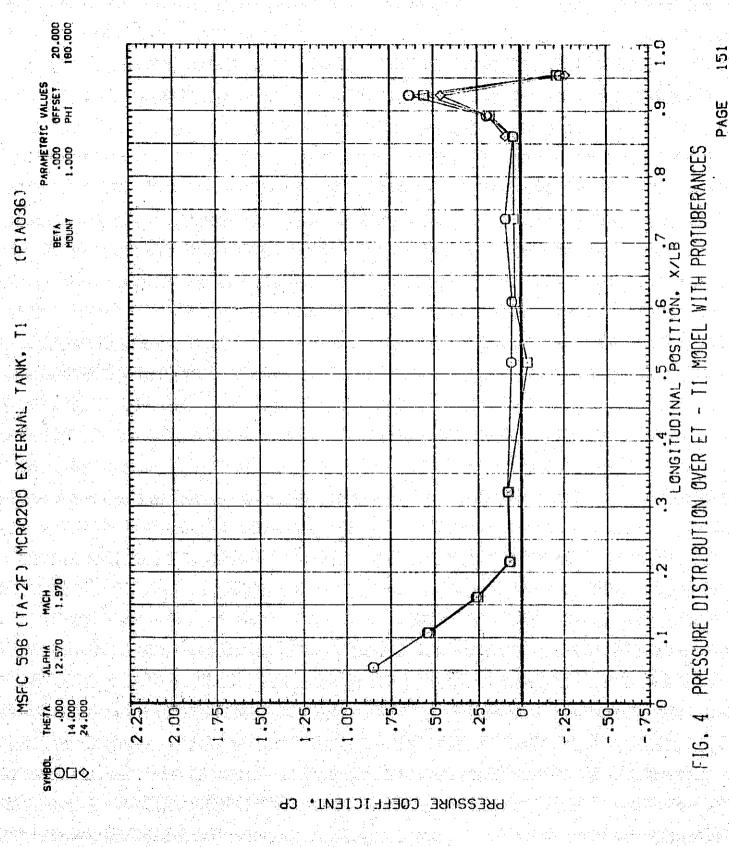
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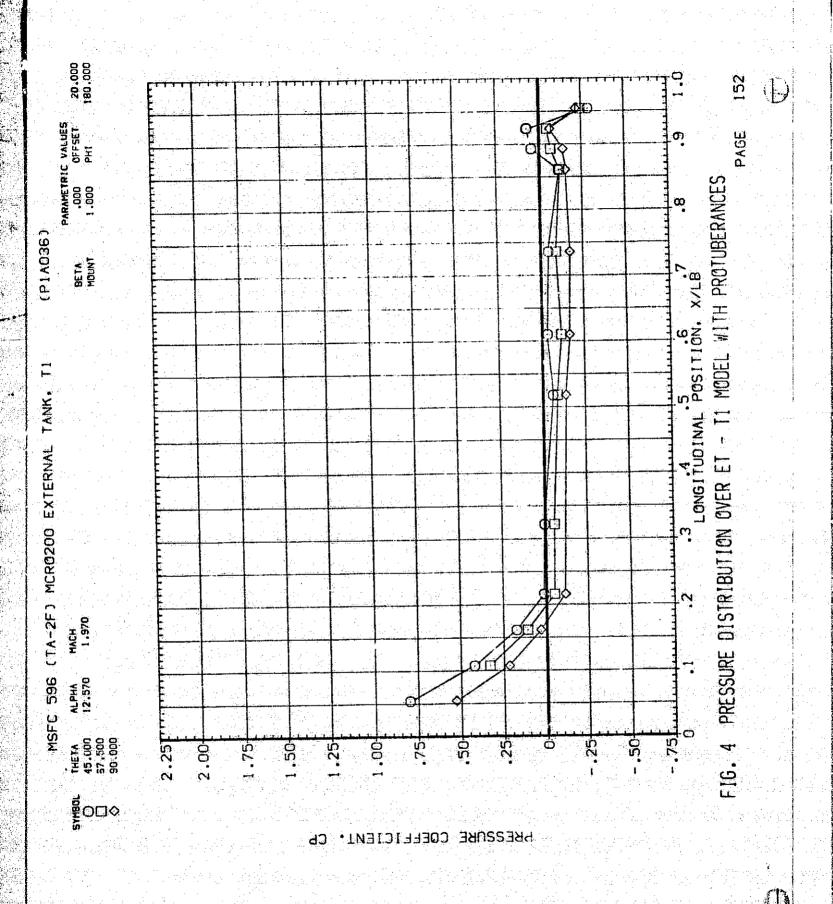
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

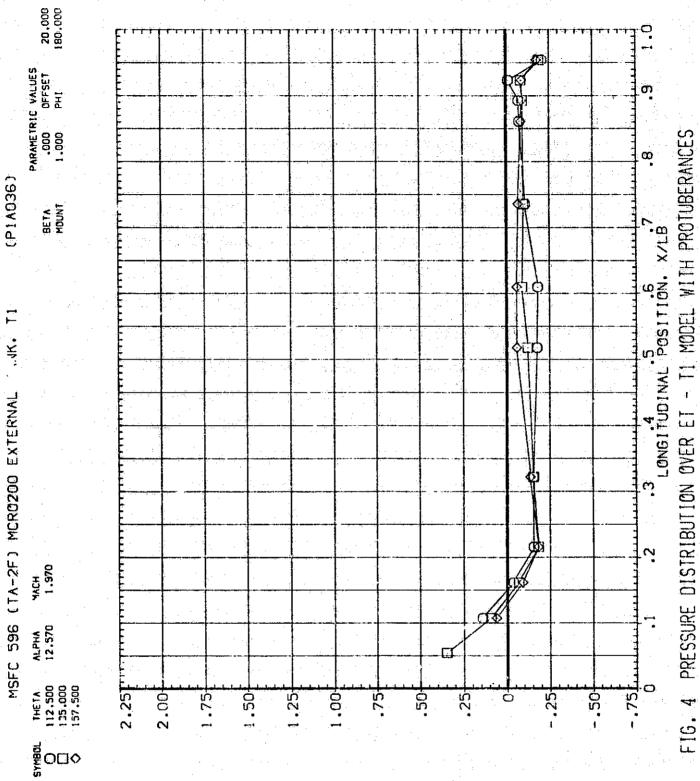
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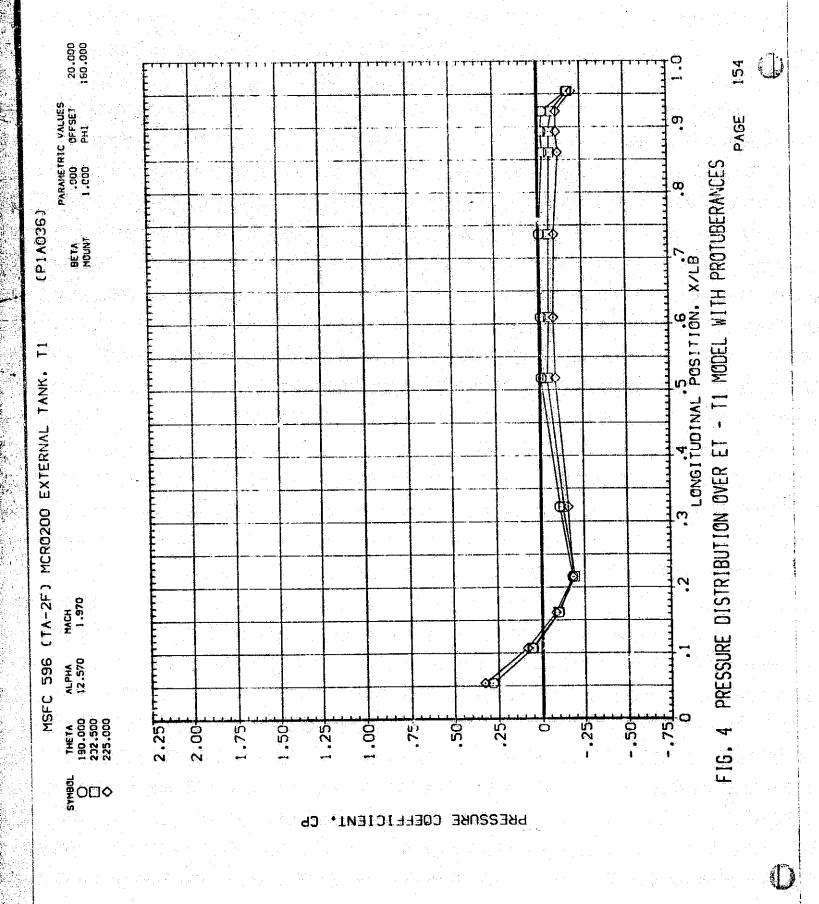


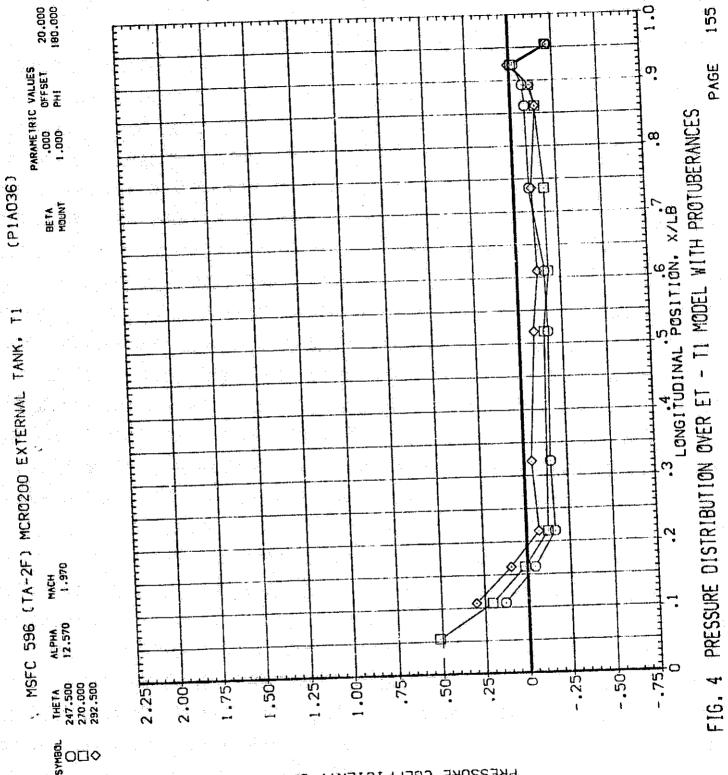
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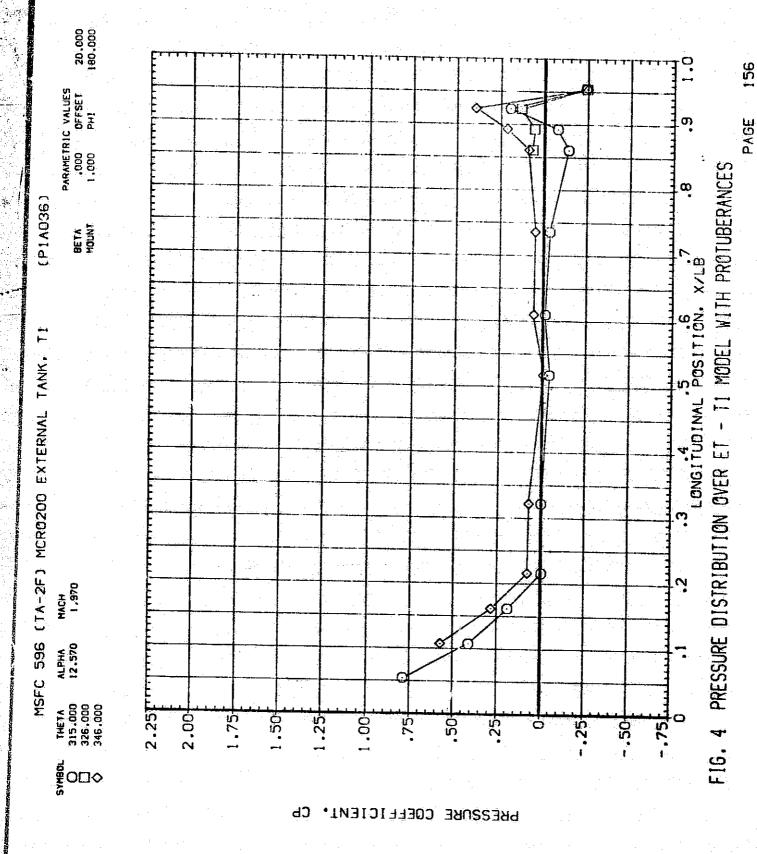
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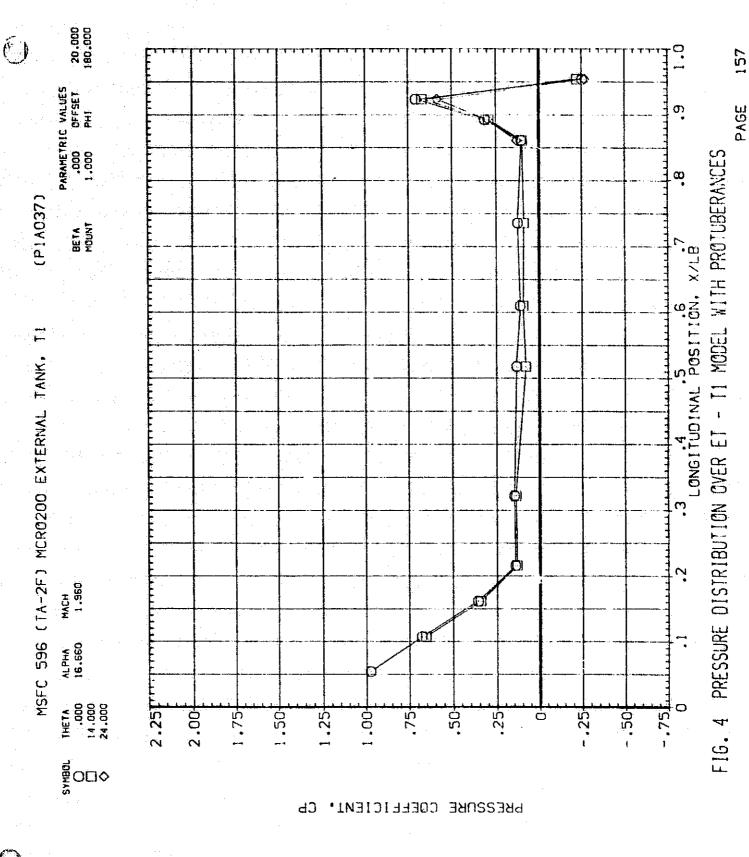


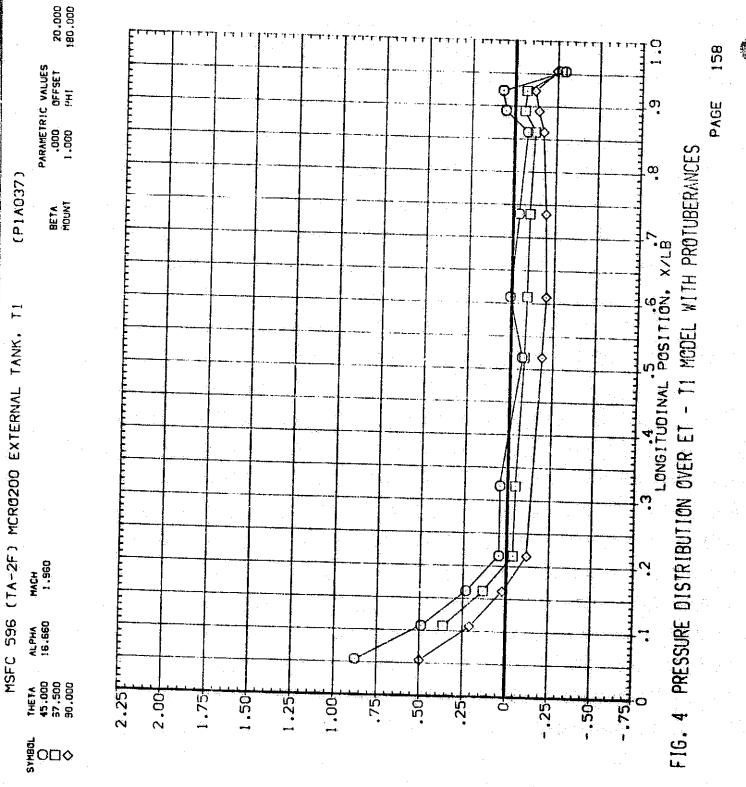


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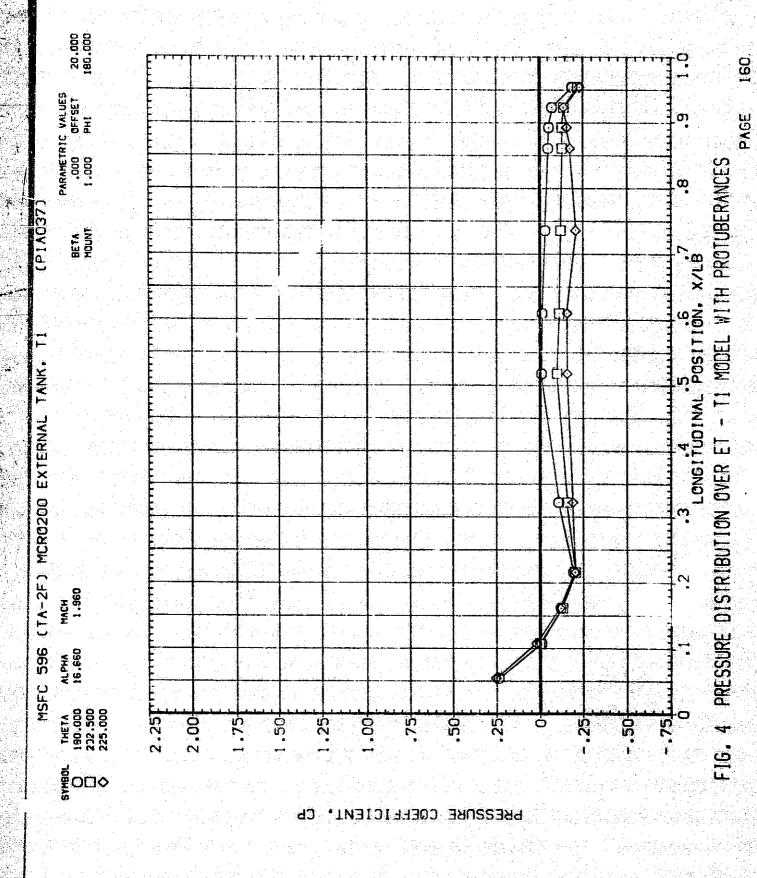
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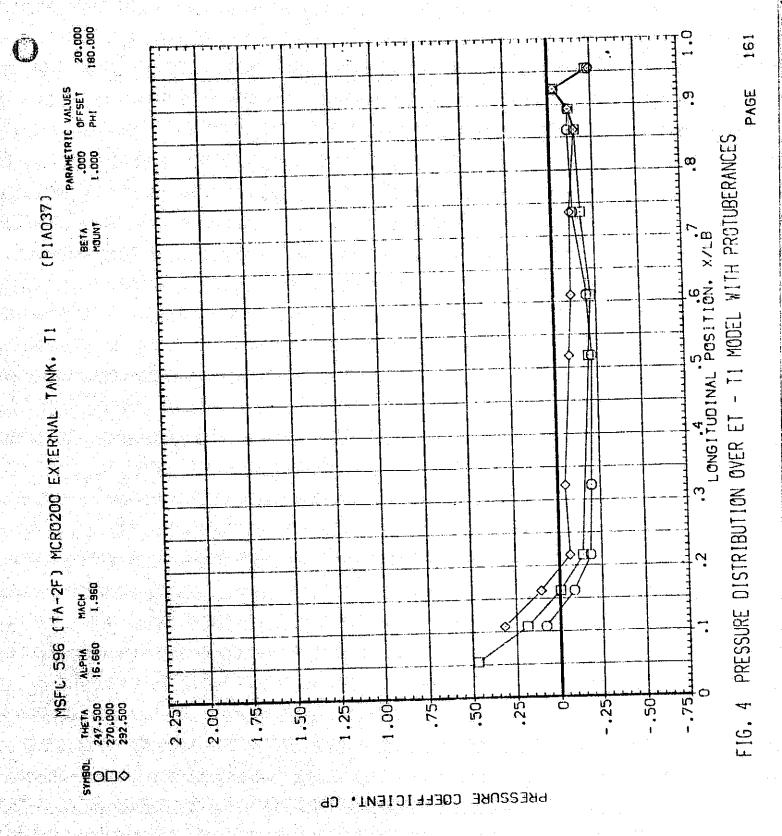


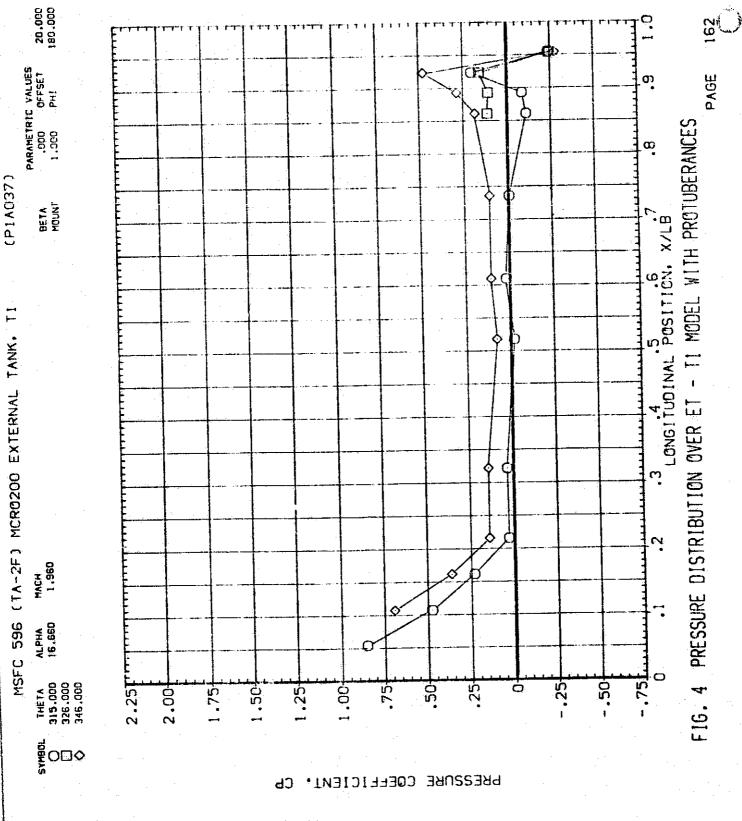


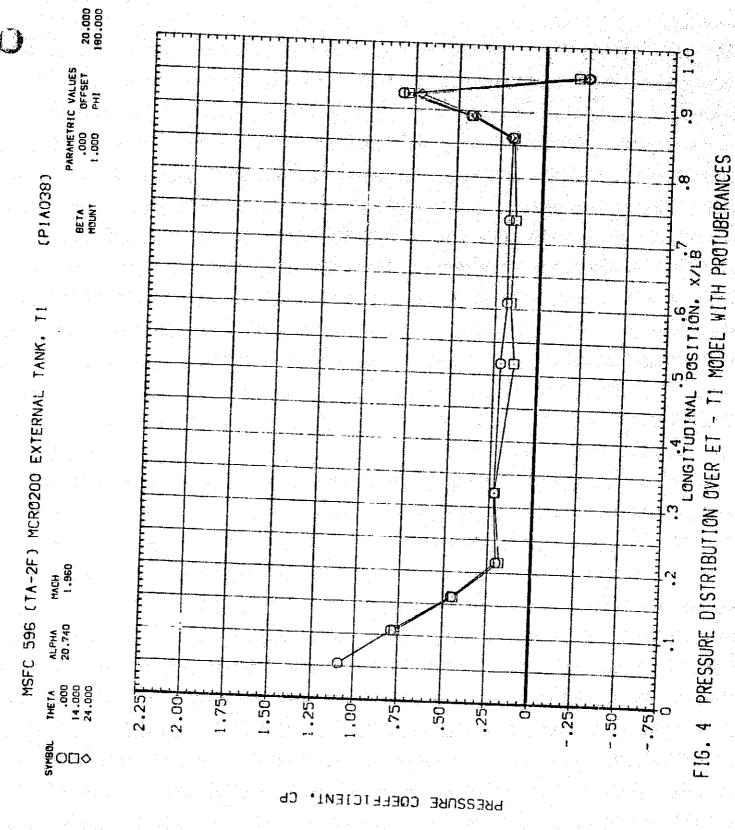
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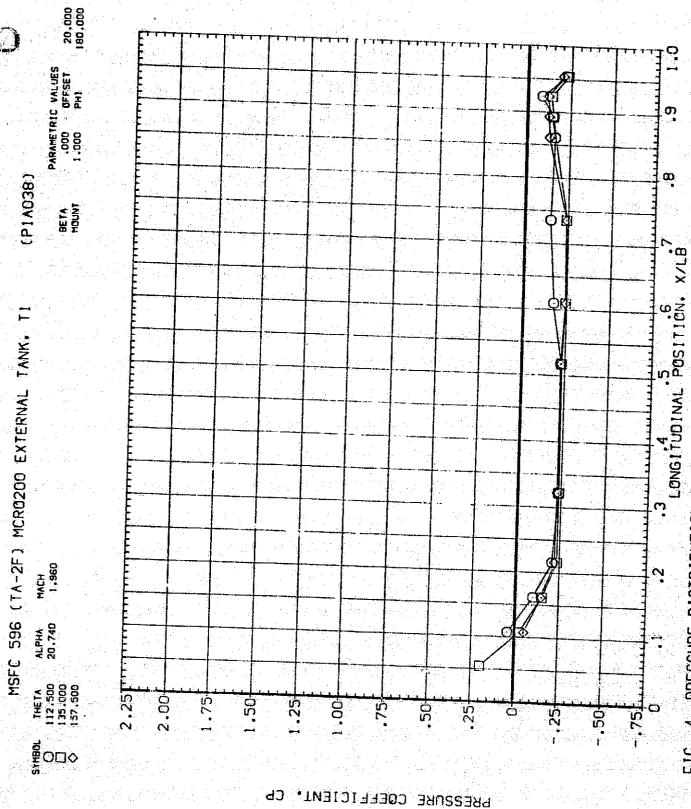
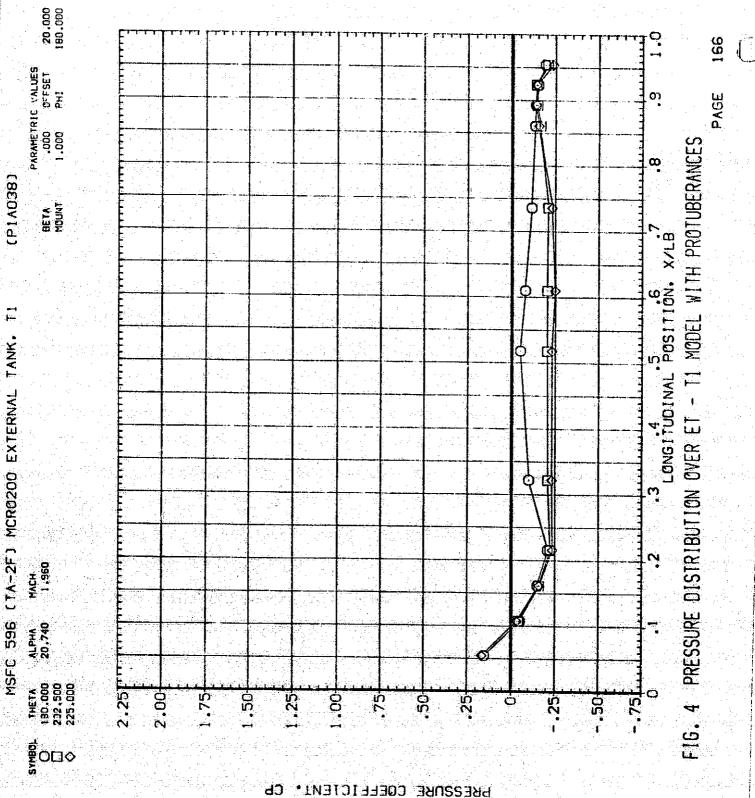
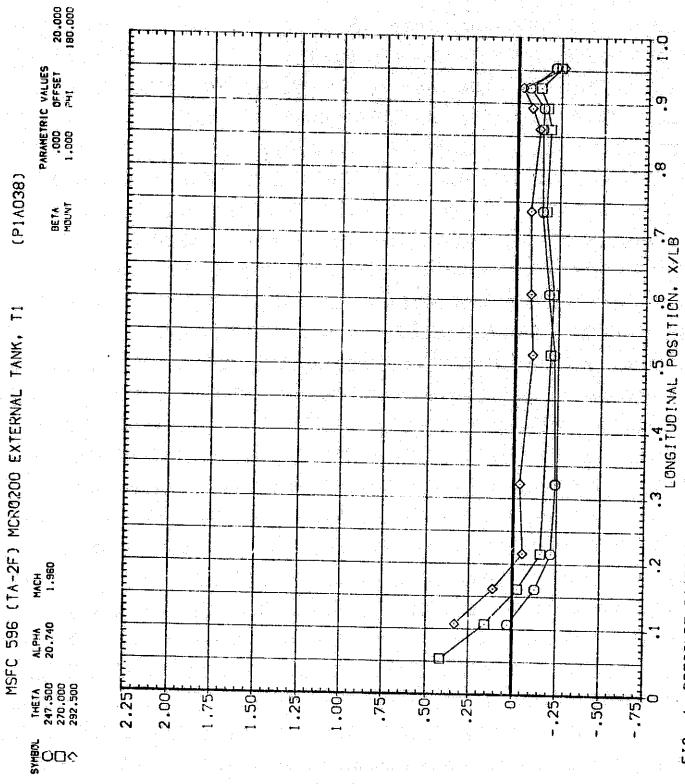


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



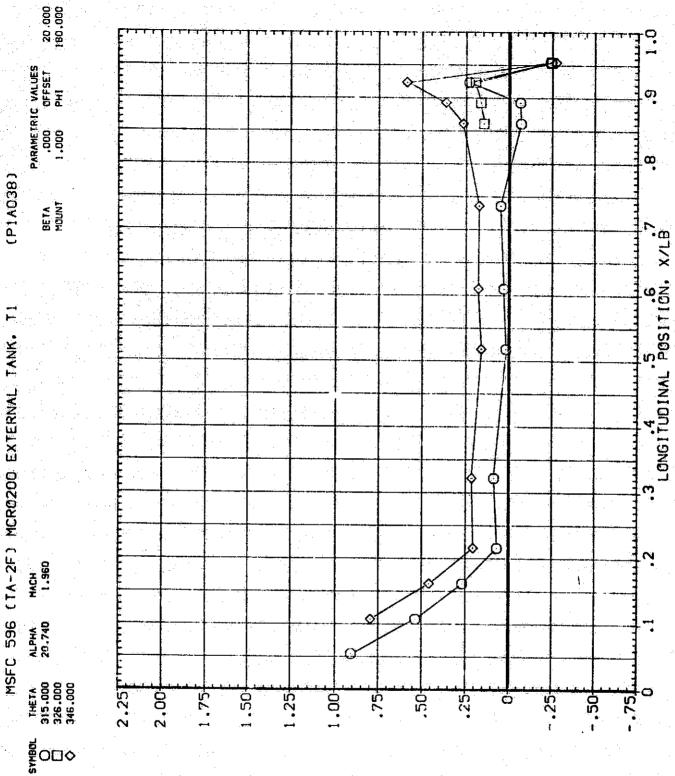


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

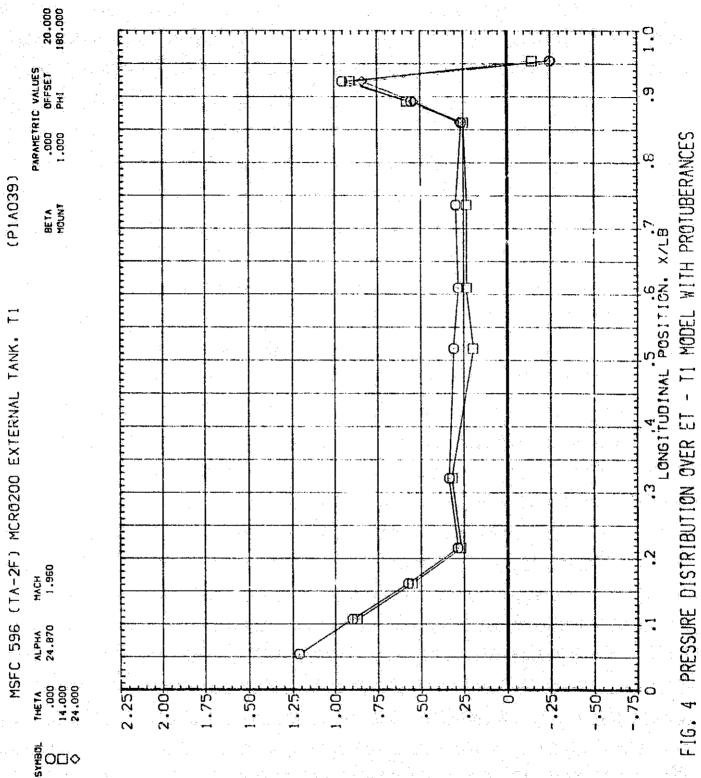
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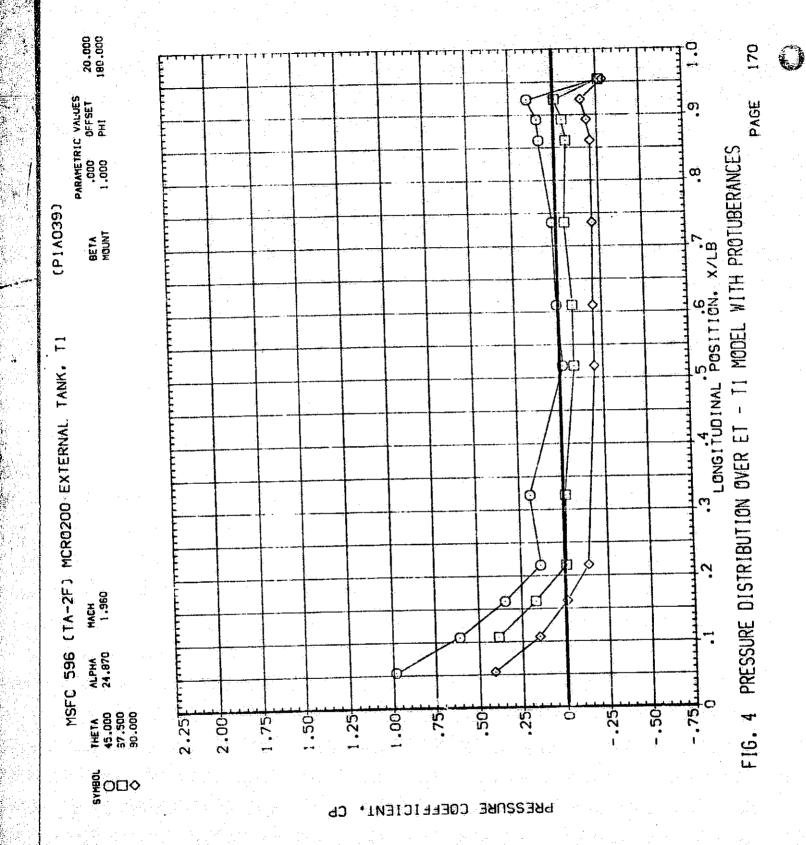
TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET





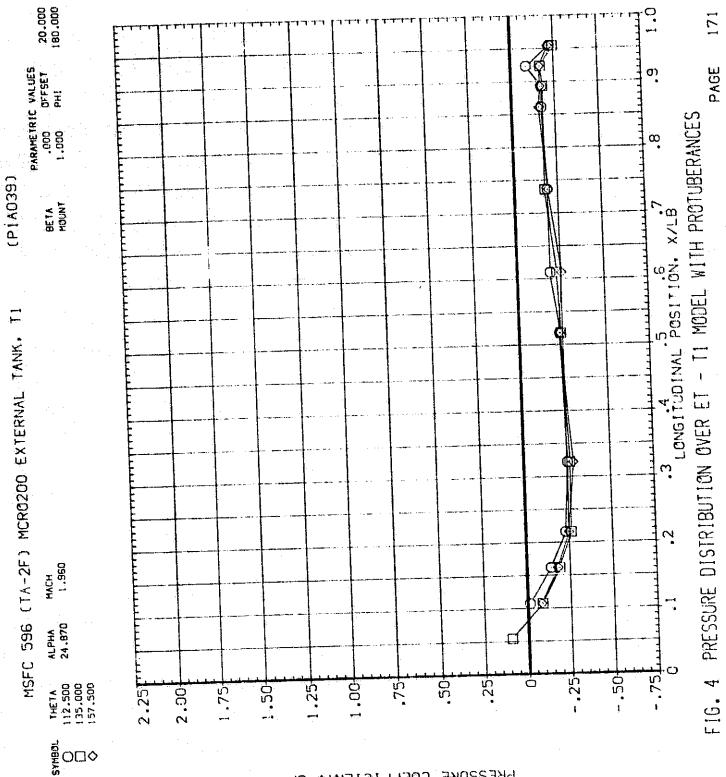
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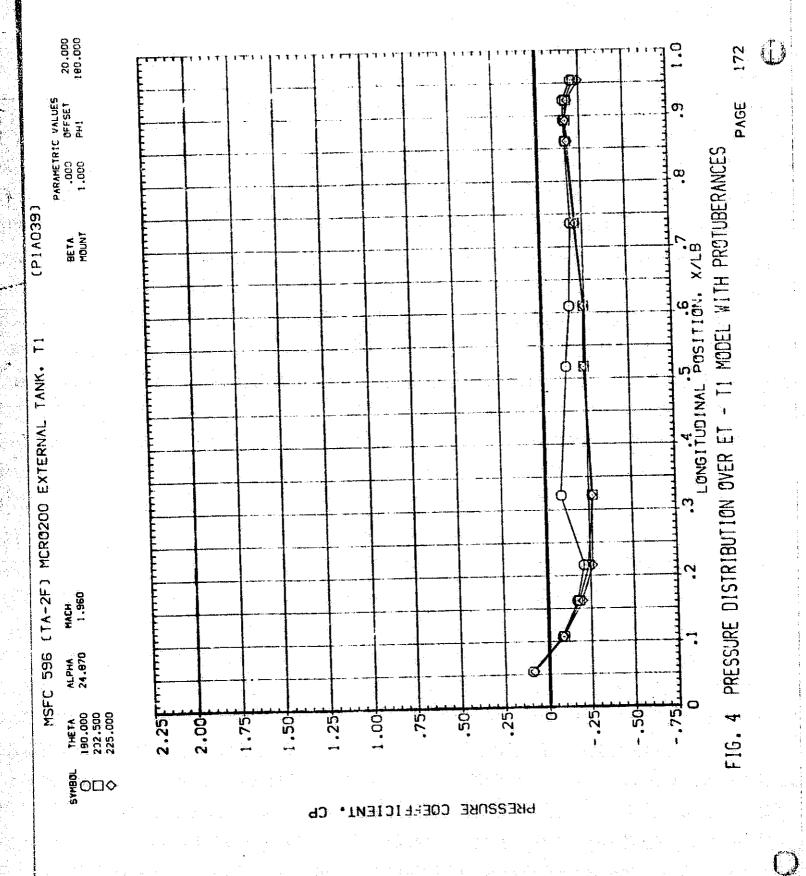


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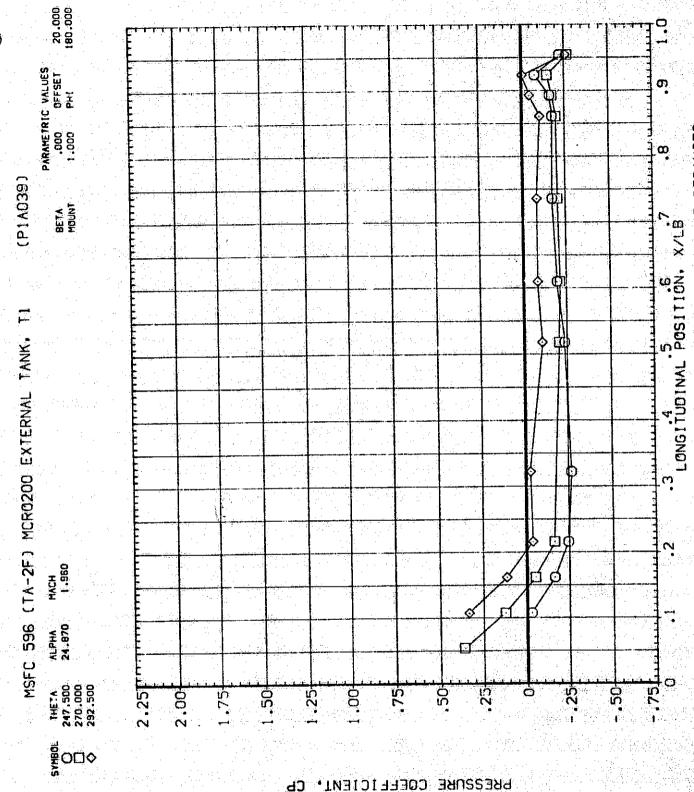
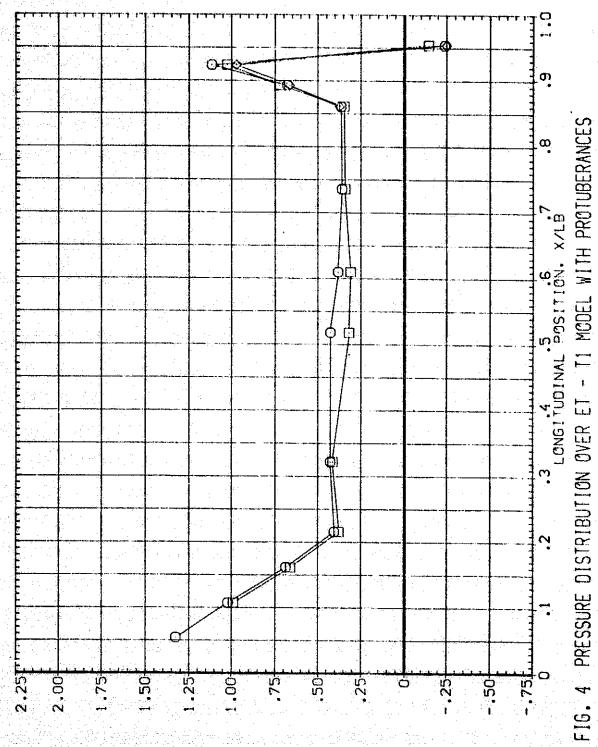


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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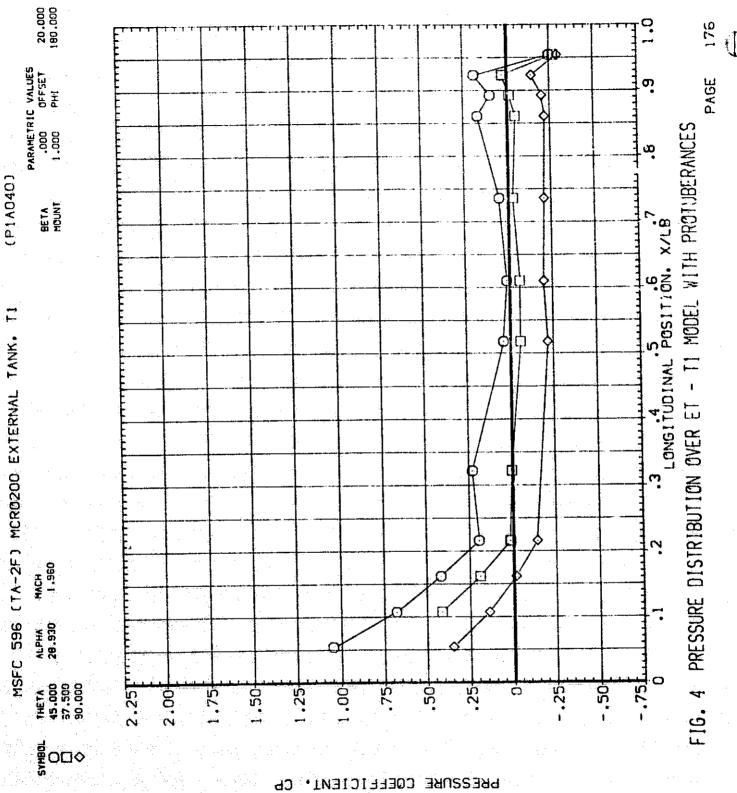
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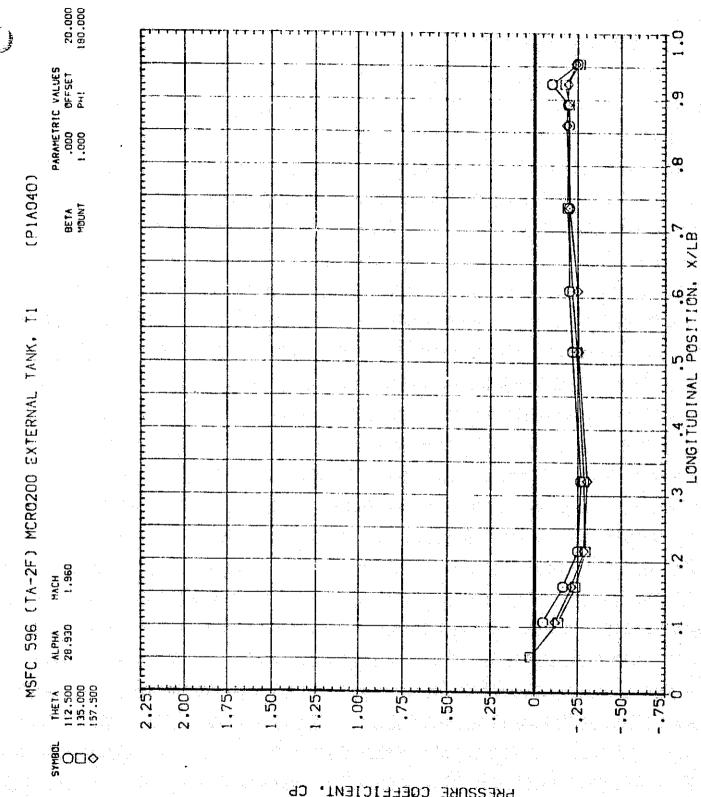
MSFC 596 (TA-2F) MCR0200 EXTERWAL TANK. TI

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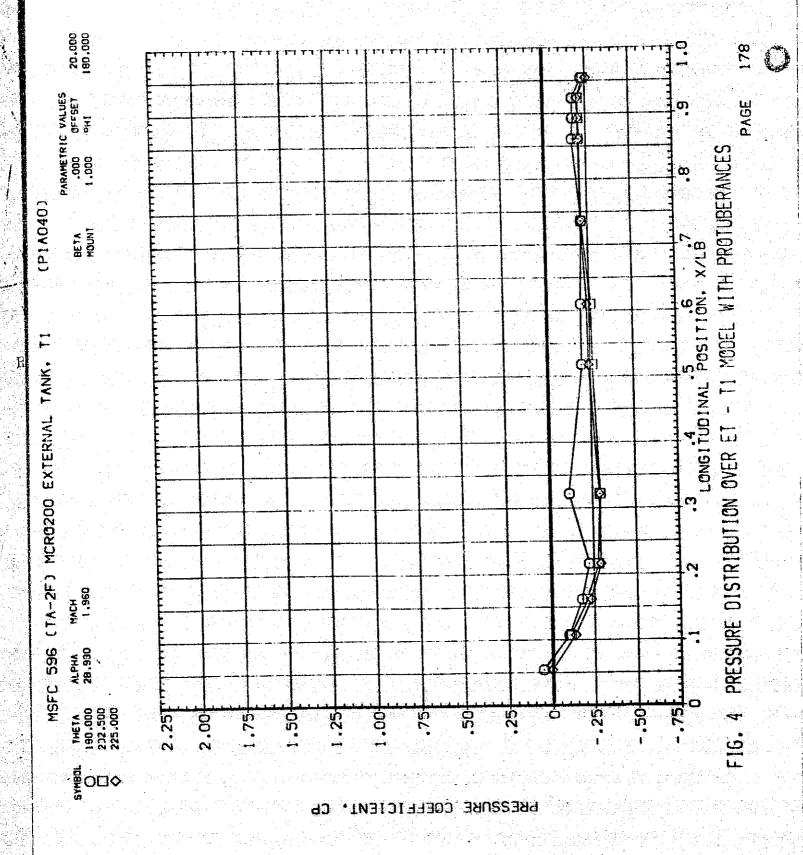
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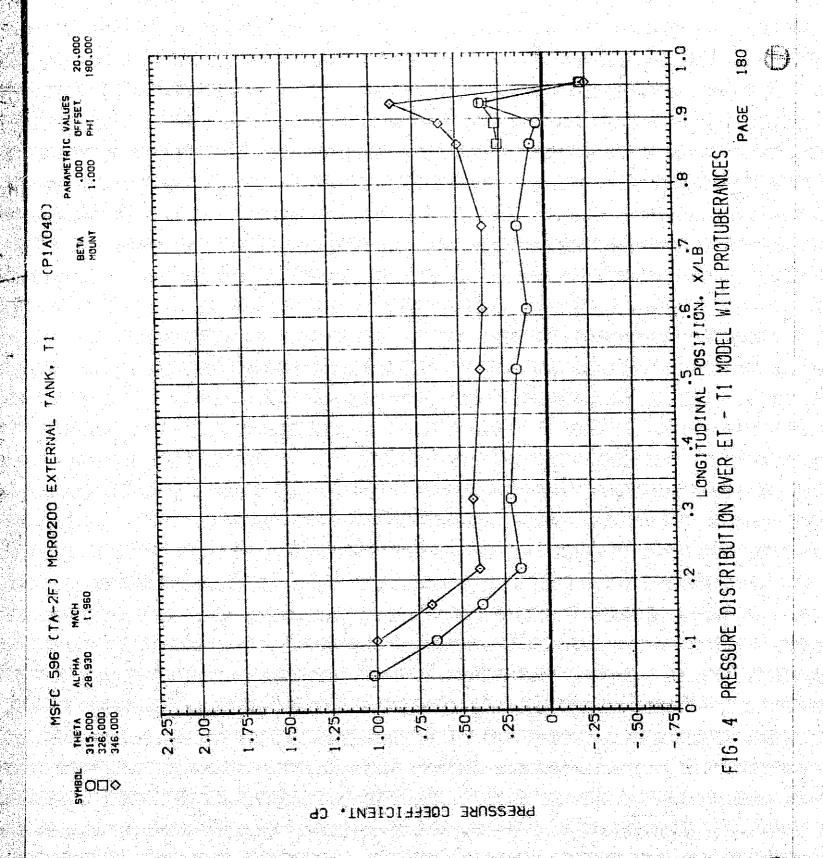
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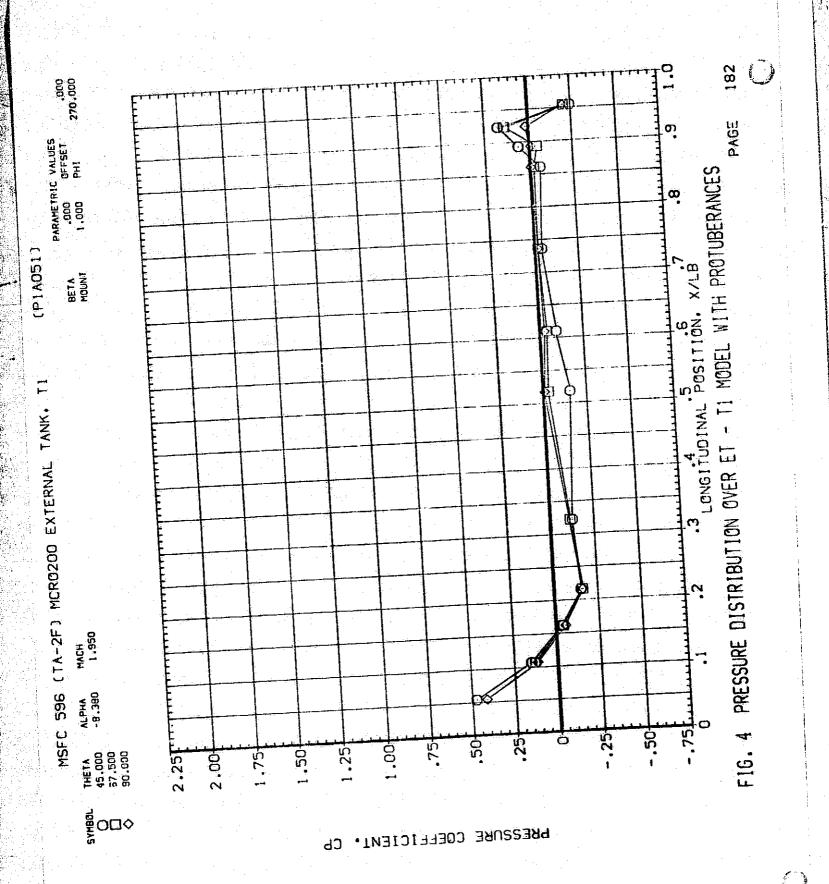


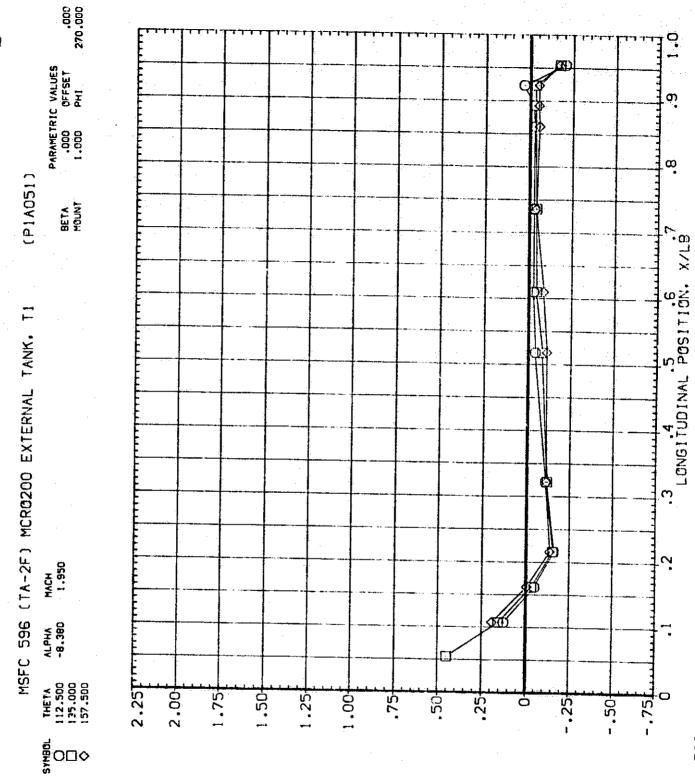


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

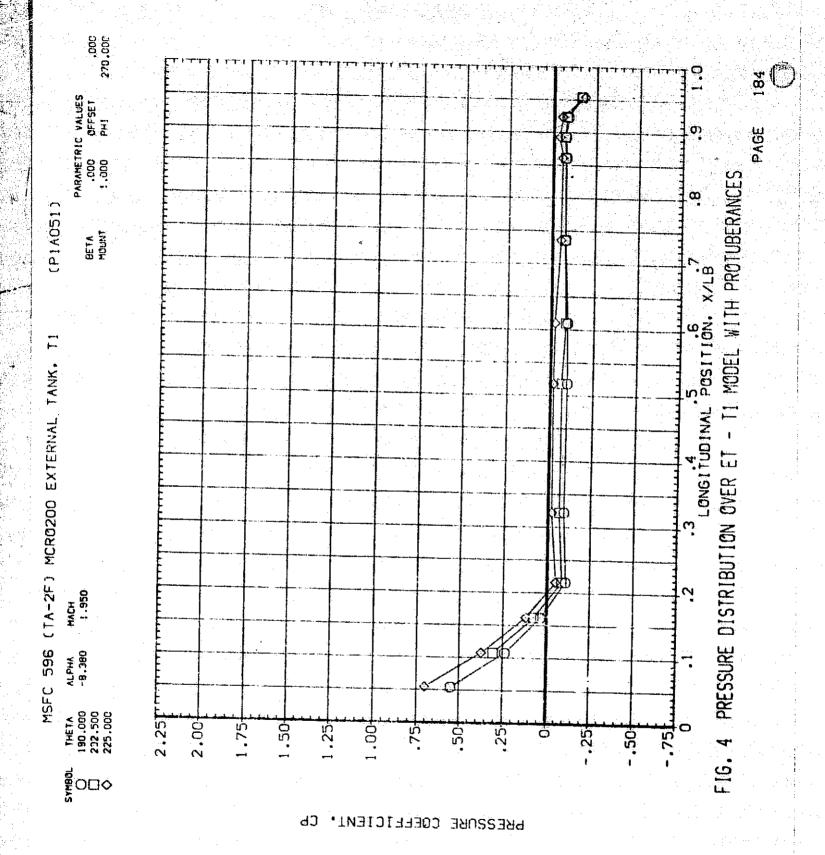
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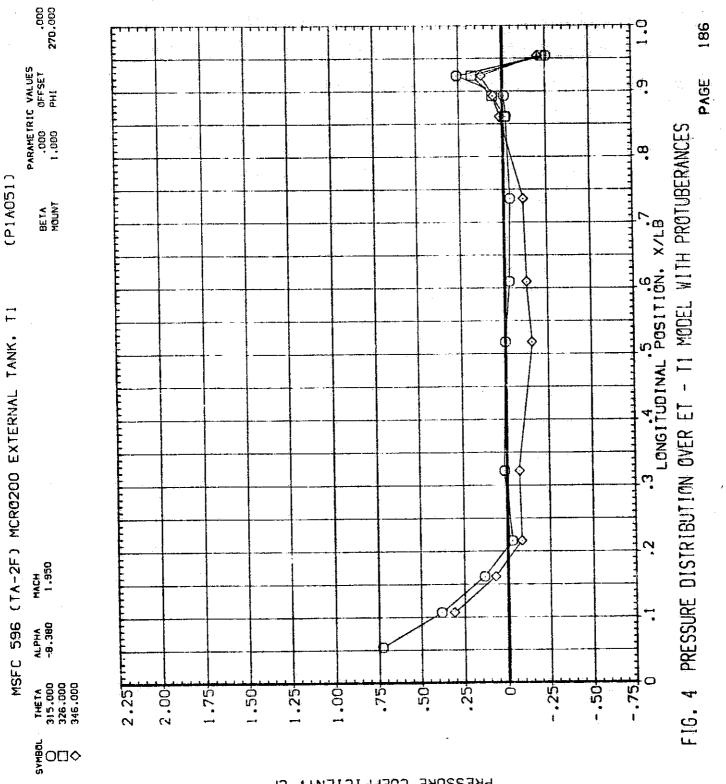
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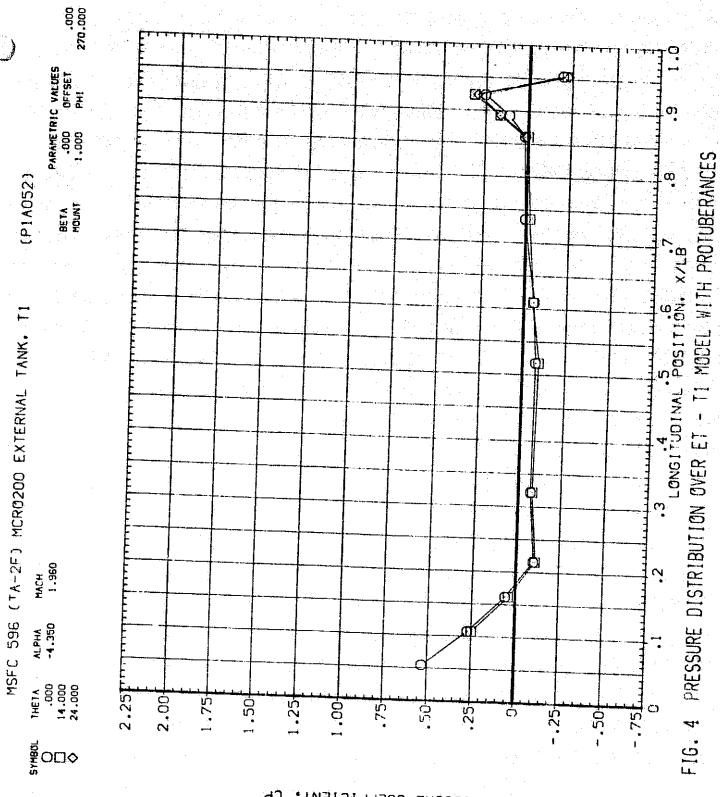


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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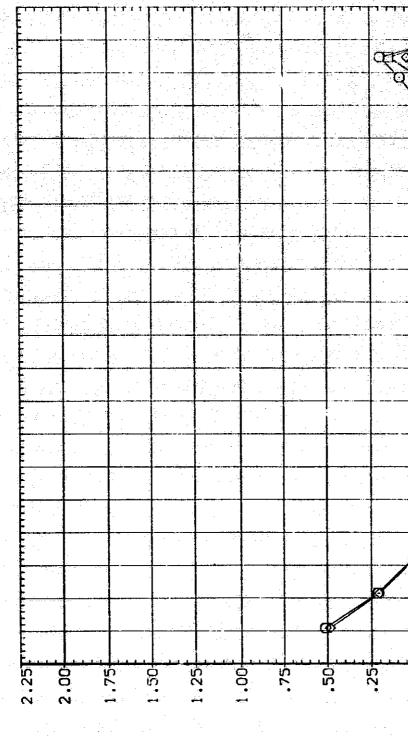
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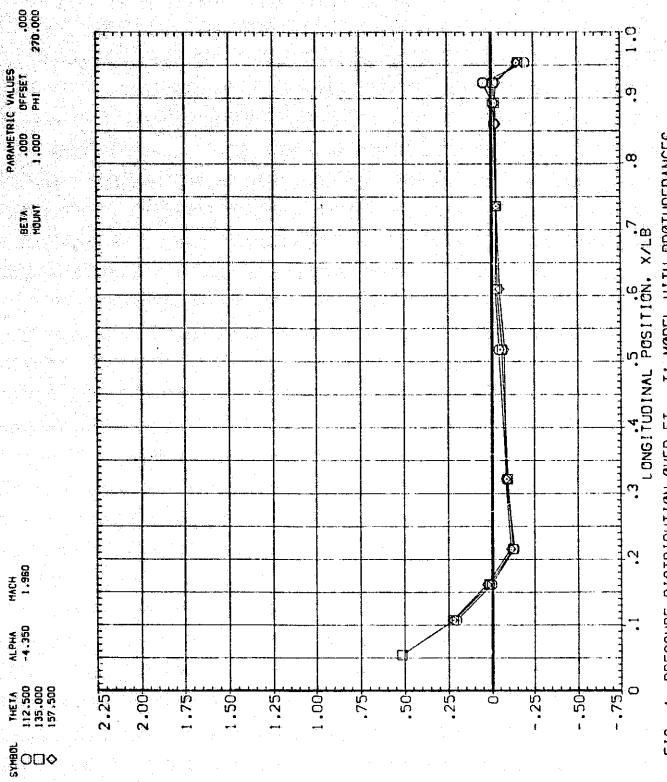
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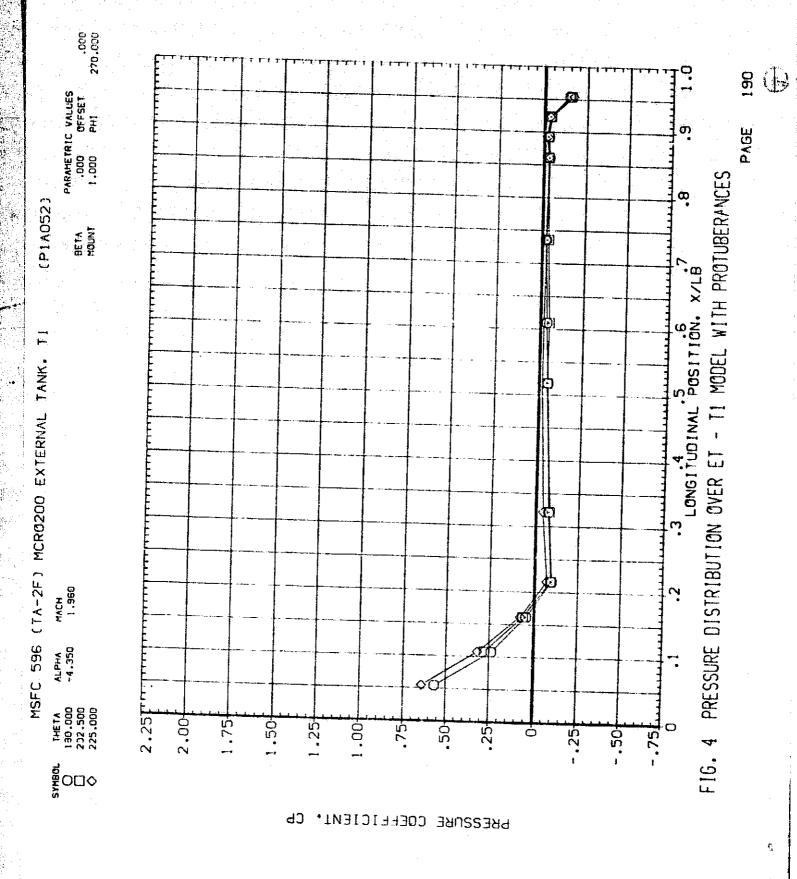
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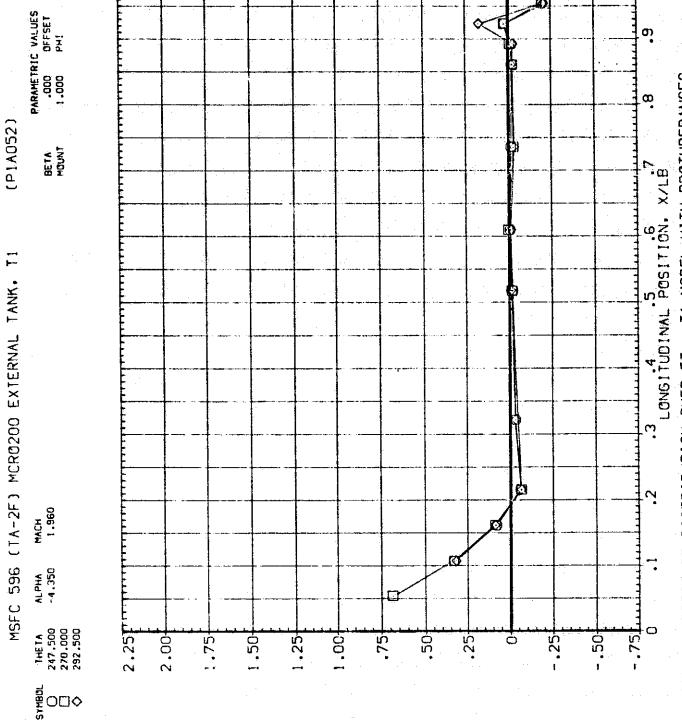


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - 71 MODEL WITH PROTUBERANCES

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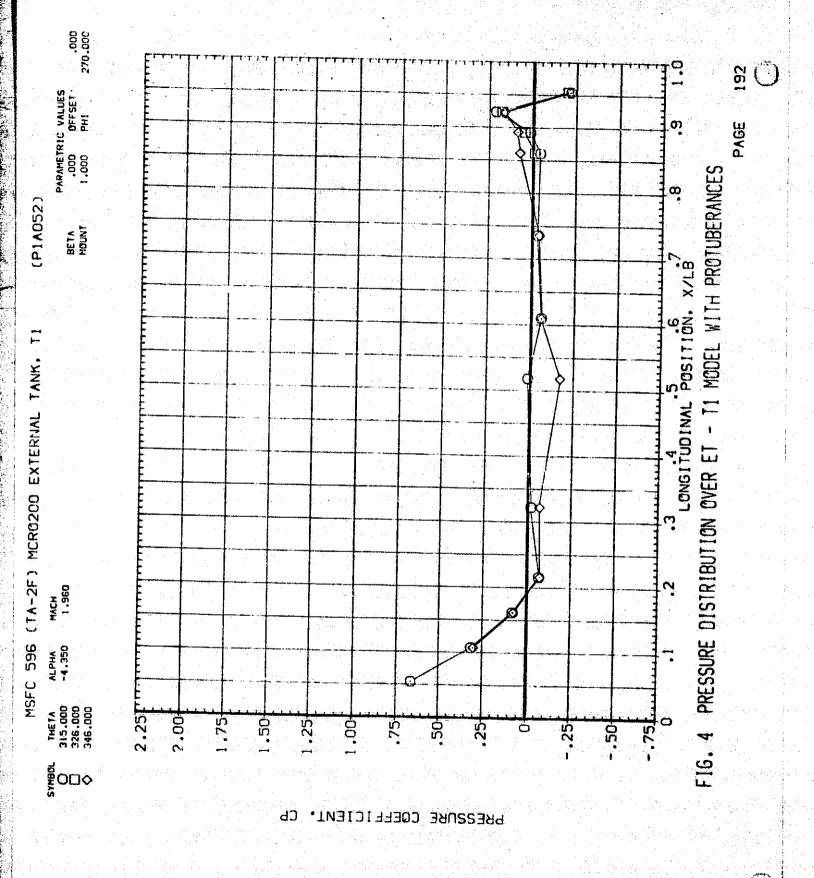




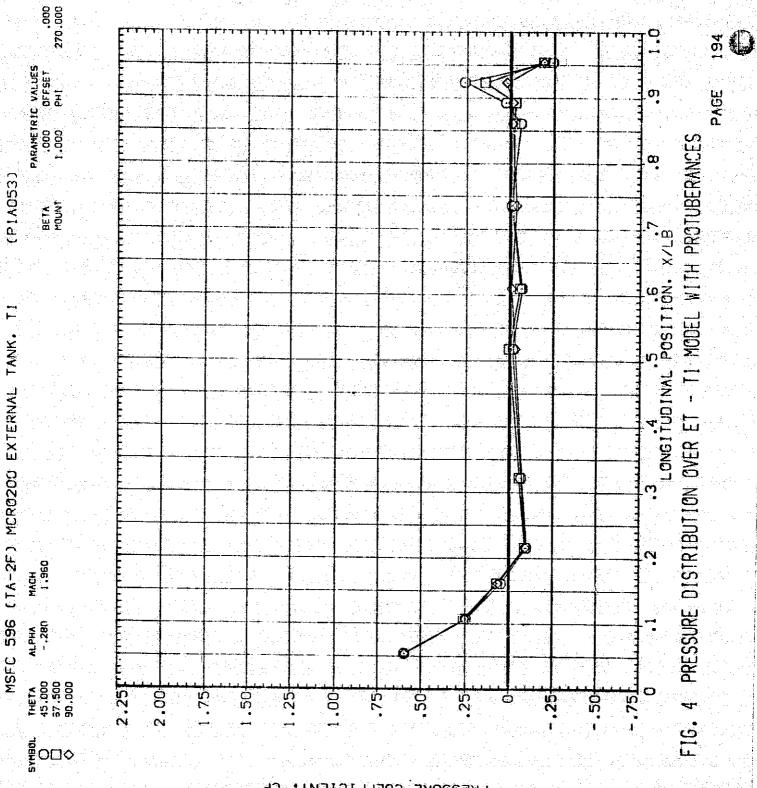
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES



F16. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

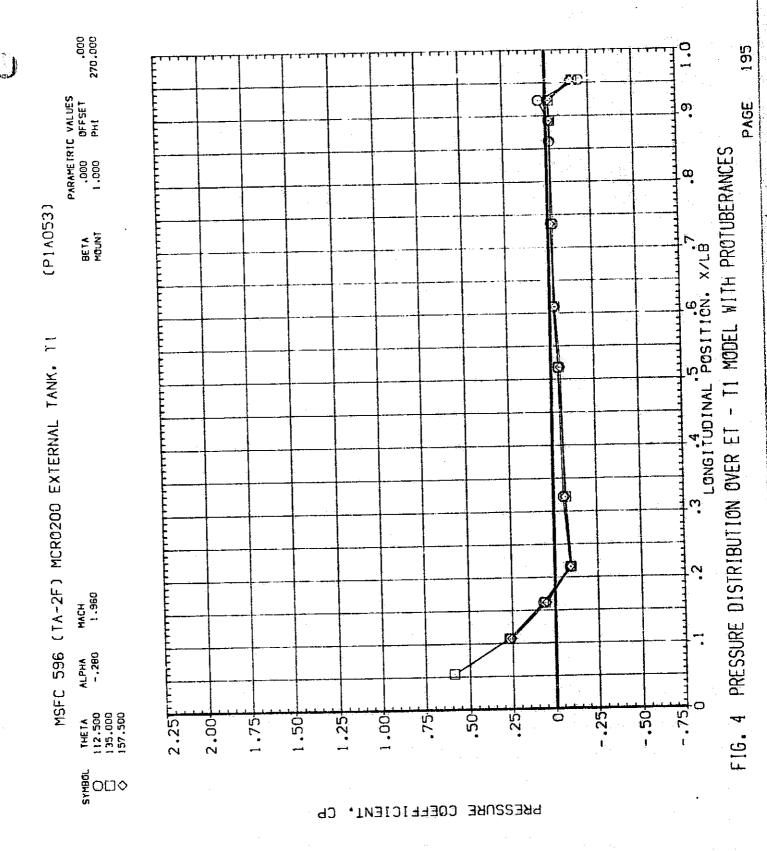


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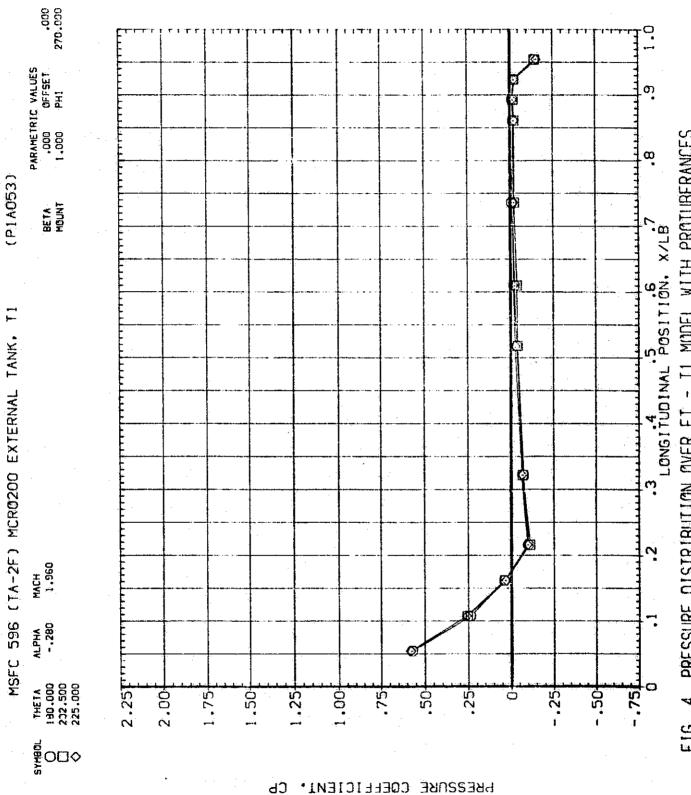


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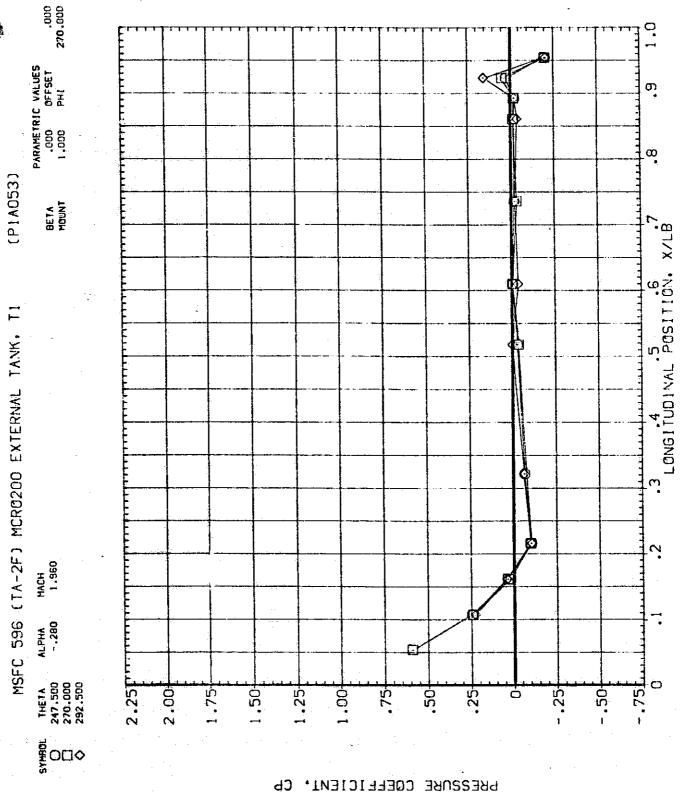
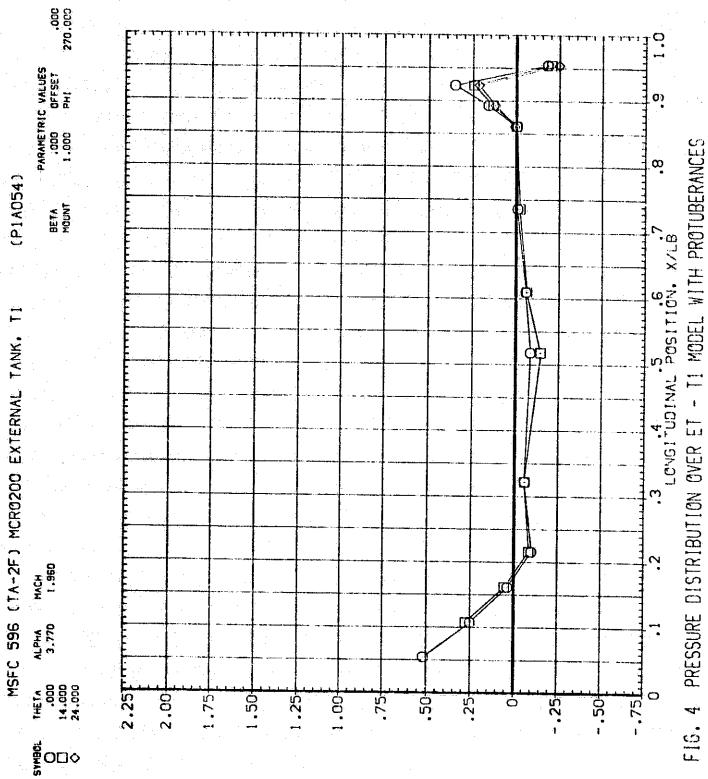
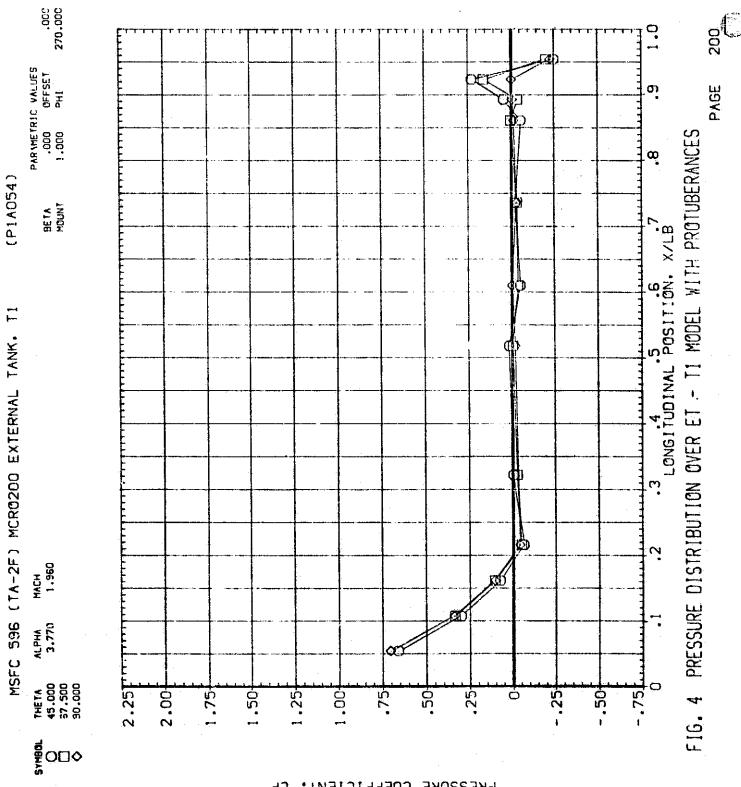


FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

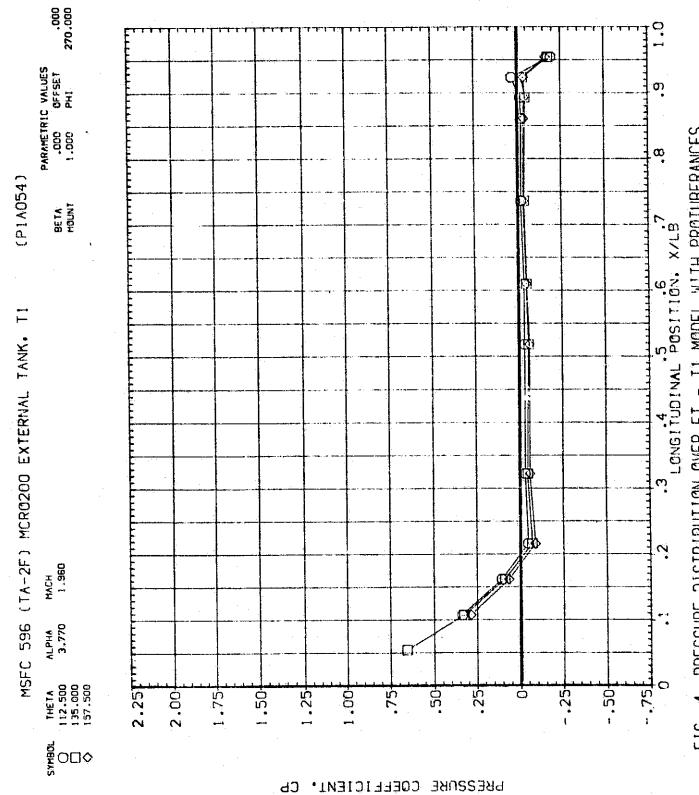
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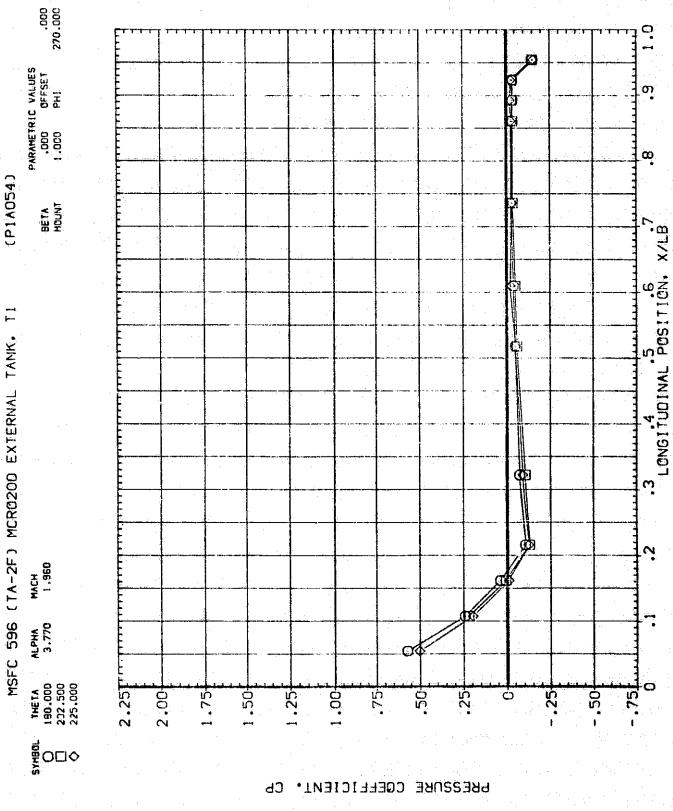
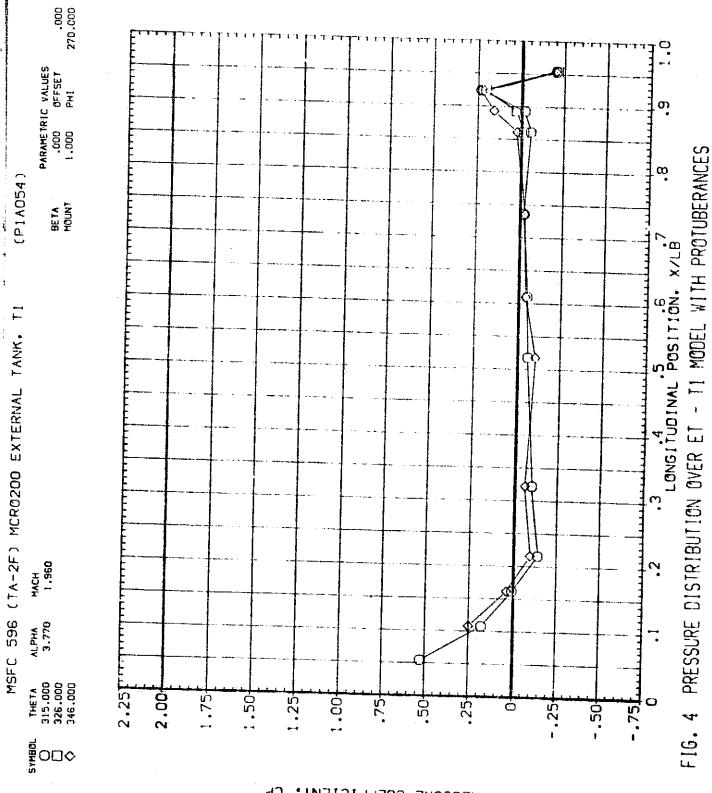


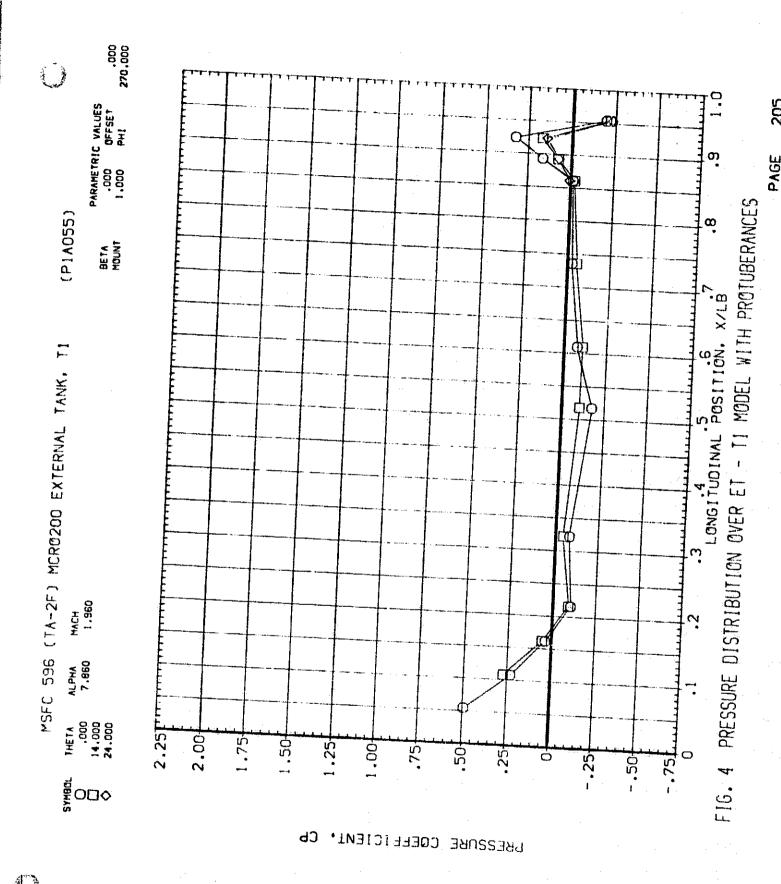
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

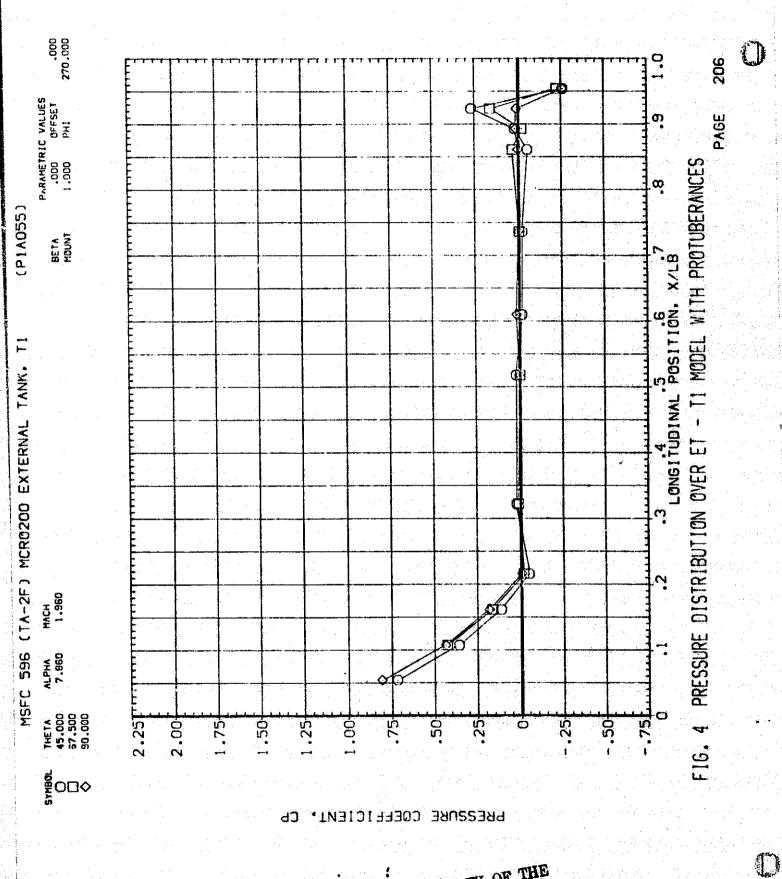
- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



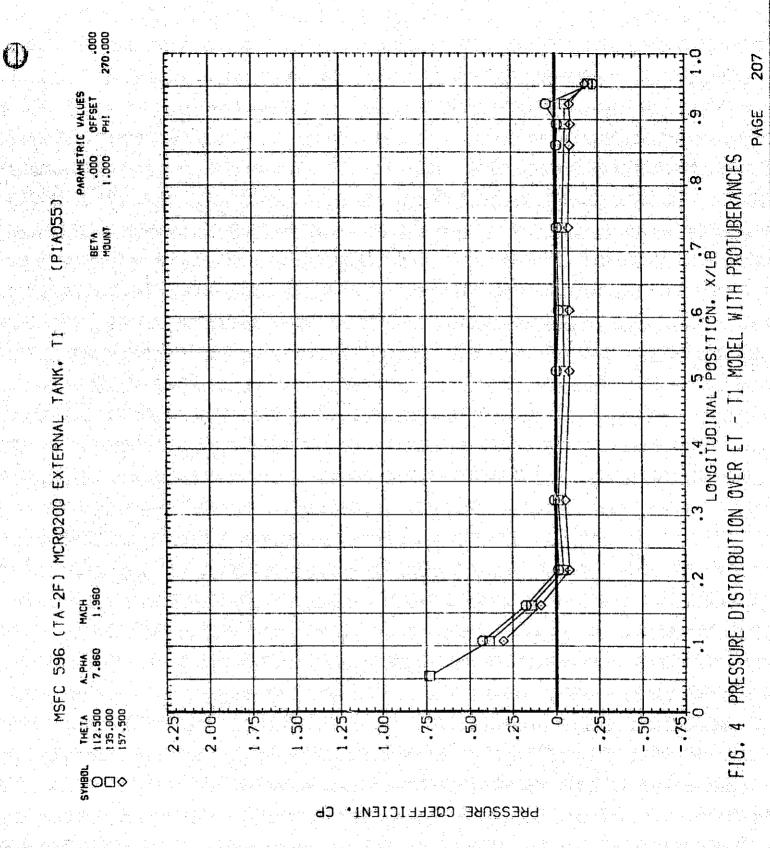
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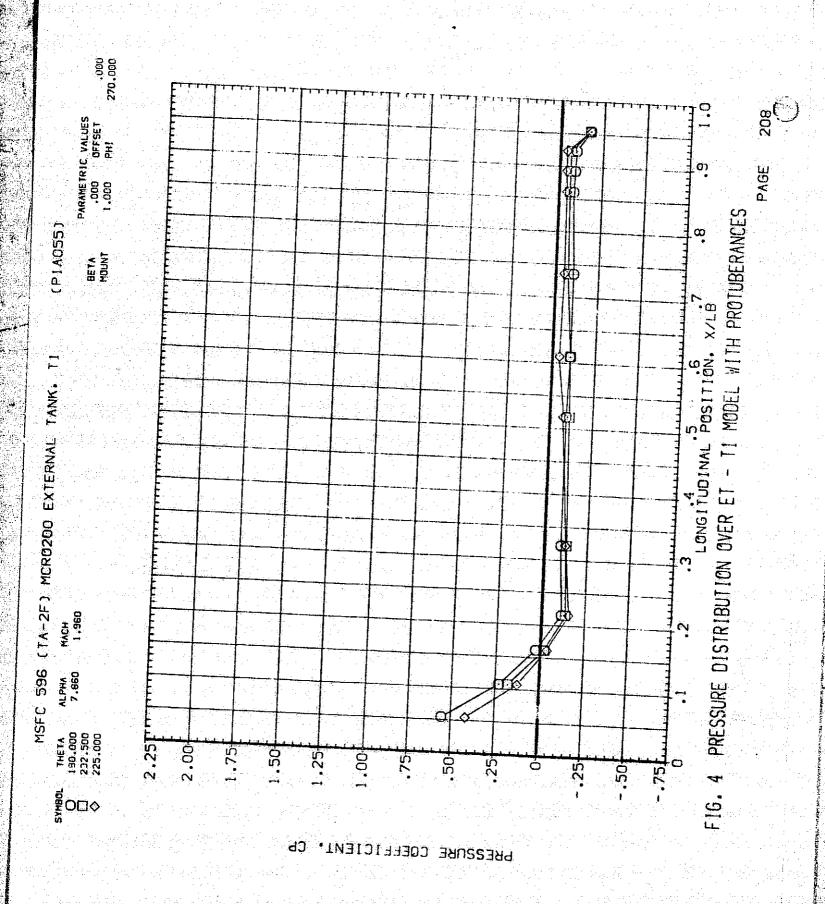
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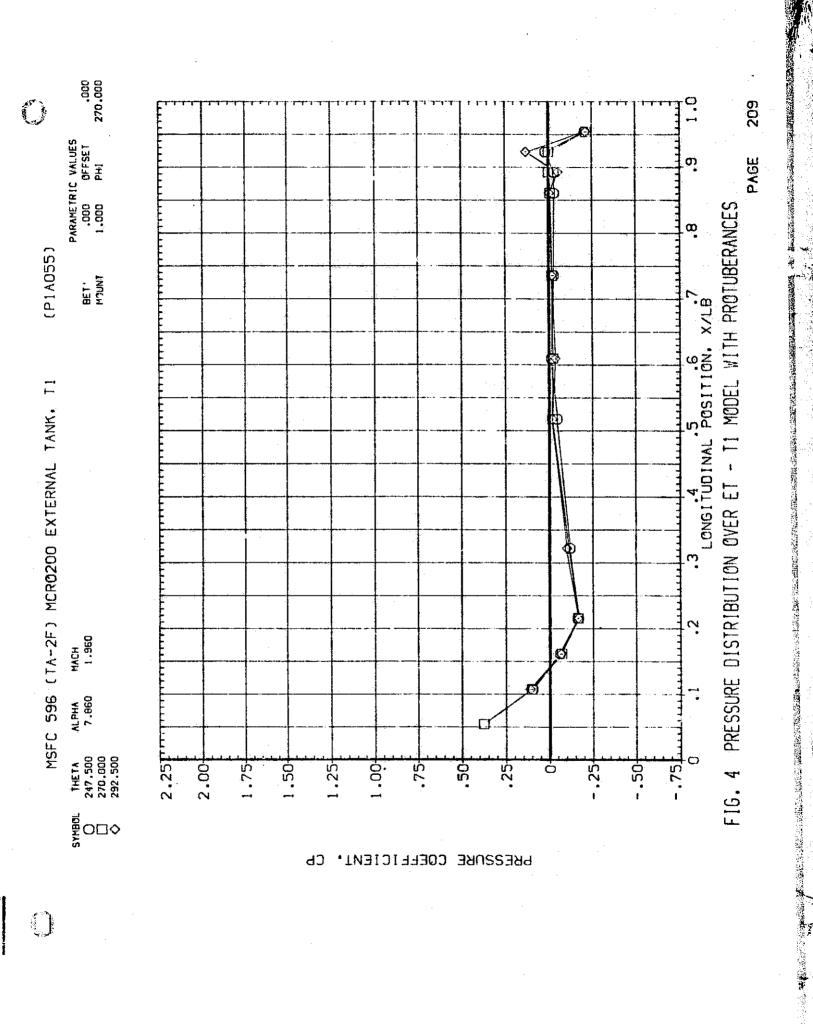


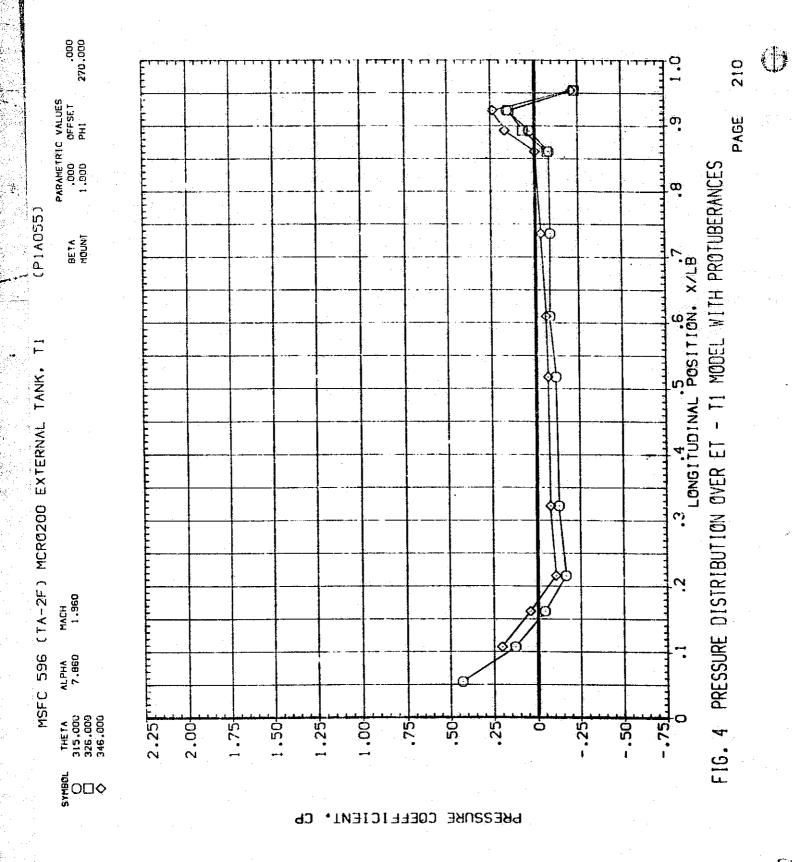


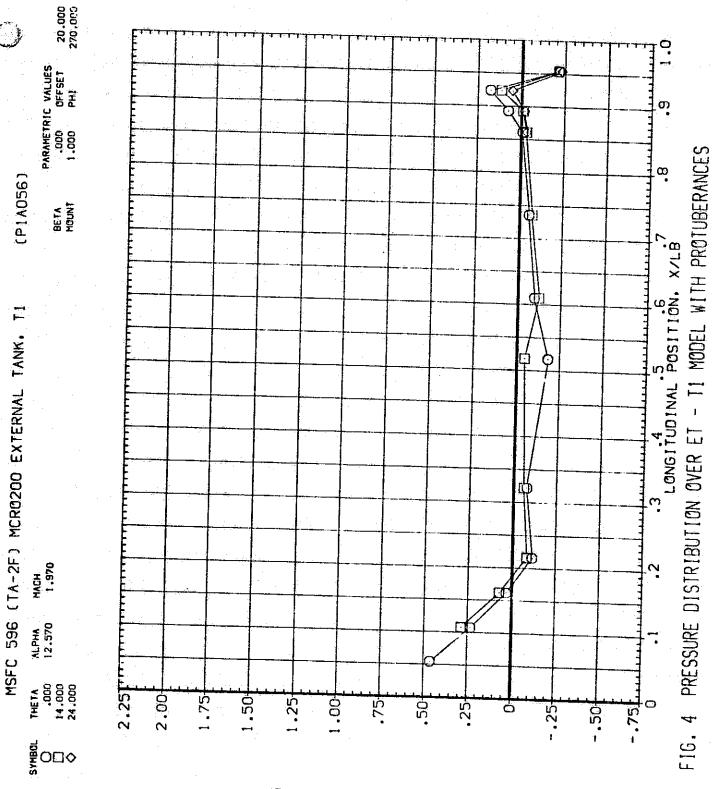
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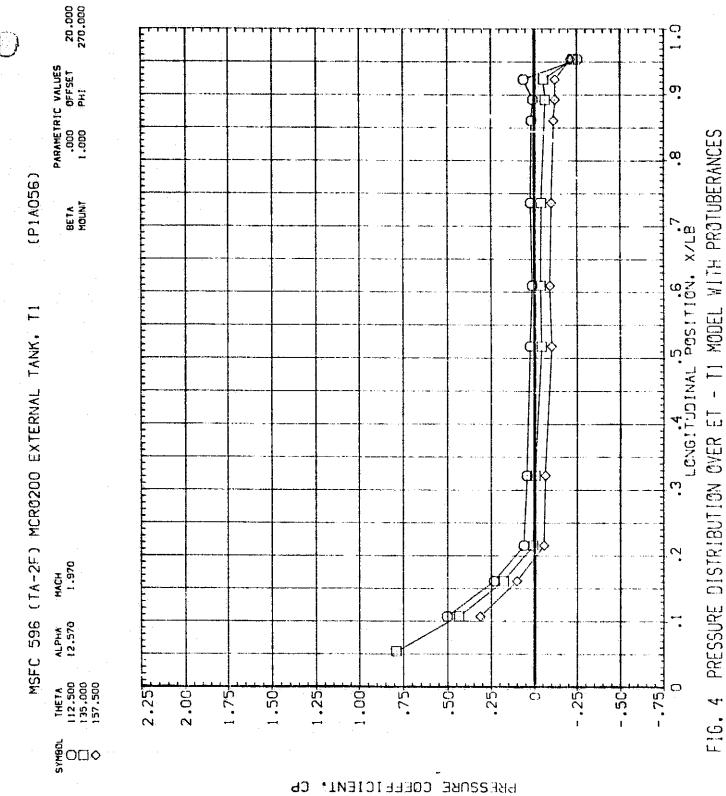


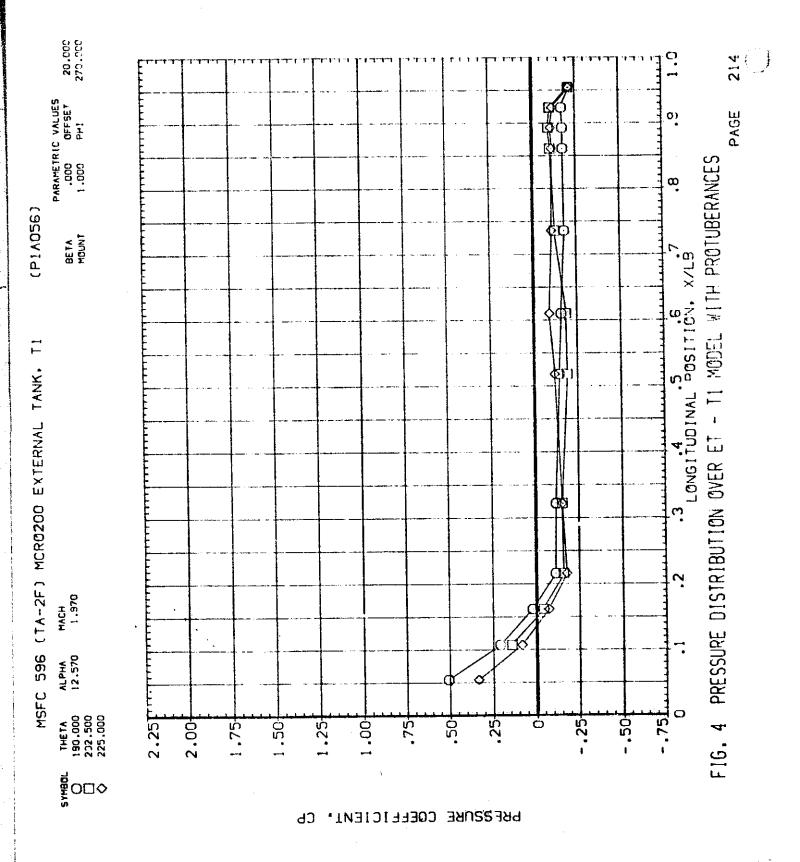


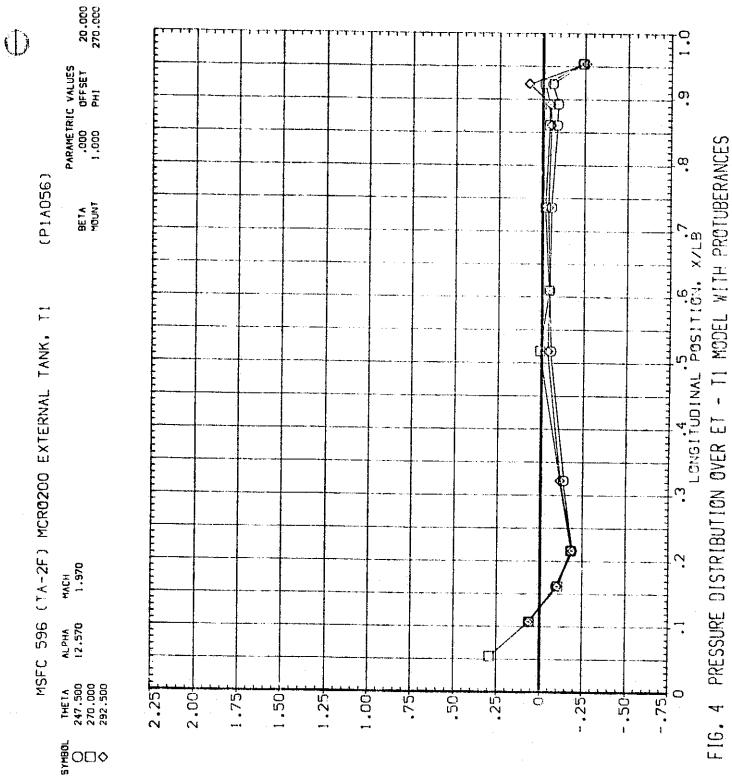


PRESSURE COEFFICIENT, CP

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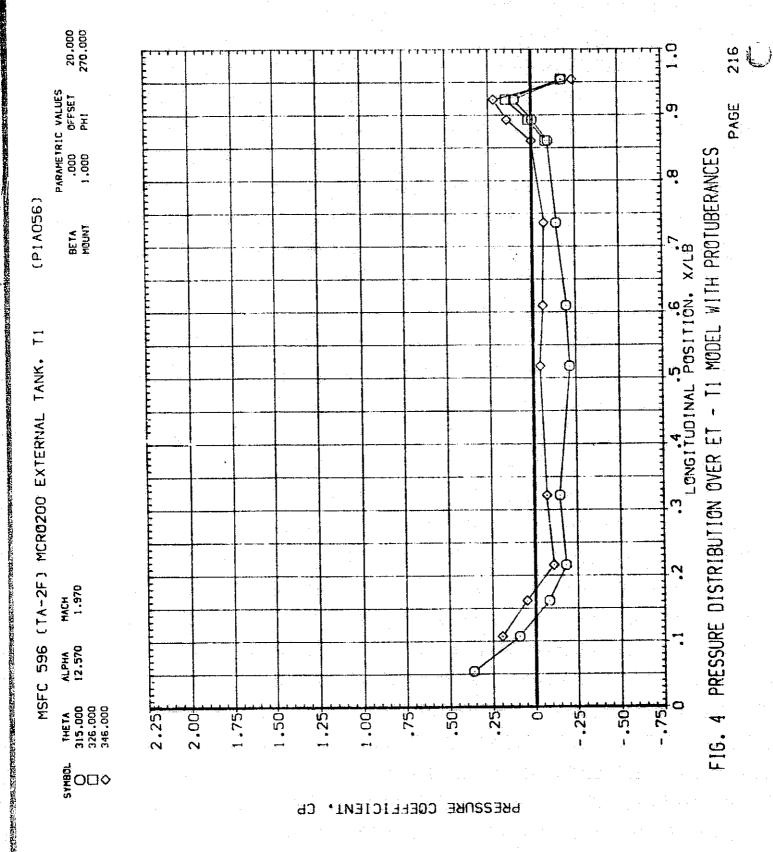






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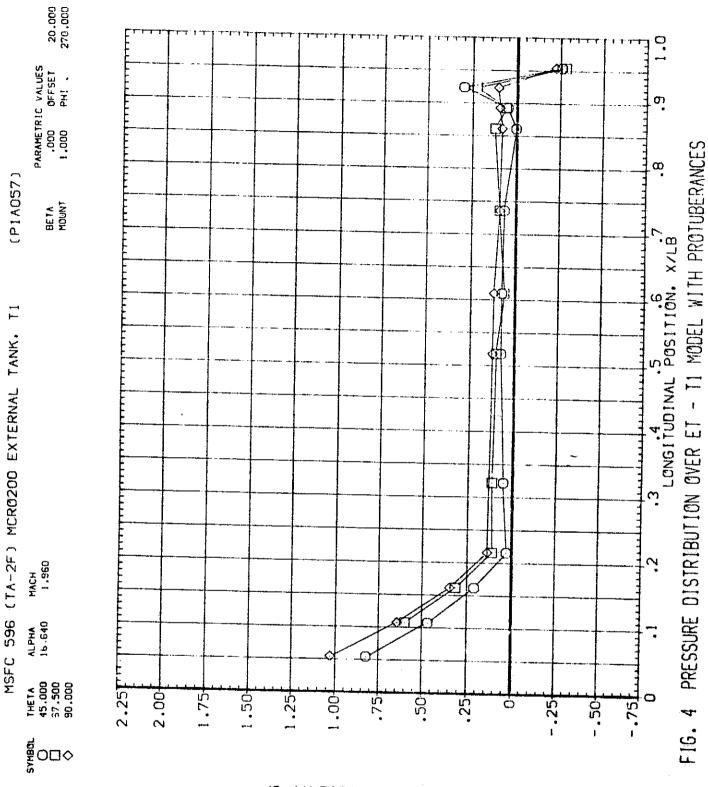
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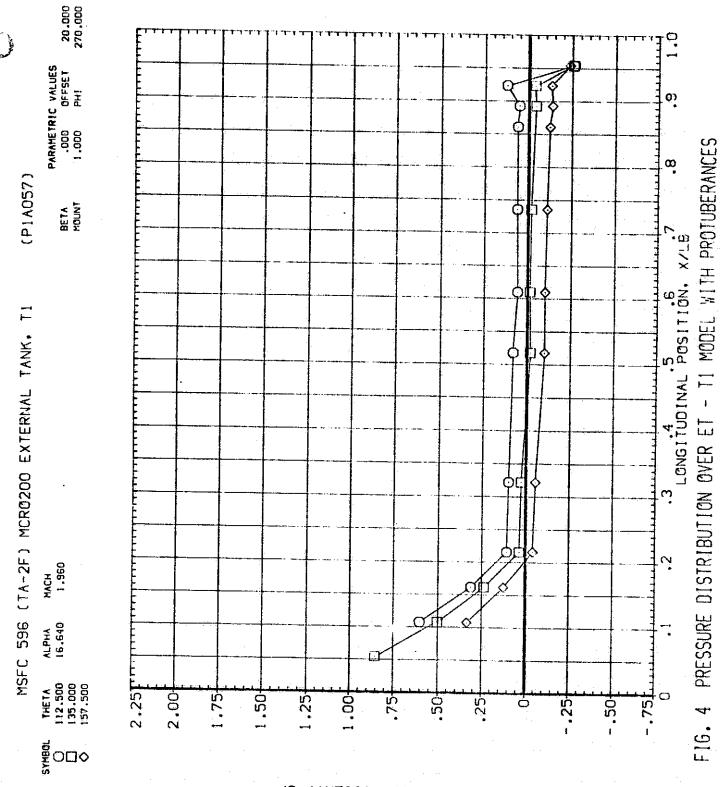
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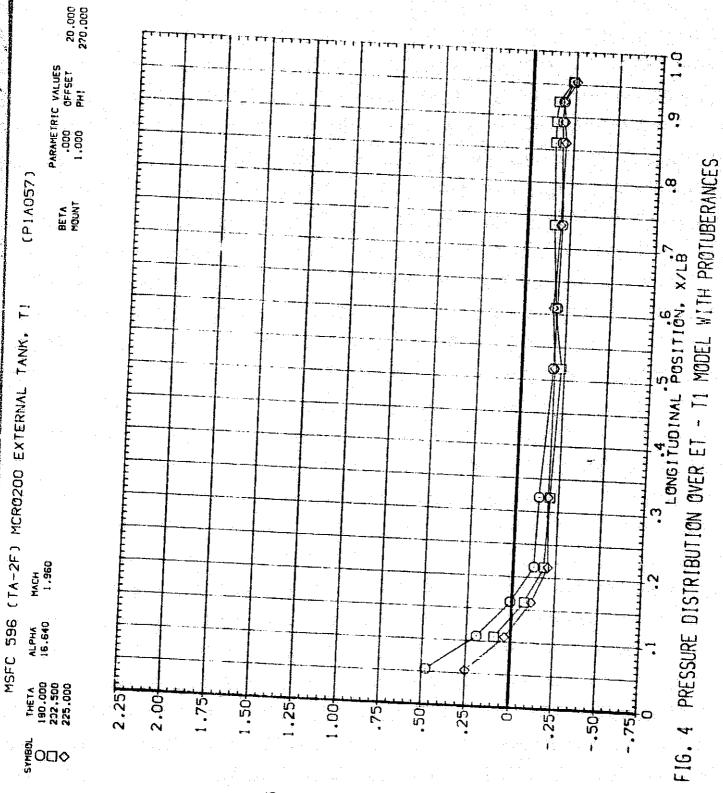


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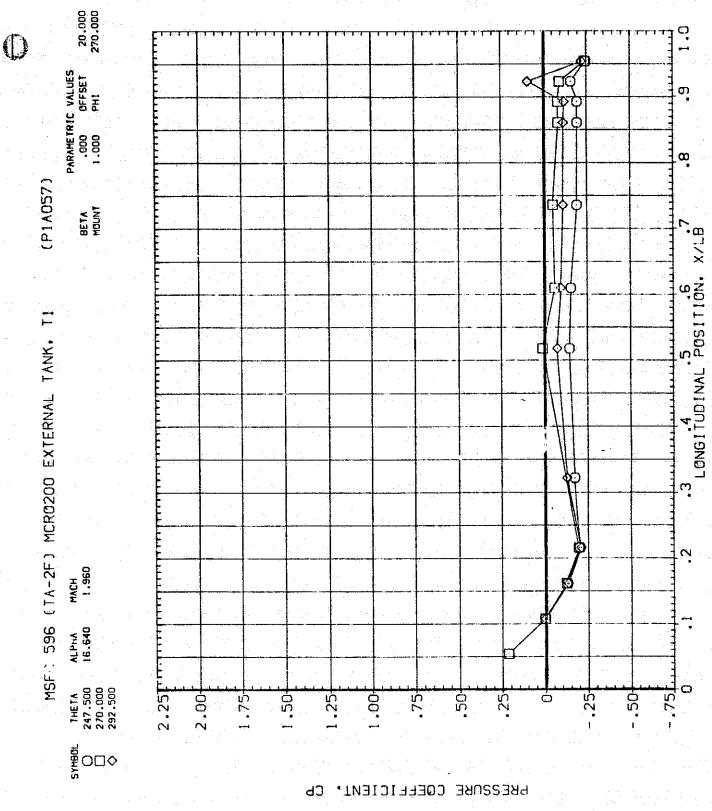
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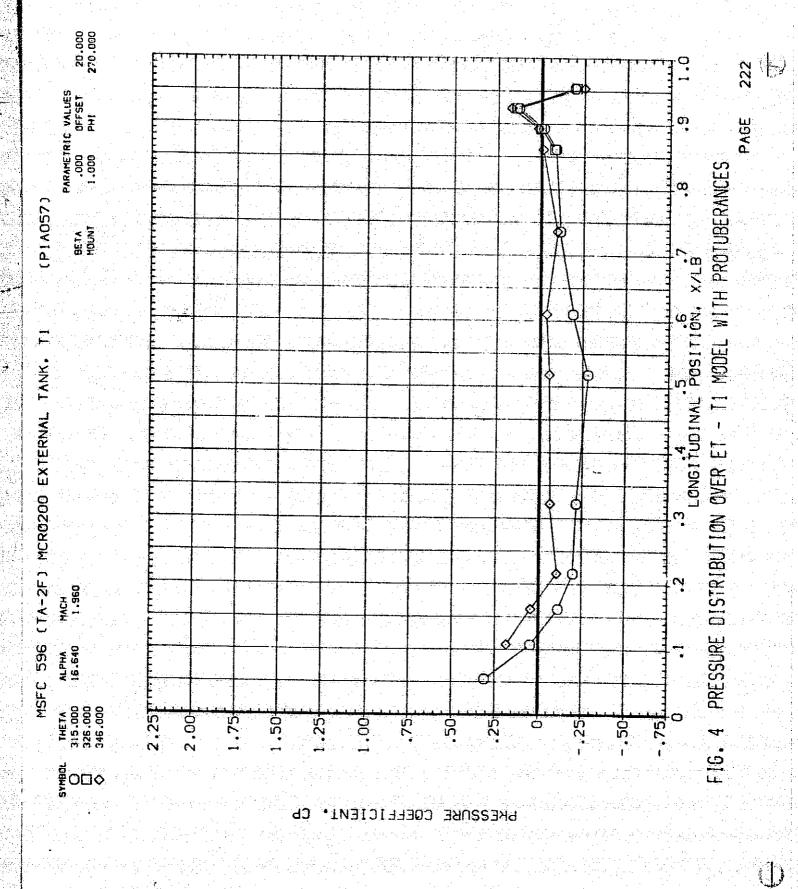
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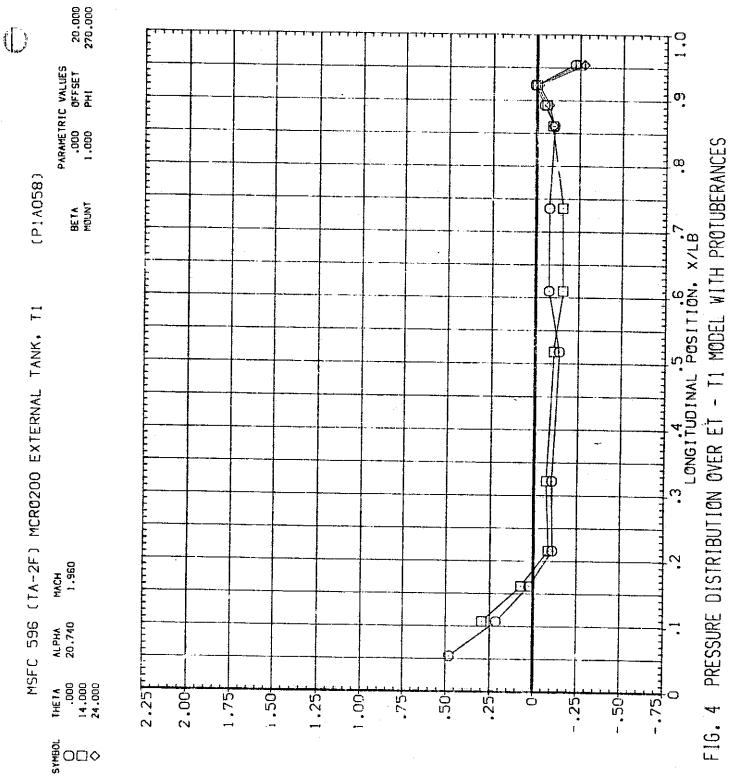
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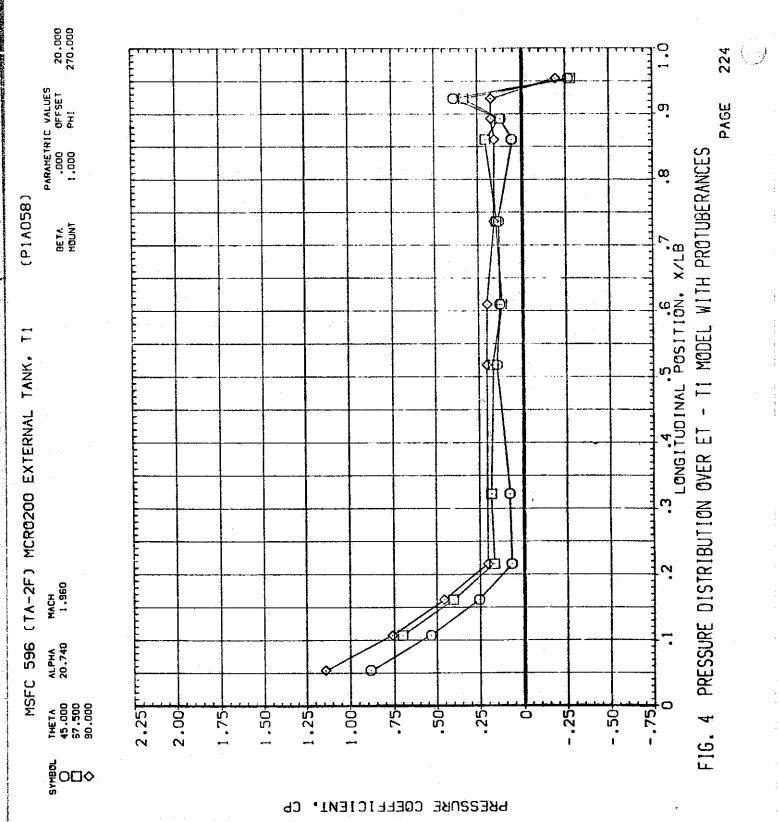
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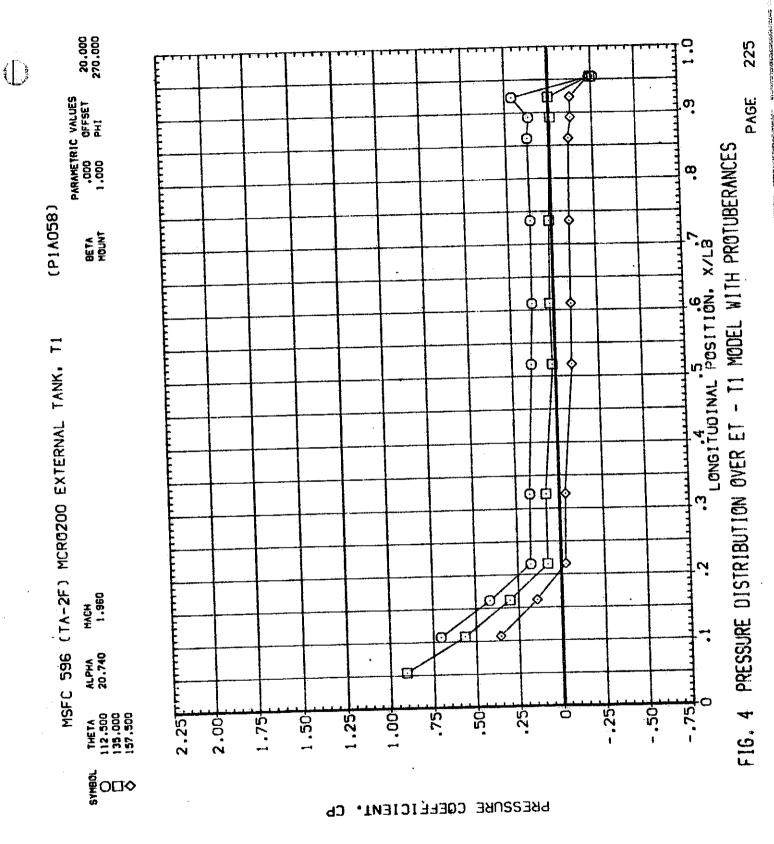


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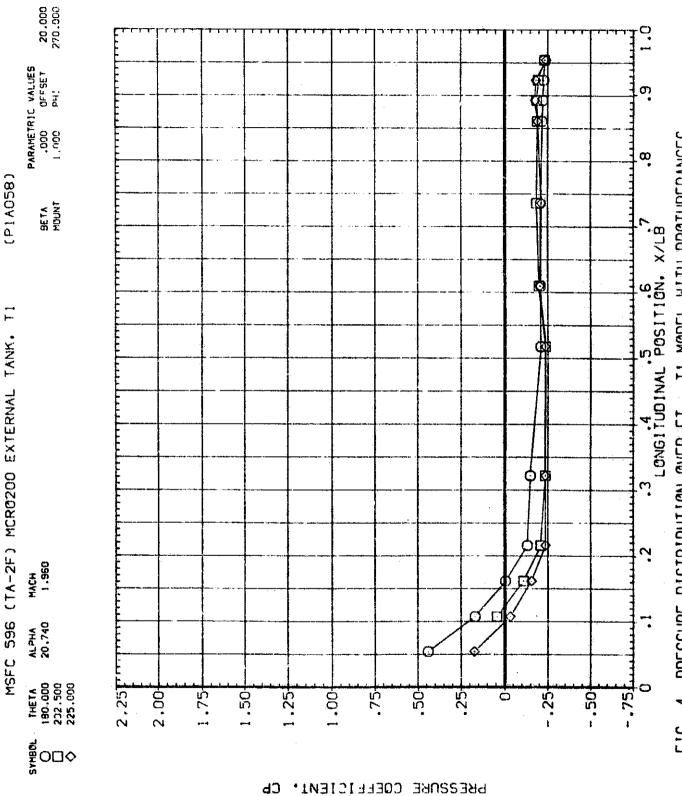


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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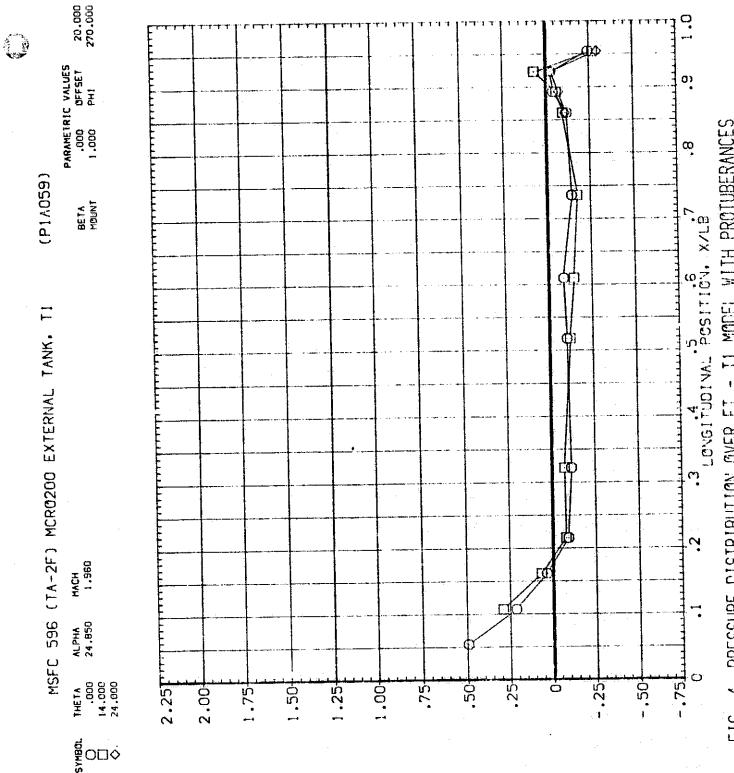
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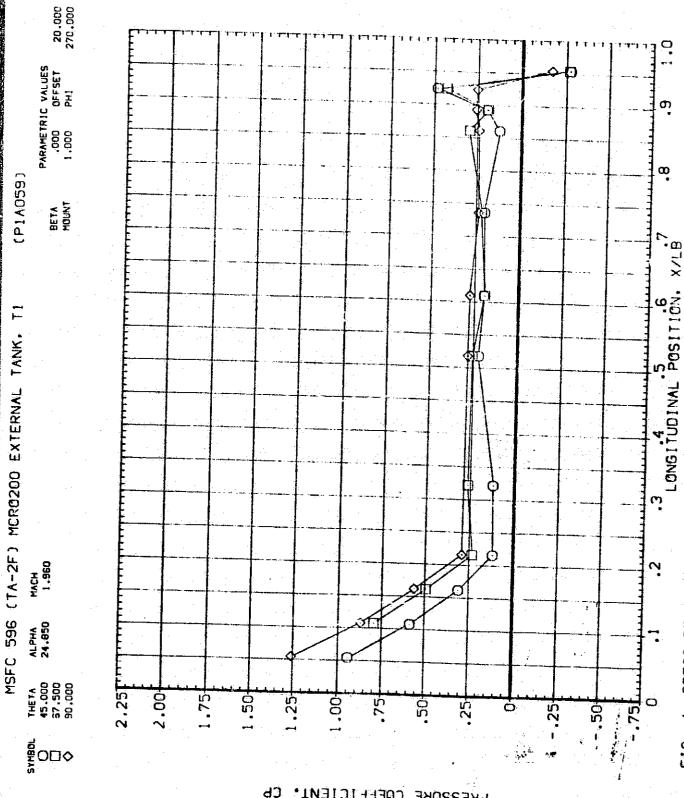


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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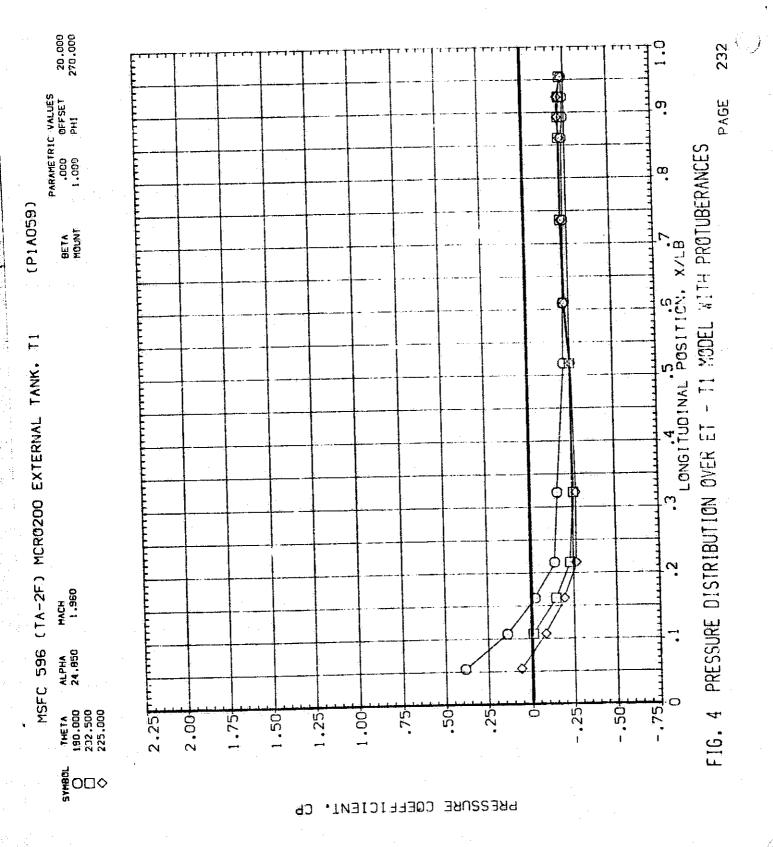
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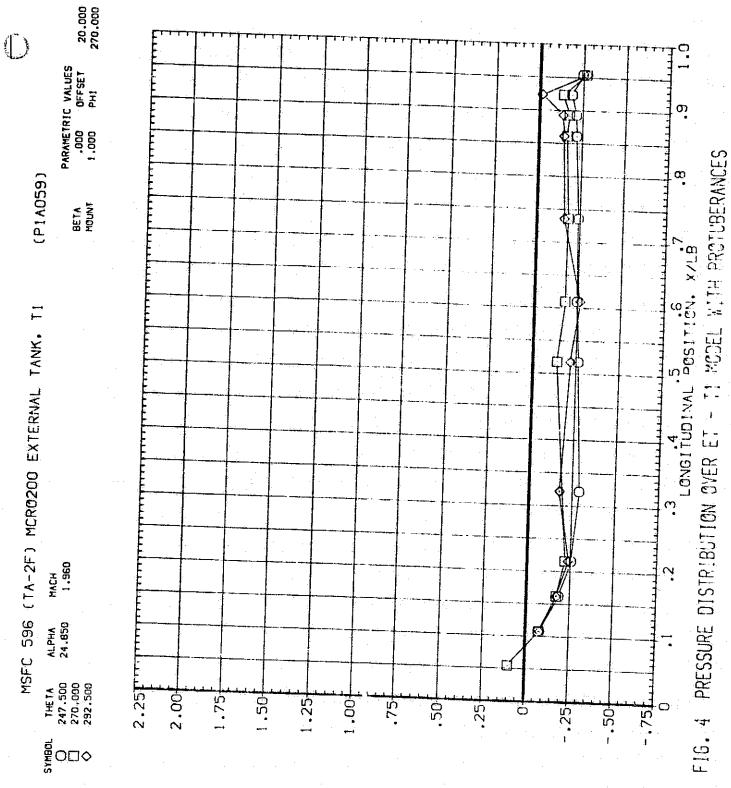
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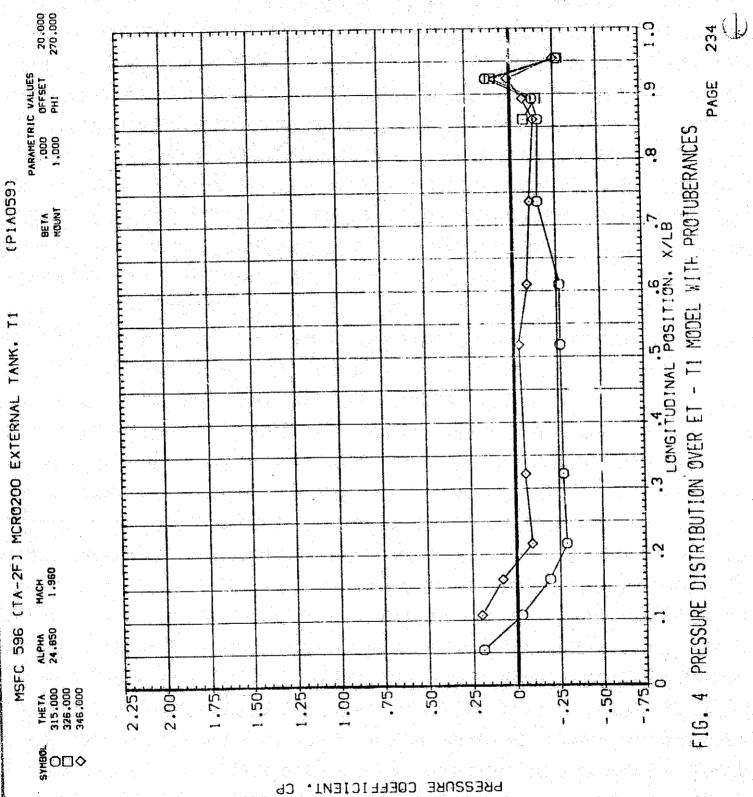
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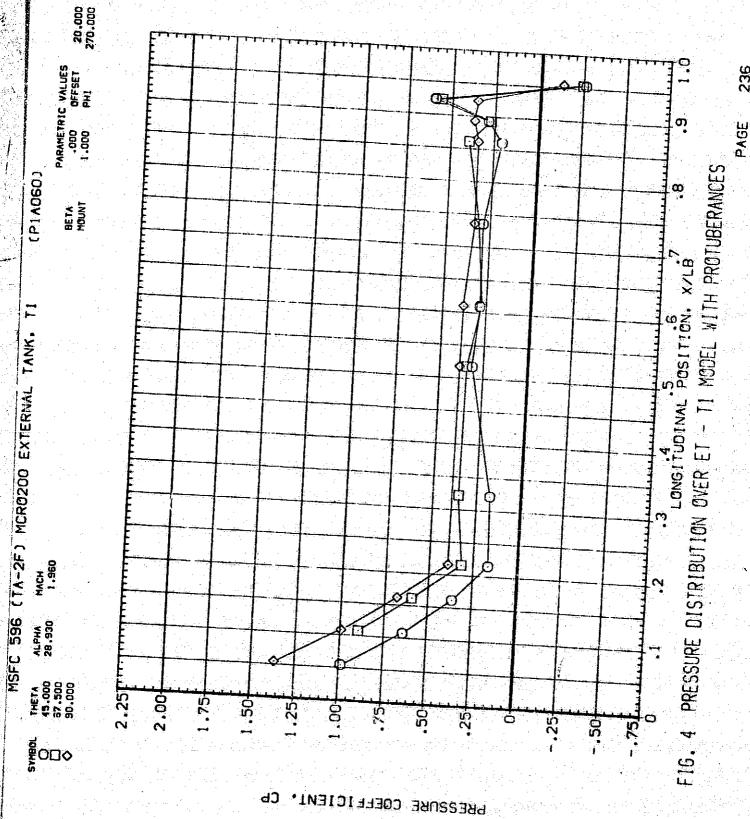
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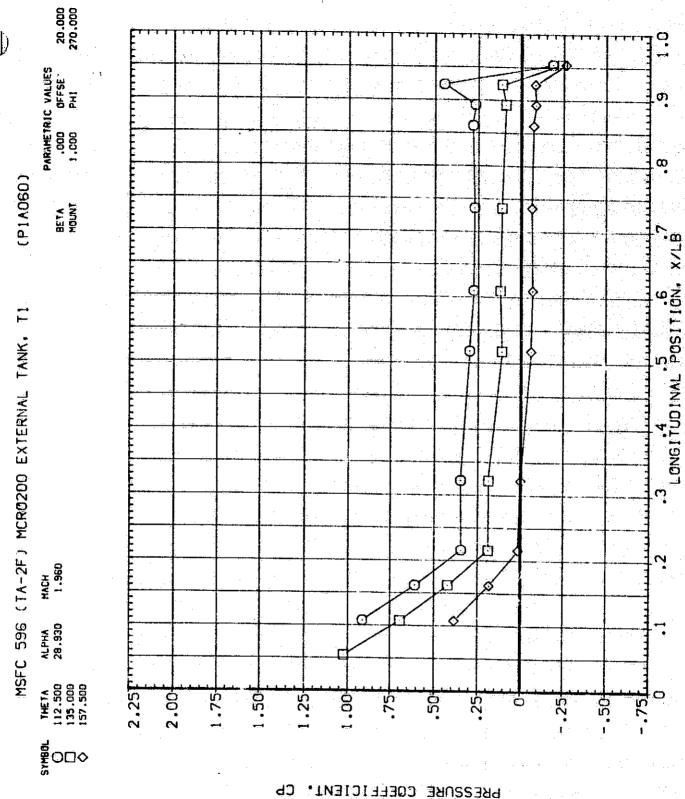


FIG. 4 PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES

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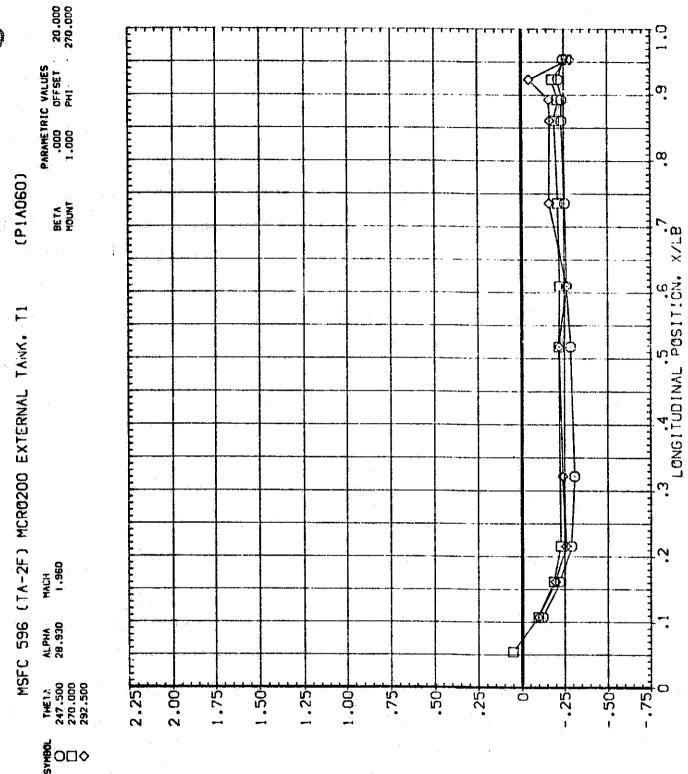
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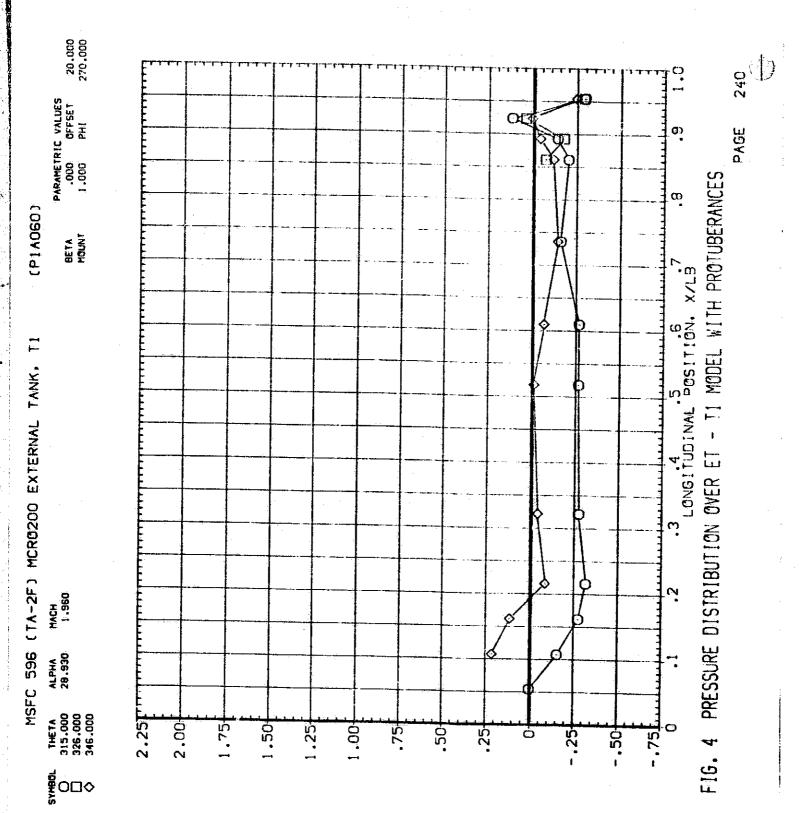
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FIG. 4 PRESSURE DISTRIBUTION OVER

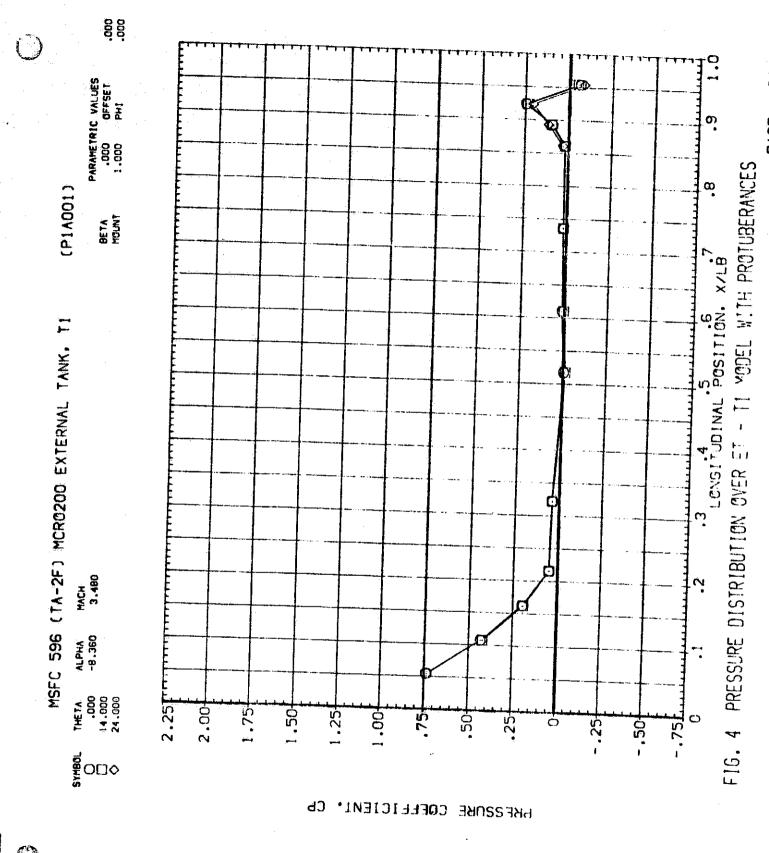


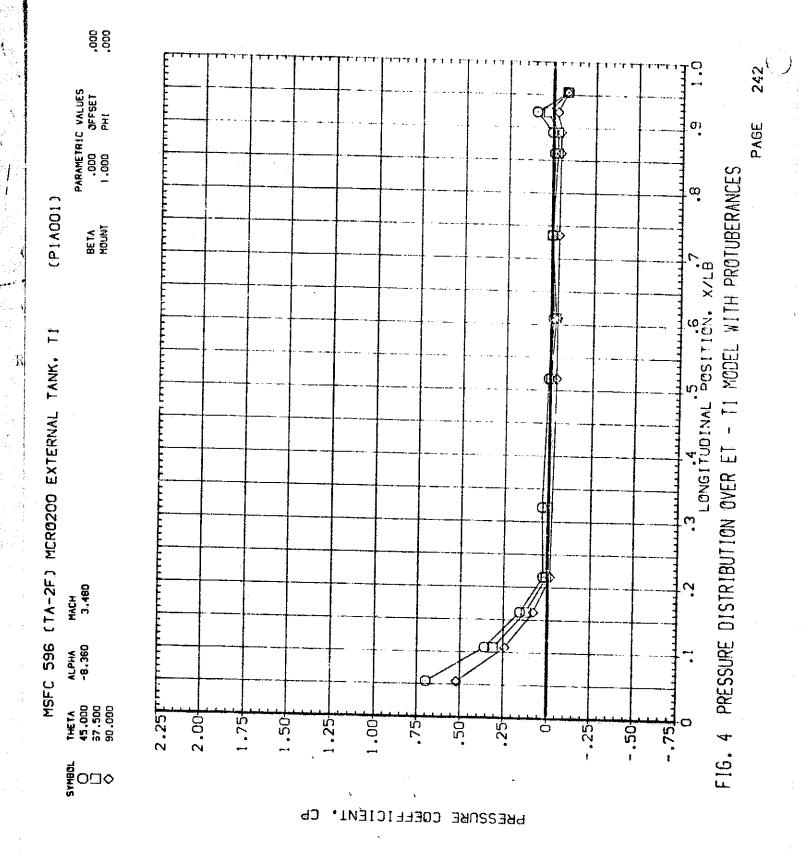
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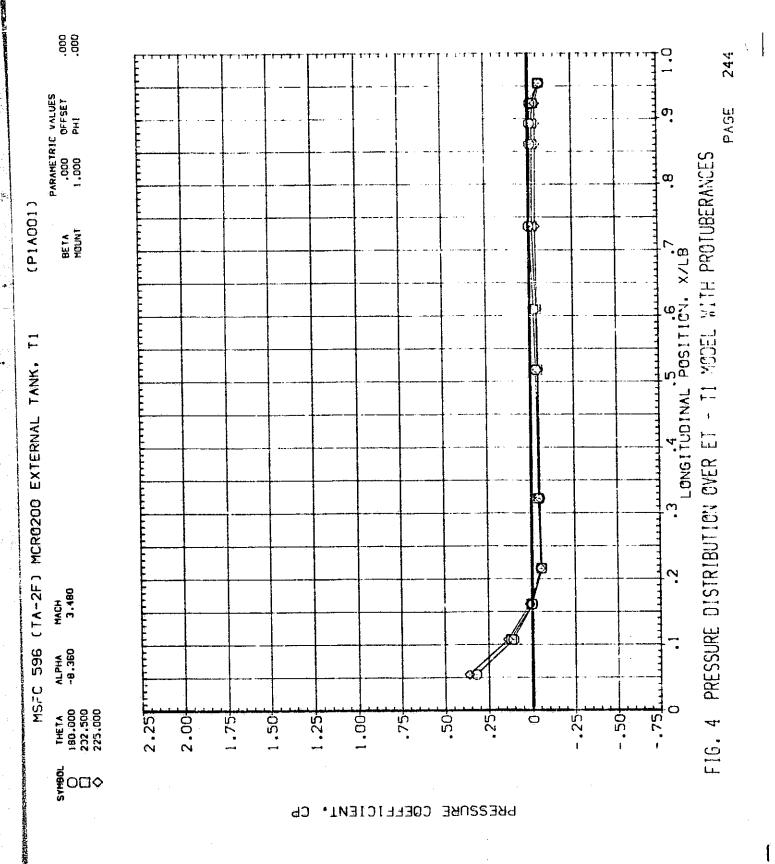
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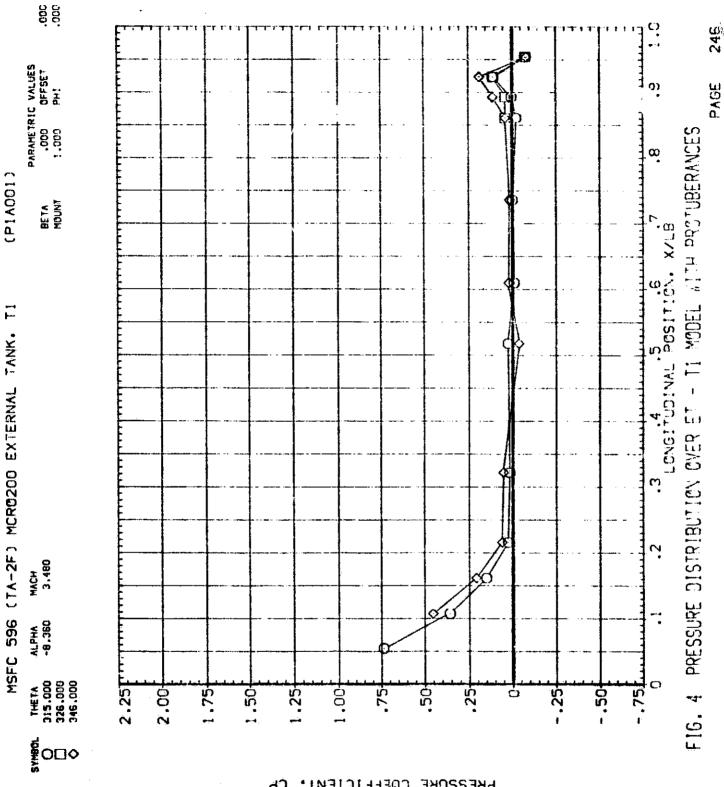
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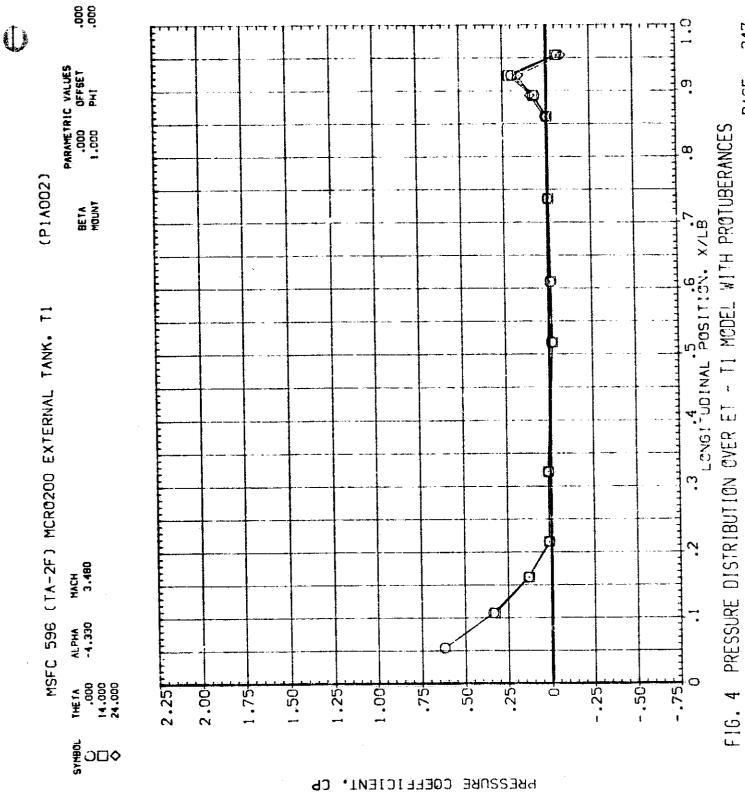
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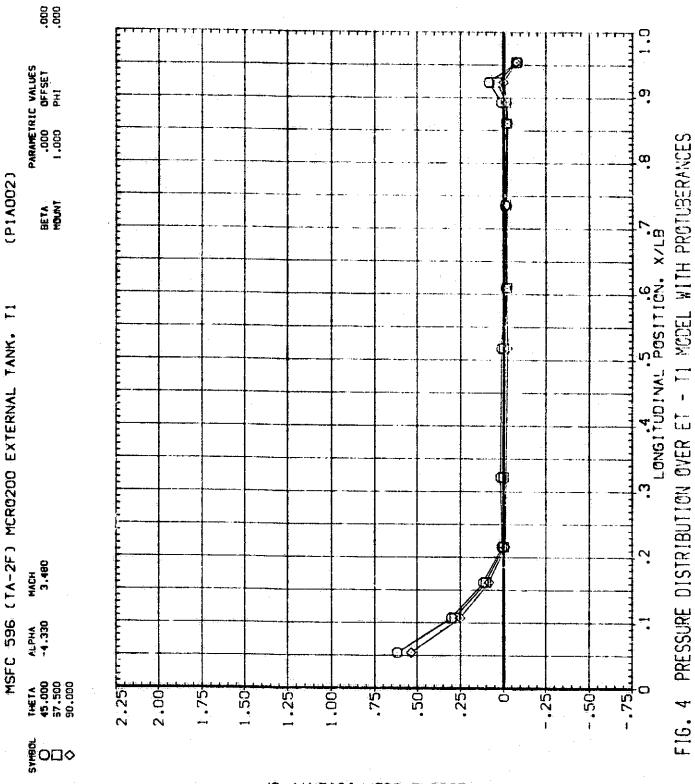


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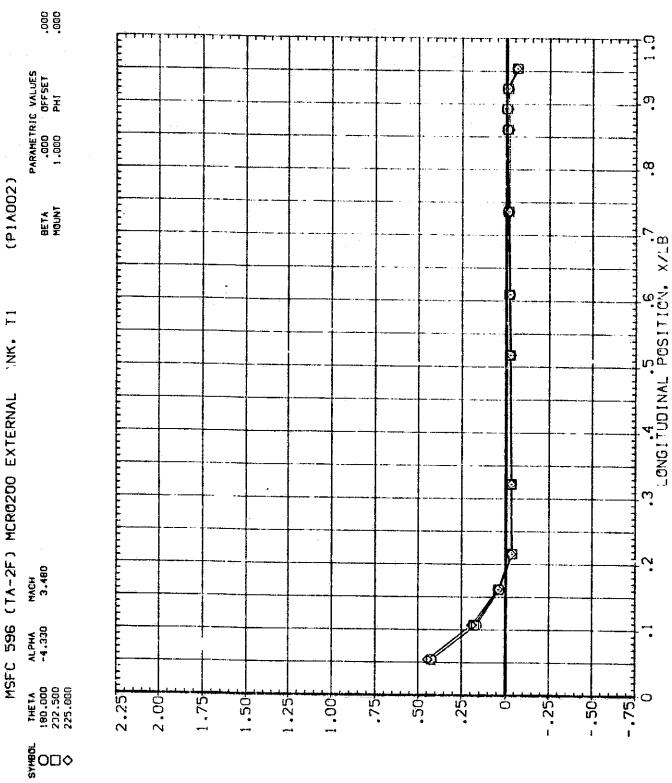




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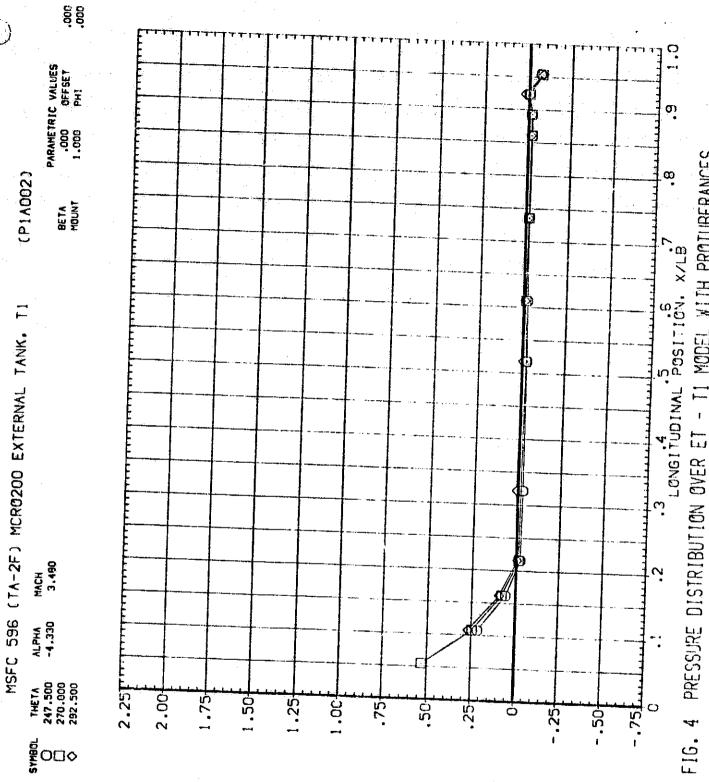
FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

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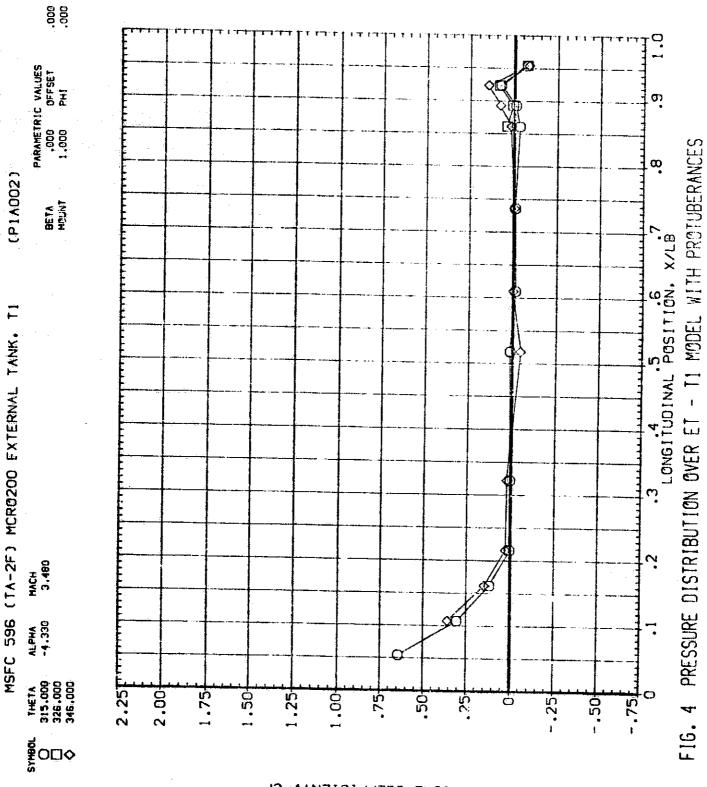
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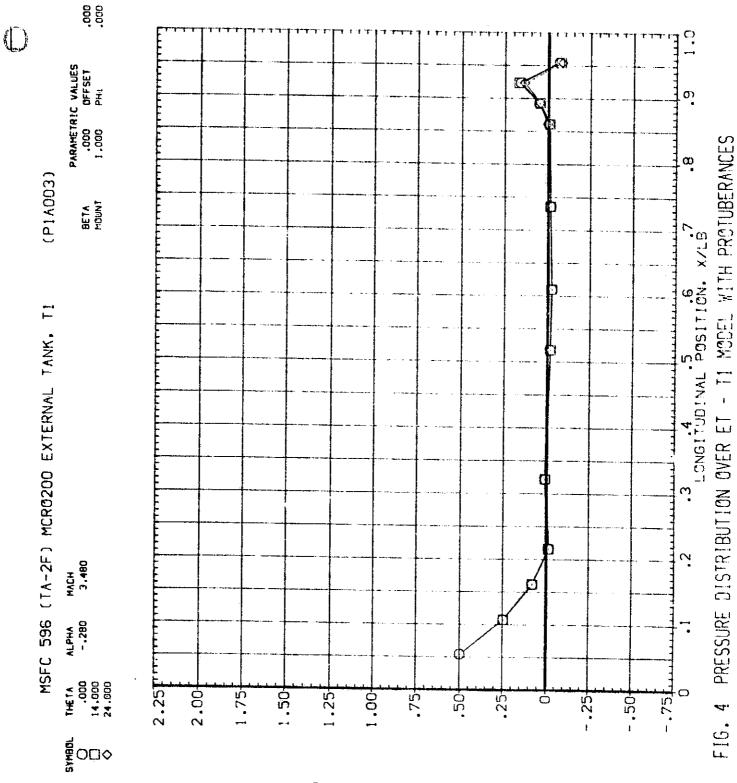


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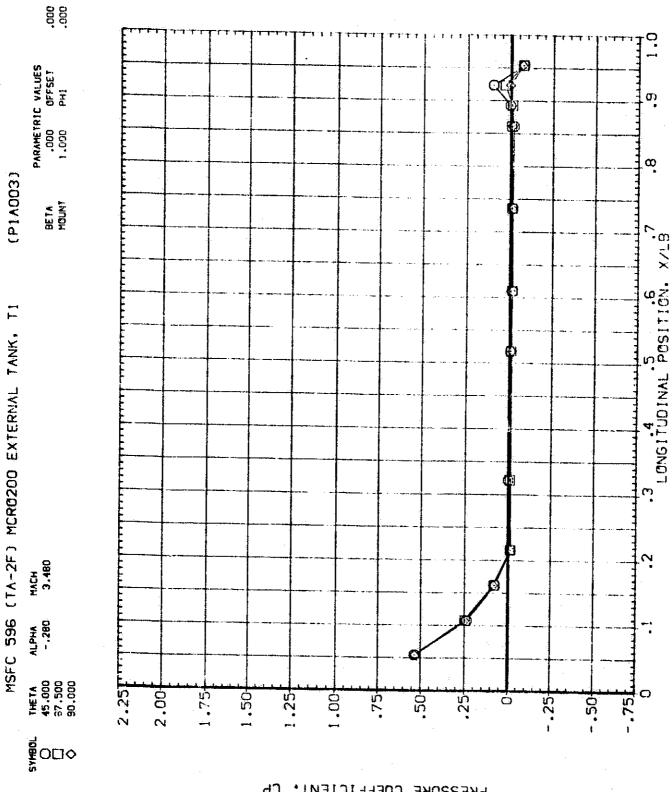
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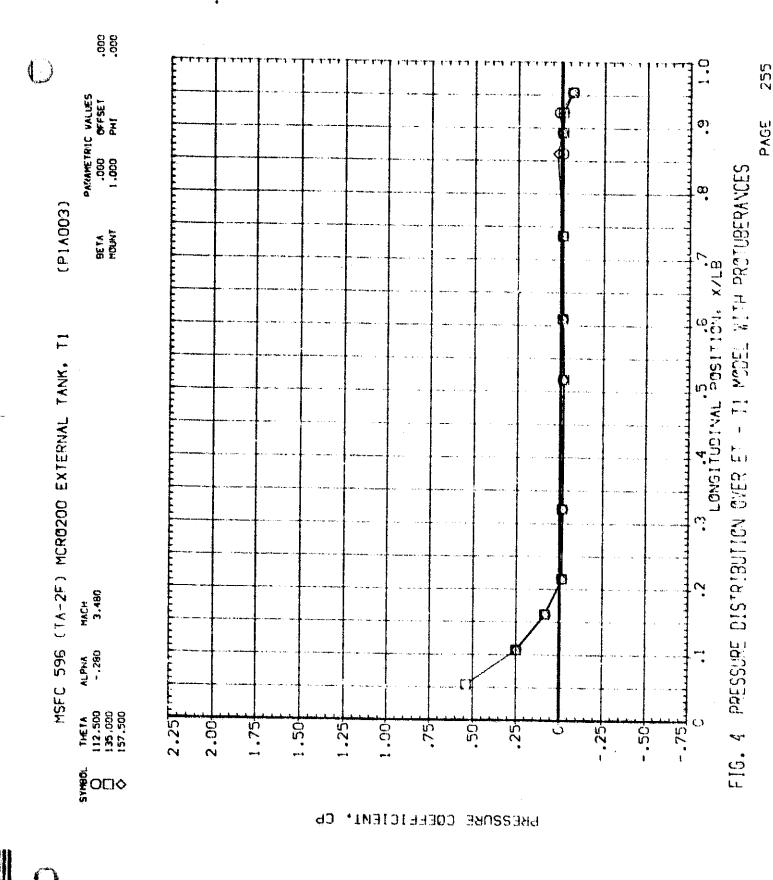
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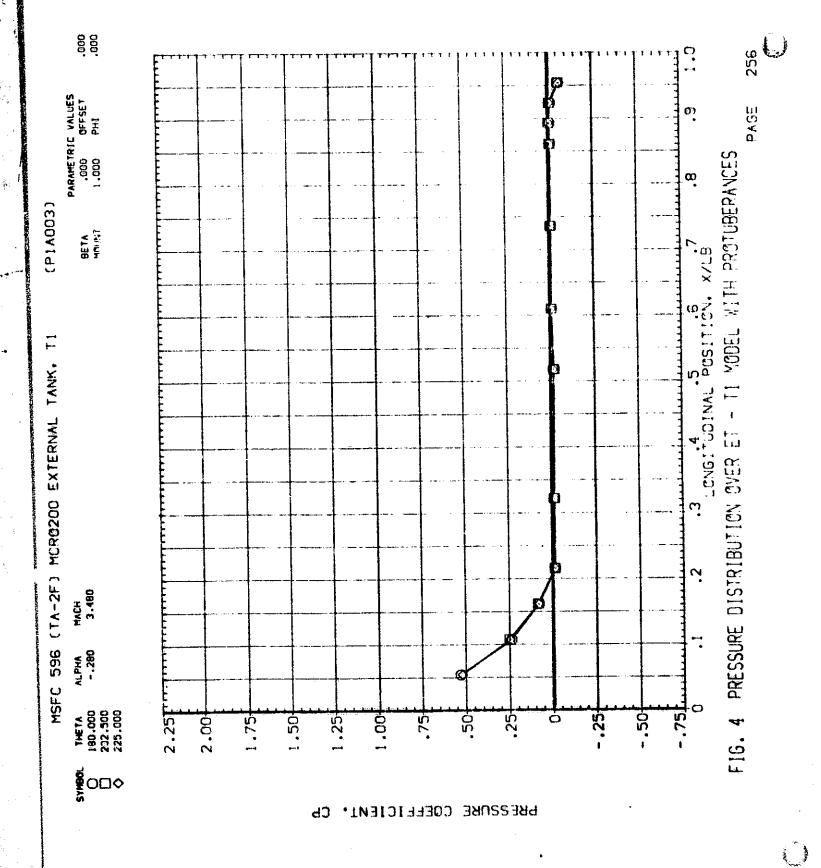
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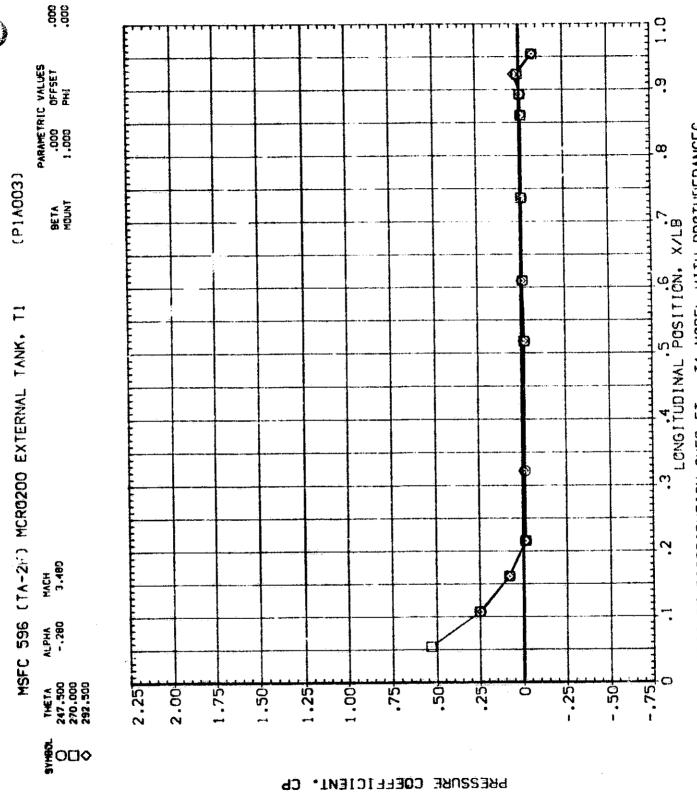
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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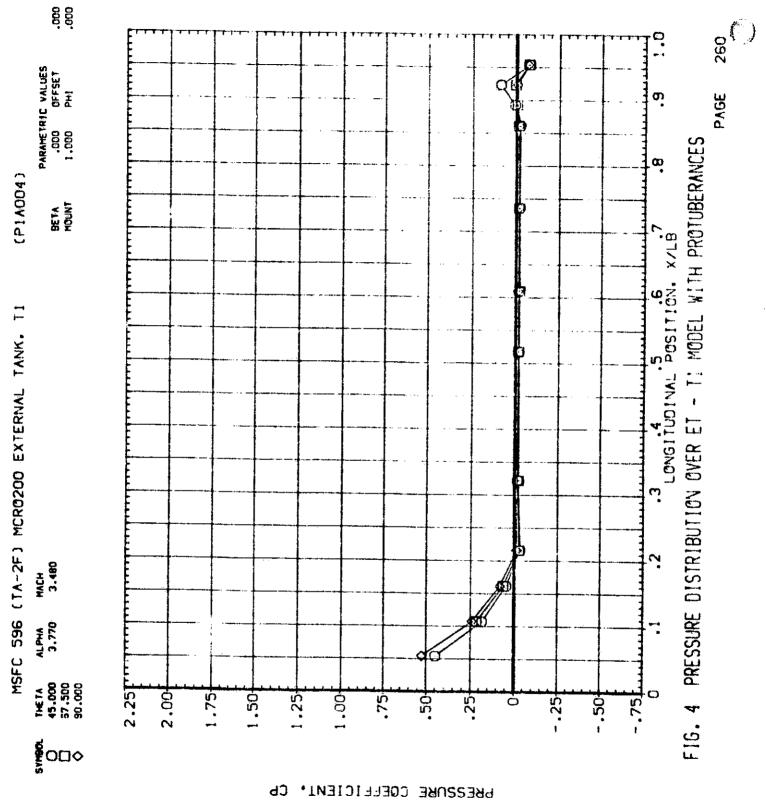
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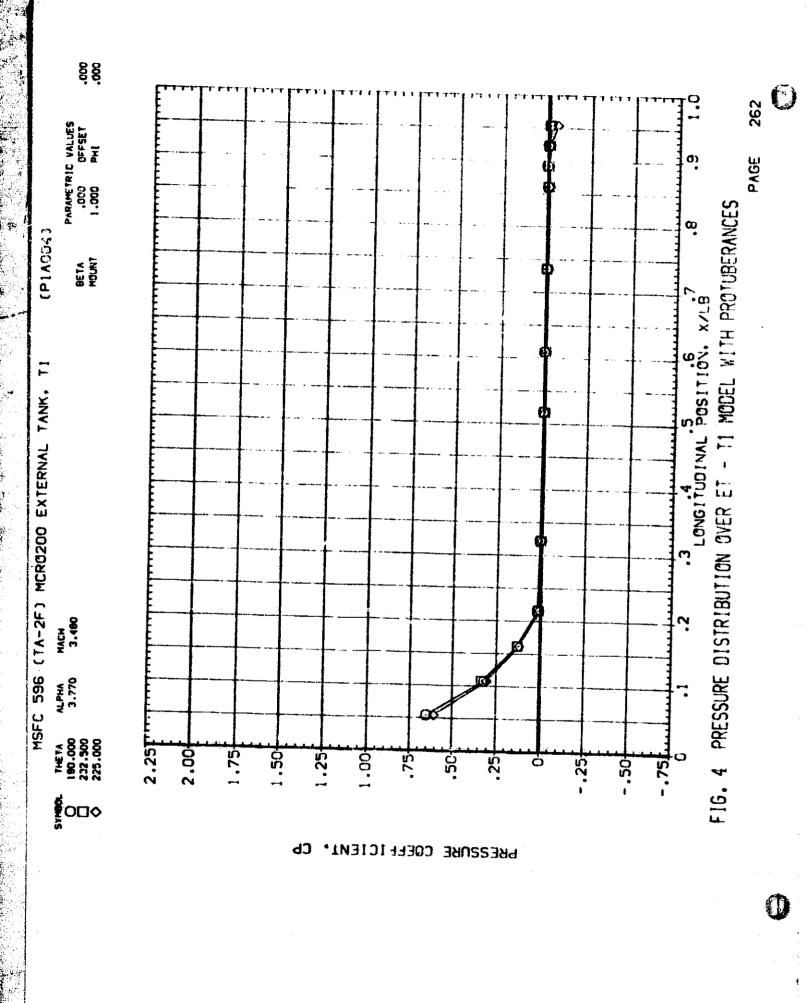
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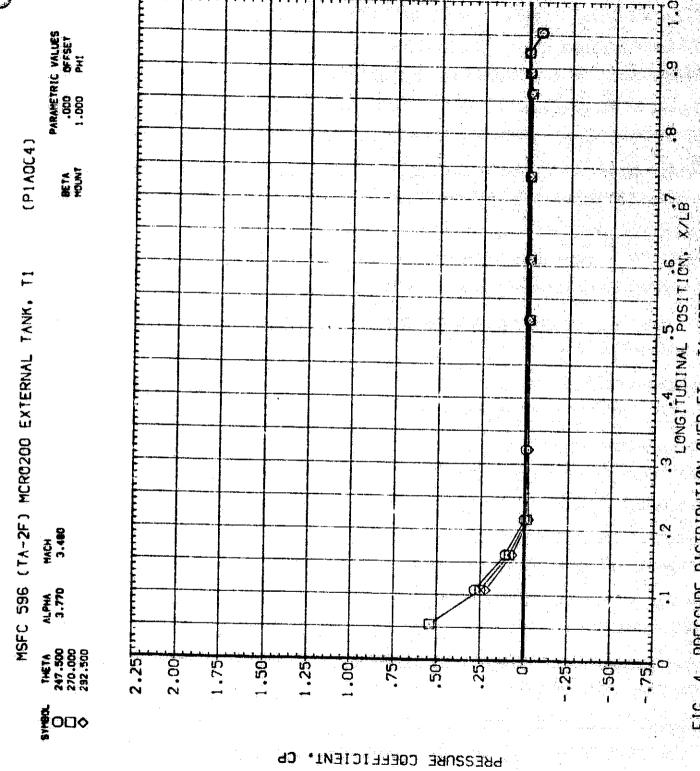
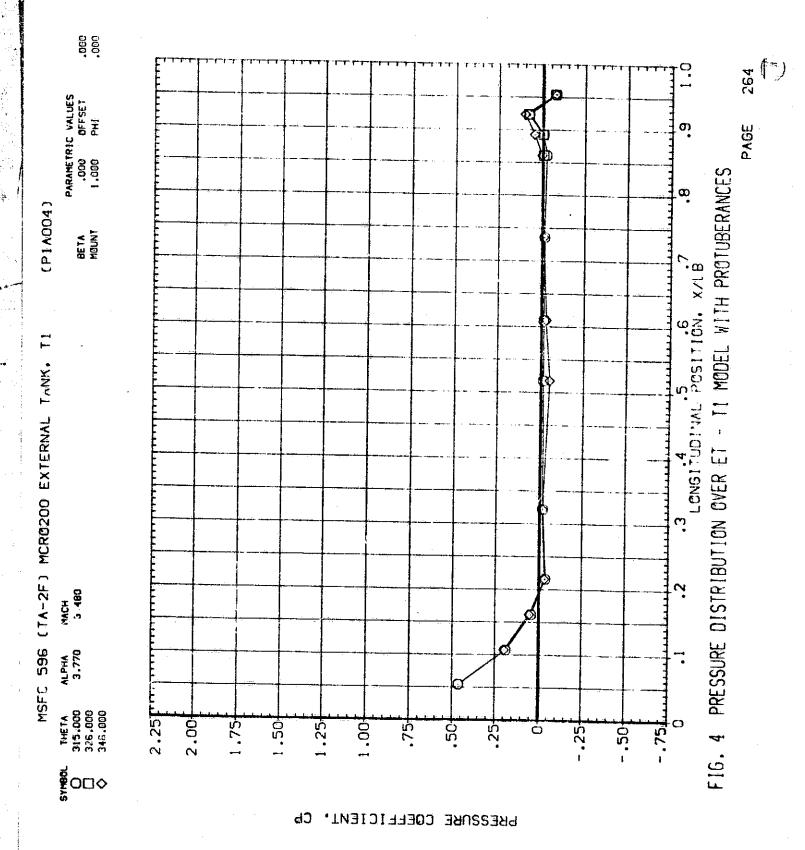


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



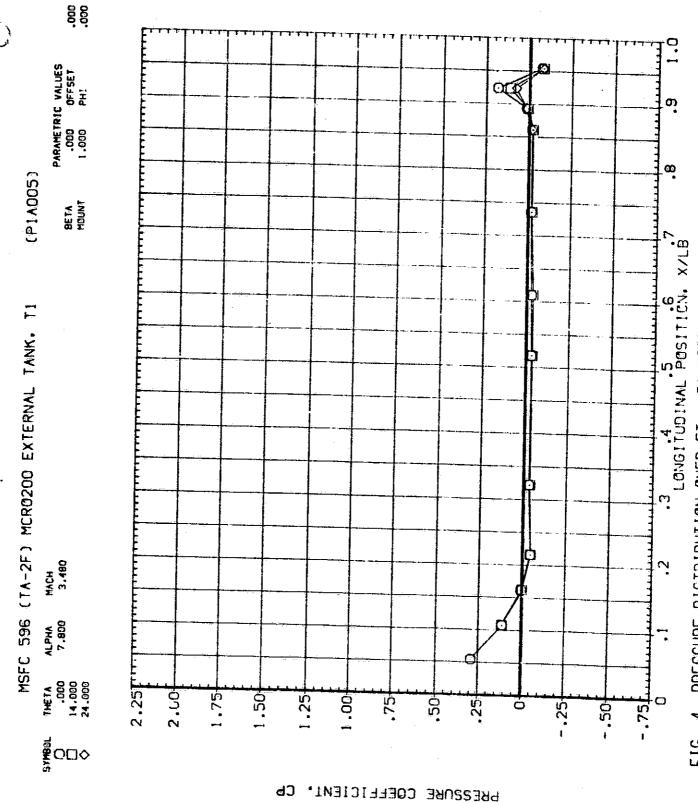
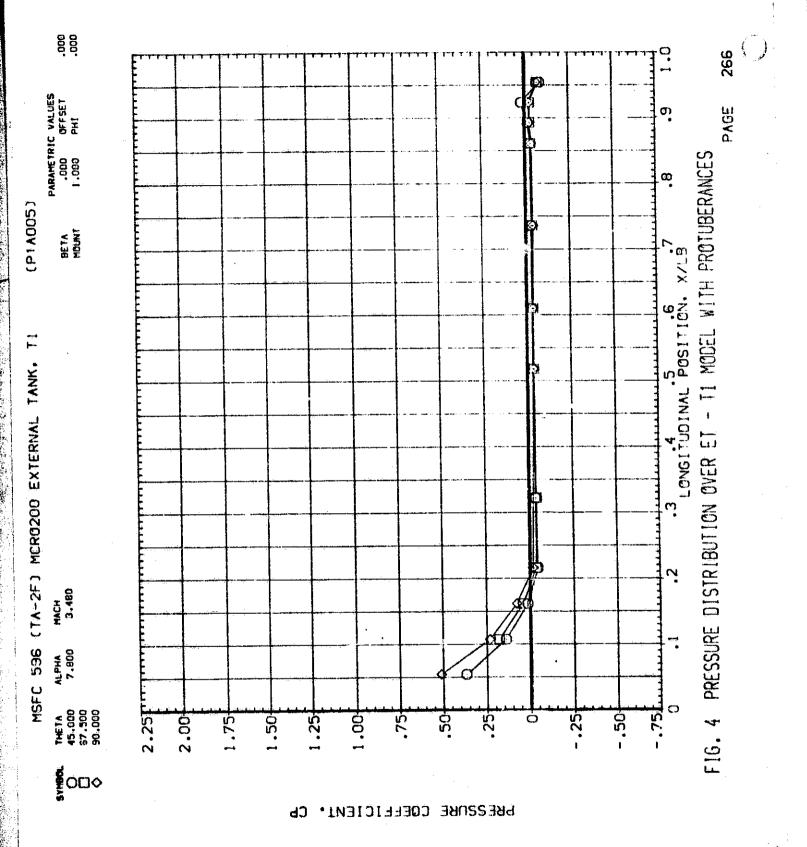


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



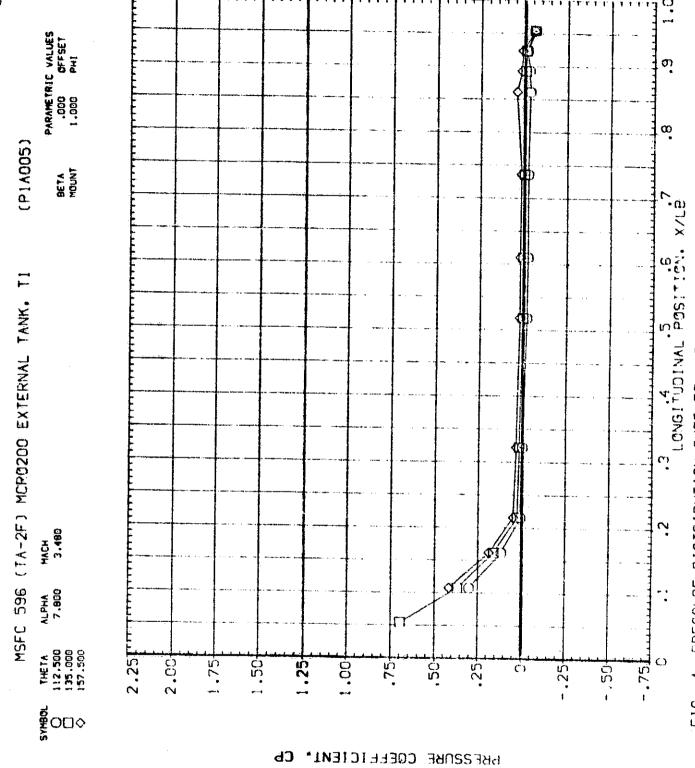
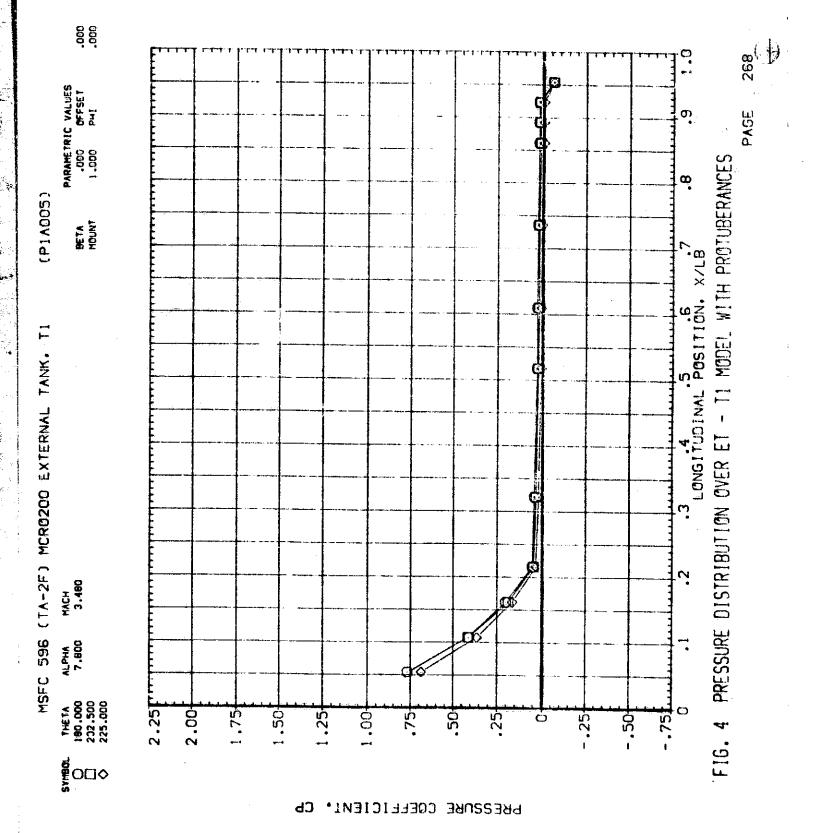
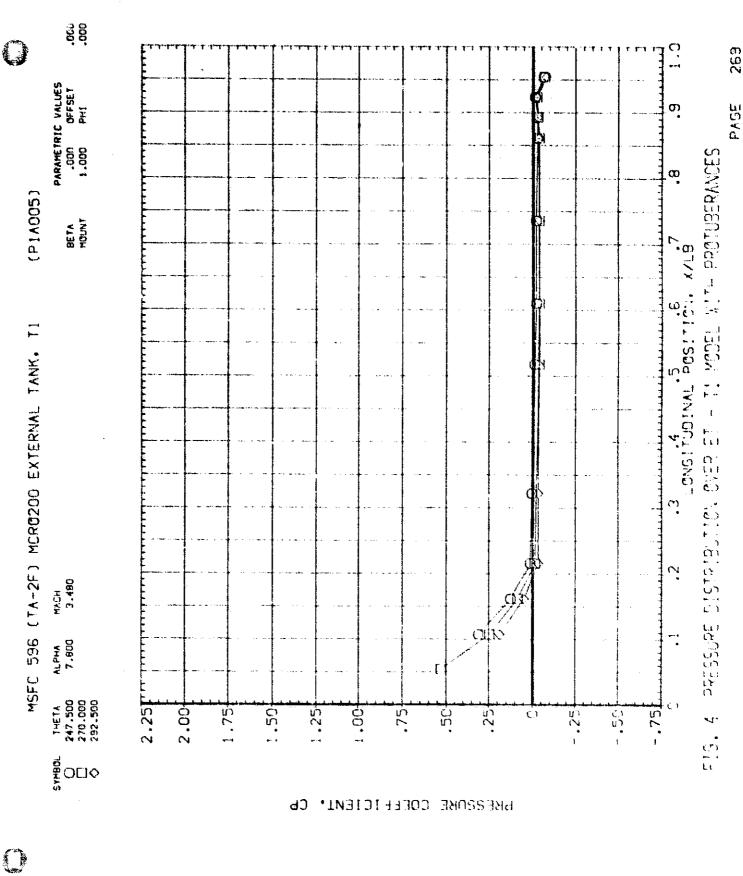


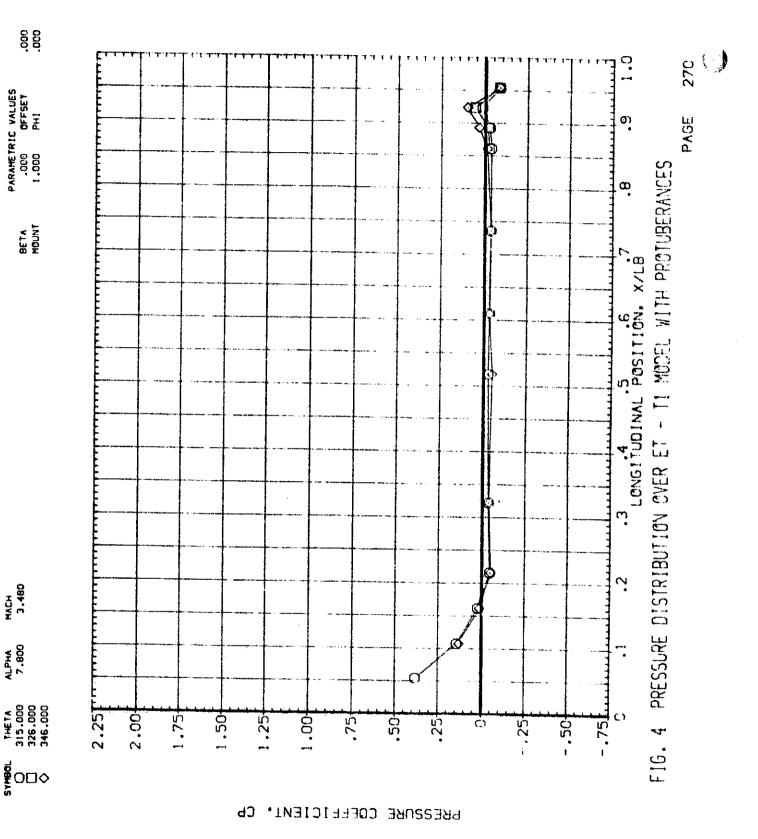
FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1 MACH 3.480 ALPHA 12.520 THETA .000 14.000 24.000

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

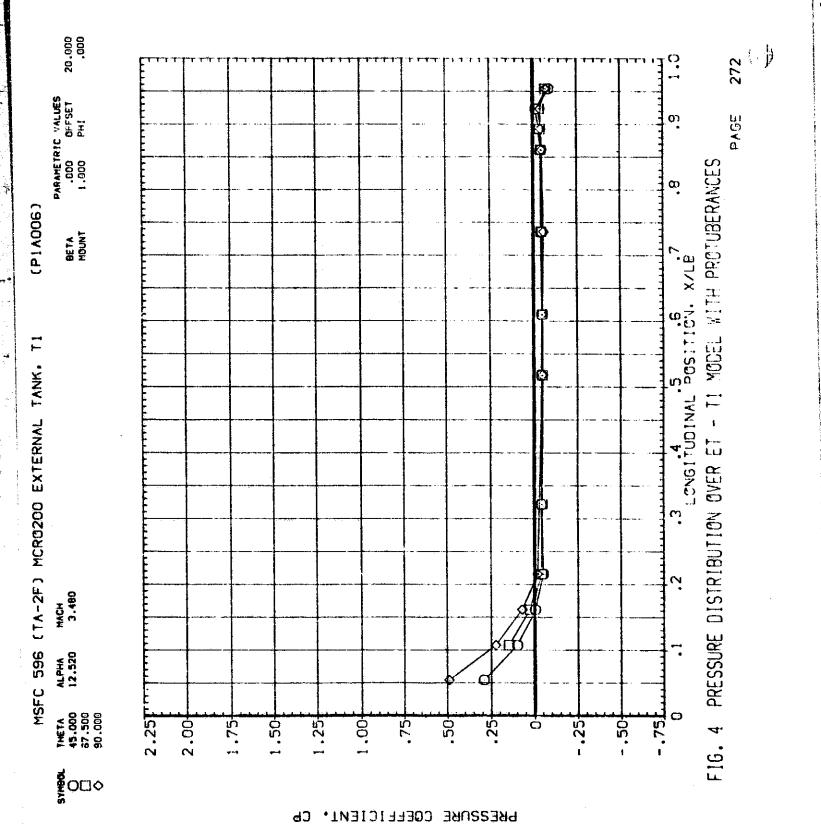
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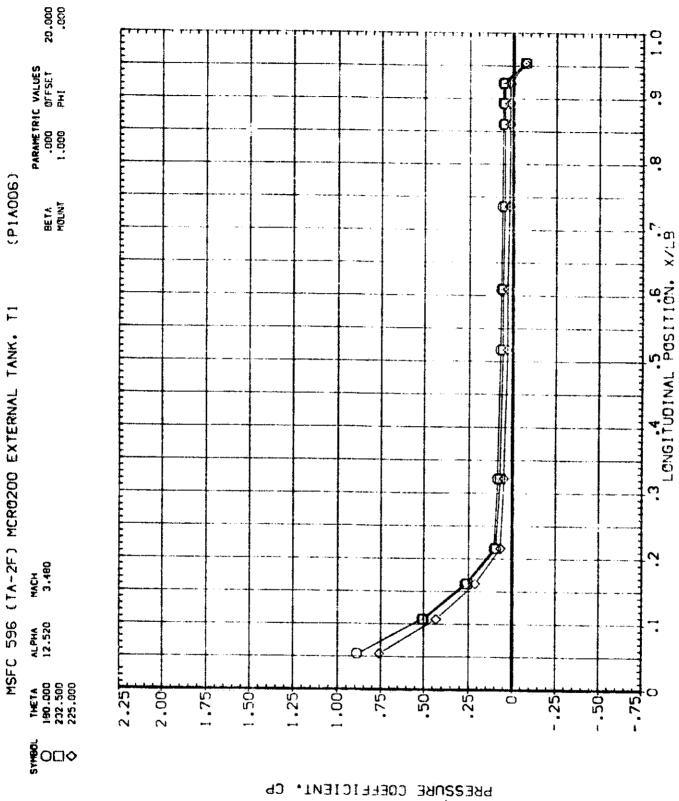


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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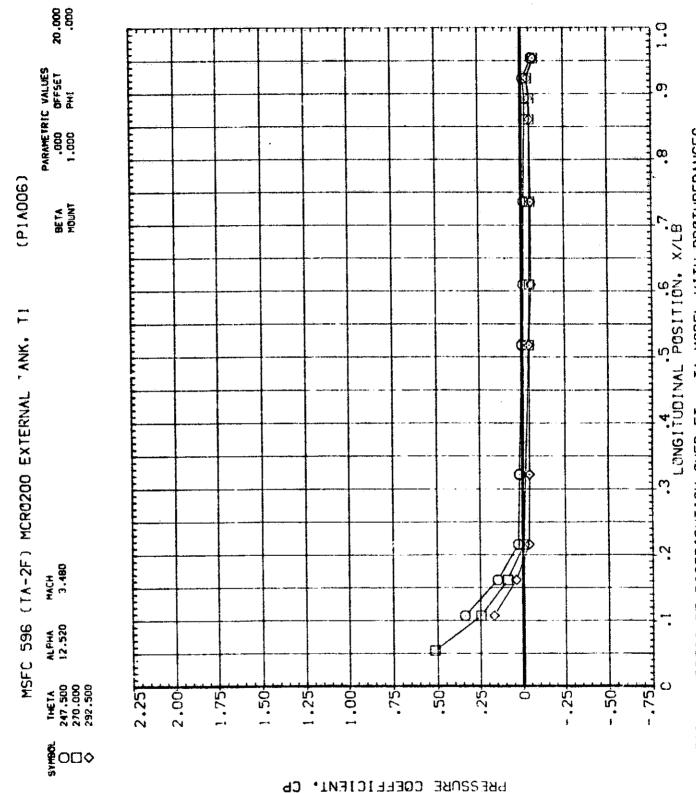
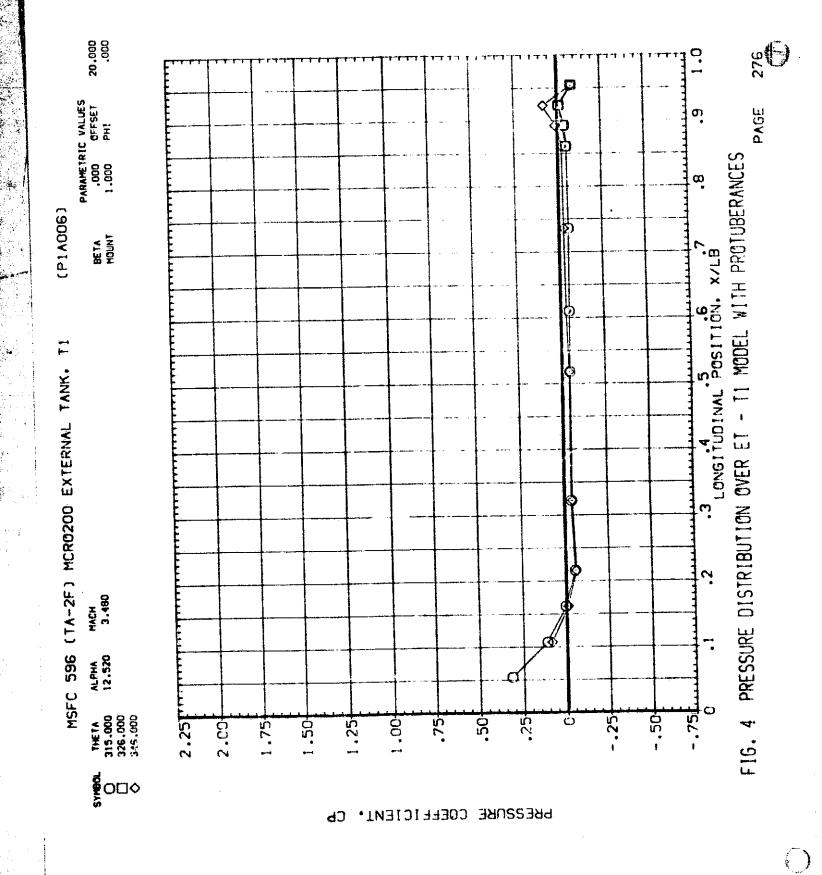
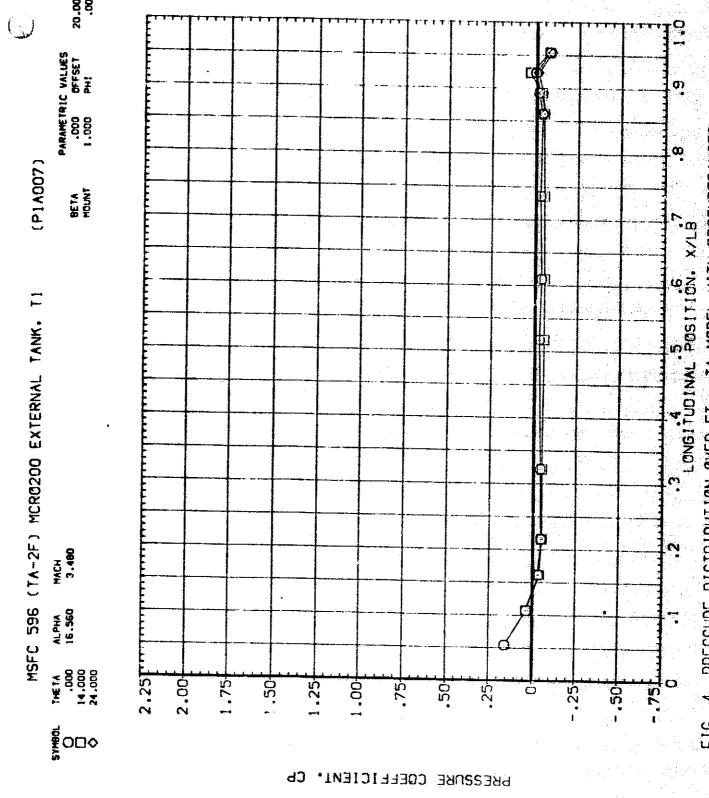


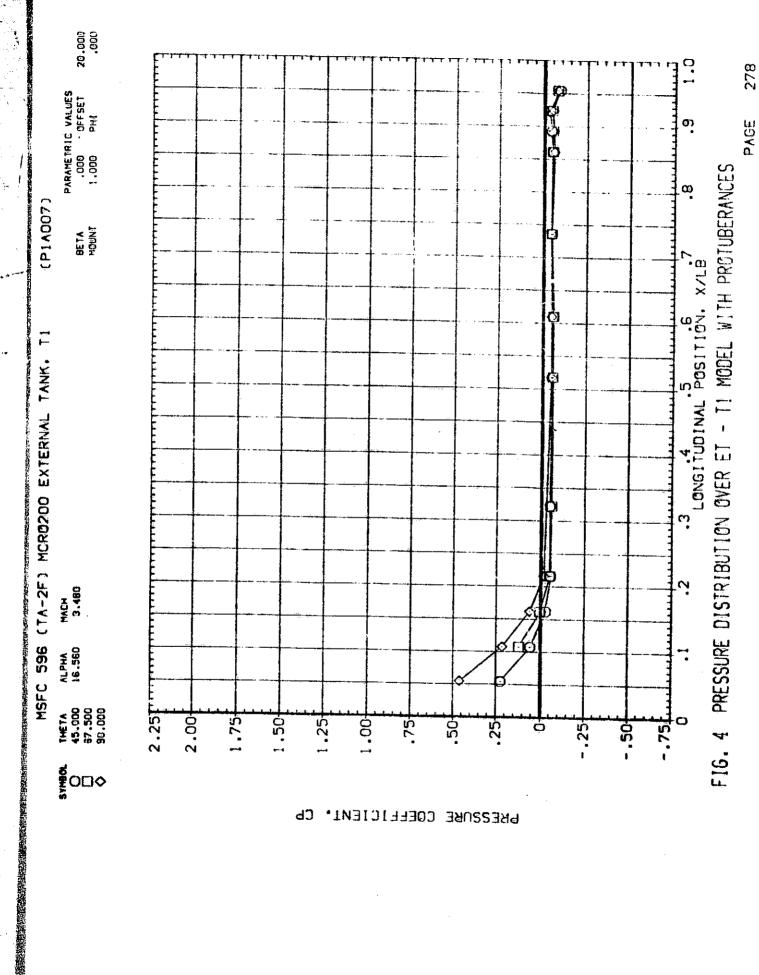
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

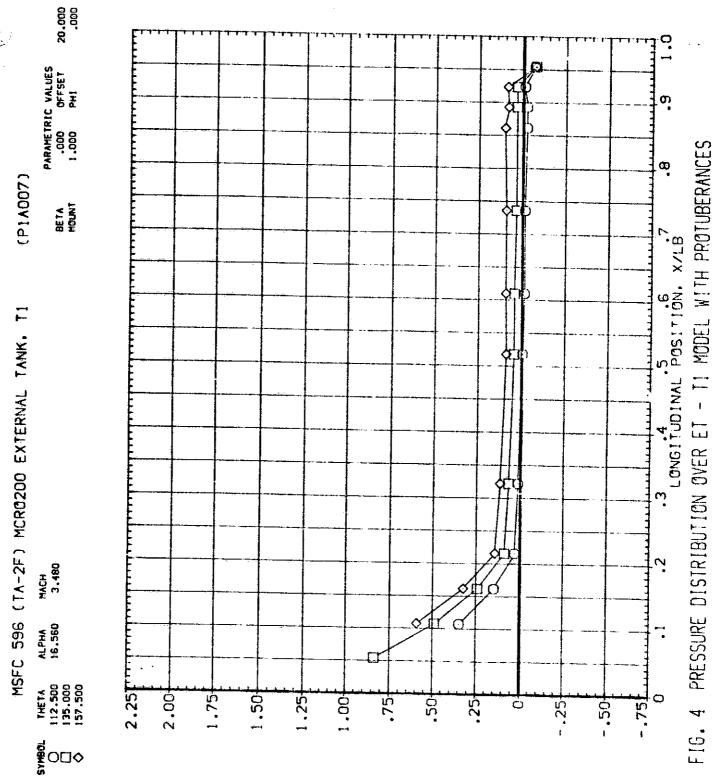




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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

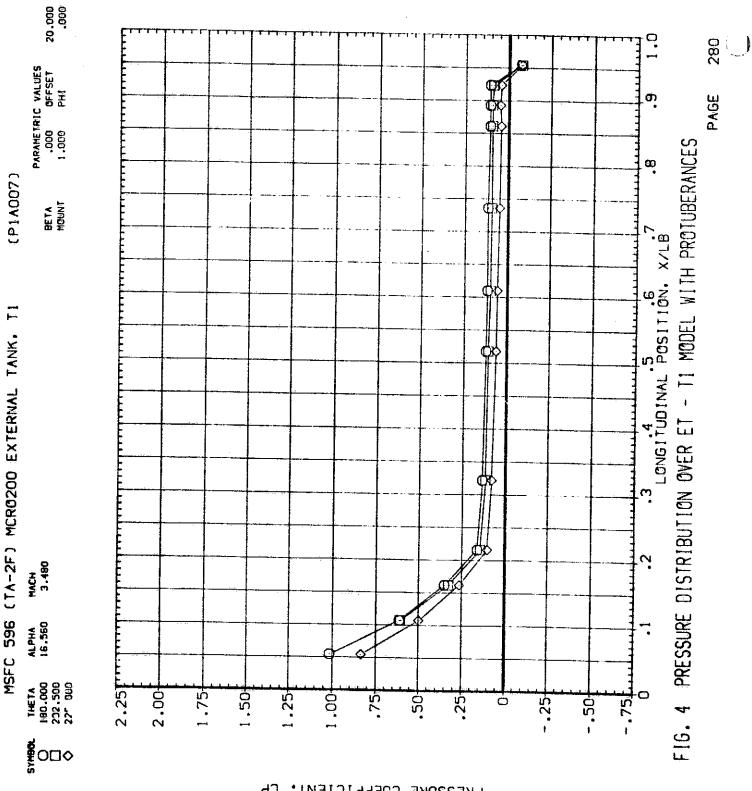




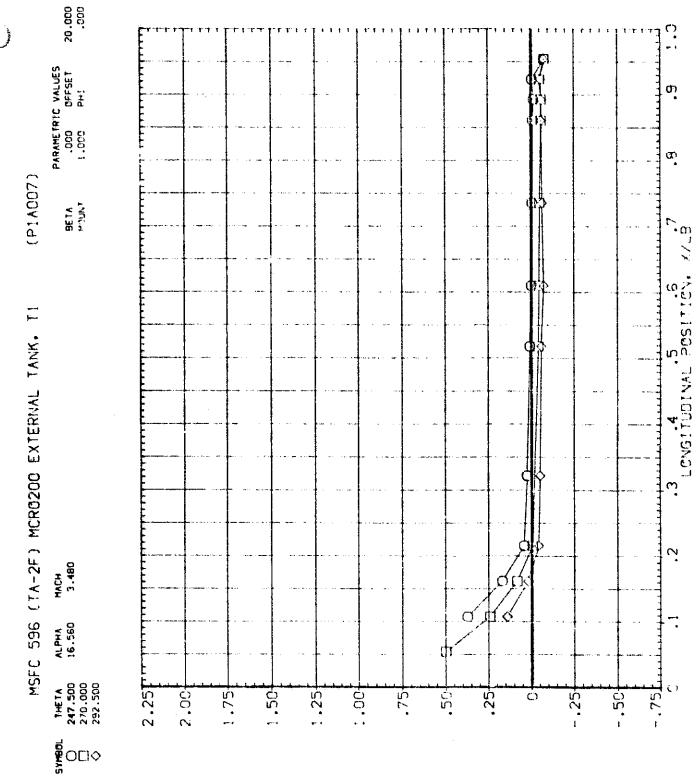
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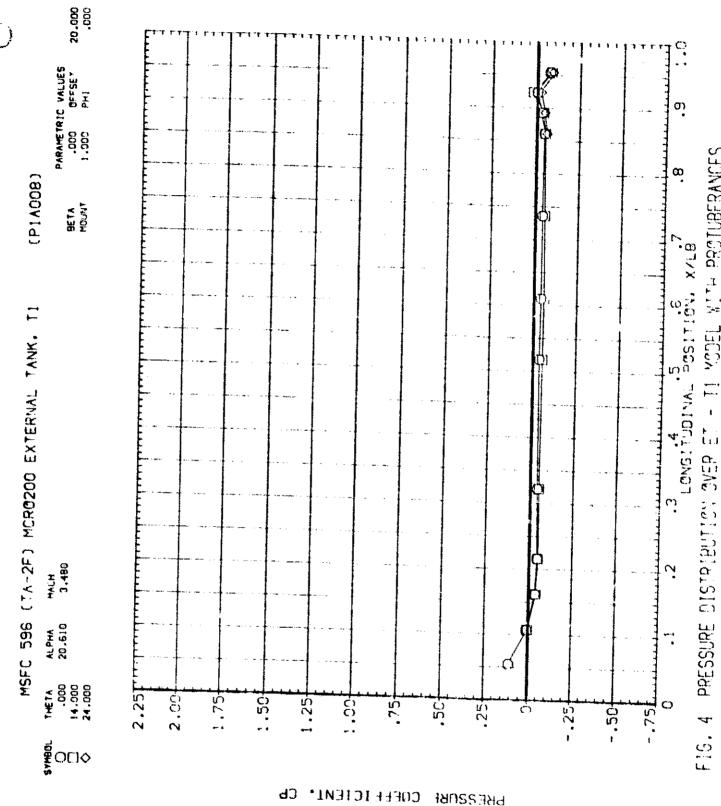


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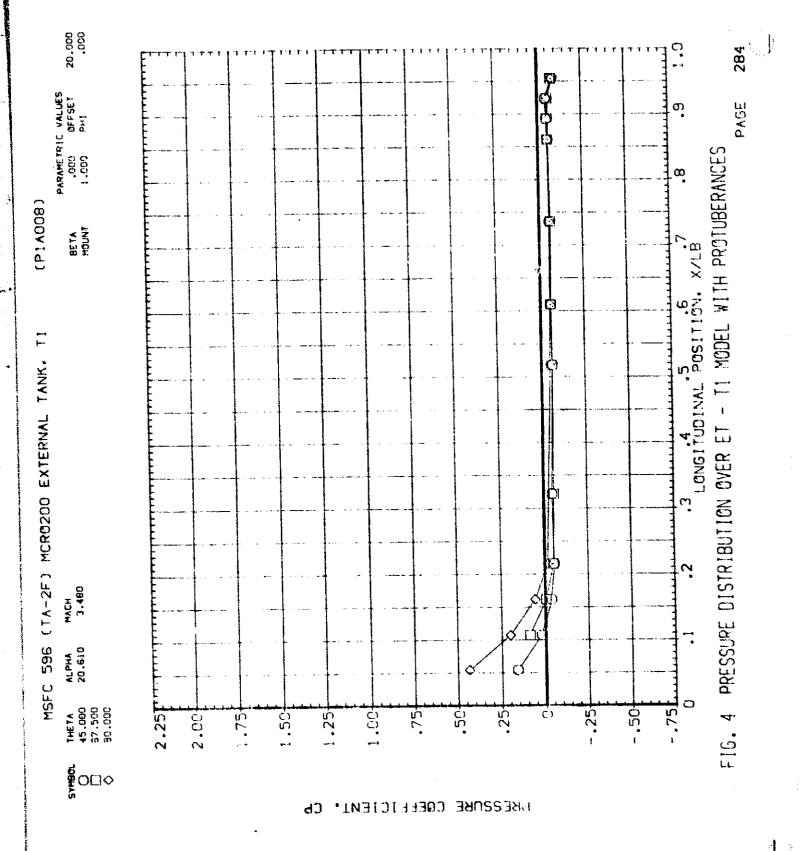
MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



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FIG. 4 PRESSURE DISTRIBUTION OVER

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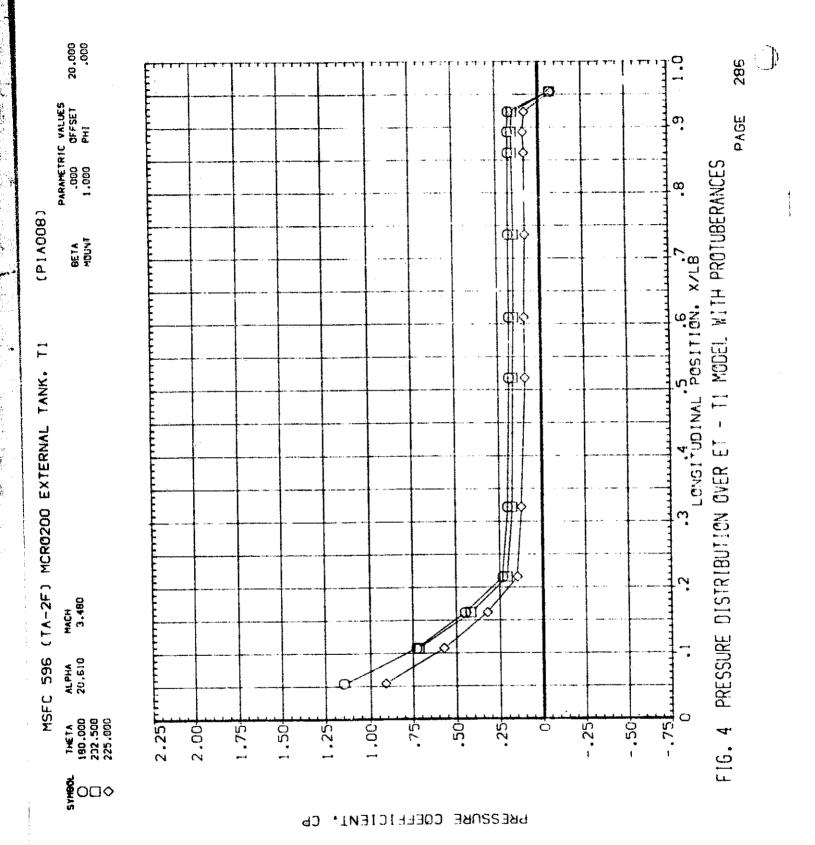
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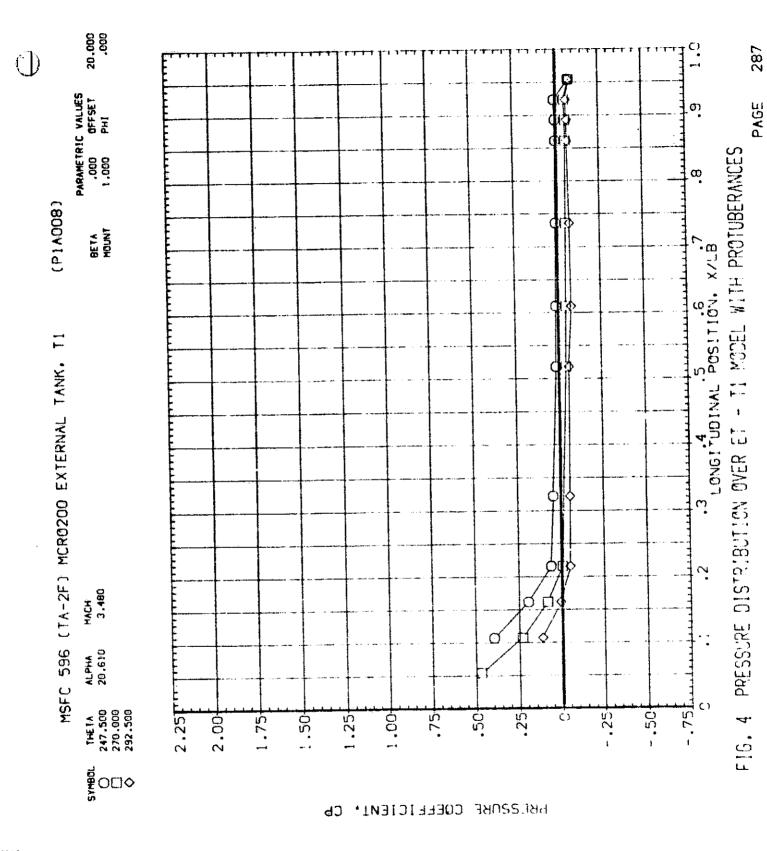
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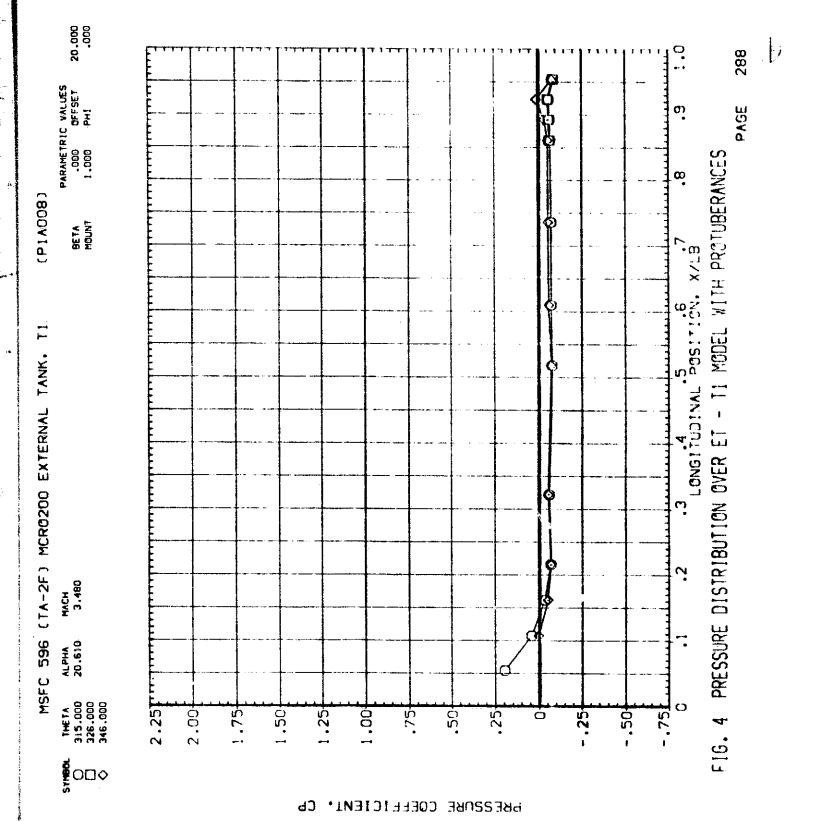
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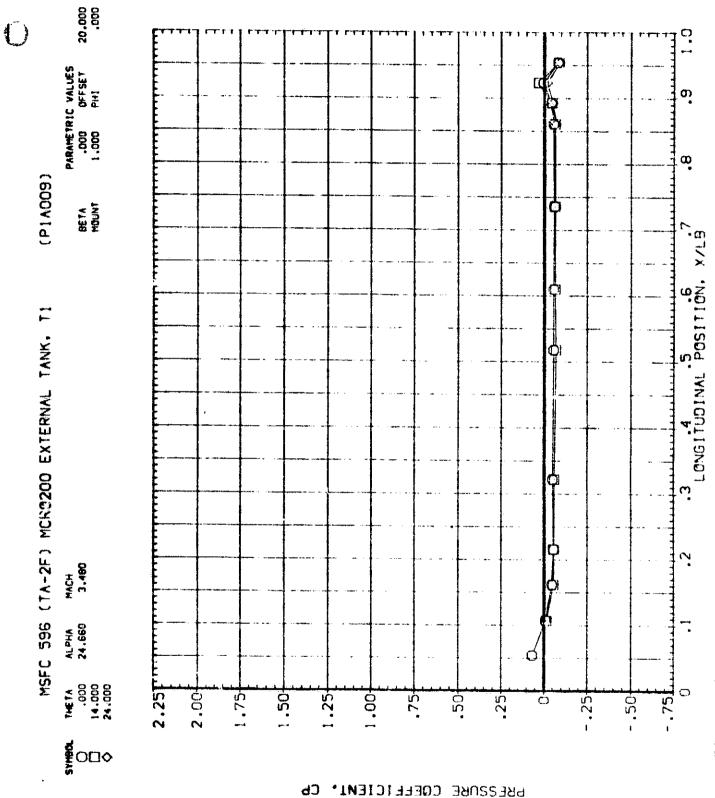
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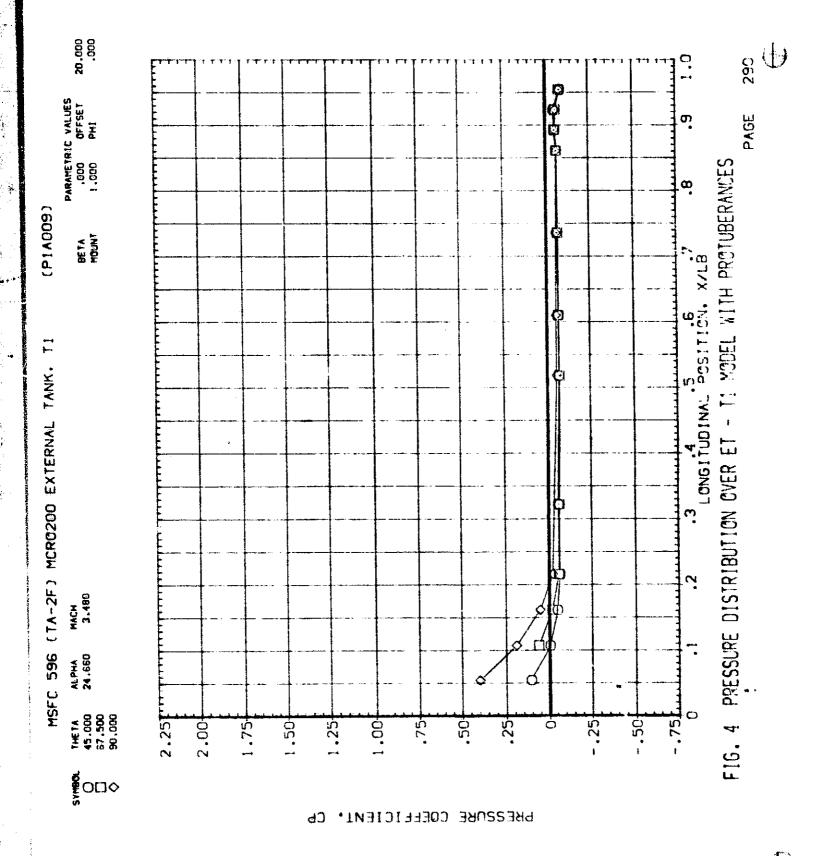








TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION CVER ET



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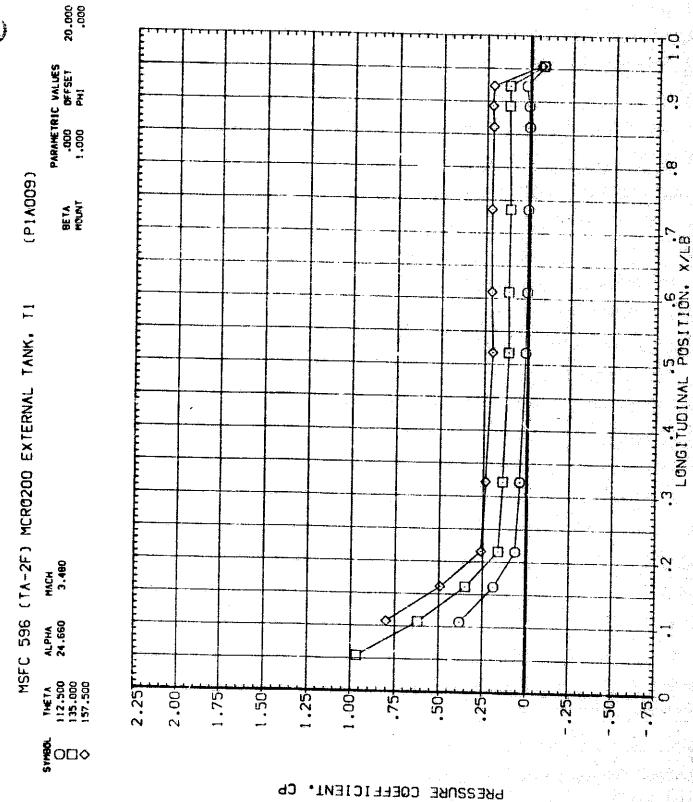


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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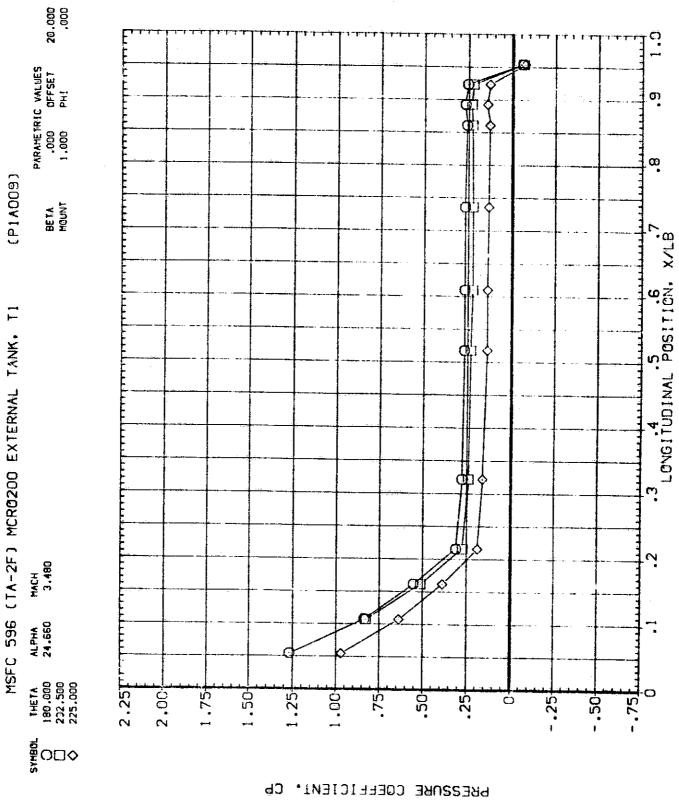
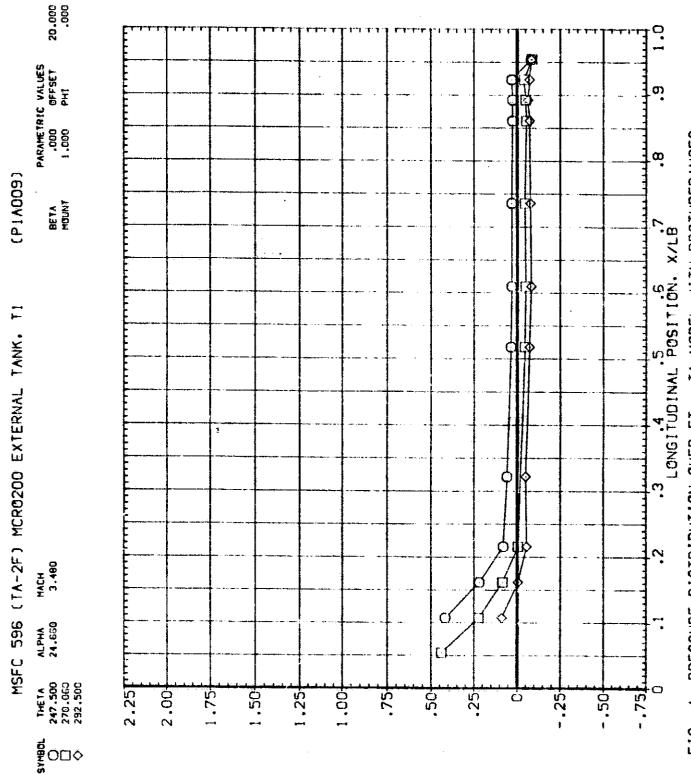


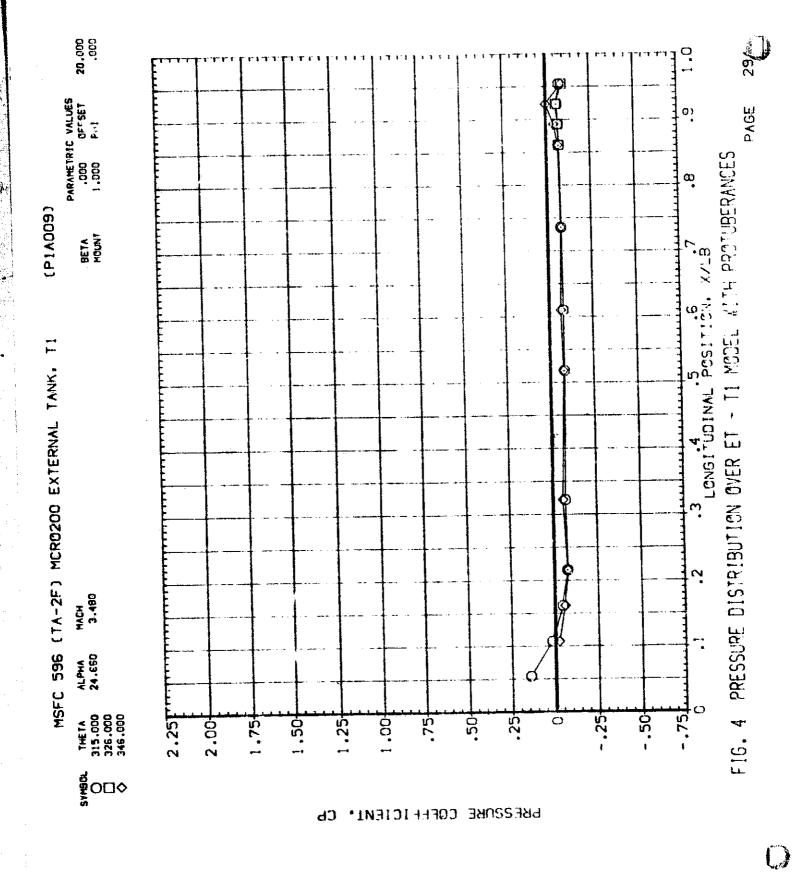
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

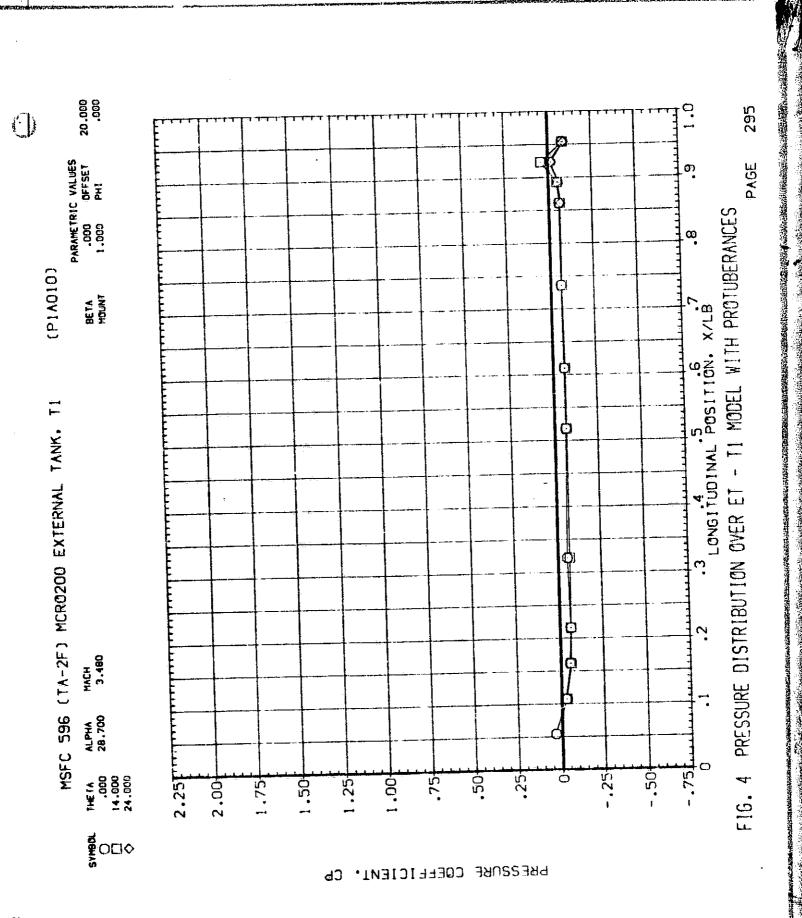
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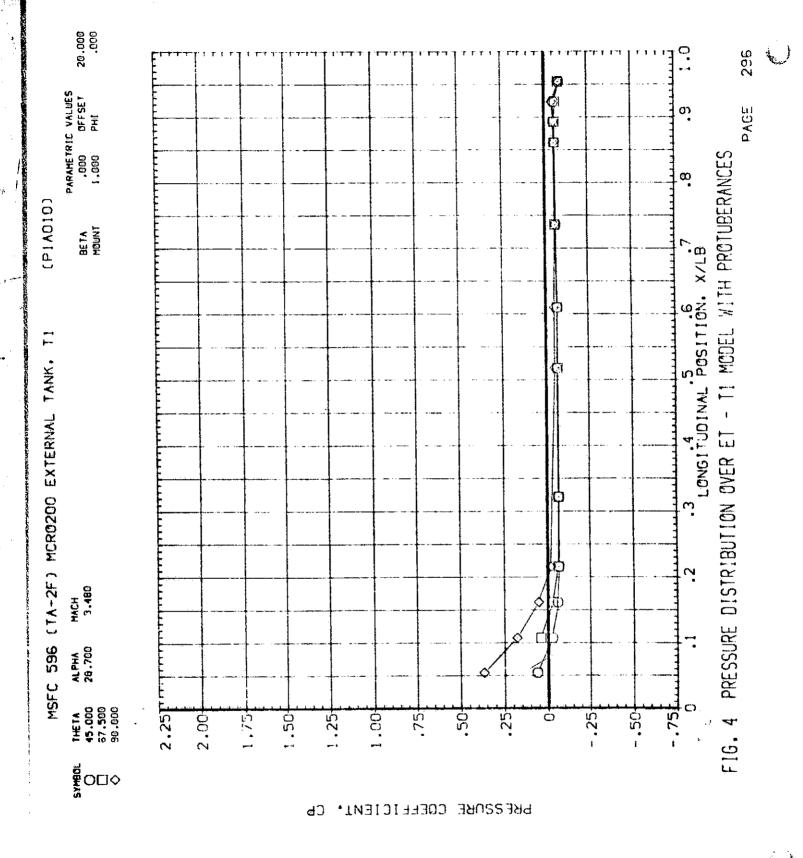
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

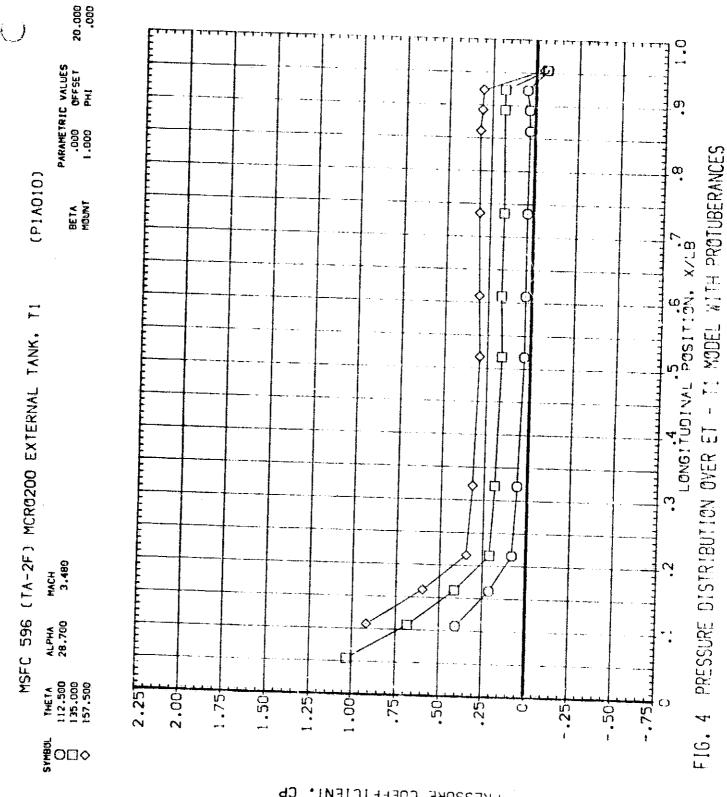




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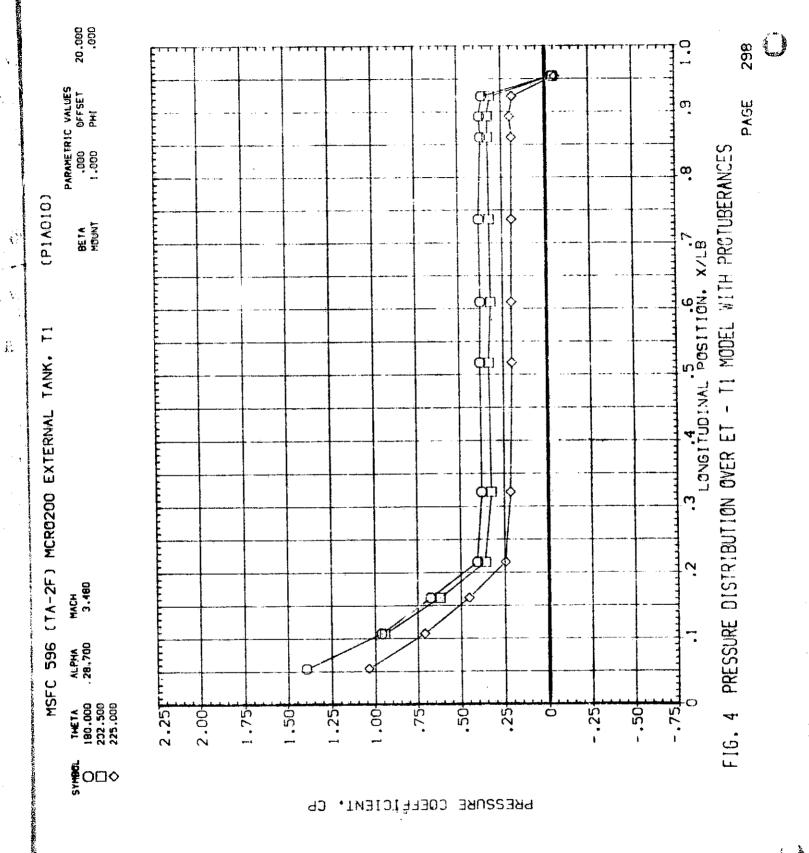


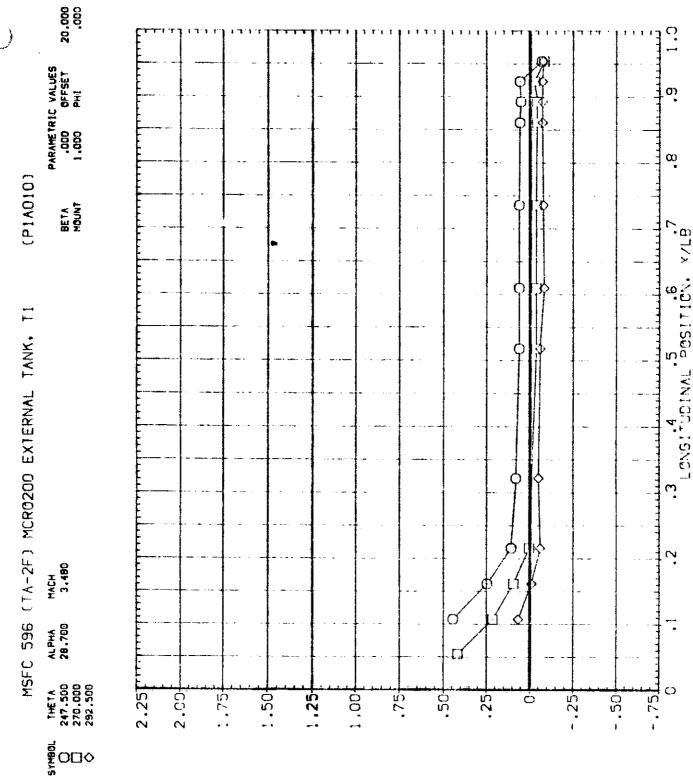
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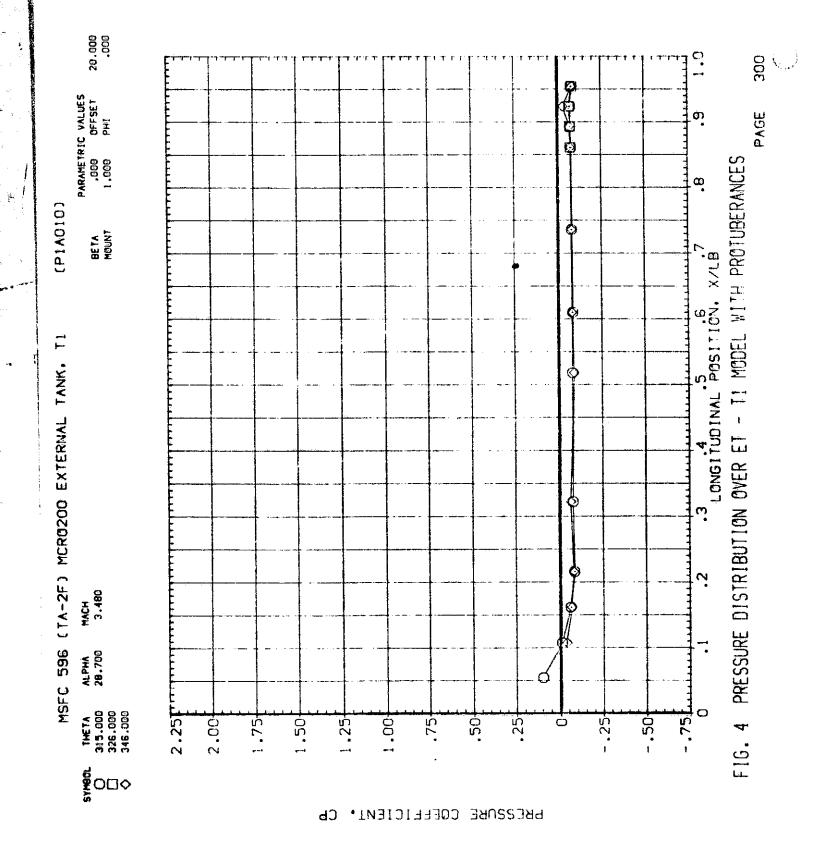
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II MODEL WITH PROTUBERANCES

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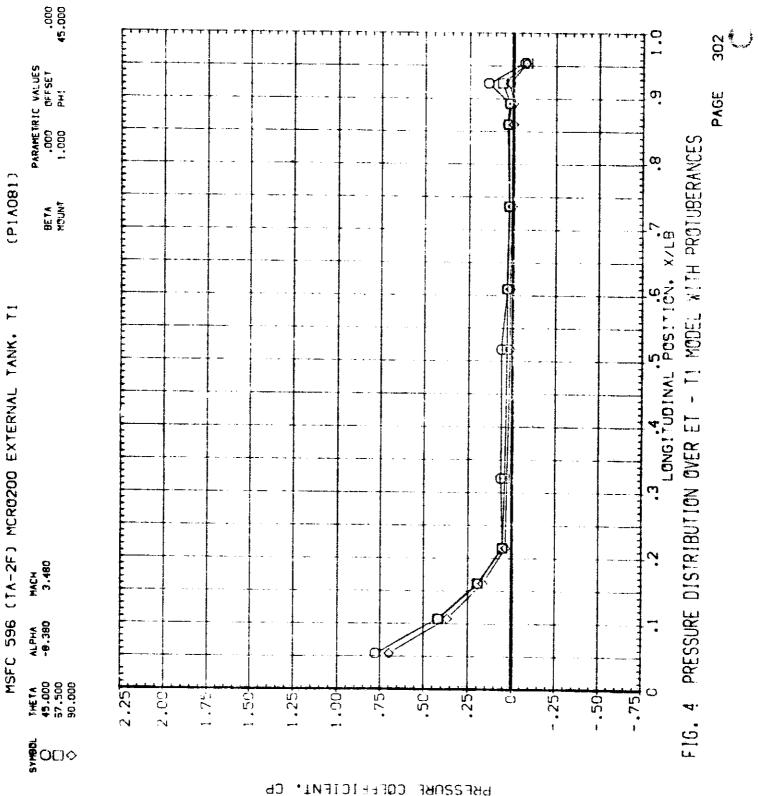
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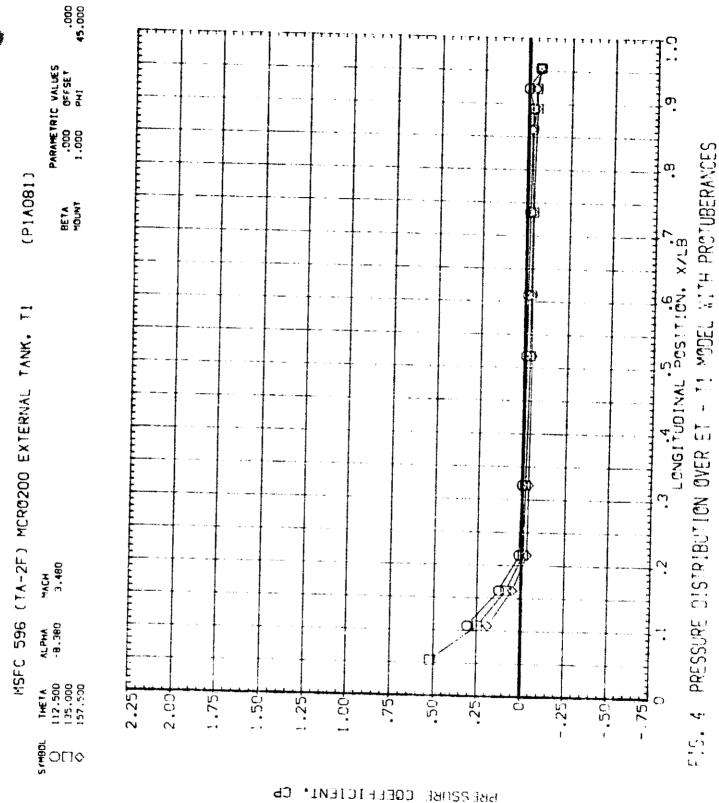
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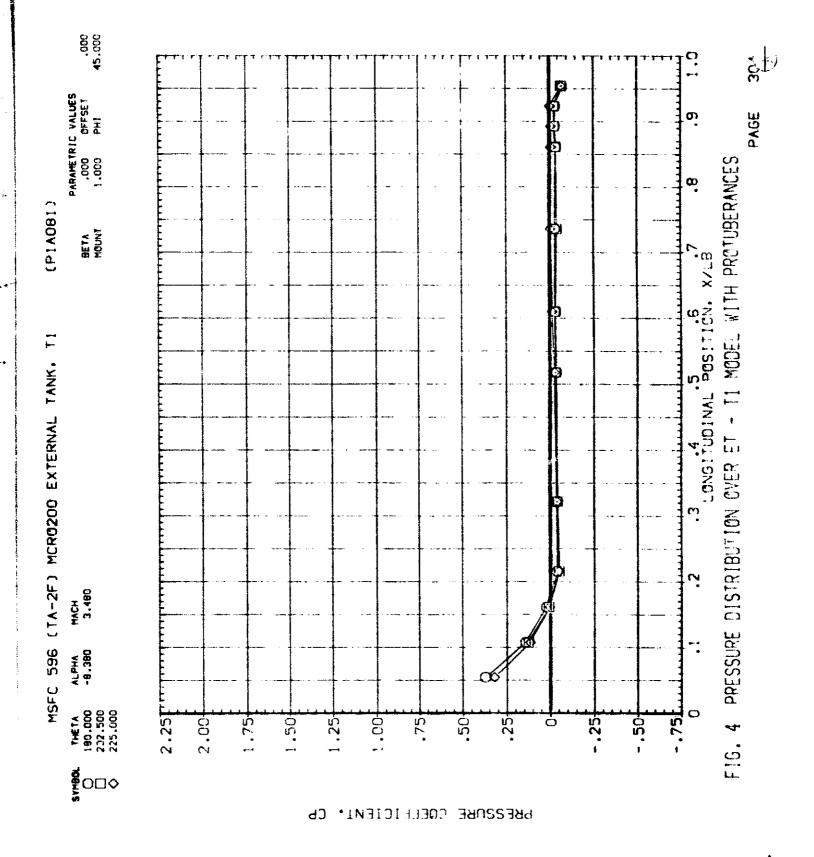


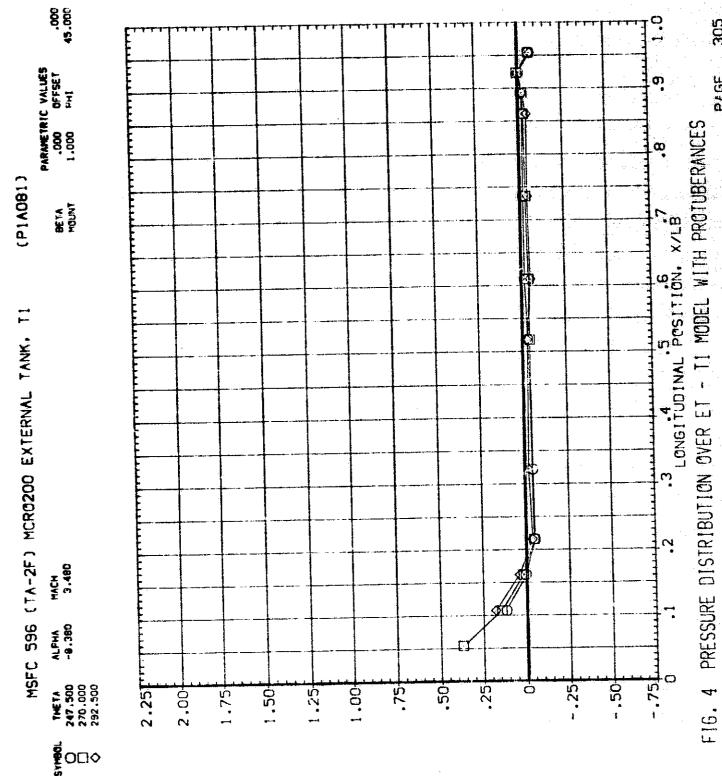
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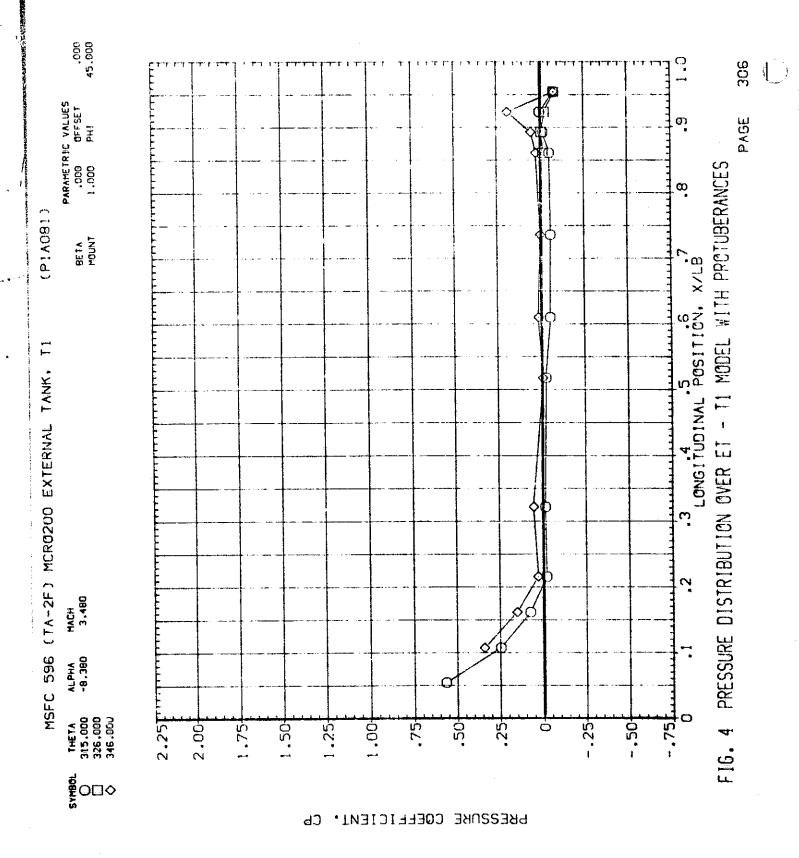




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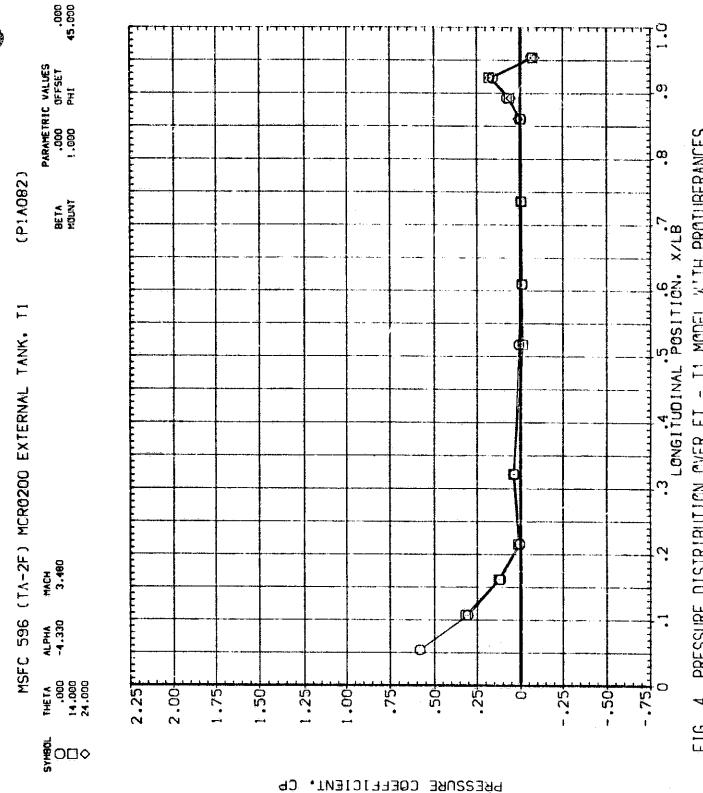
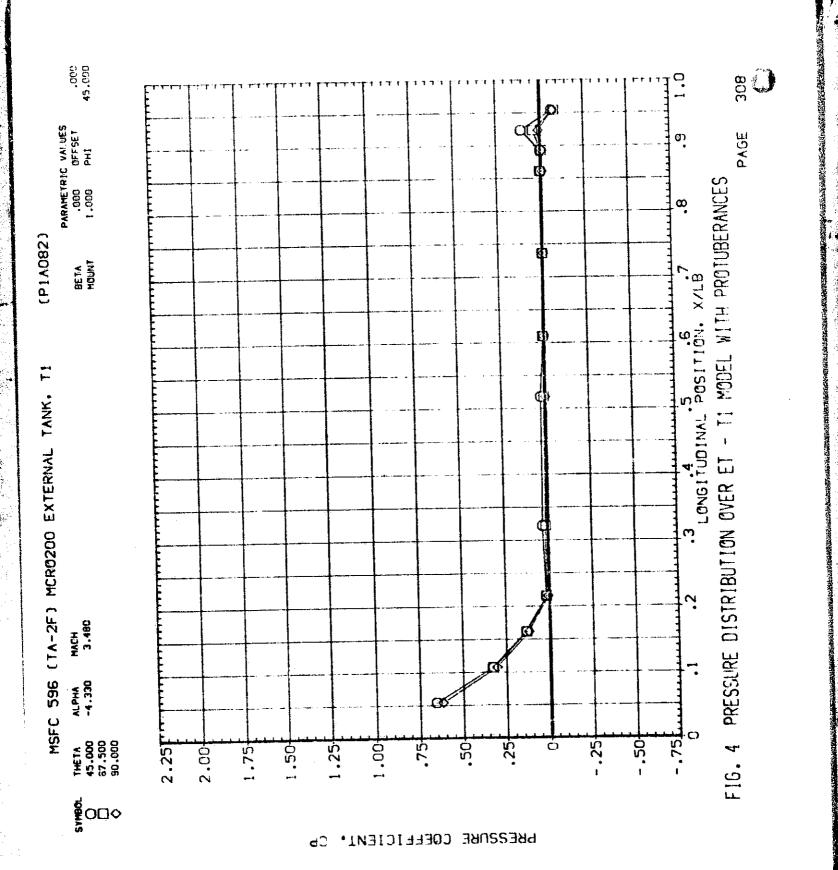


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

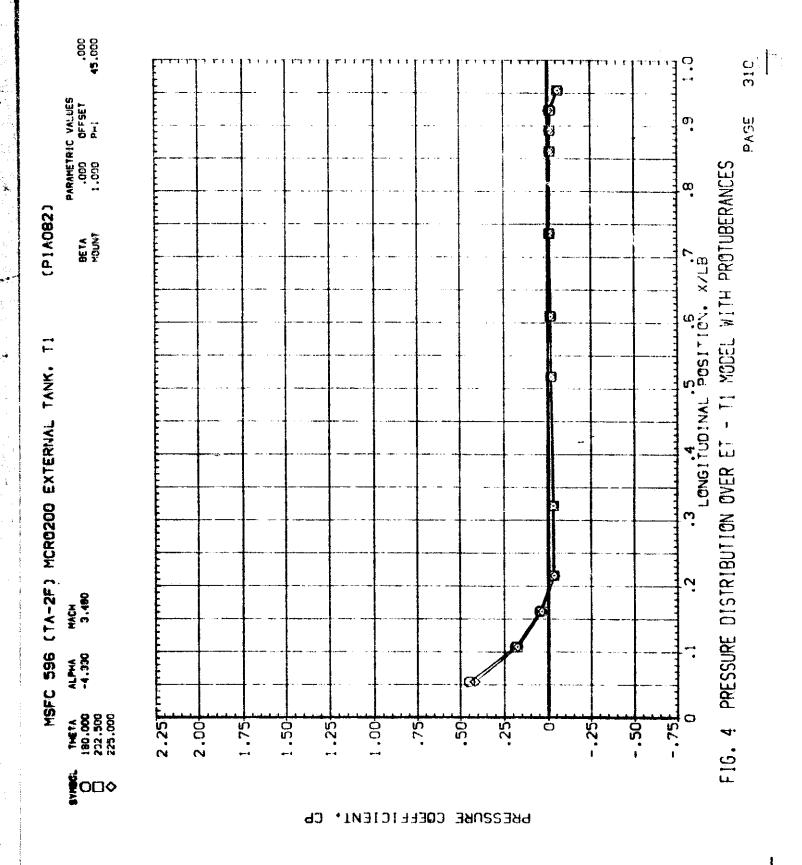


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FIG. 4 PRESSURE DISTRIBUTION OVER ET



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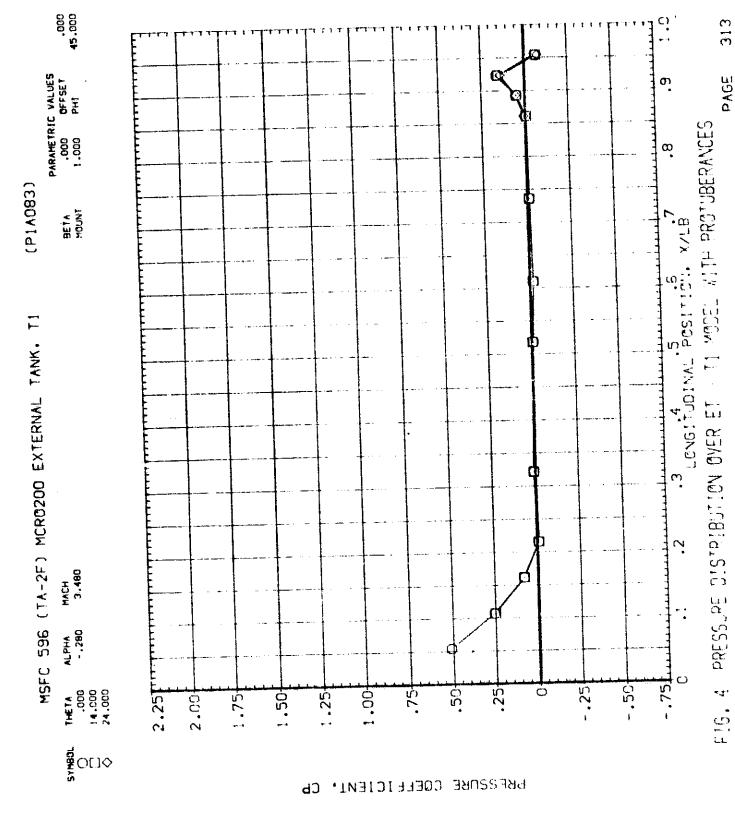
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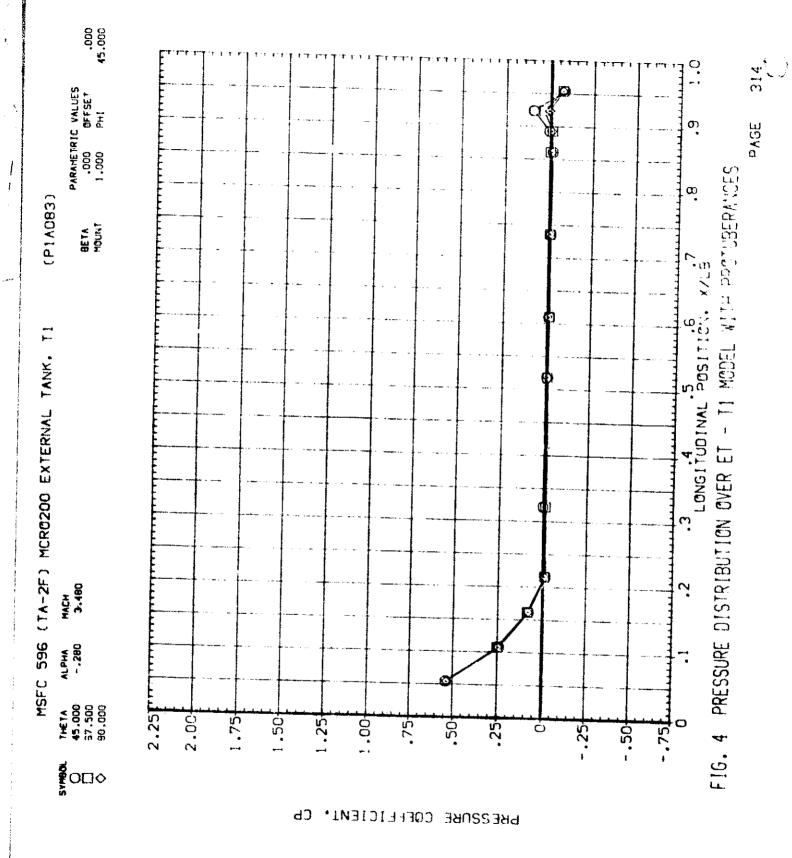
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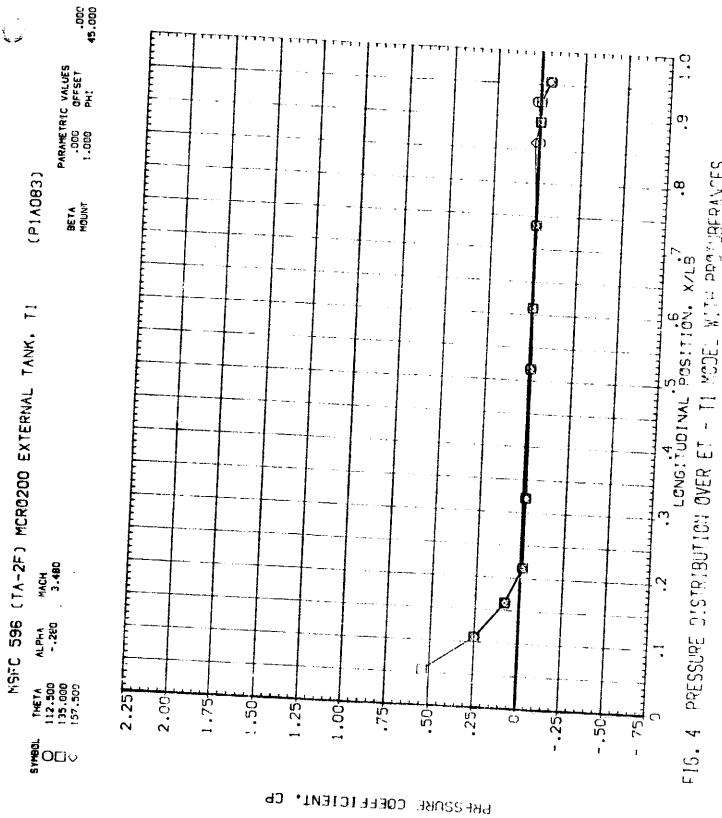
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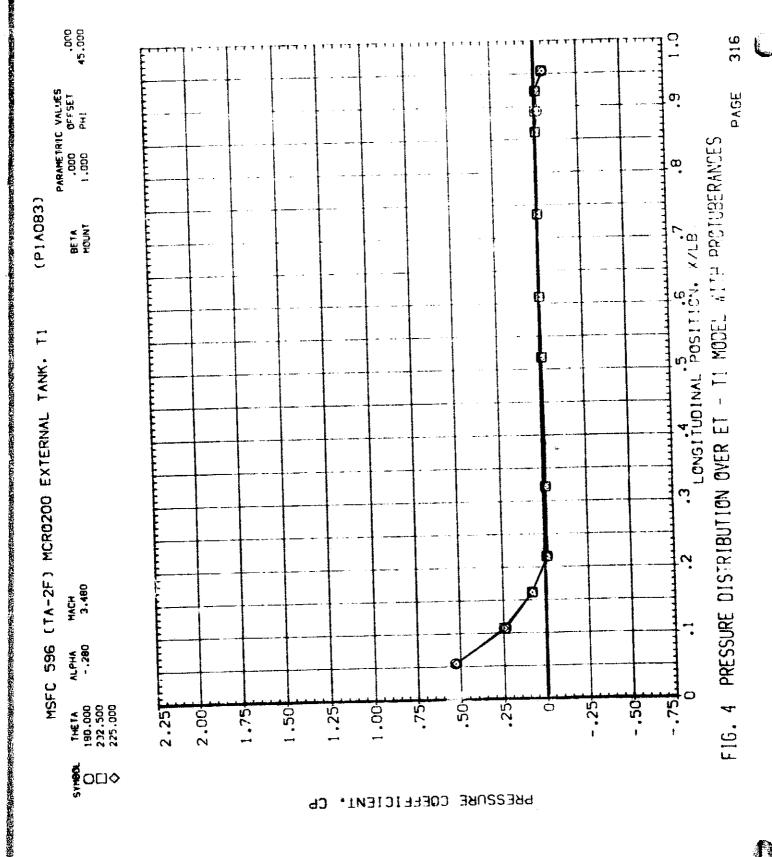


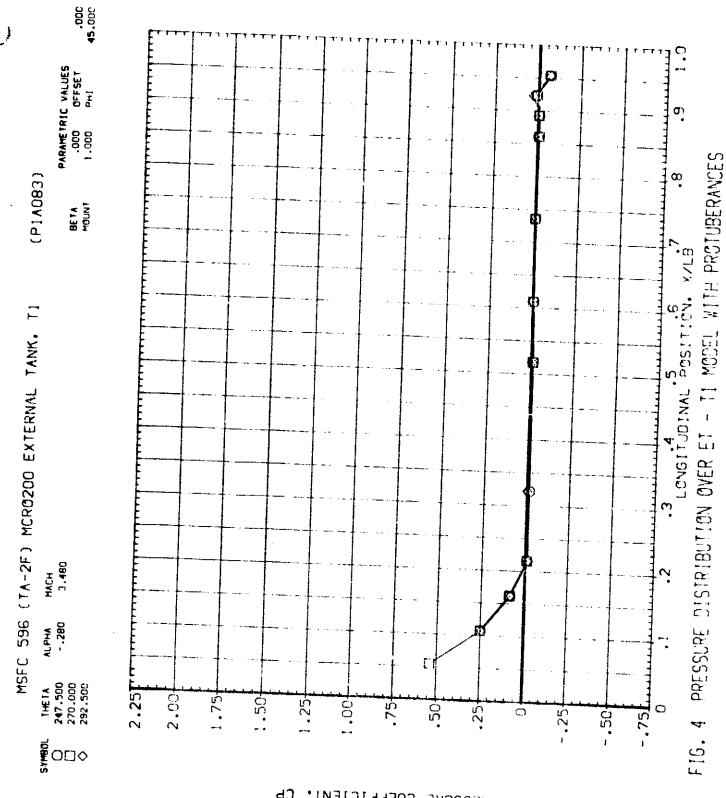




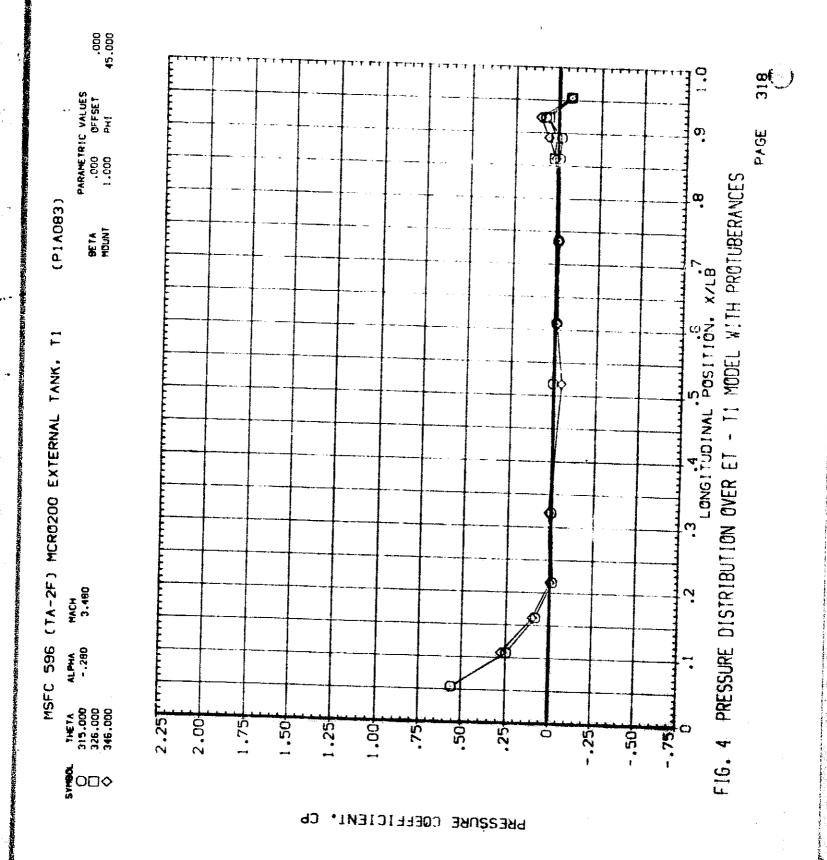
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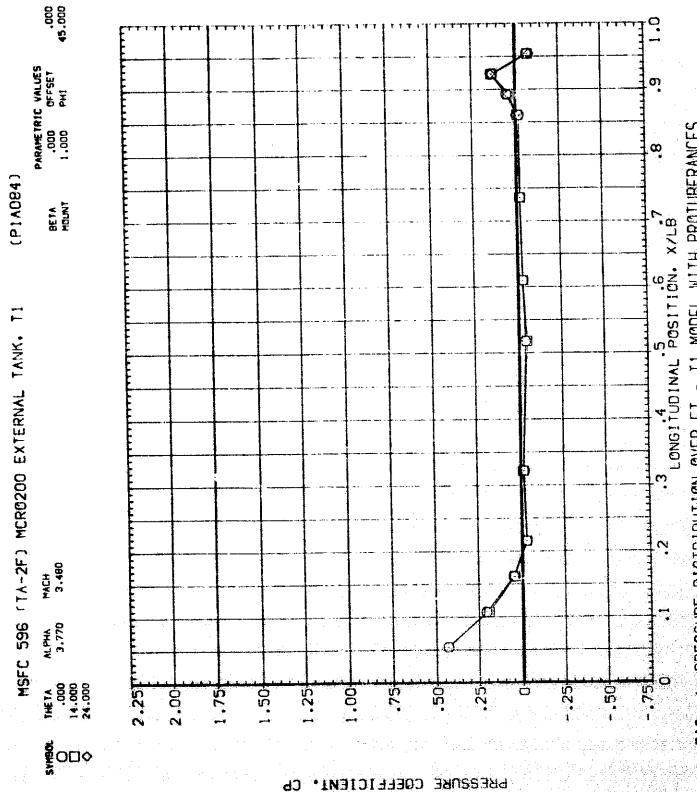
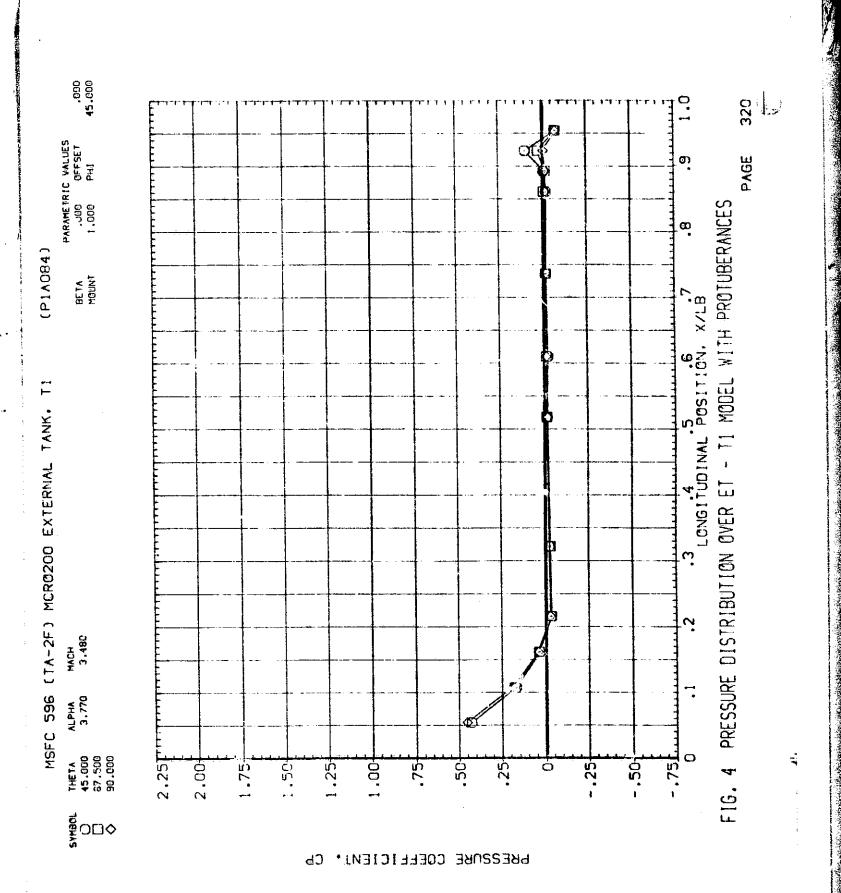


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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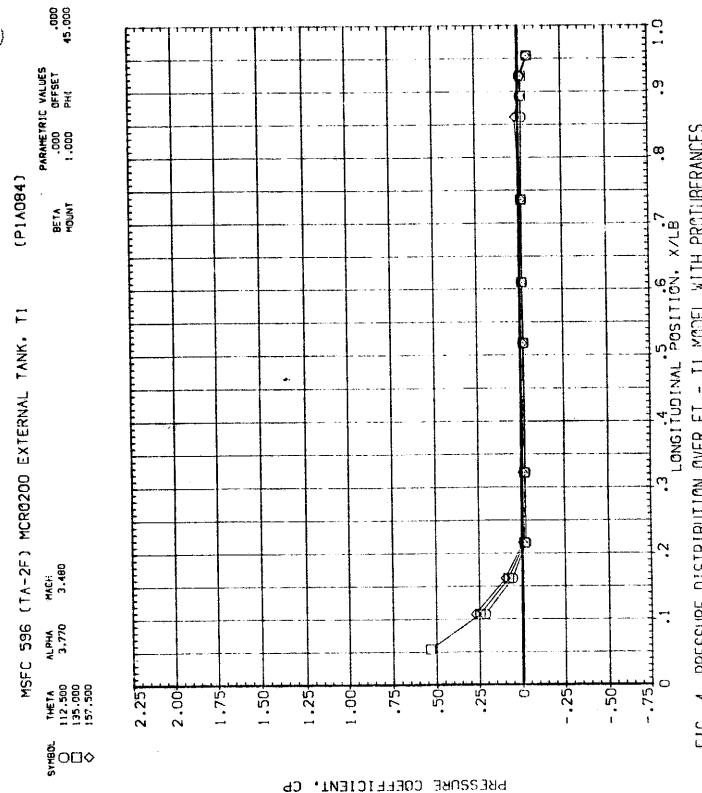
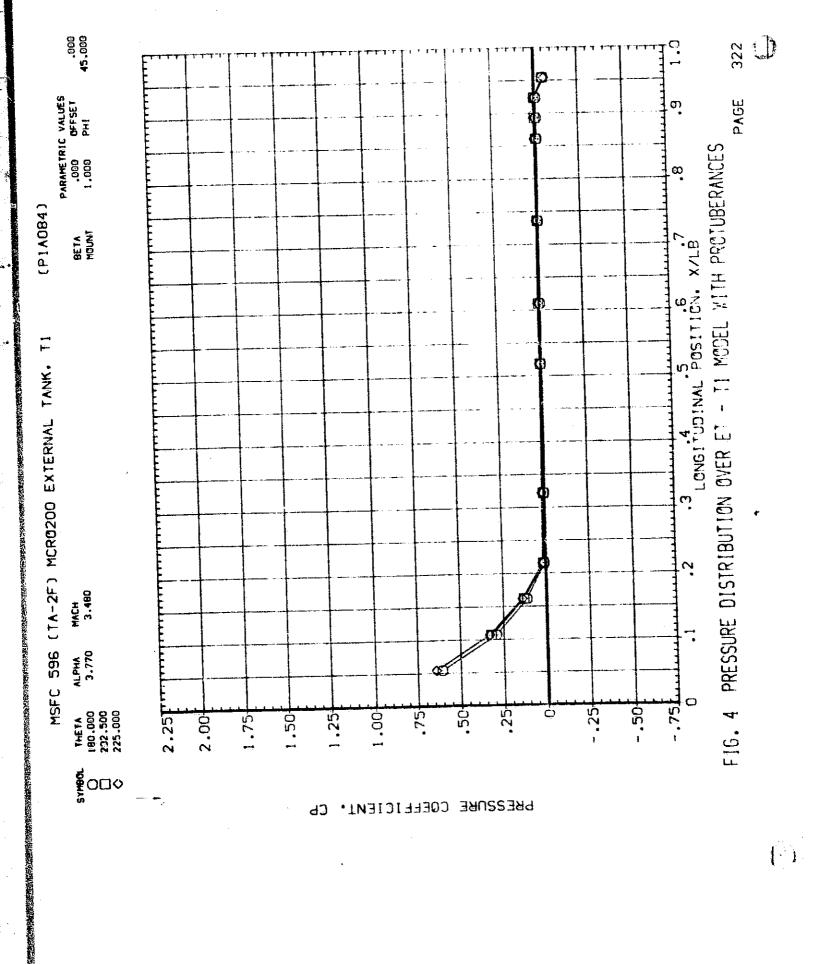
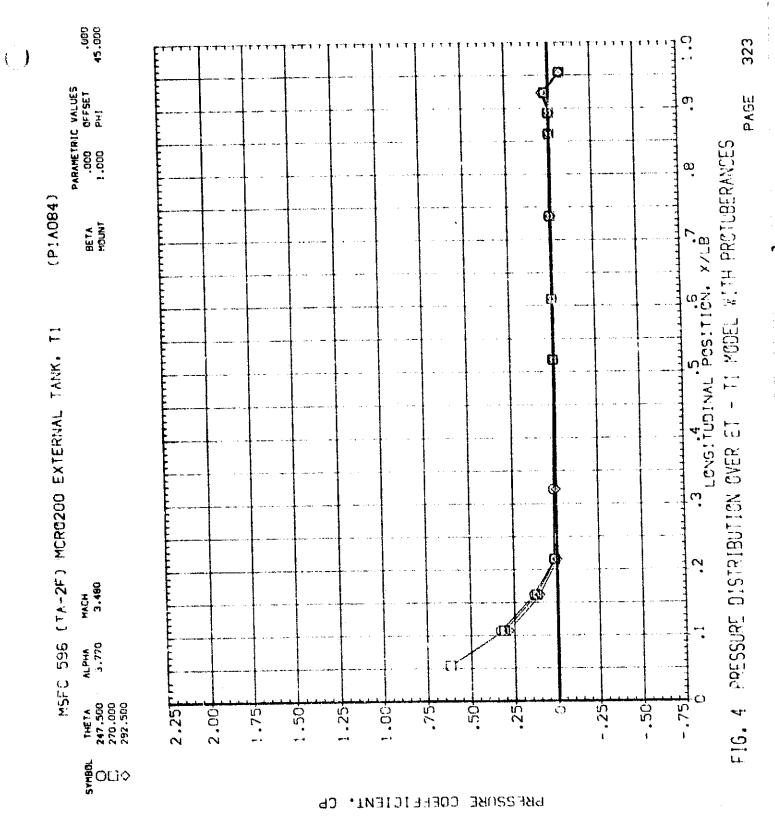


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

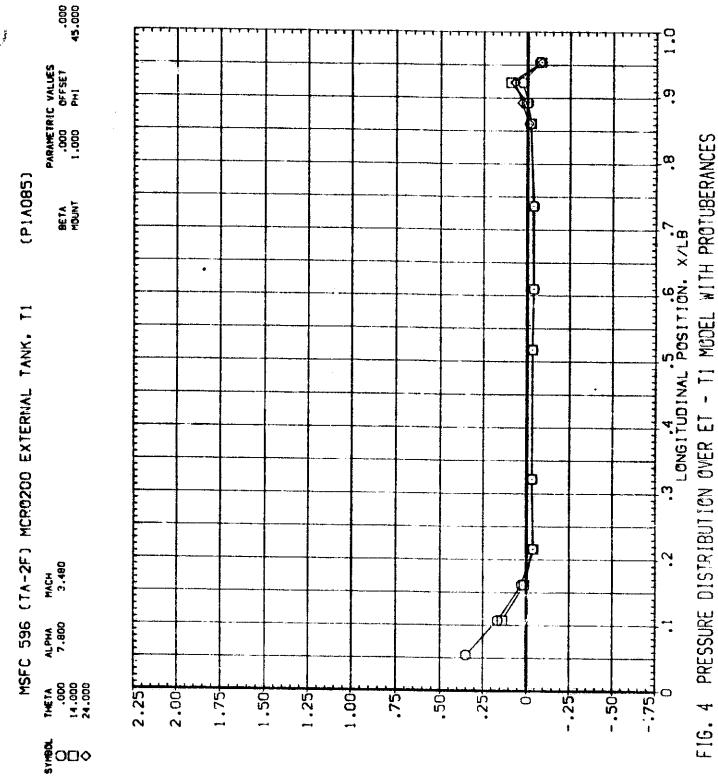
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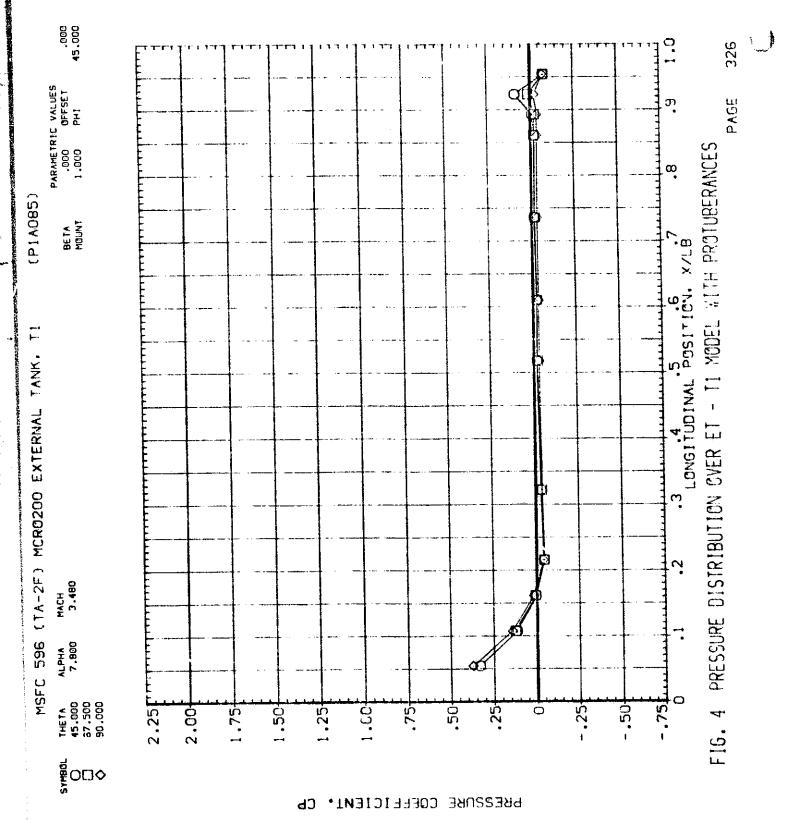
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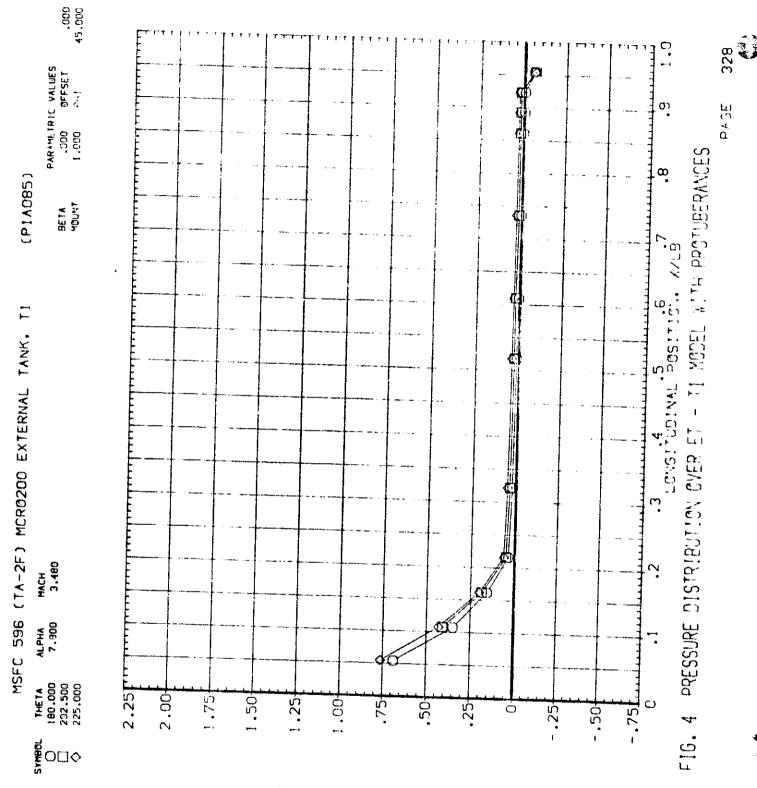
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II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET -

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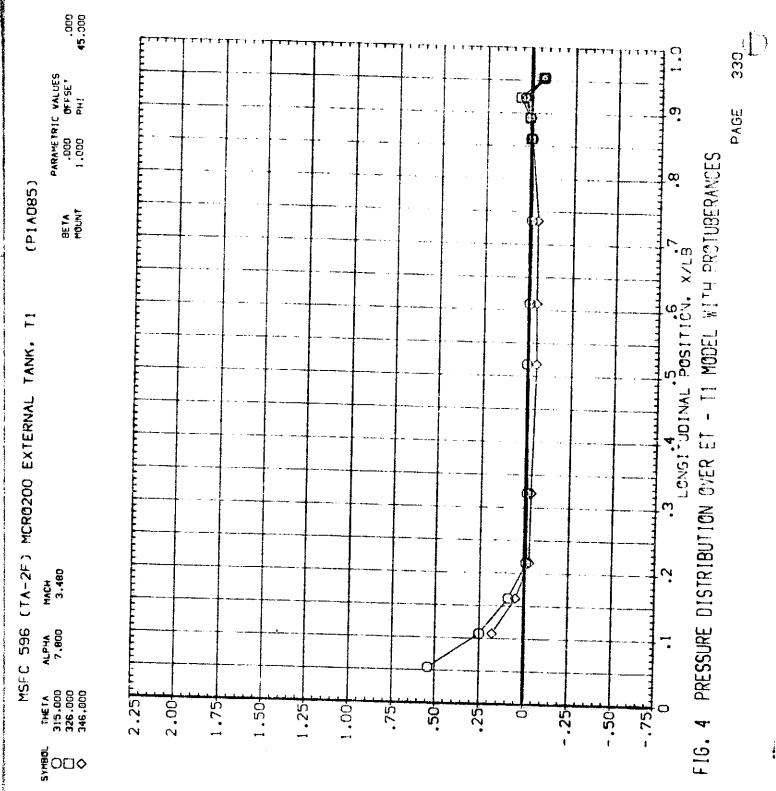


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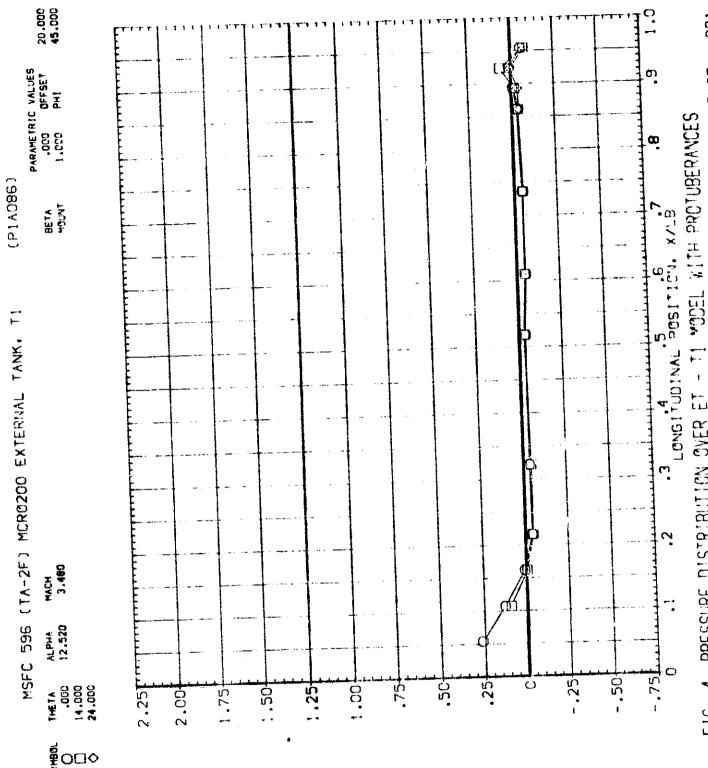
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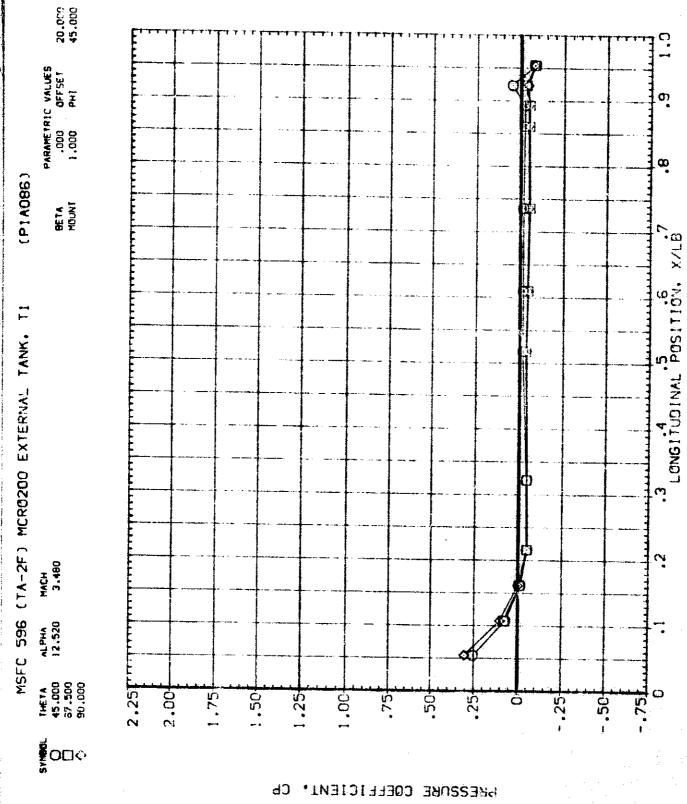
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FIG. 4 PRESSURE DISTRIBUTION OVER ET



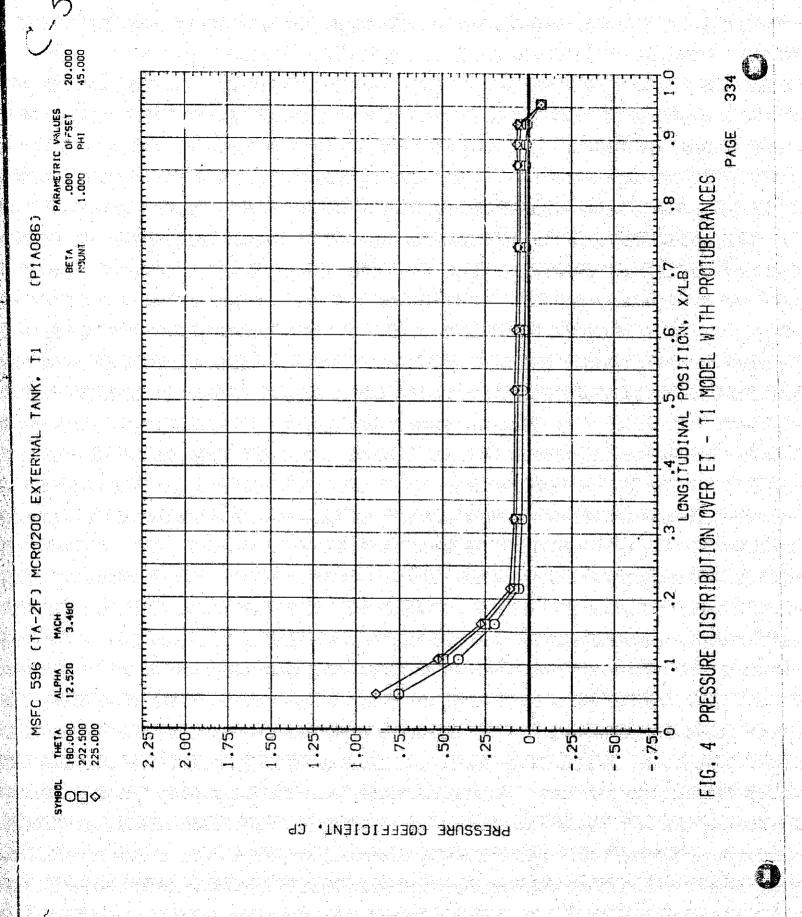
- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

FIG: 4 PRESSURE DISTRIBUTION OVER ET - TI MEDEL WITH PROTUBERANCES

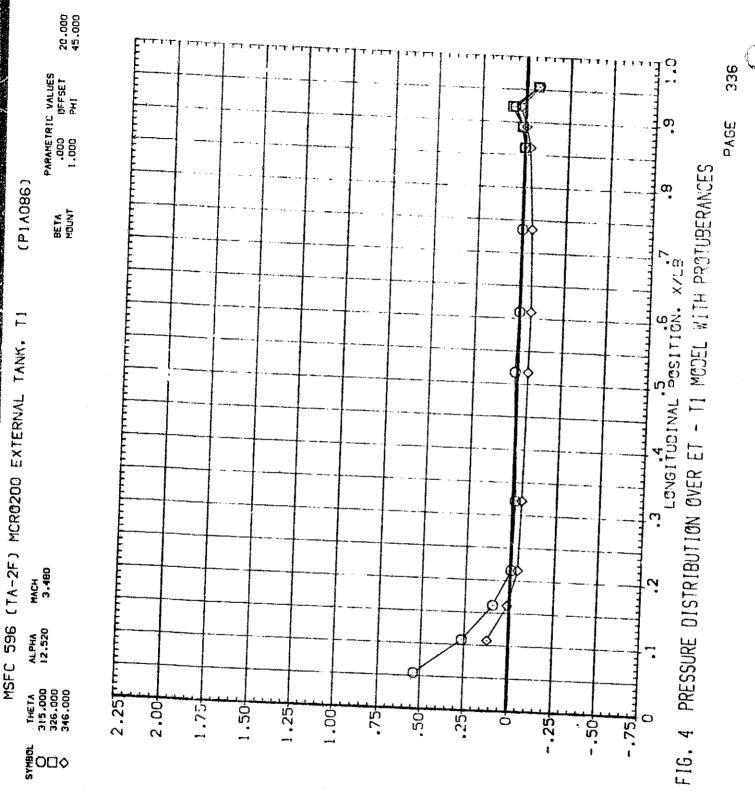
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

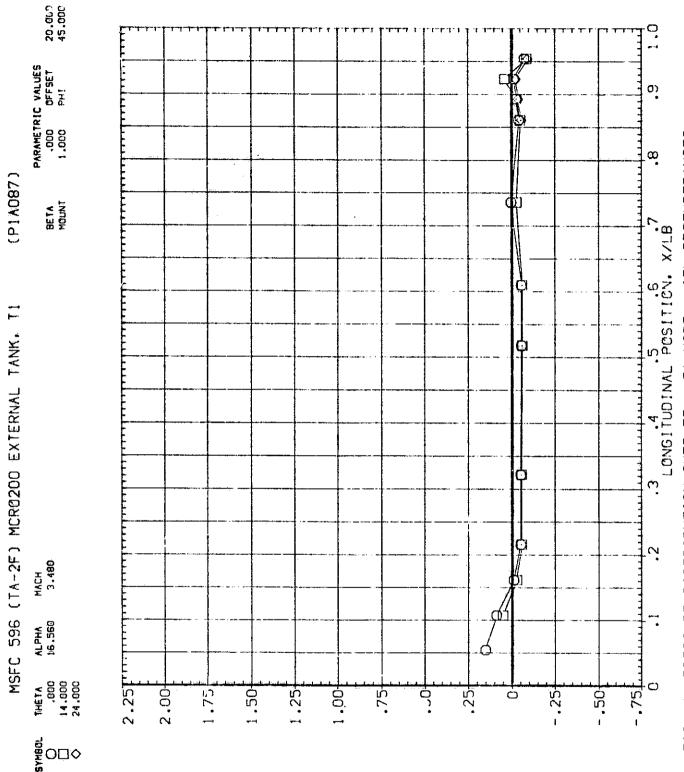
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TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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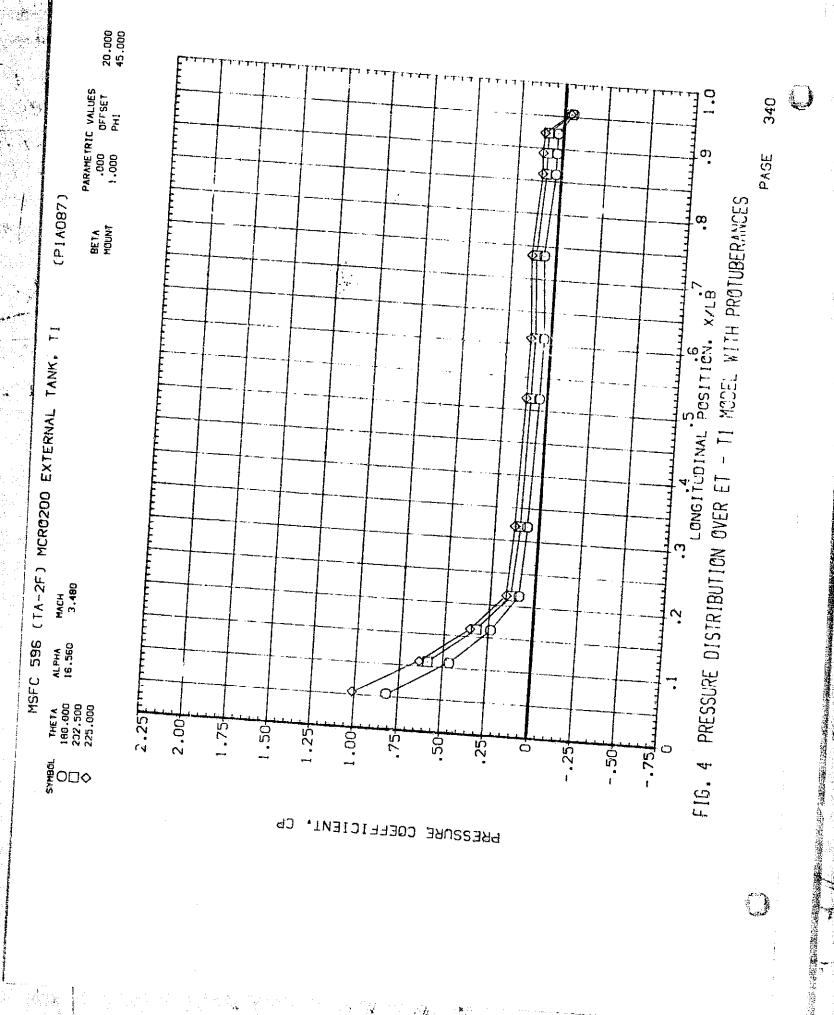
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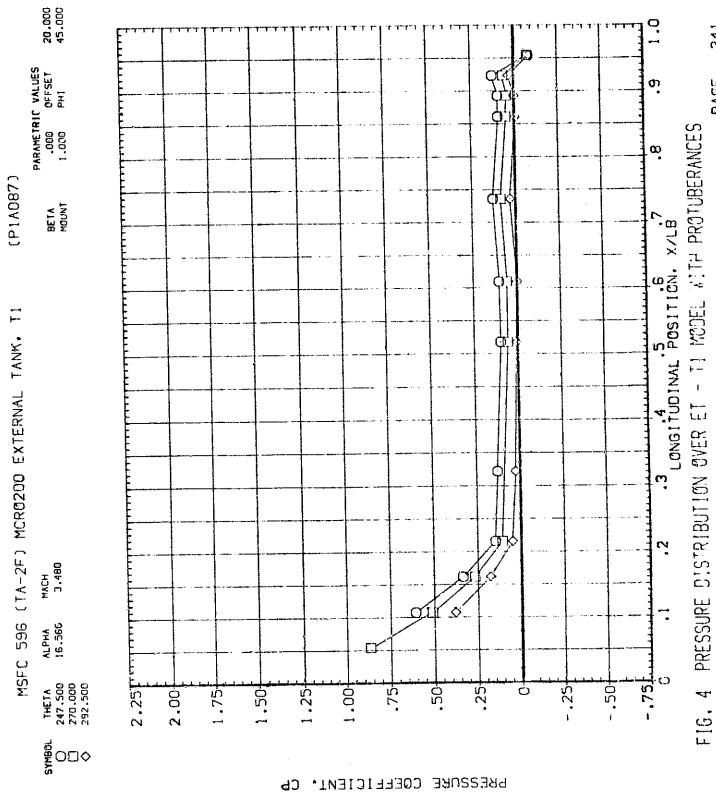
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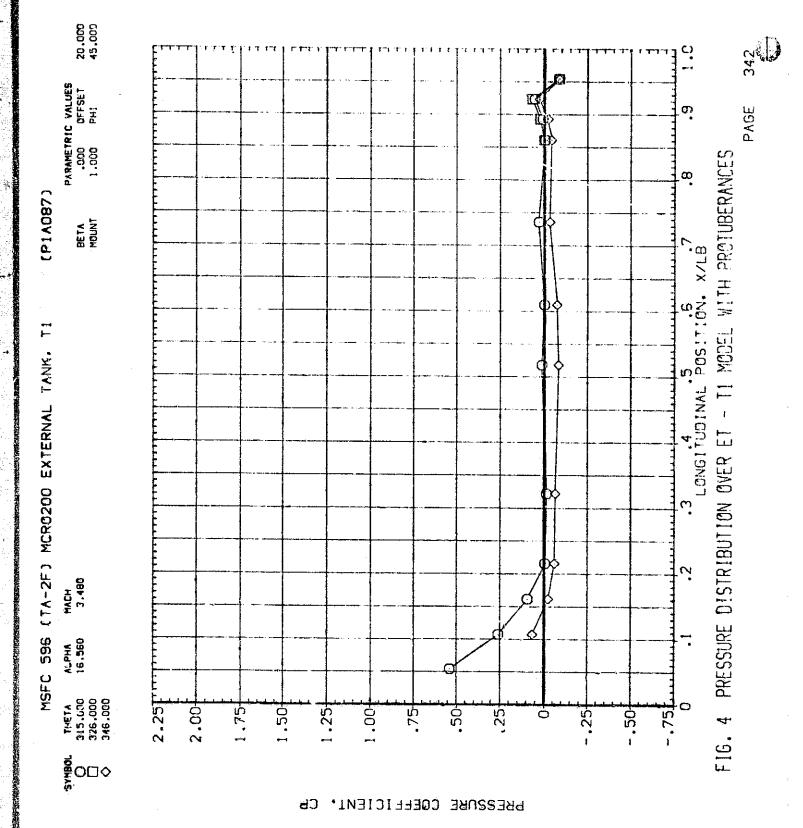
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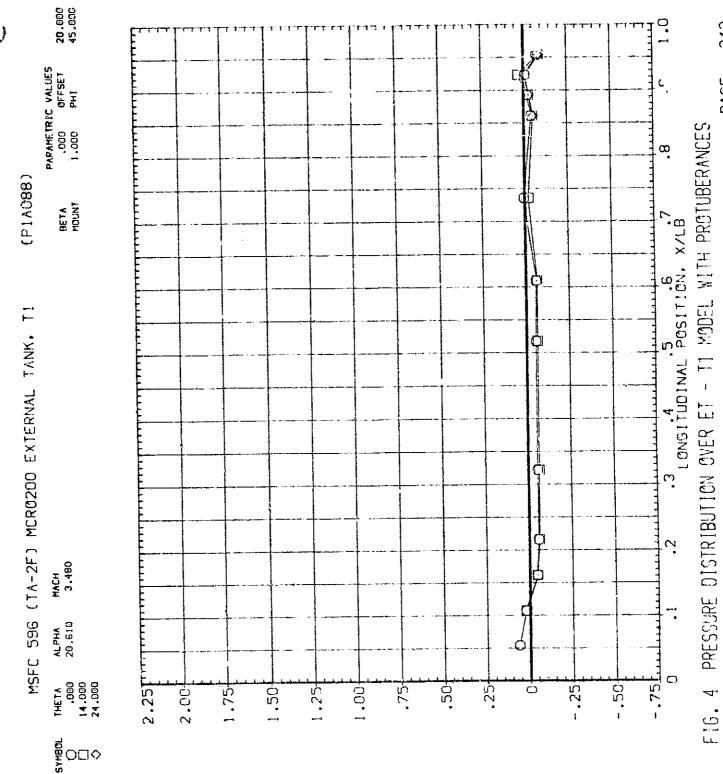






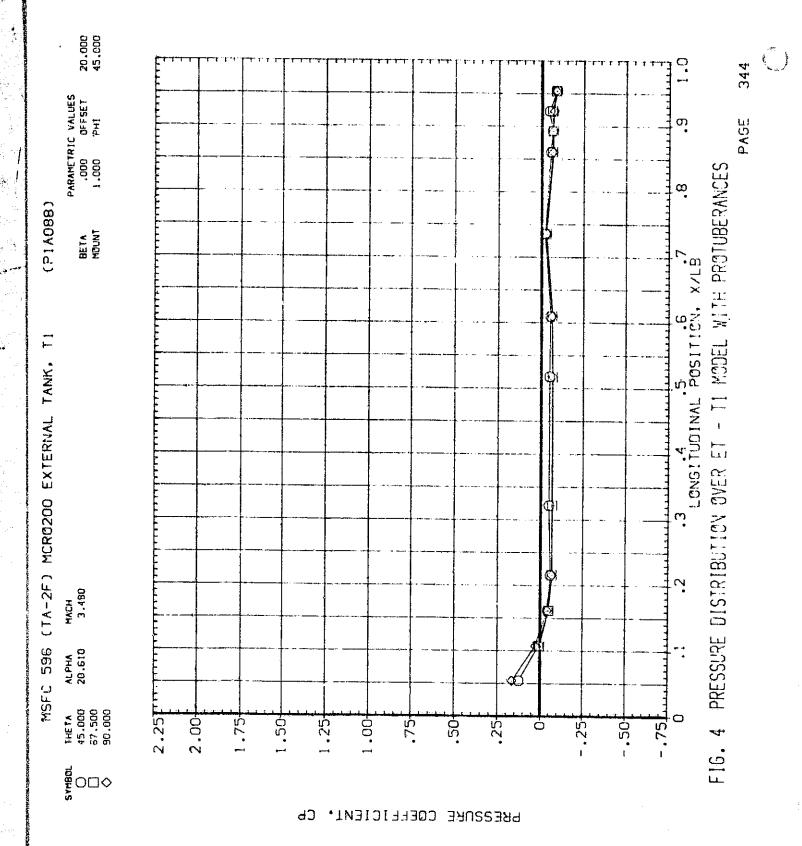
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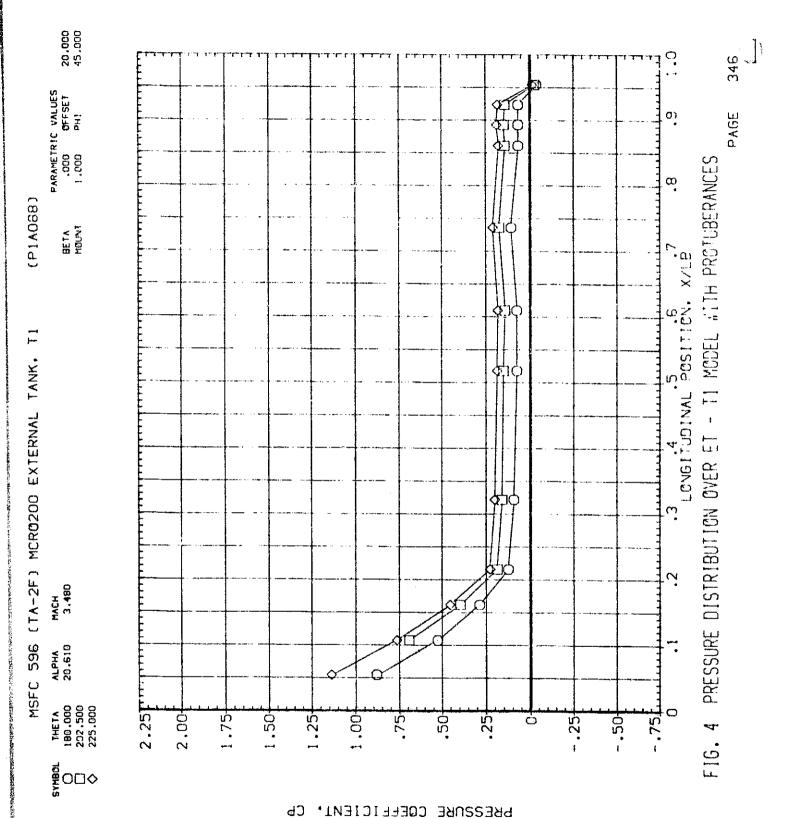
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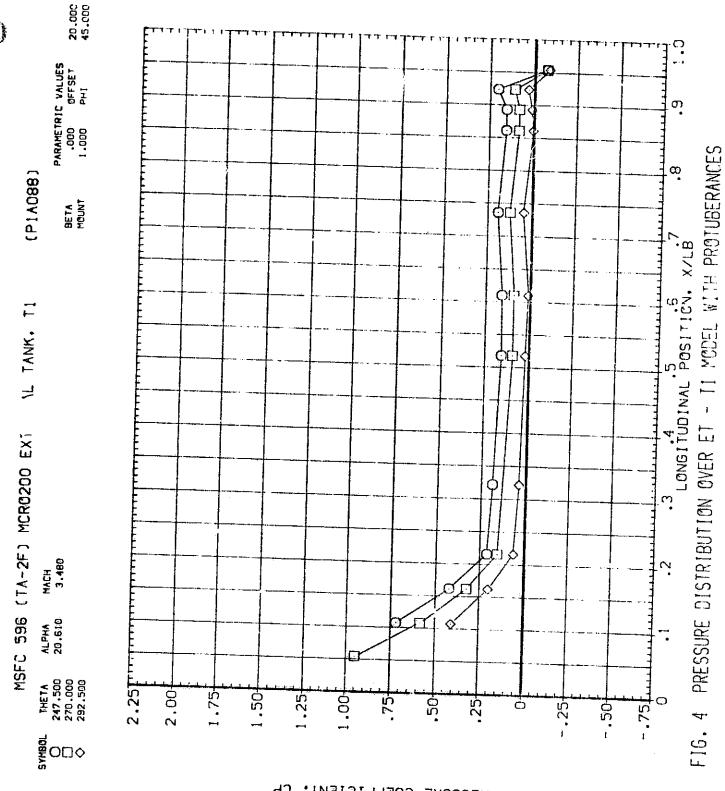
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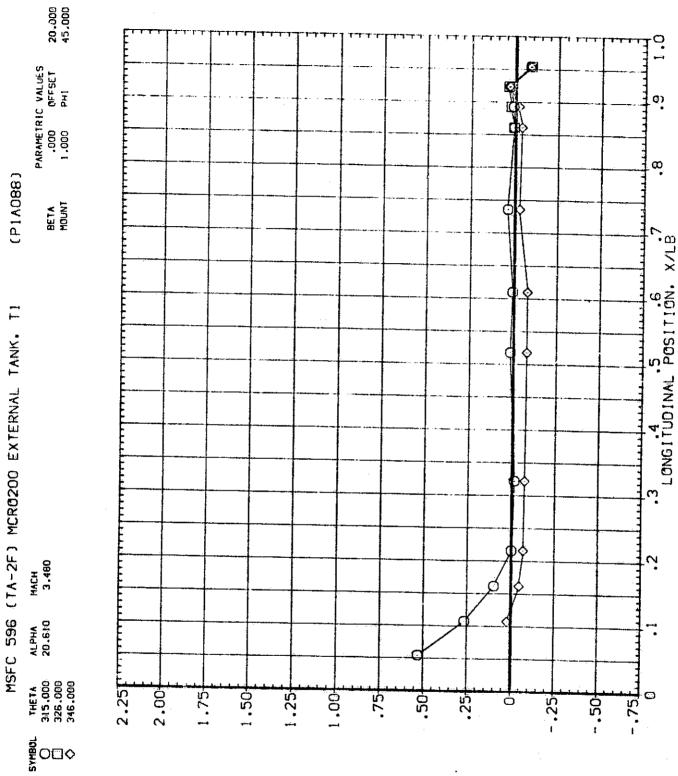
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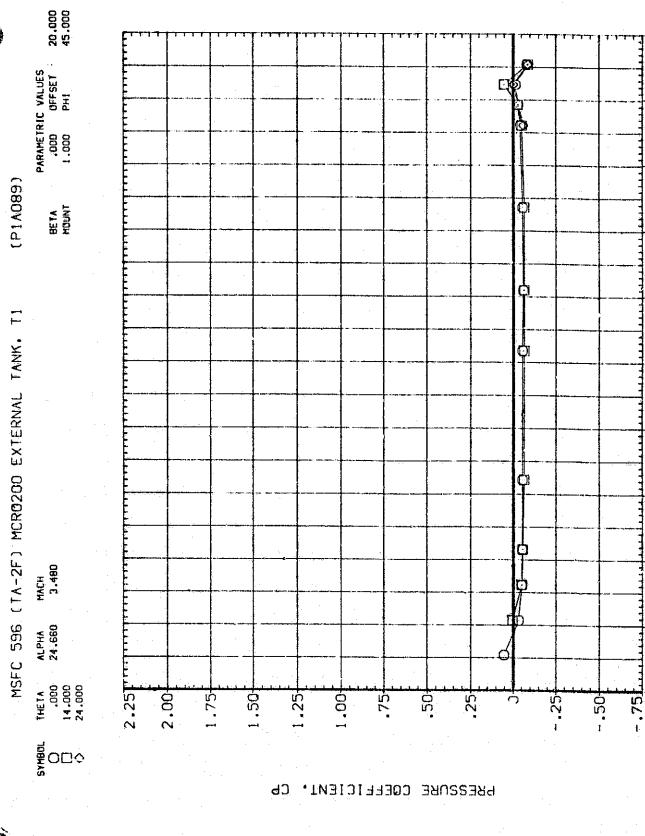


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TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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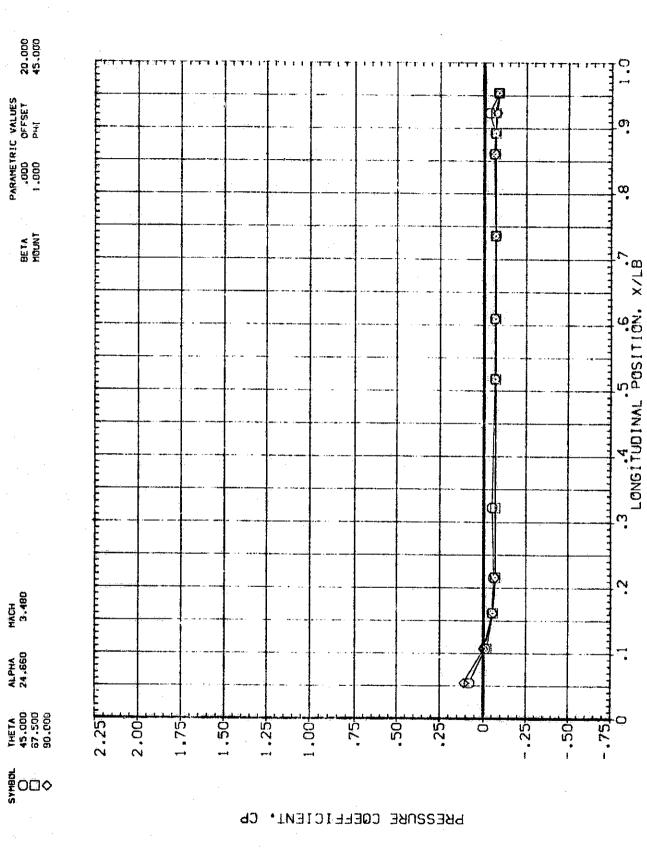


II MODEL WITH PROTUBERANCES 1 FIG. 4 PRESSURE DISTRIBUTION OVER ET

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(TA-2F) MCR0200 EXTERNAL TANK, TI

MSFC 596

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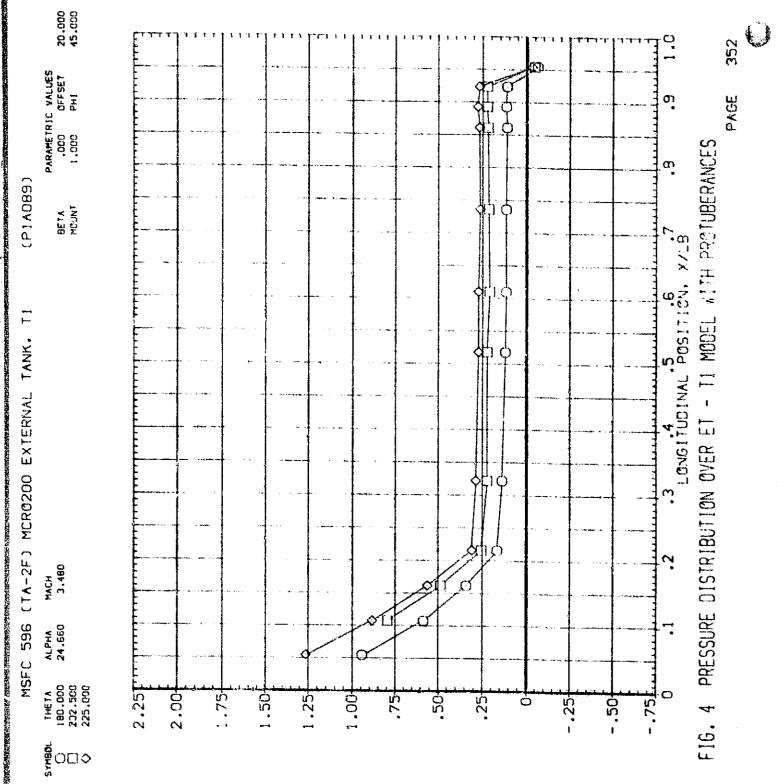
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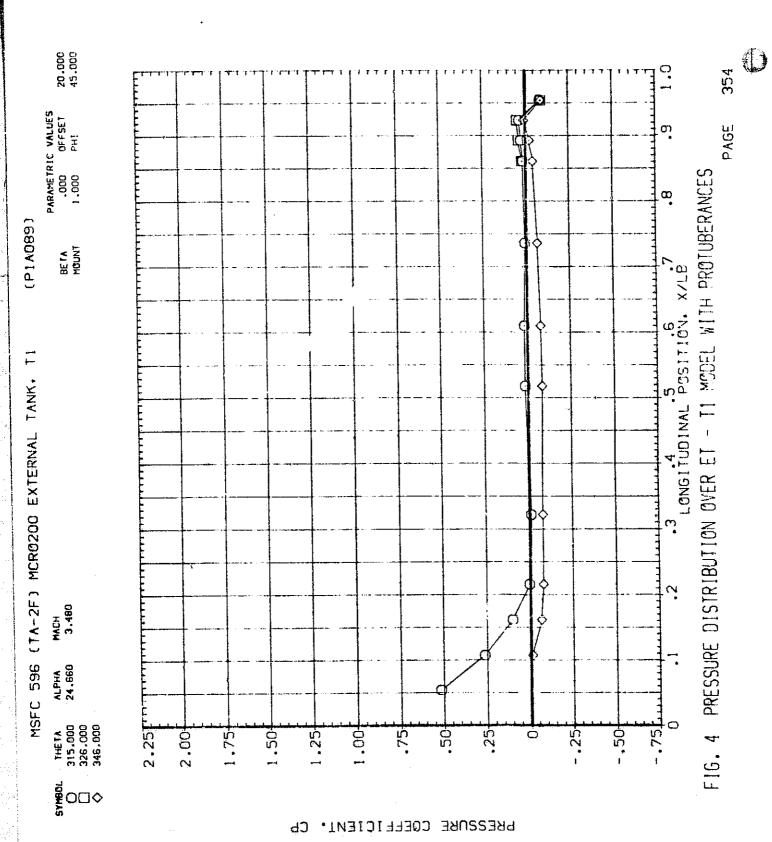
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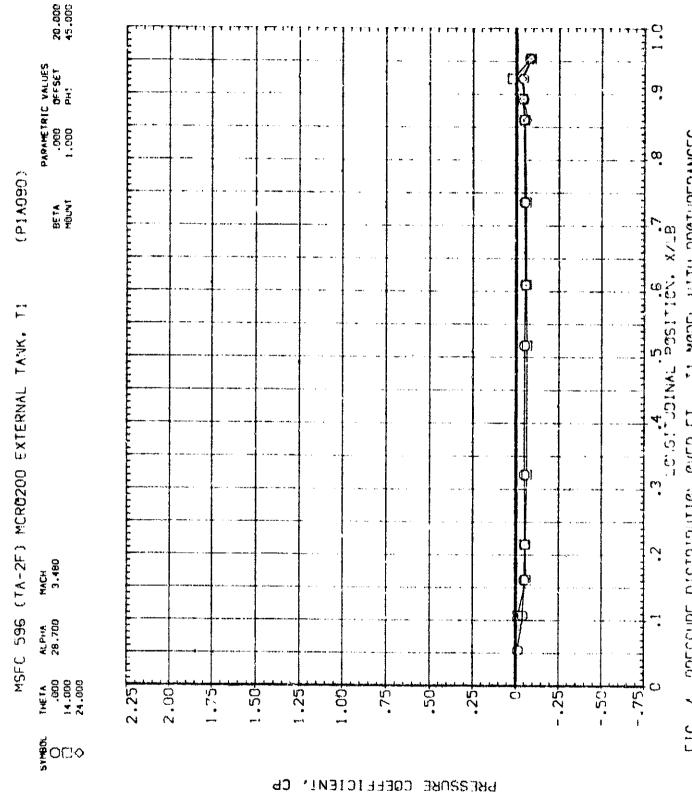


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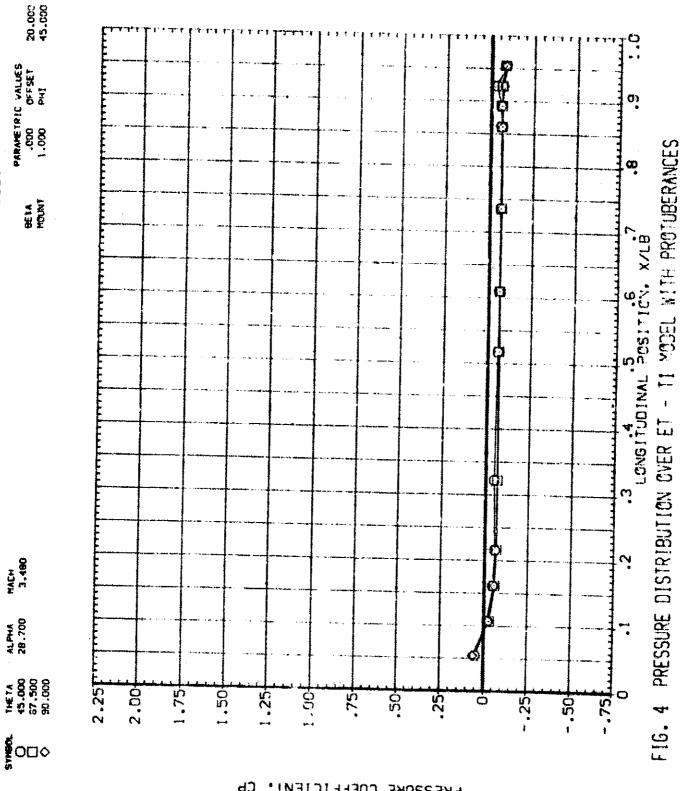
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1.000 PHI စာ FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES (P1A089) δ BE TA MOUNT POSITION X/LB MSFC 596 (1A-2F) MCR0200 EXTERNAL TANK, TI LONGITUDINAL MACH 3.480 ALPHA 24.660 2.25下 7HETA 247.500 270.000 292.500 1.25 1.00 .75 50 .25 -.75 1.75 - 25 -.50 2.00 1.50 Ö age O□◇ PRESSURE COEFFICIENT,

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FIS. 4 PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES



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MSFC 596 (IA-2F) MCR0200 EXTERNAL TANK, TI

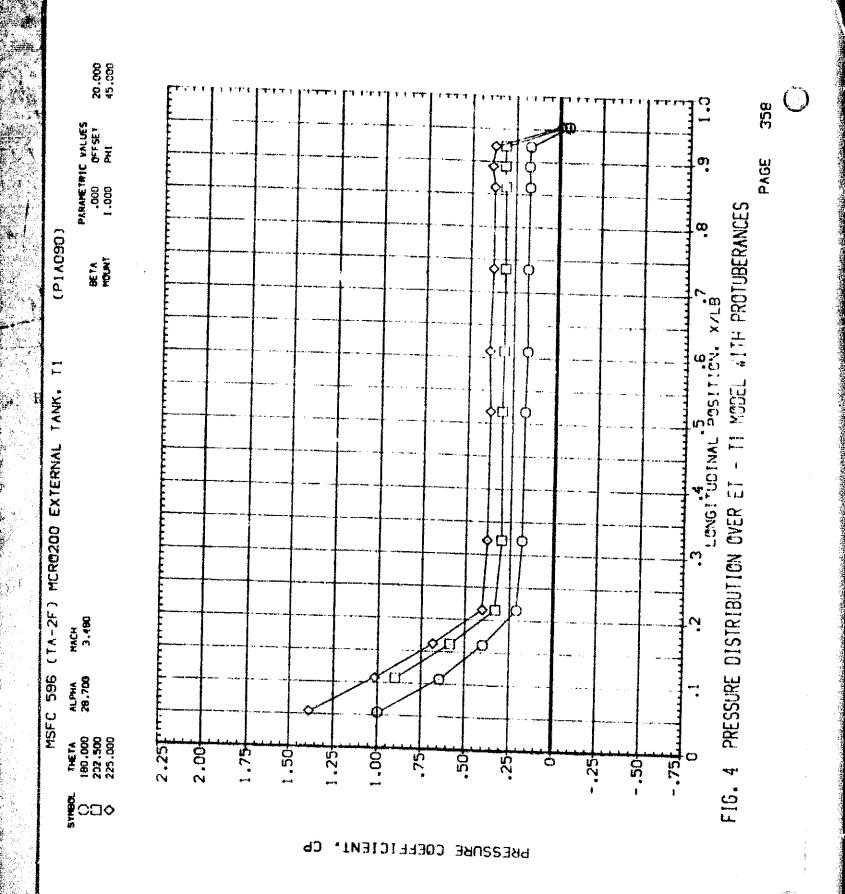
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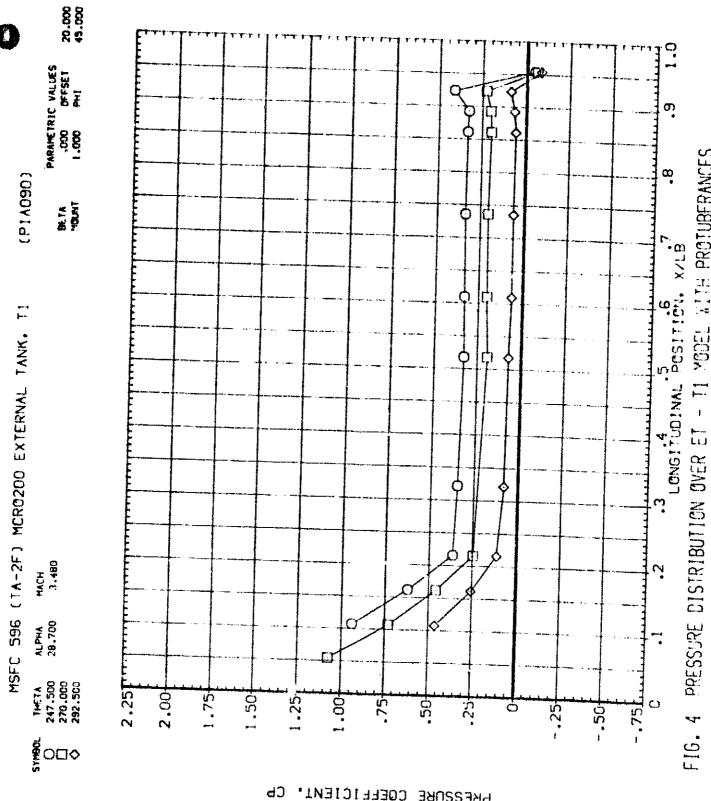
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II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



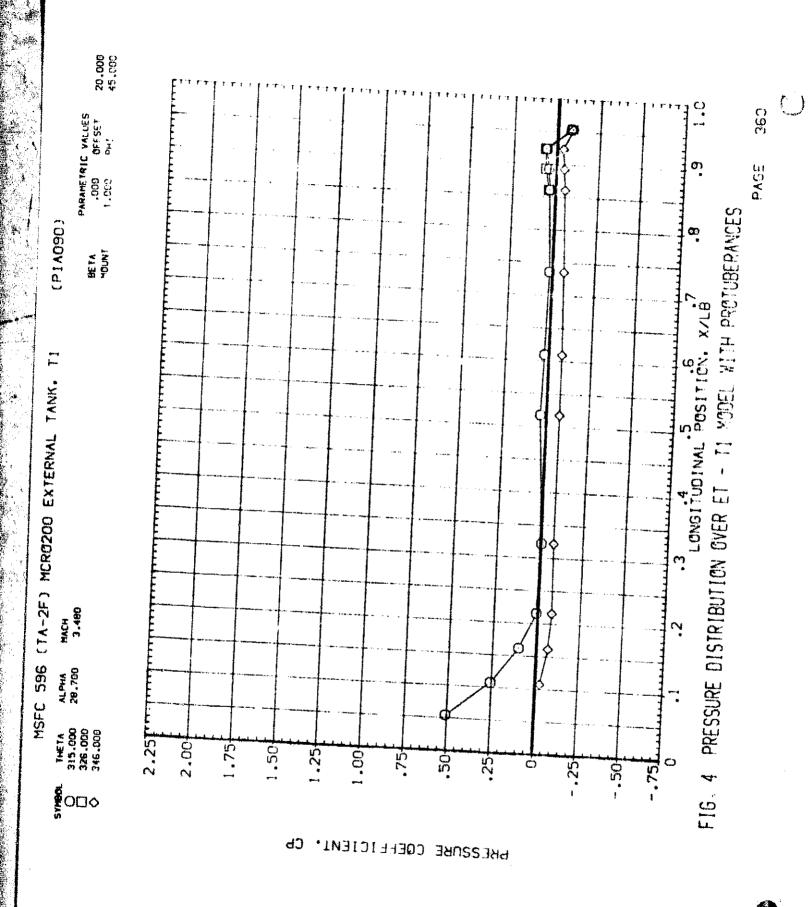
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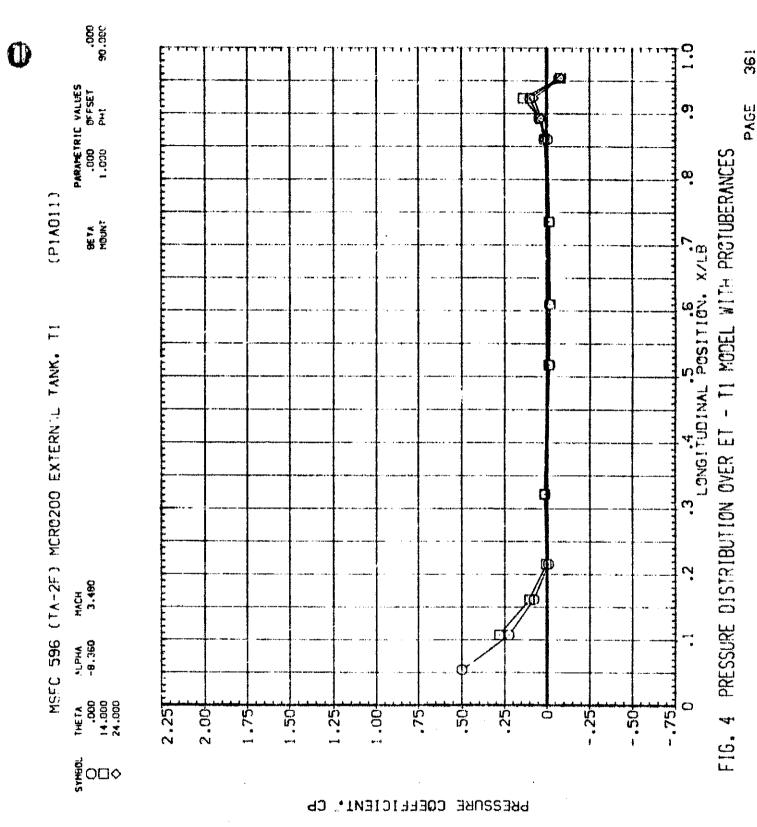


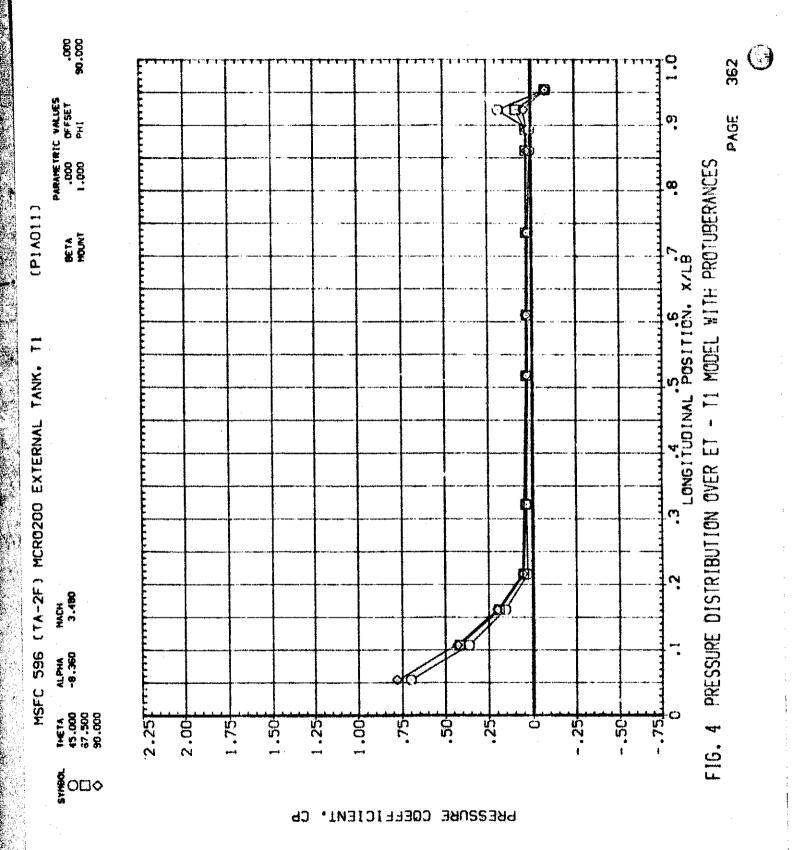
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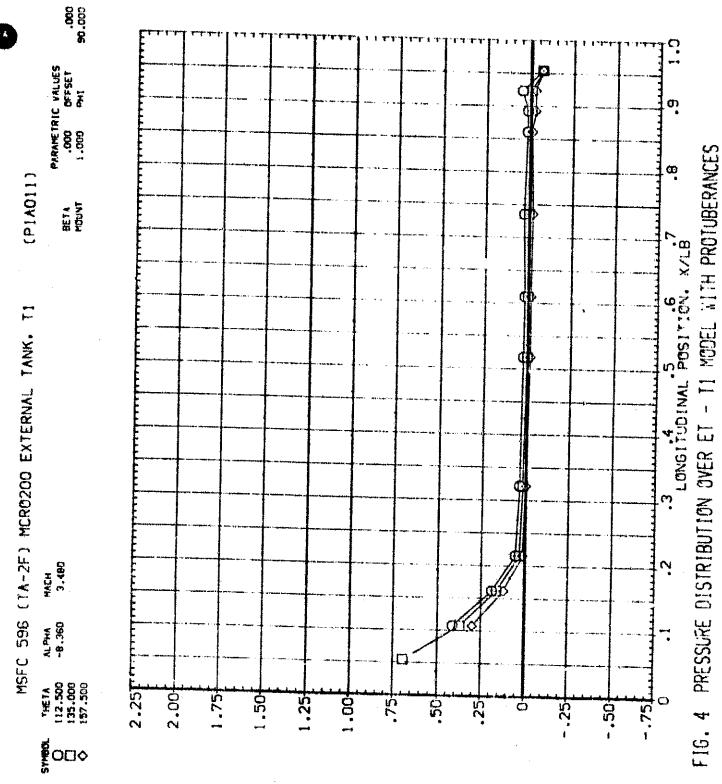
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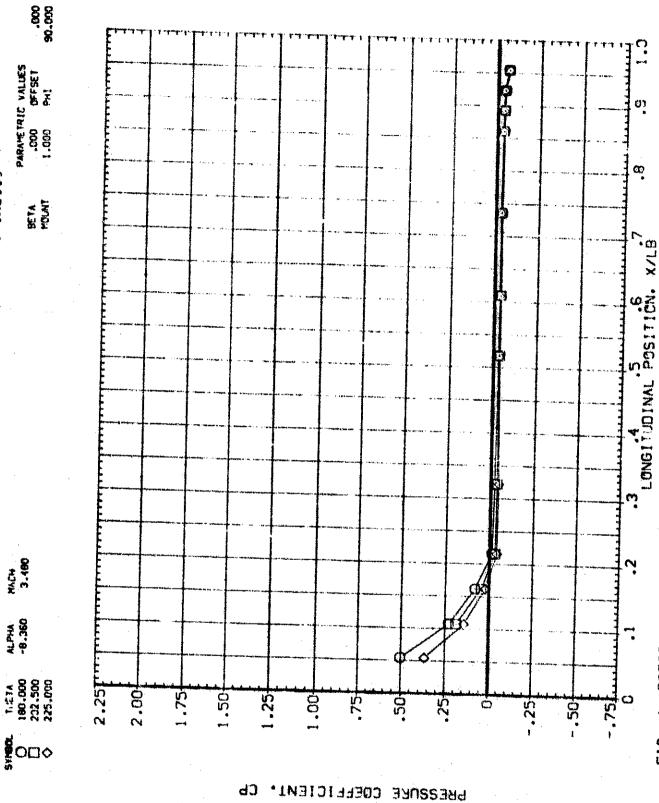








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MSEC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI

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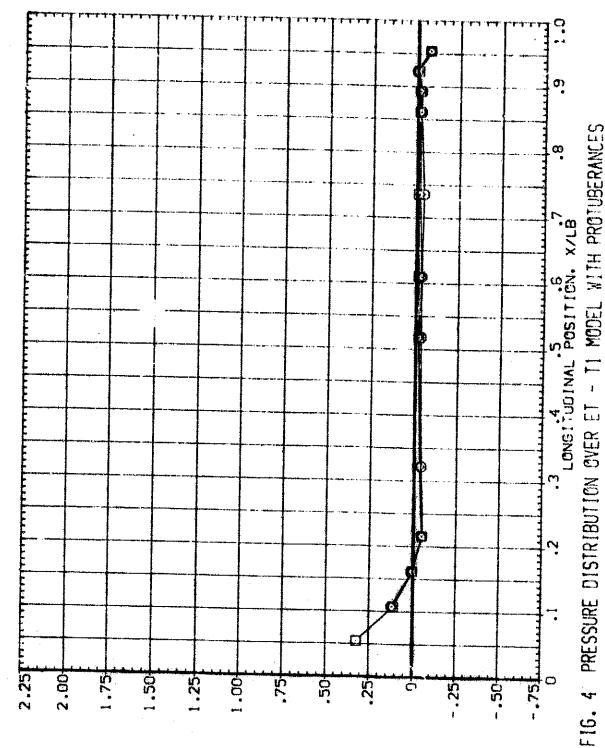
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II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER

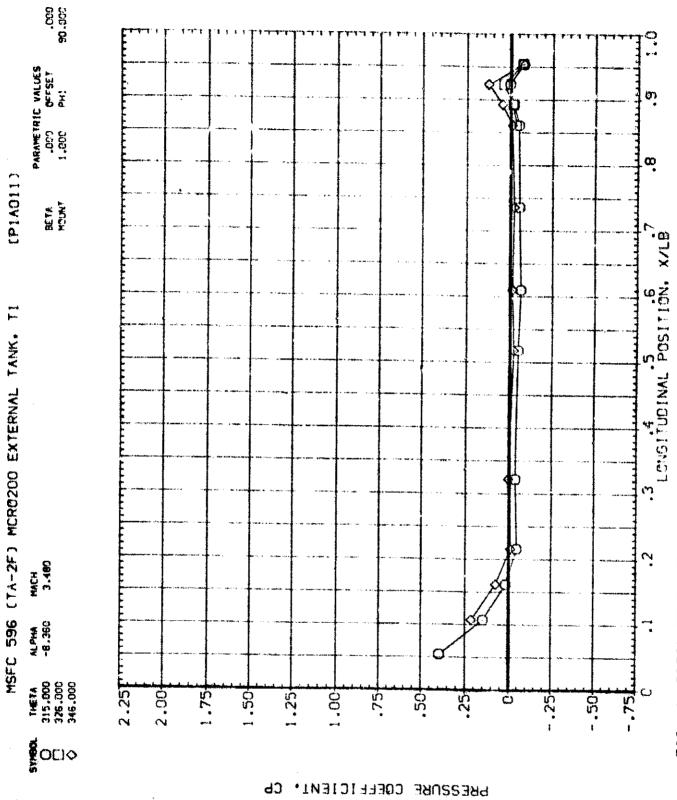
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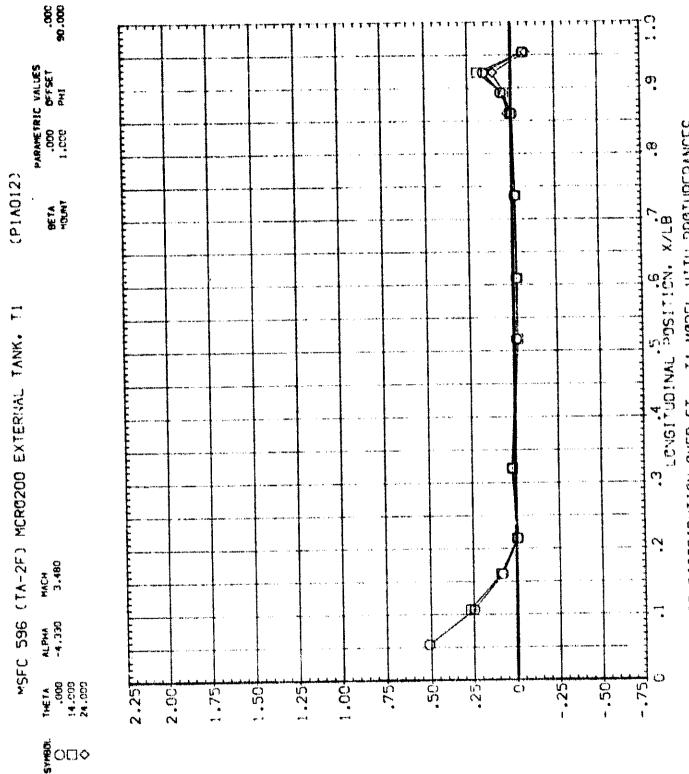


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- 11 MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION CVER

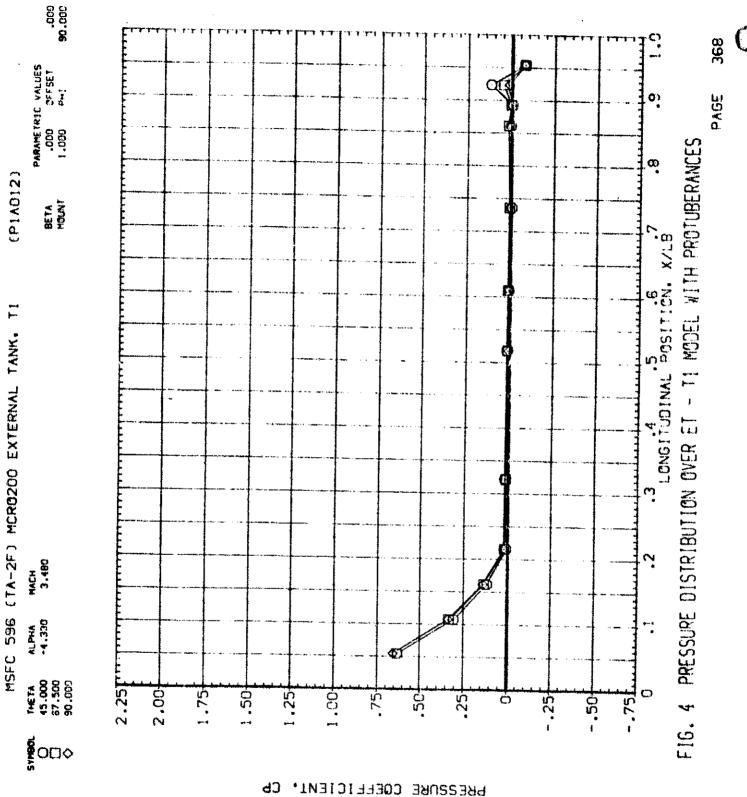


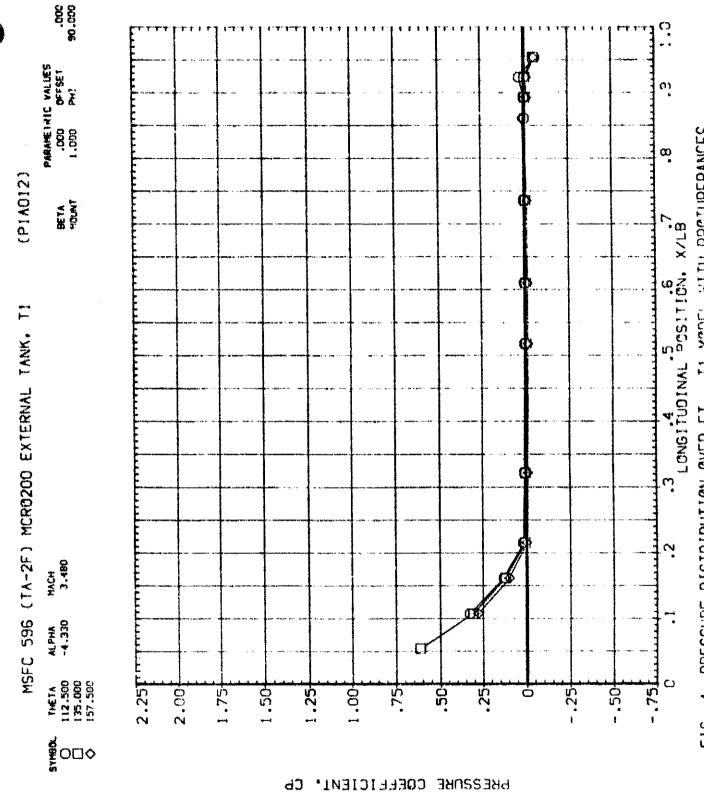


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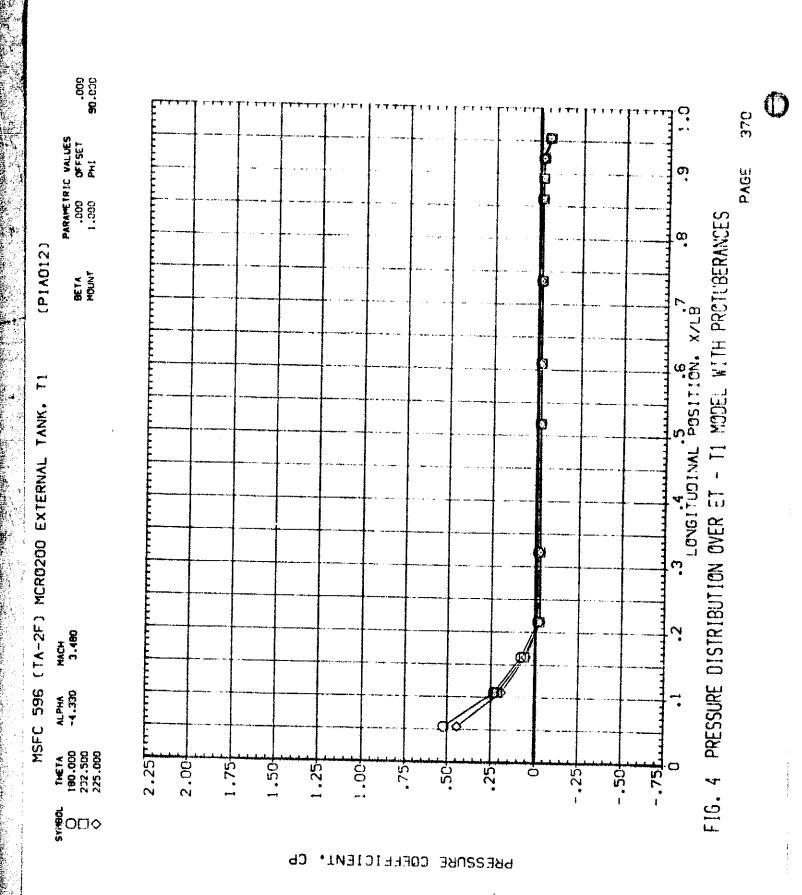
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PASE FIS. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

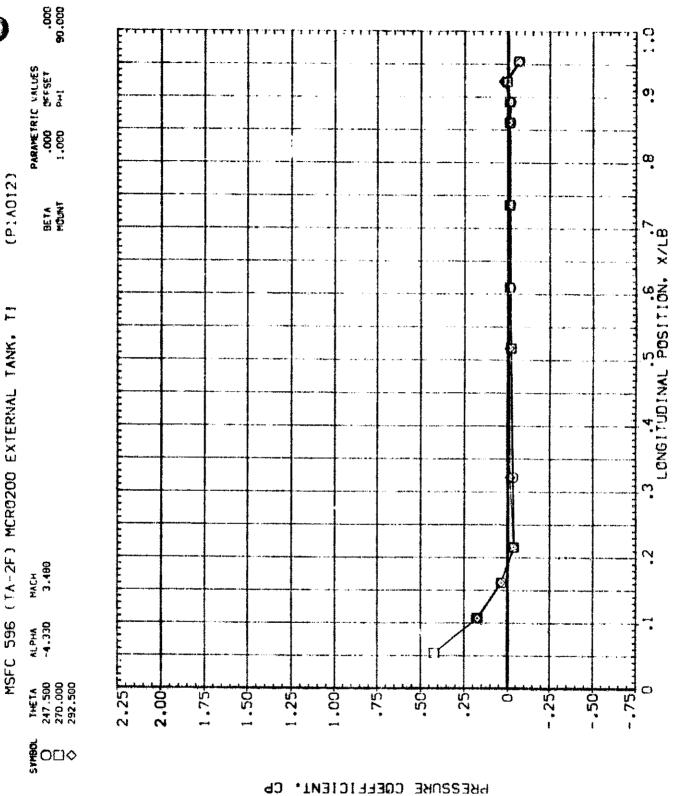




- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



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- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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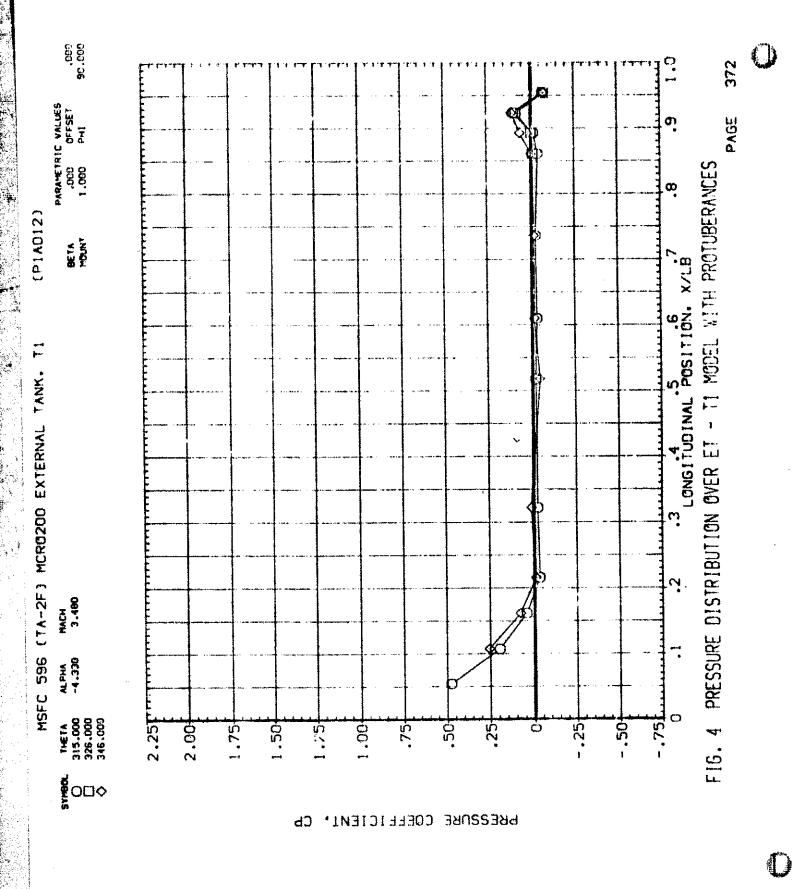


FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

.000 .000 .000 374 PARAMETRIC VALUES .000 OFFSET 1.000 PHI တံ PASE 11 MODEL WITH PROTUBERANCES ထ (P1A013) BETA MOUNT POSITION, XALE MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI Lengi Tubinal 1 FIG. 4 PRESSURE DISTRIBUTION OVER ET ņ 3.480 ALPHA -.280 2.25F 1HETA 45.000 67.500 90.000 2.00 1.50 1.75 1.25 1.00 75 .50 25 -.25 -.50 -.75 ¥ O□◊ PRESSURE COEFFICIENT, CP

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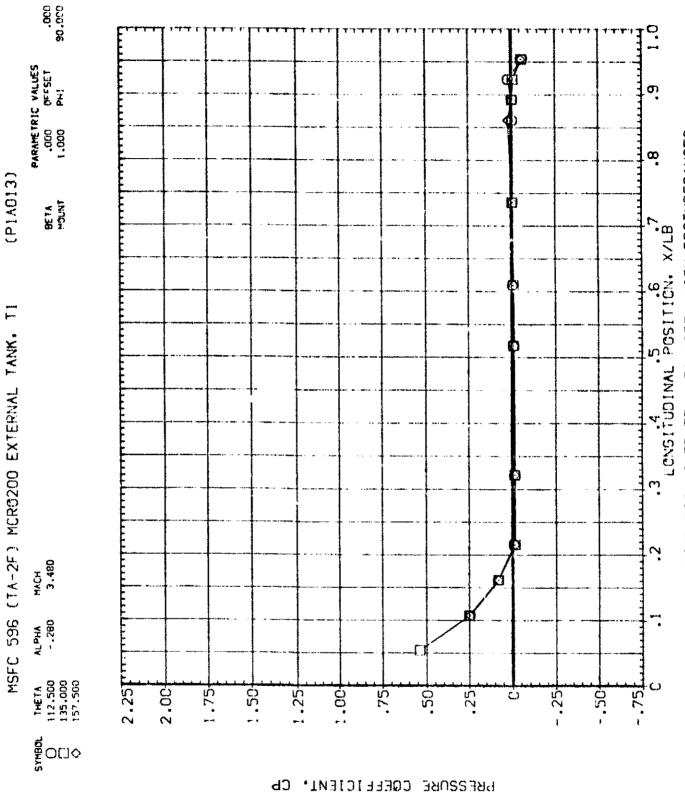


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

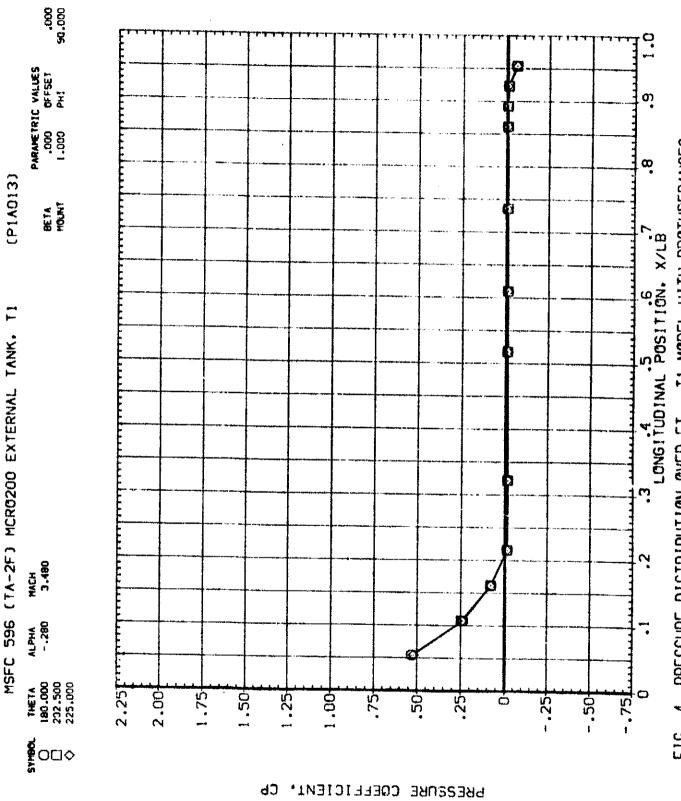
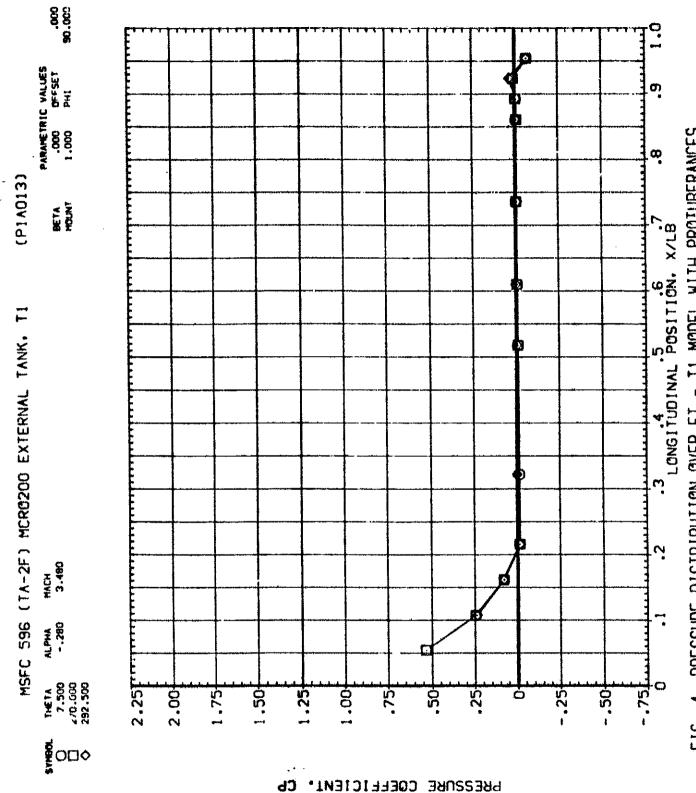


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

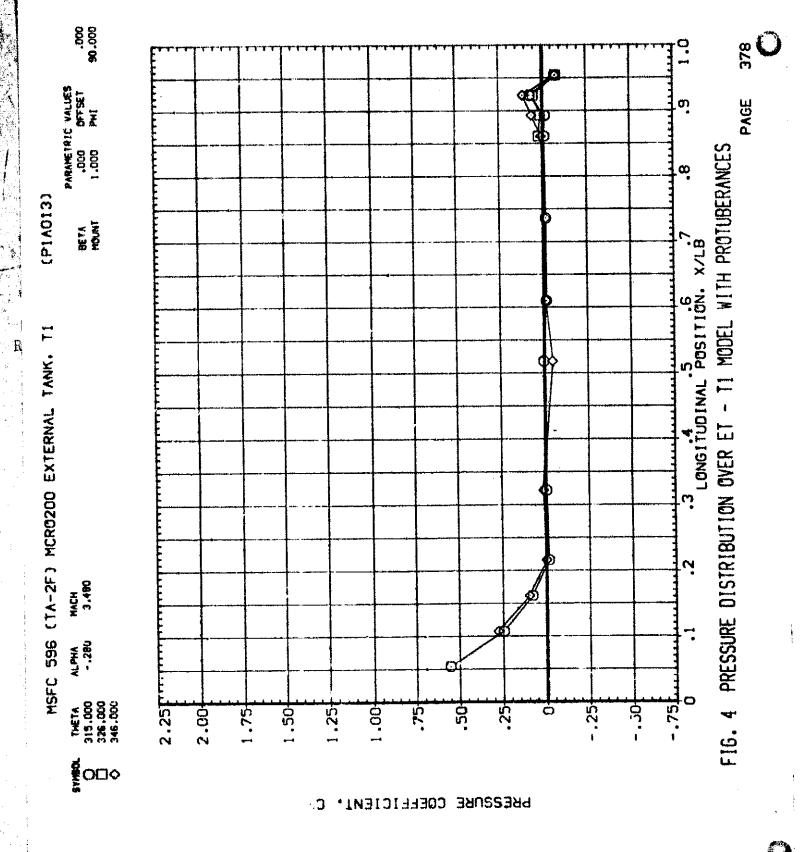
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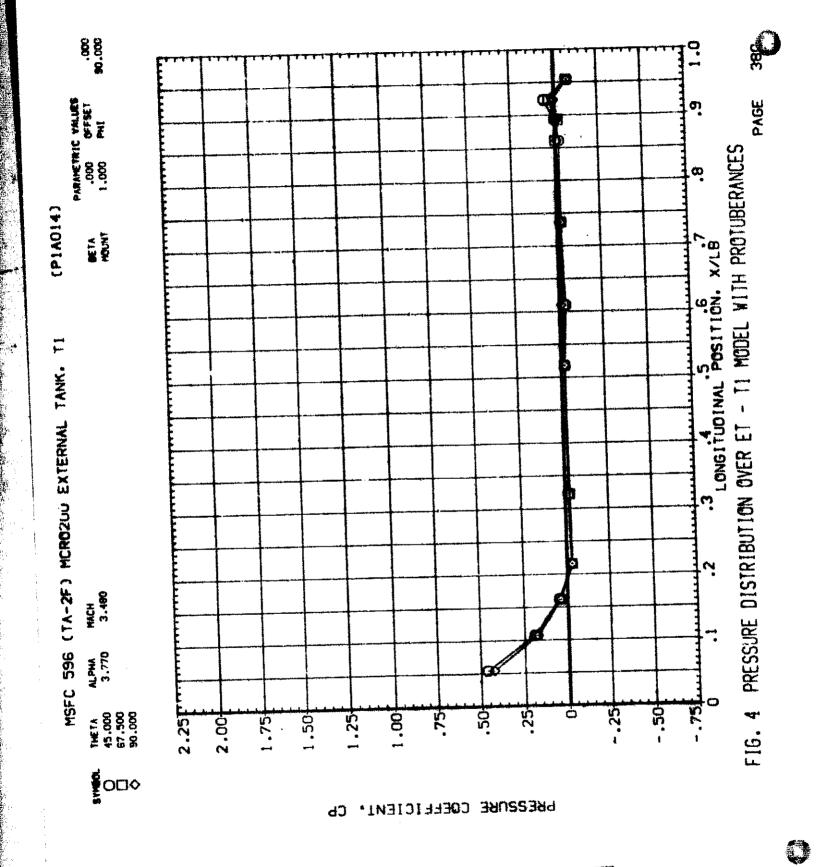
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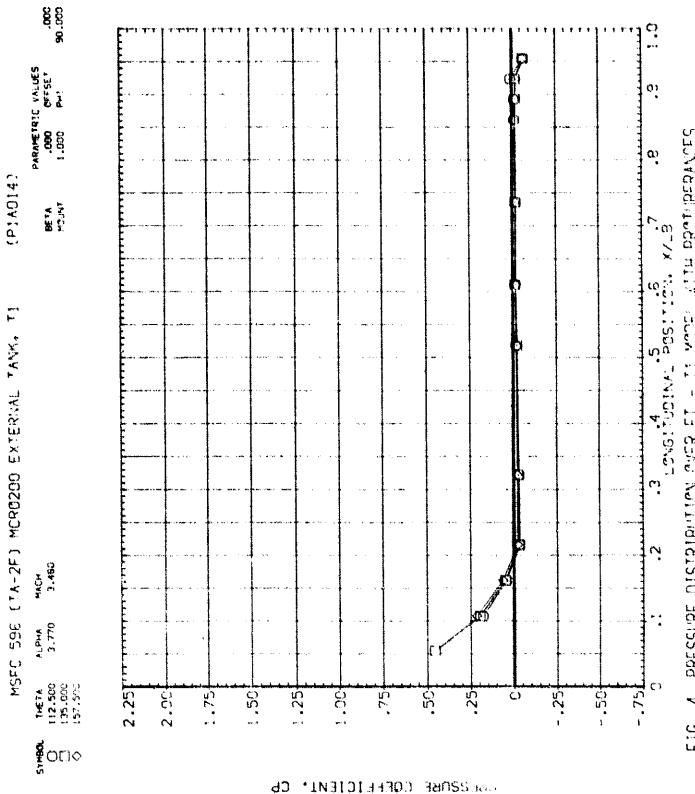


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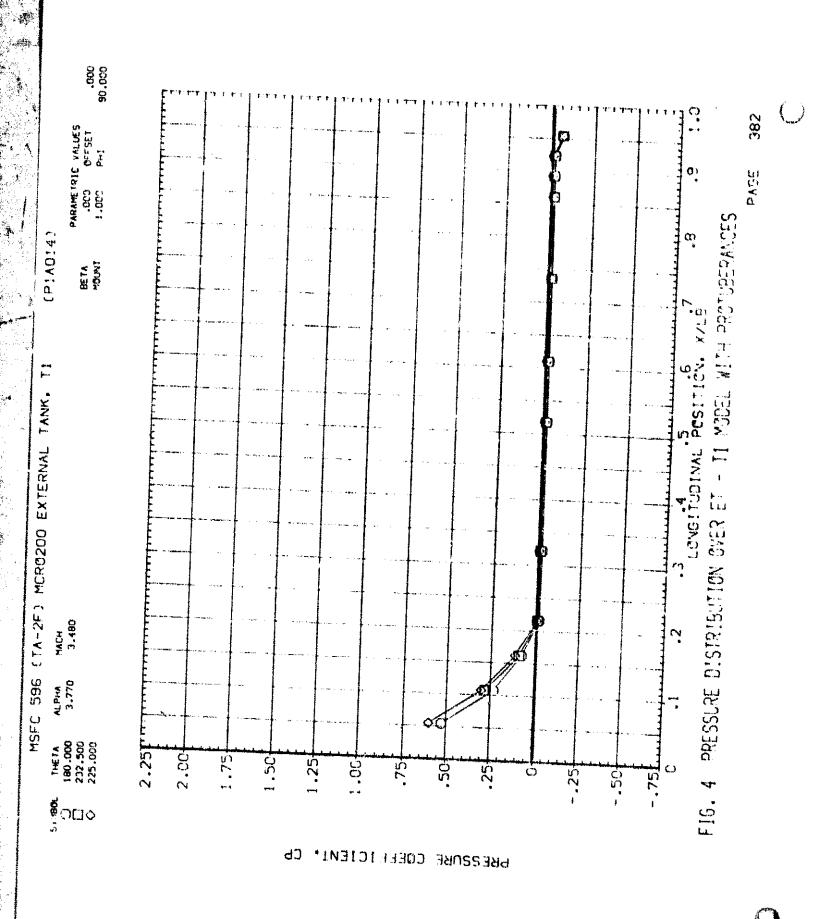


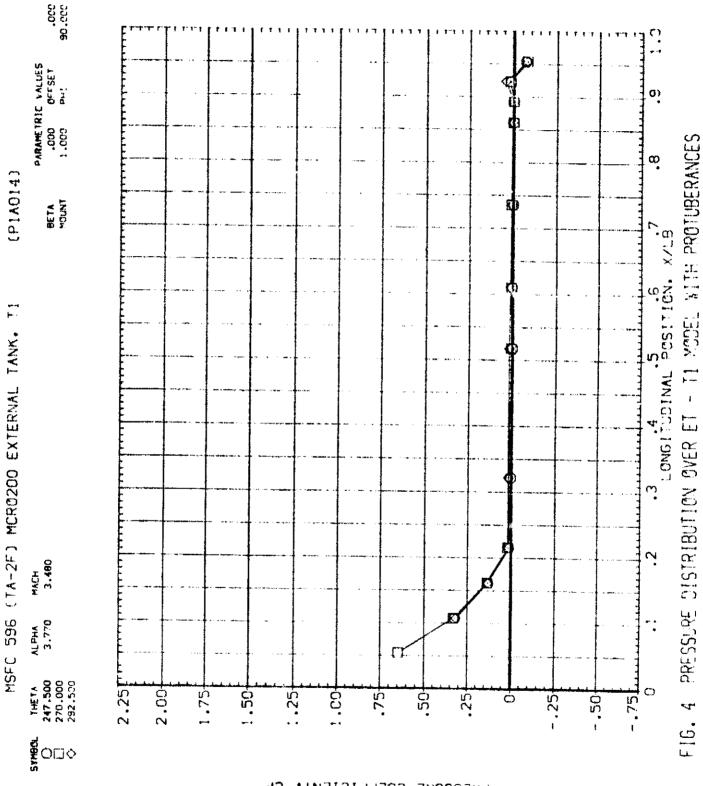
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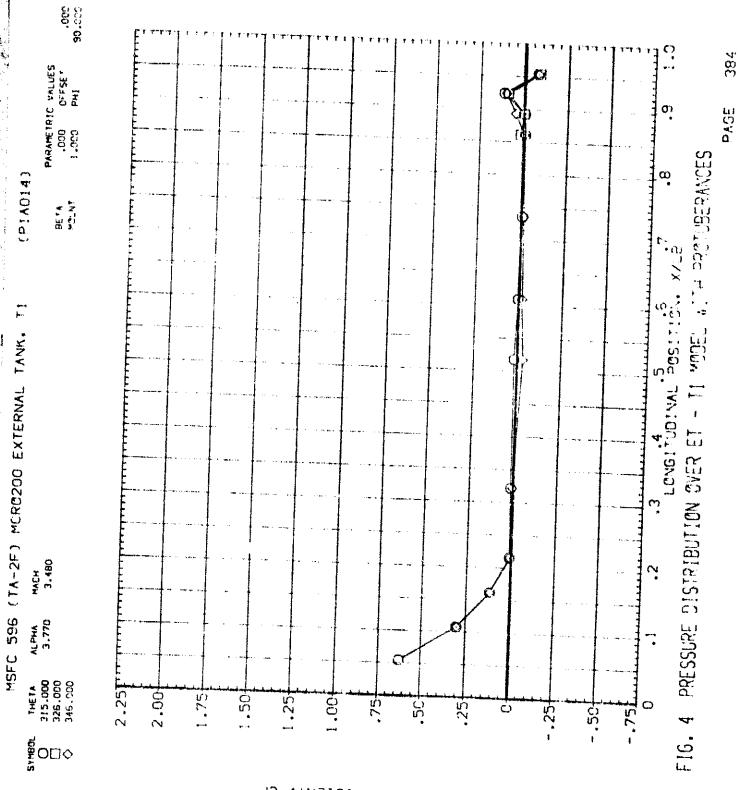
MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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II MODEL WITH PROTUBERANCES 1 PRESSURE DISTRIBUTION OVER ET

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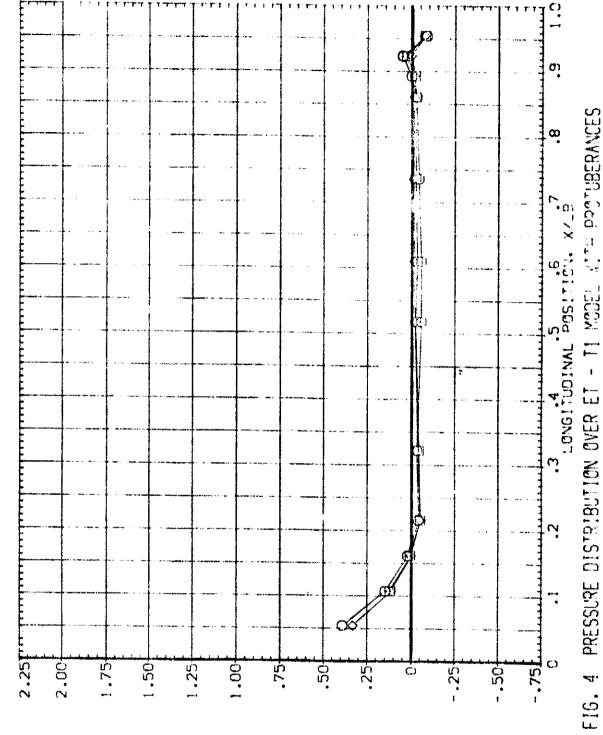
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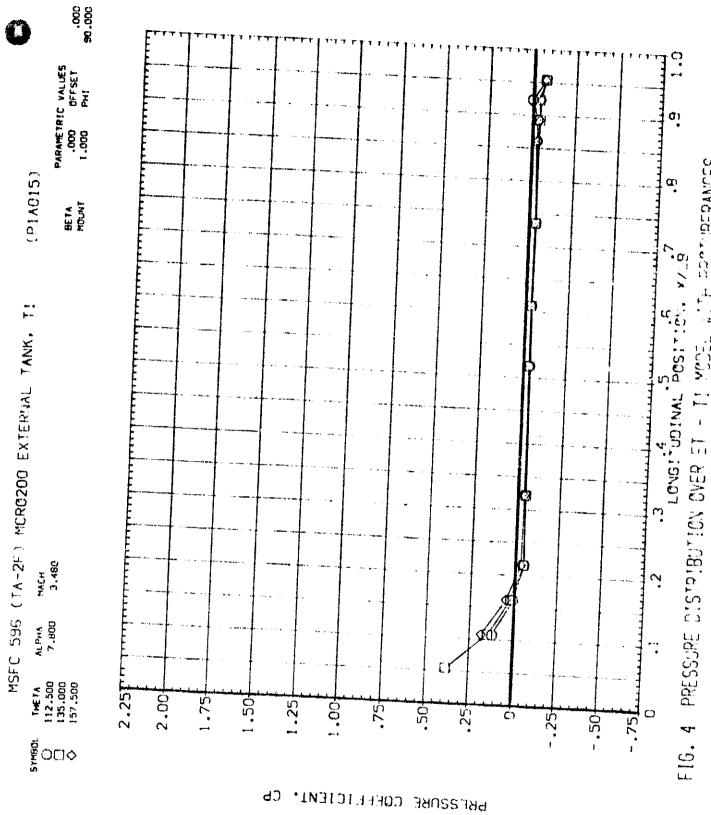
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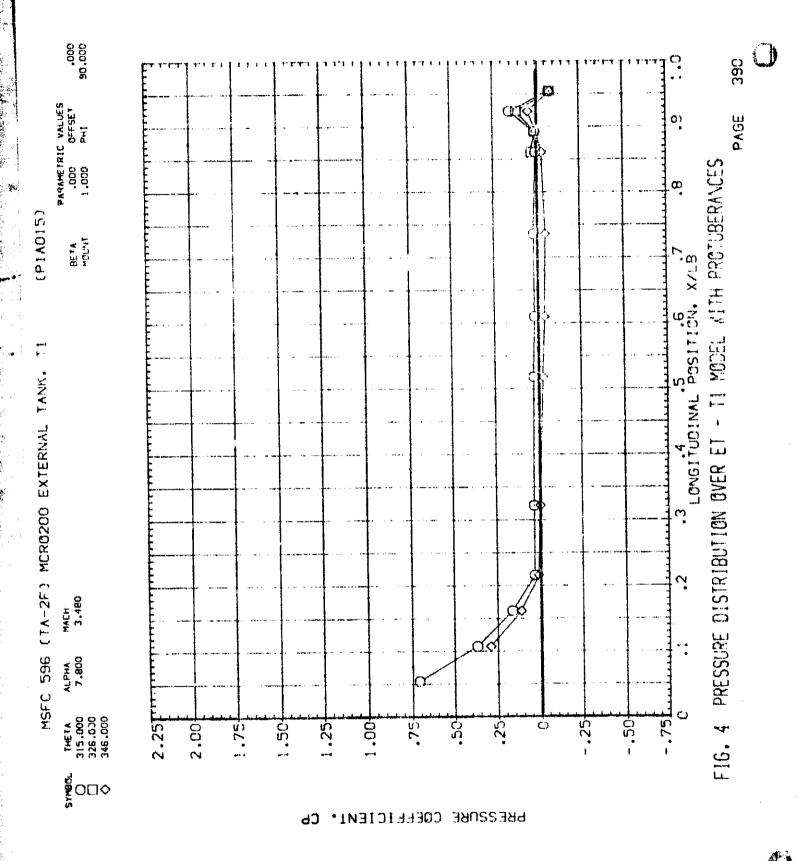
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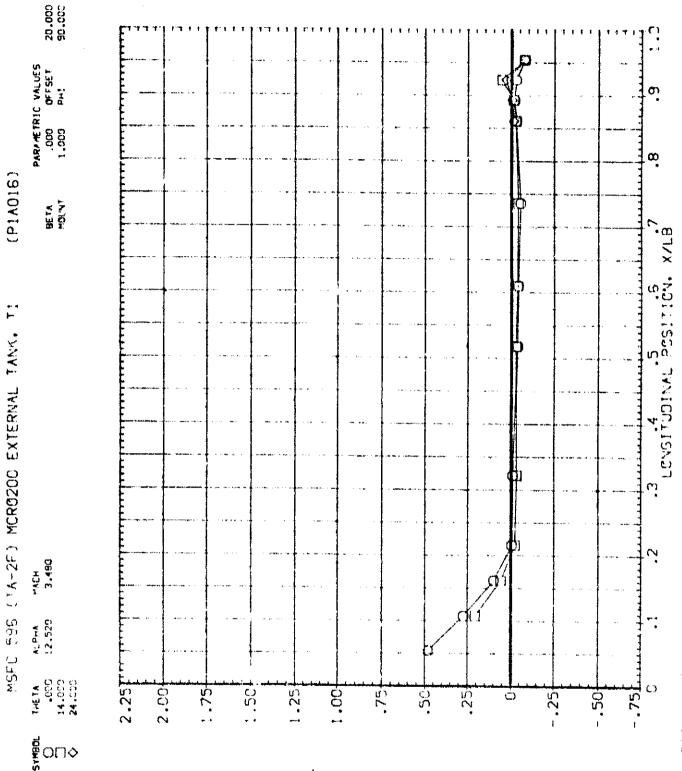
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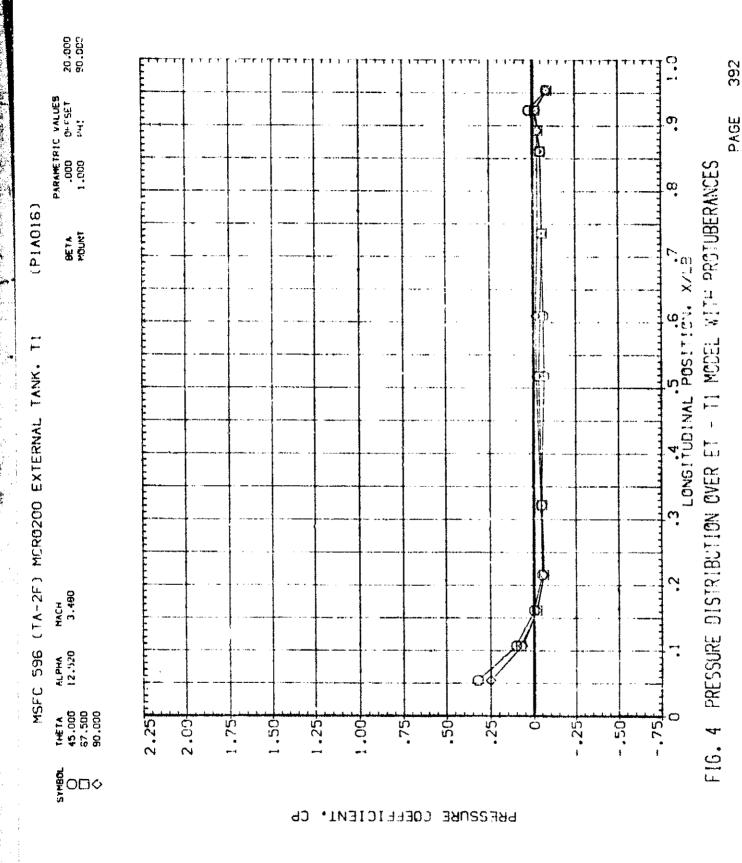




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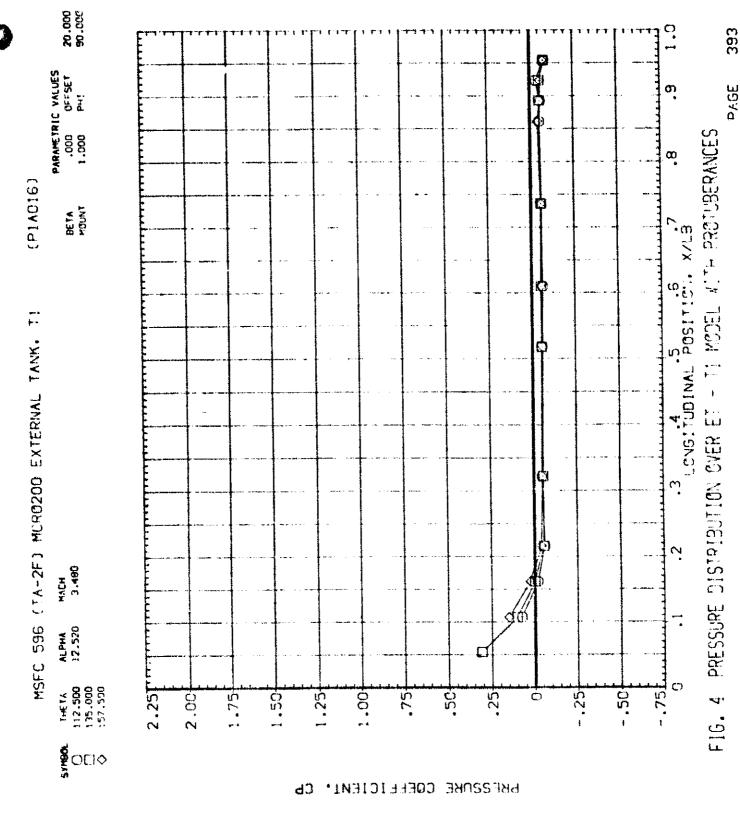
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II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

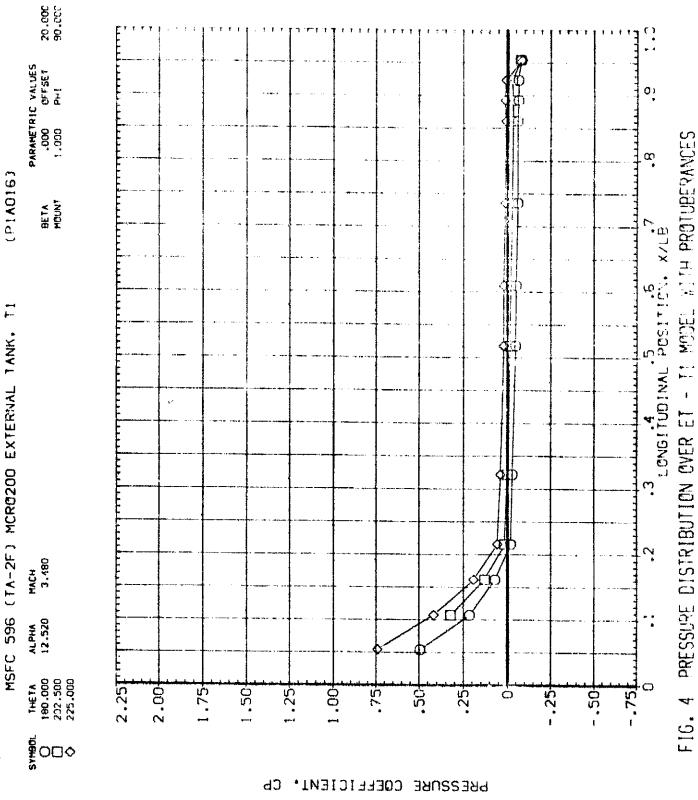


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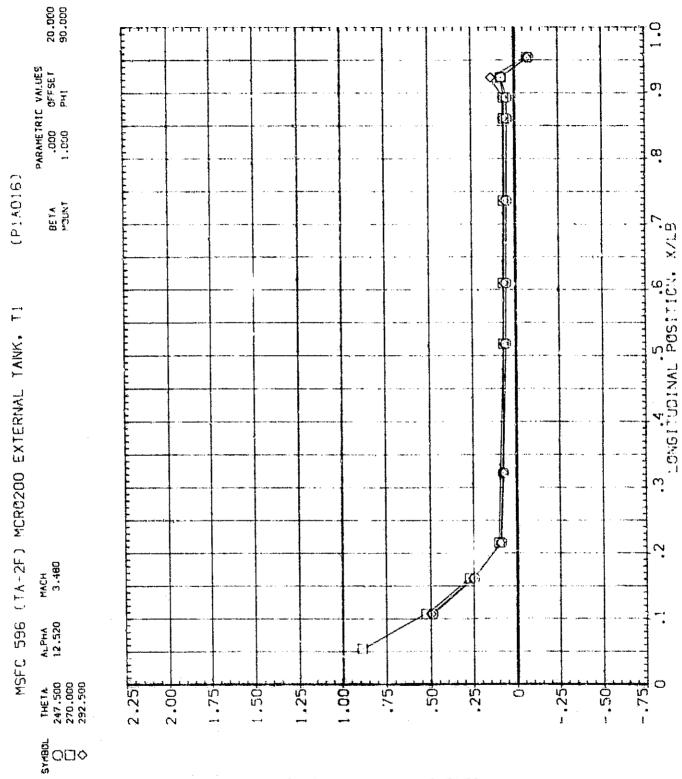
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FIG. 4 PRESSURE DISTRIBUTION OVER ET

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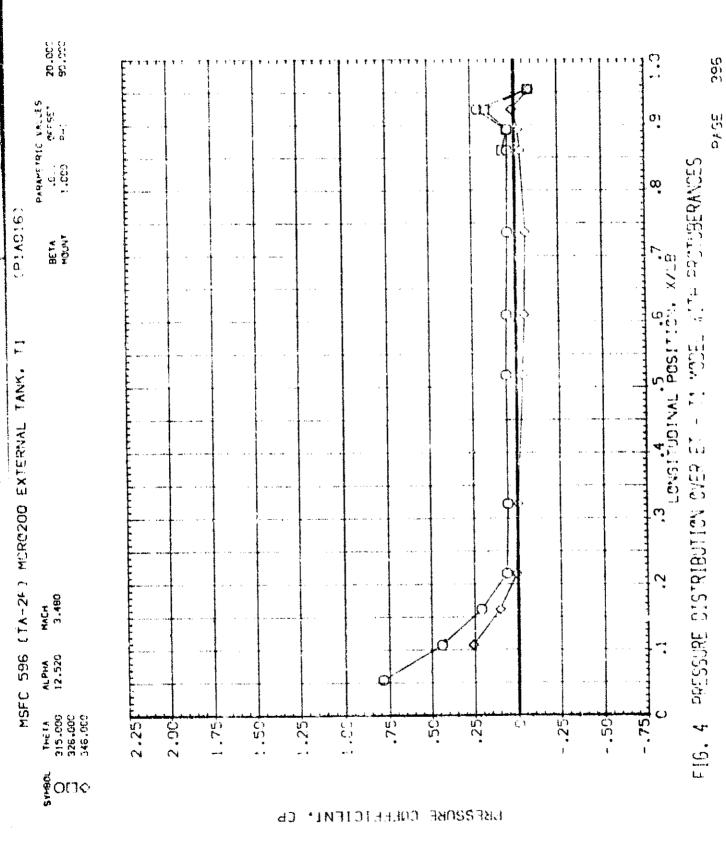


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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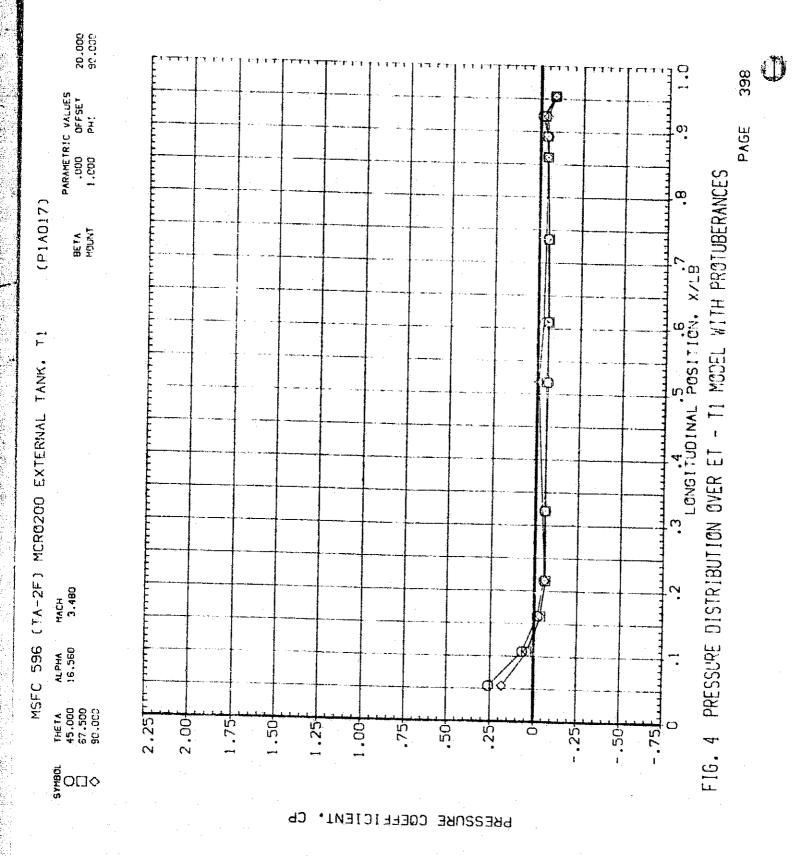
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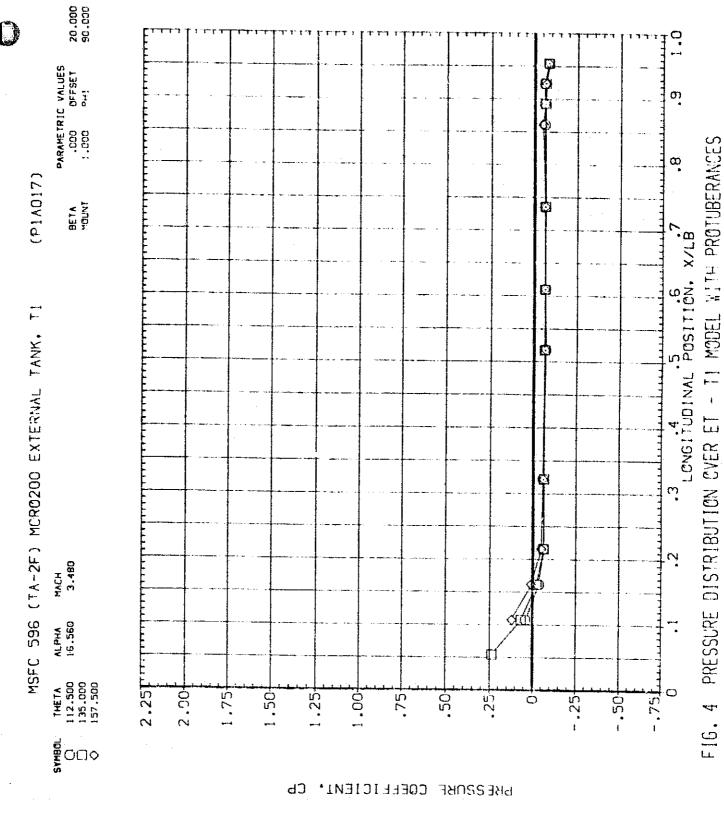
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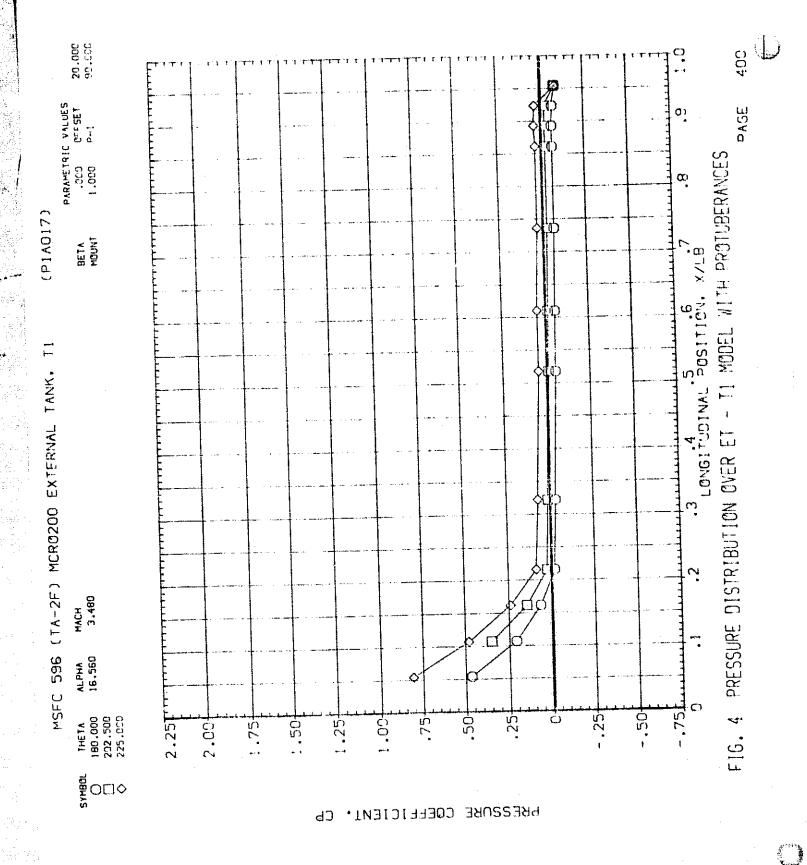
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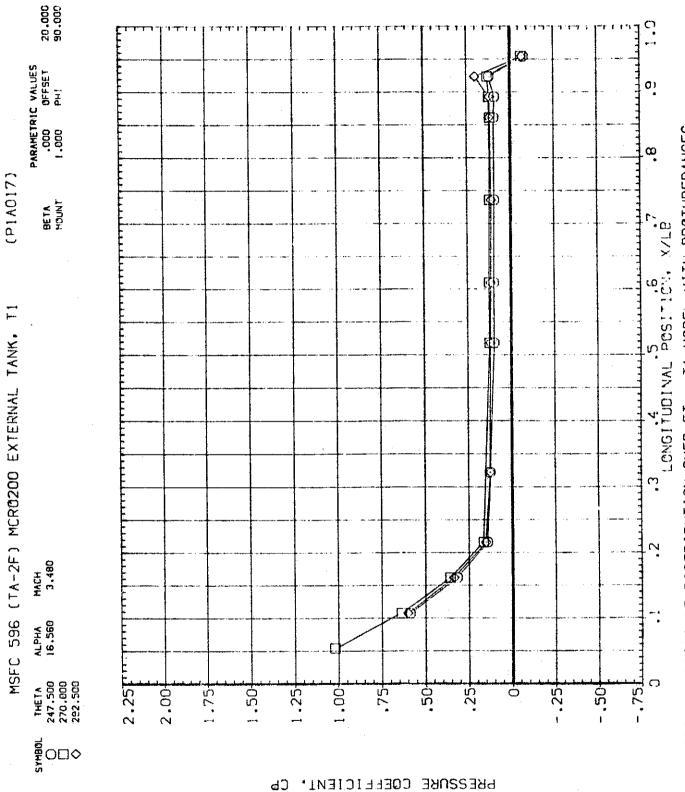
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PAGE WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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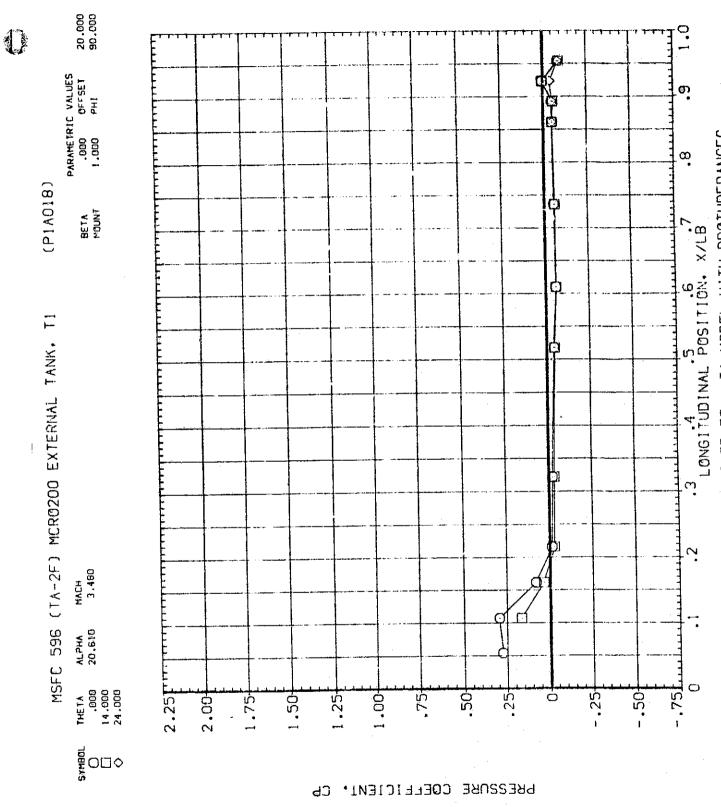
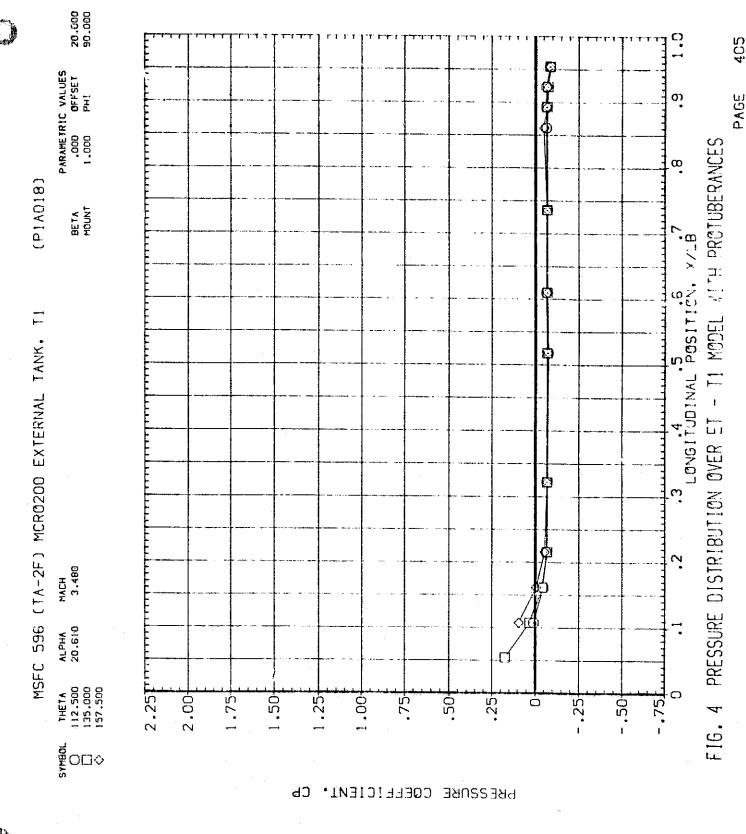
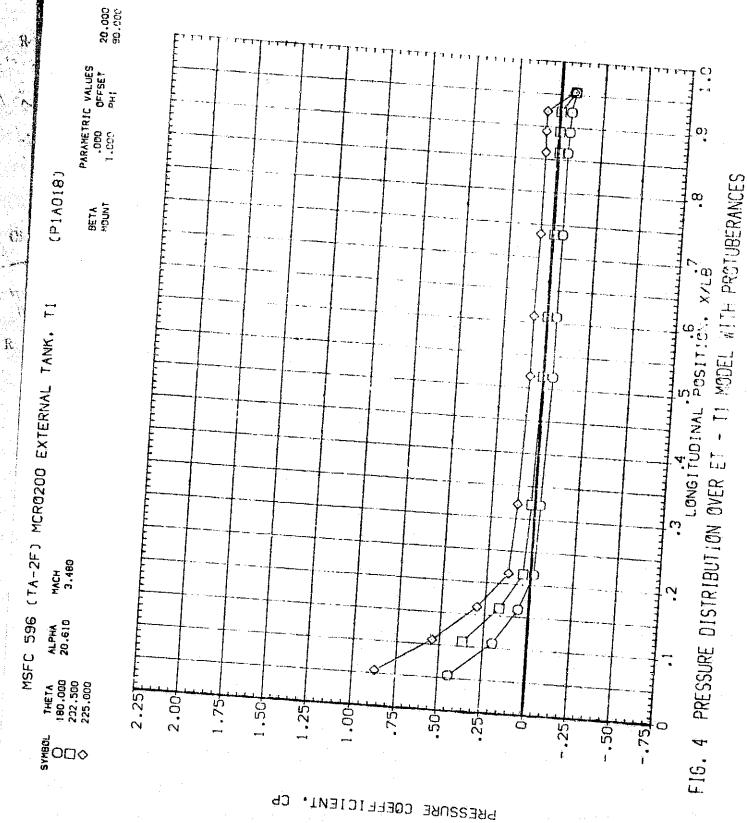


FIG. 4 PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES

- II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



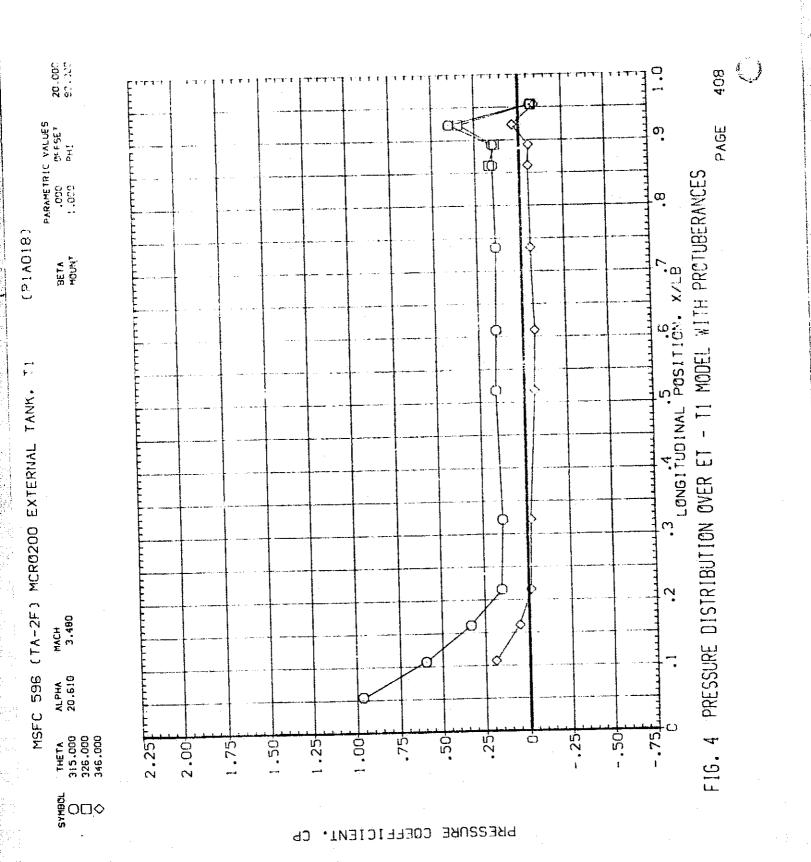


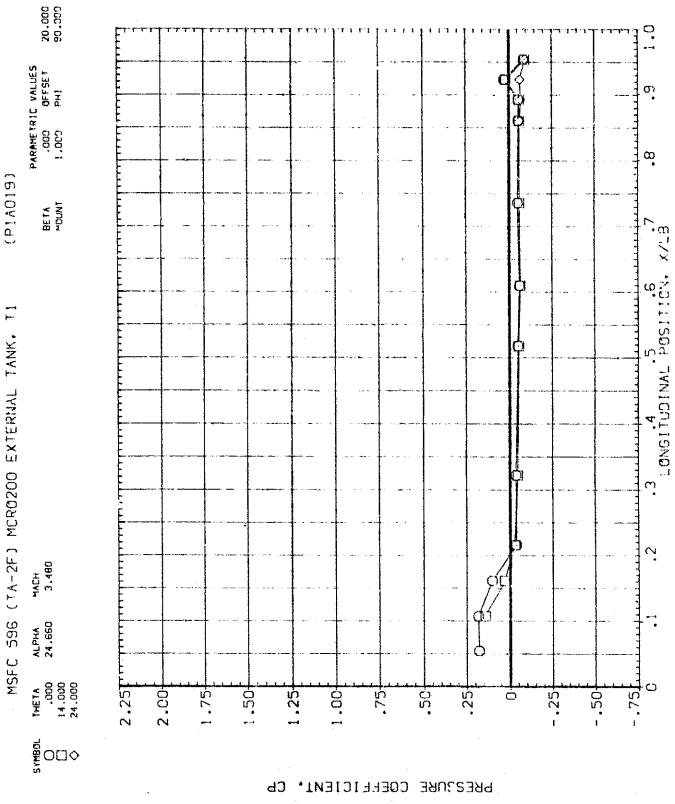
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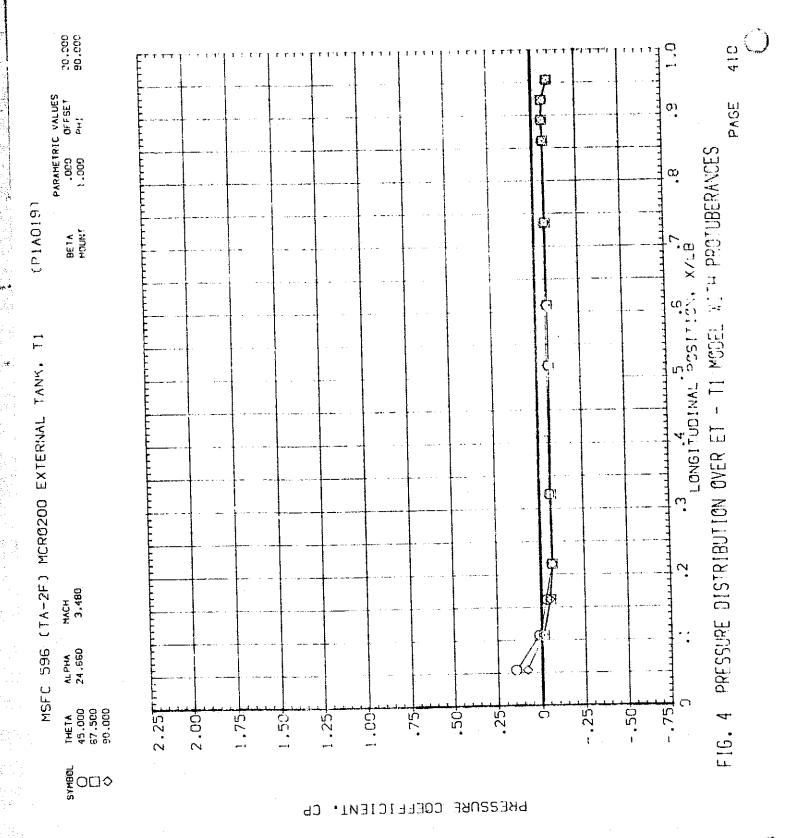
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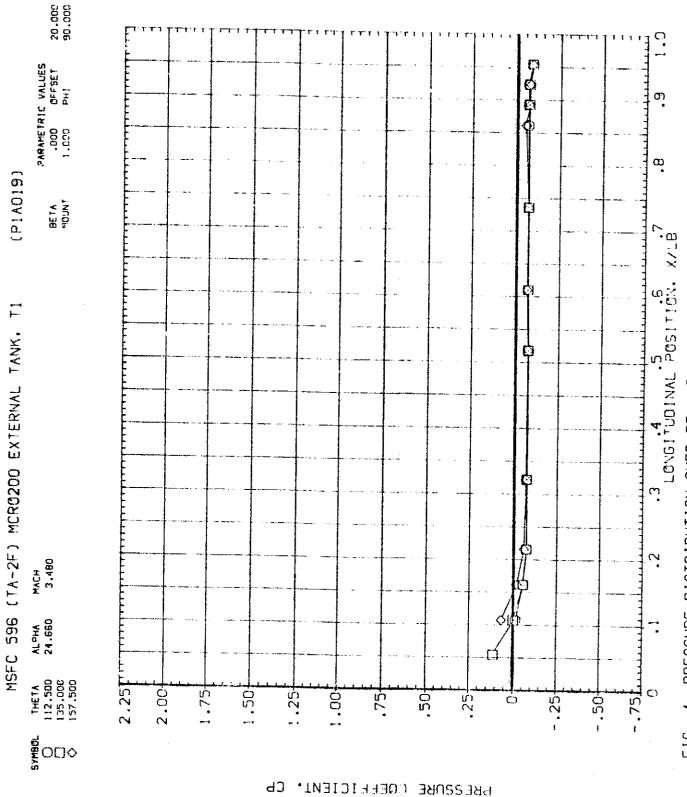




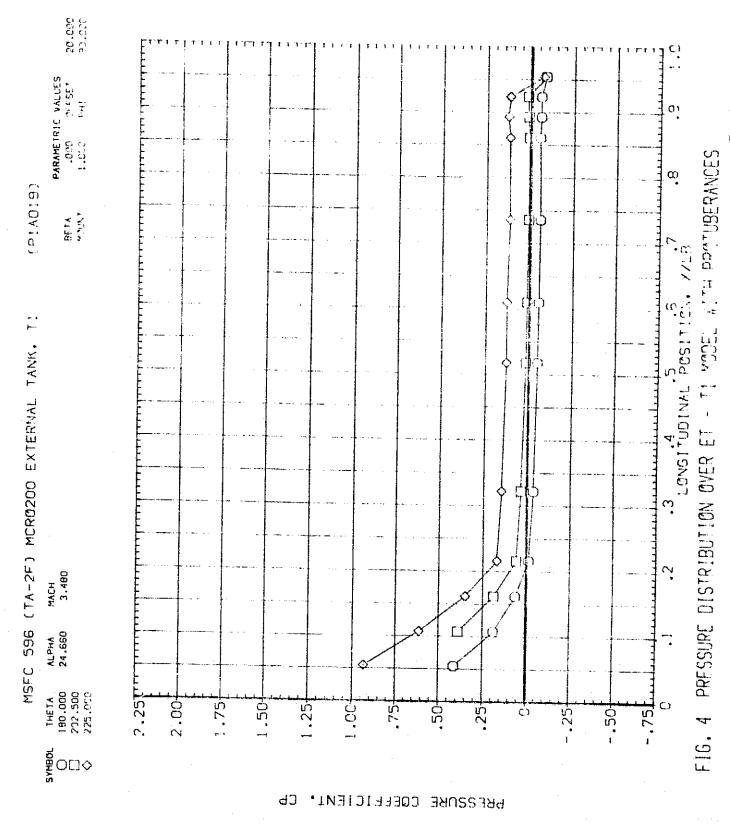
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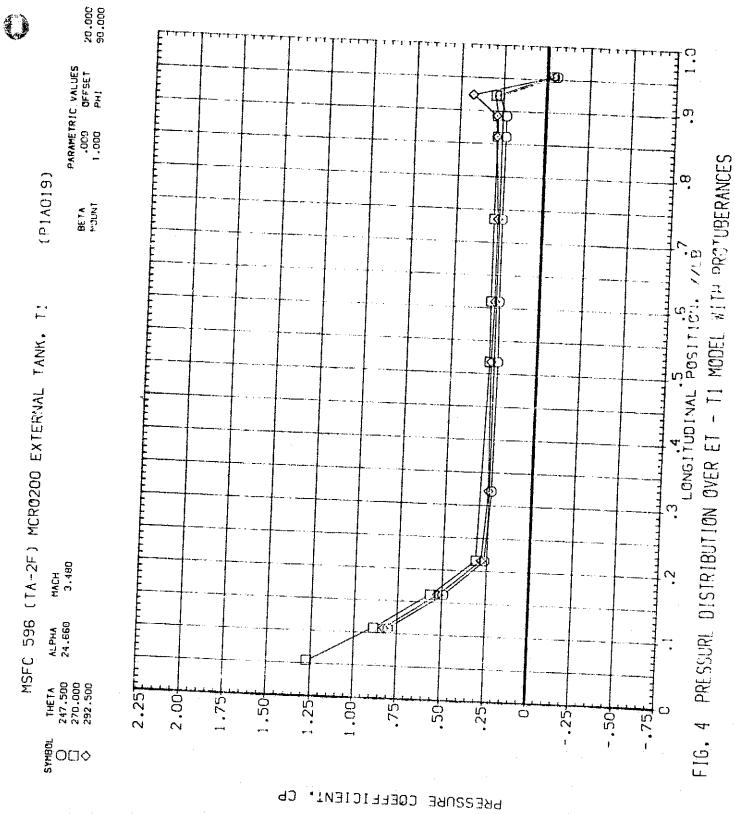


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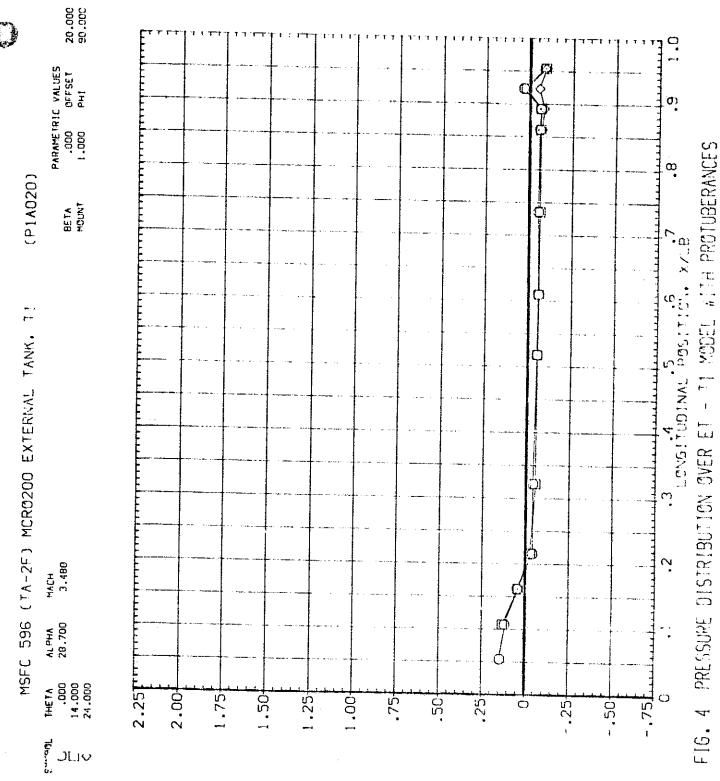


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PAGE 412

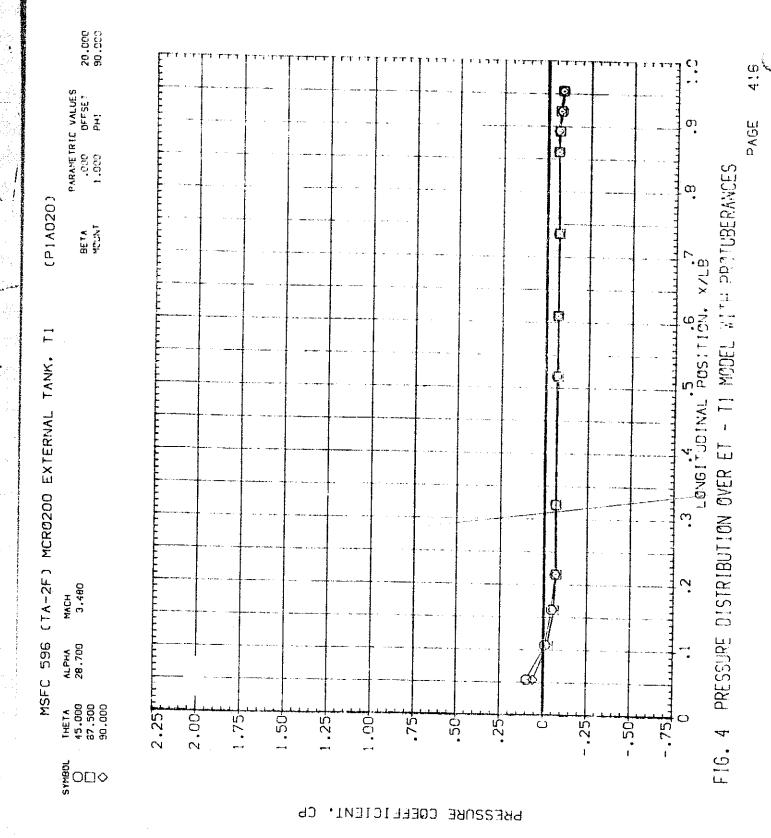


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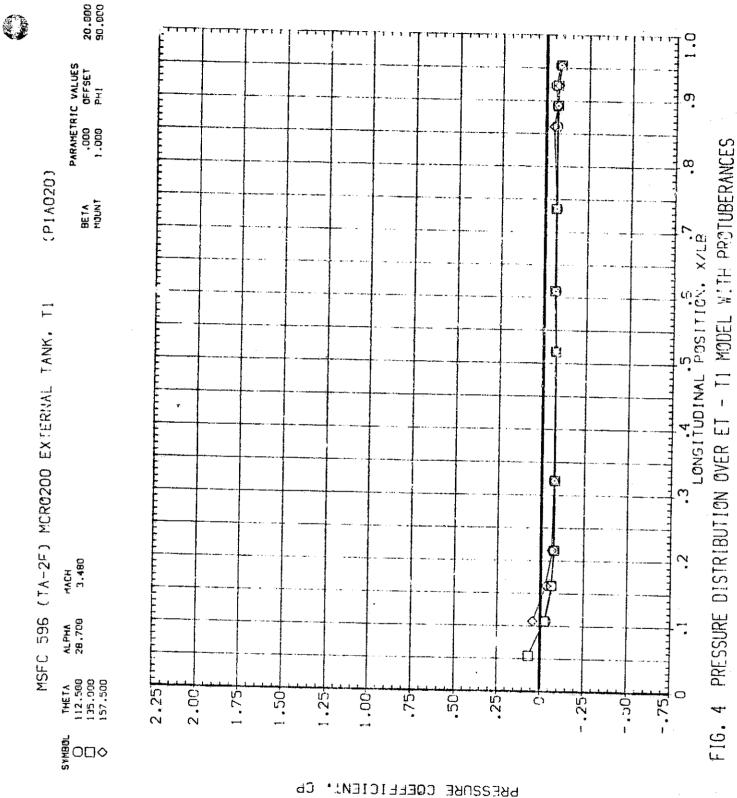


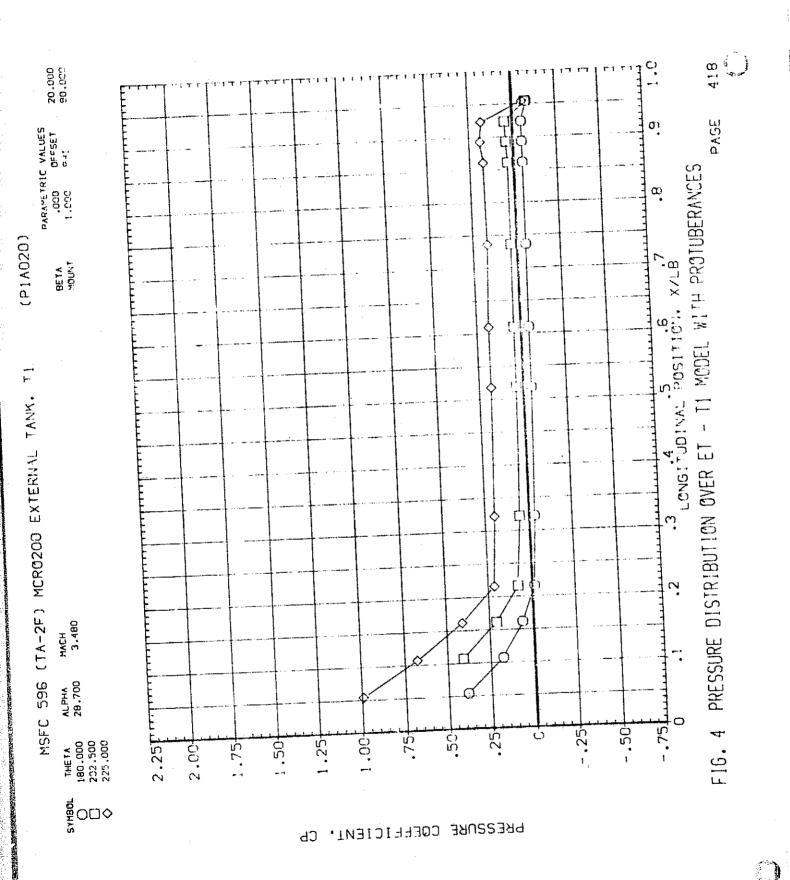
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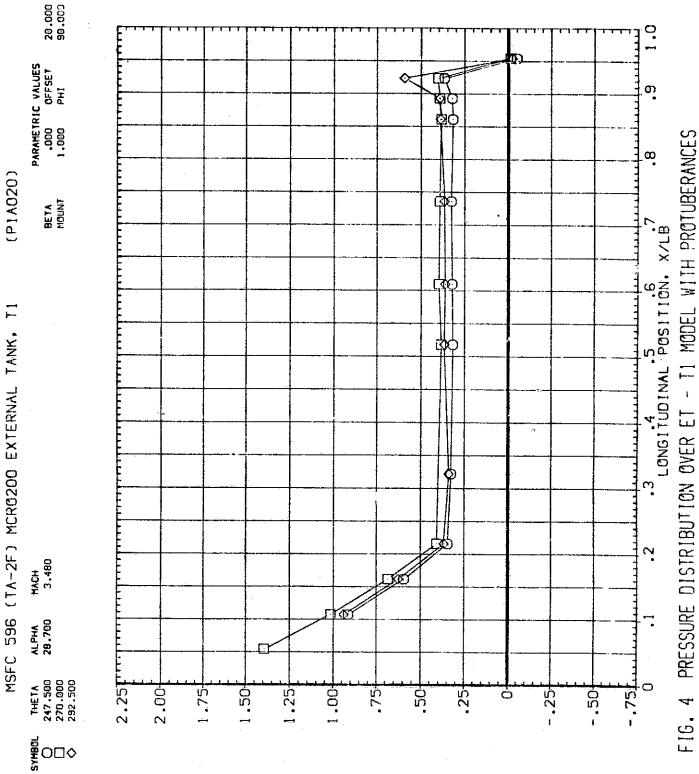












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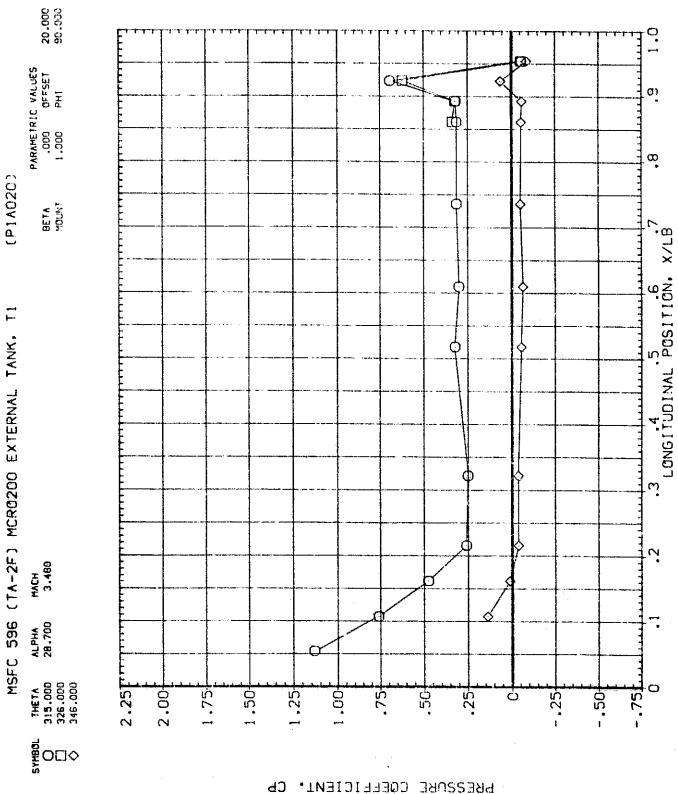


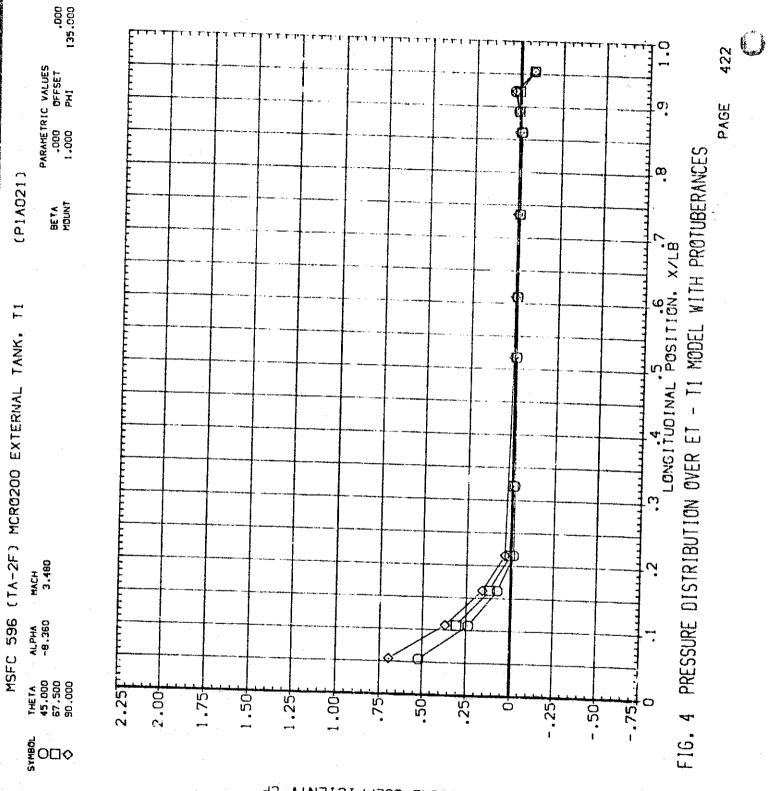
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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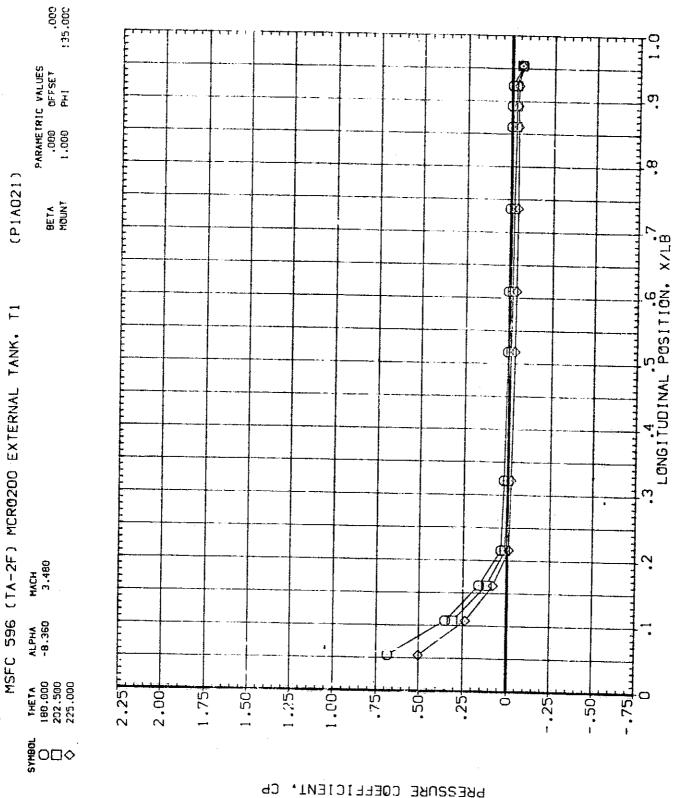
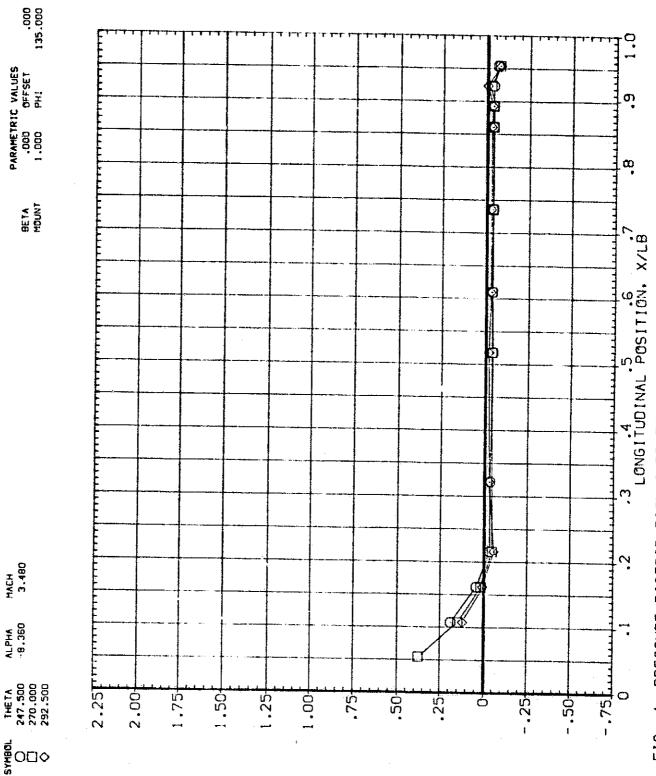


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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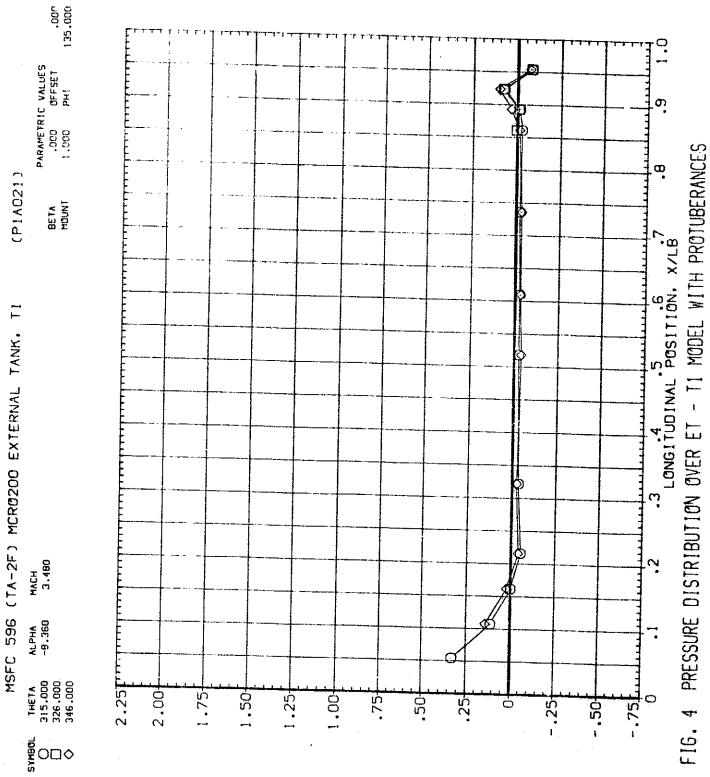
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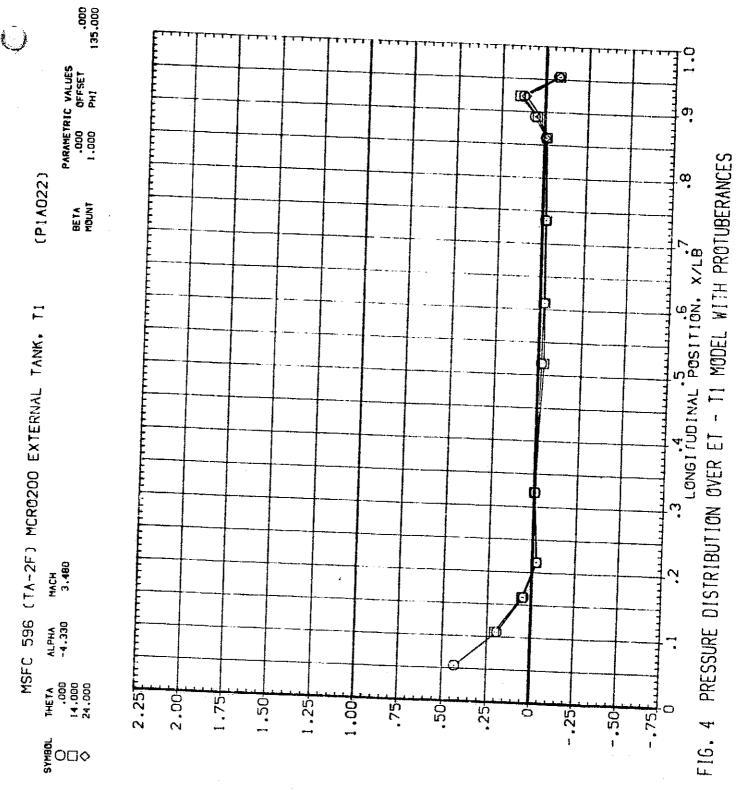
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



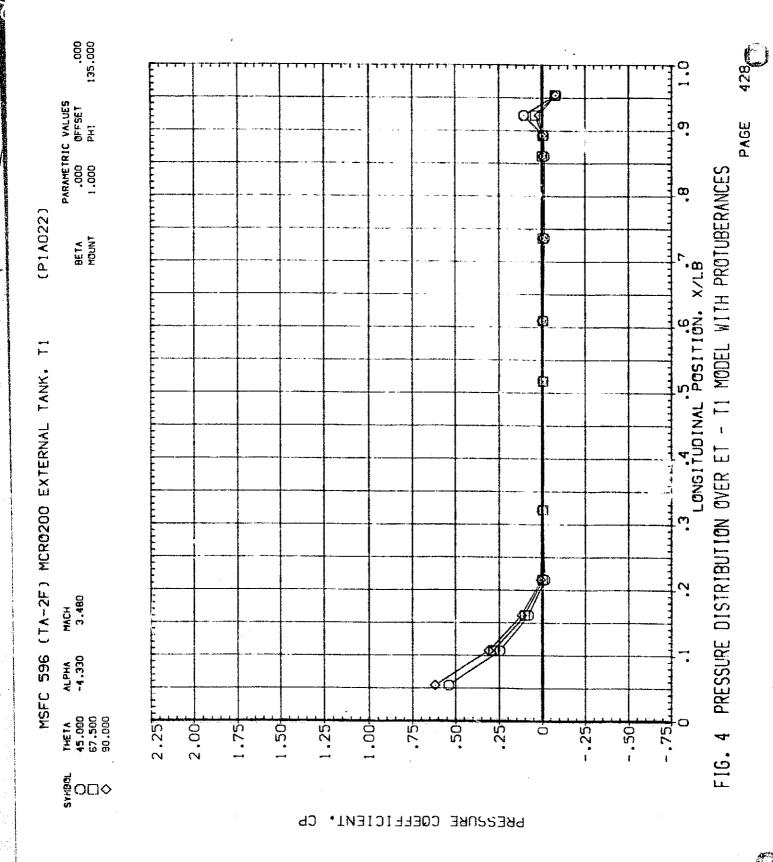
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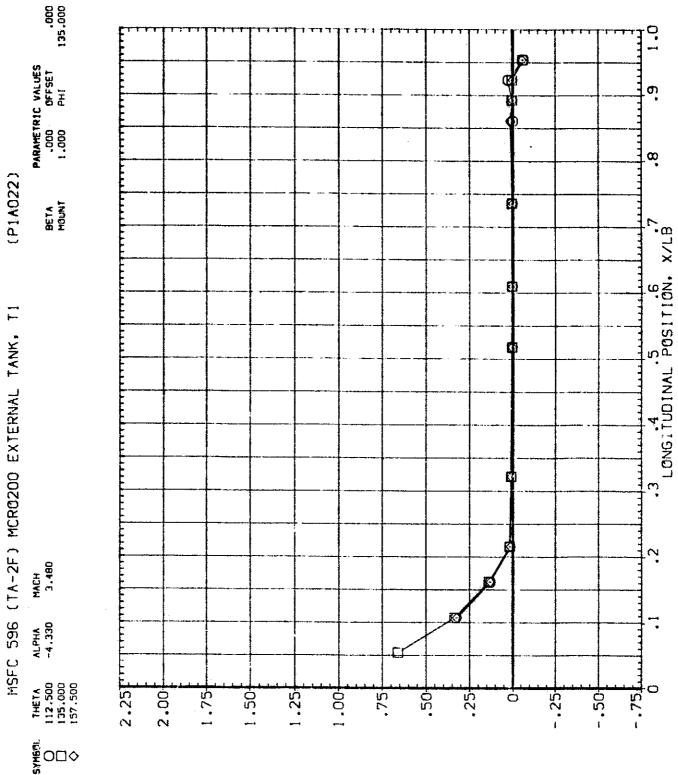
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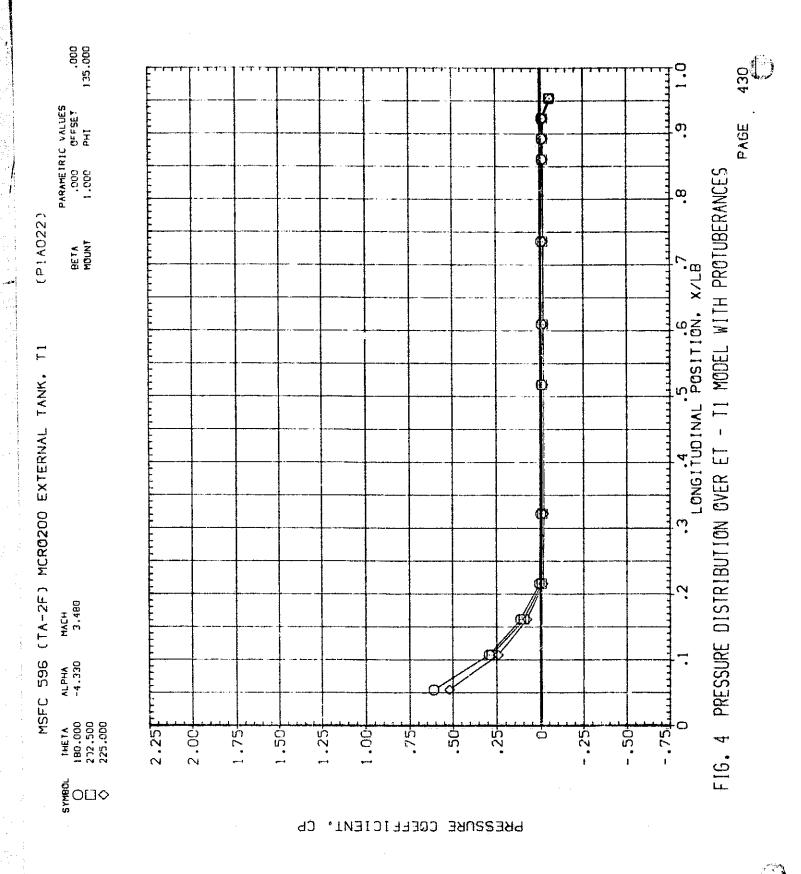


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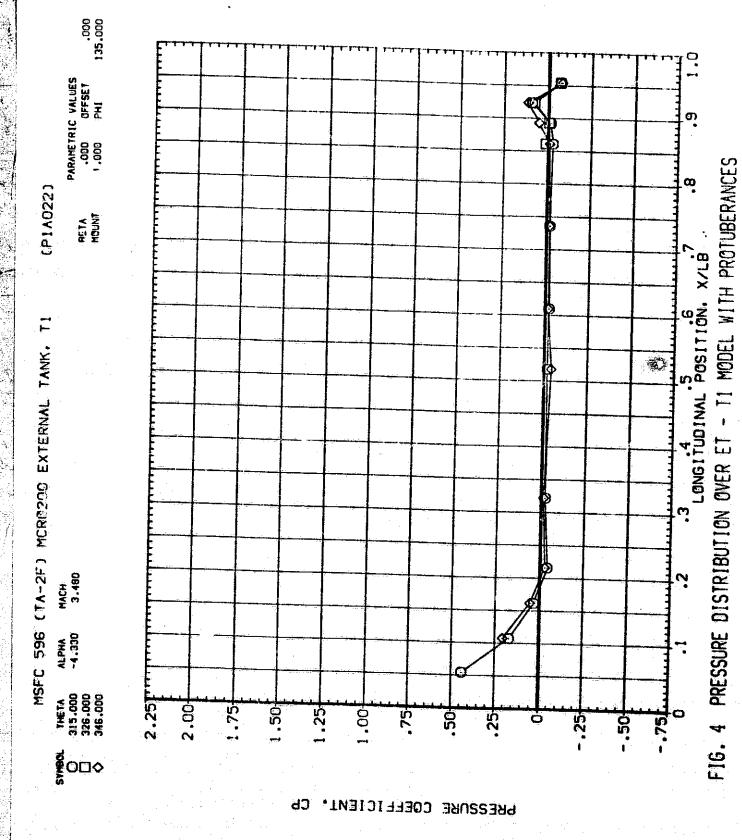
- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



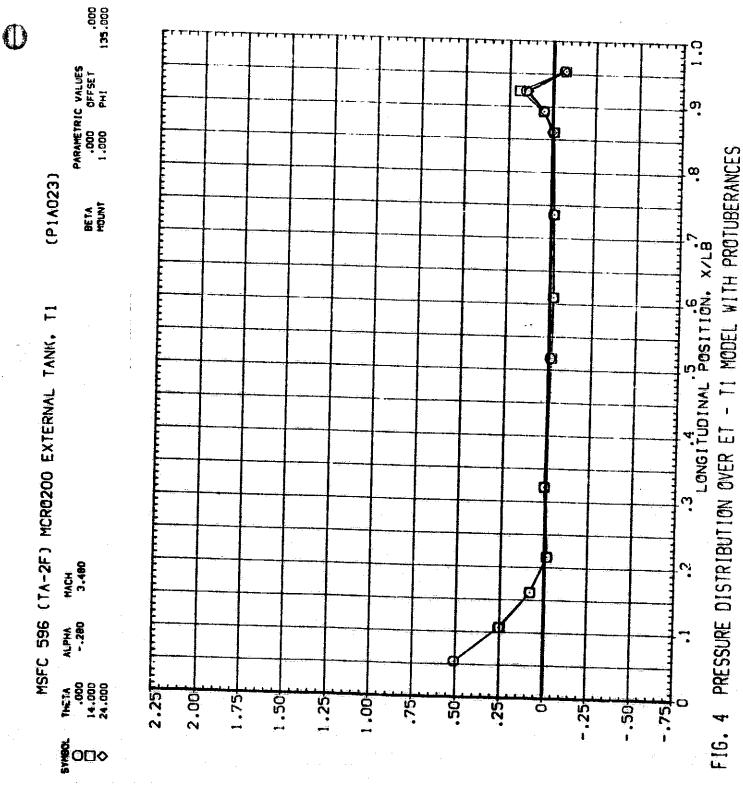
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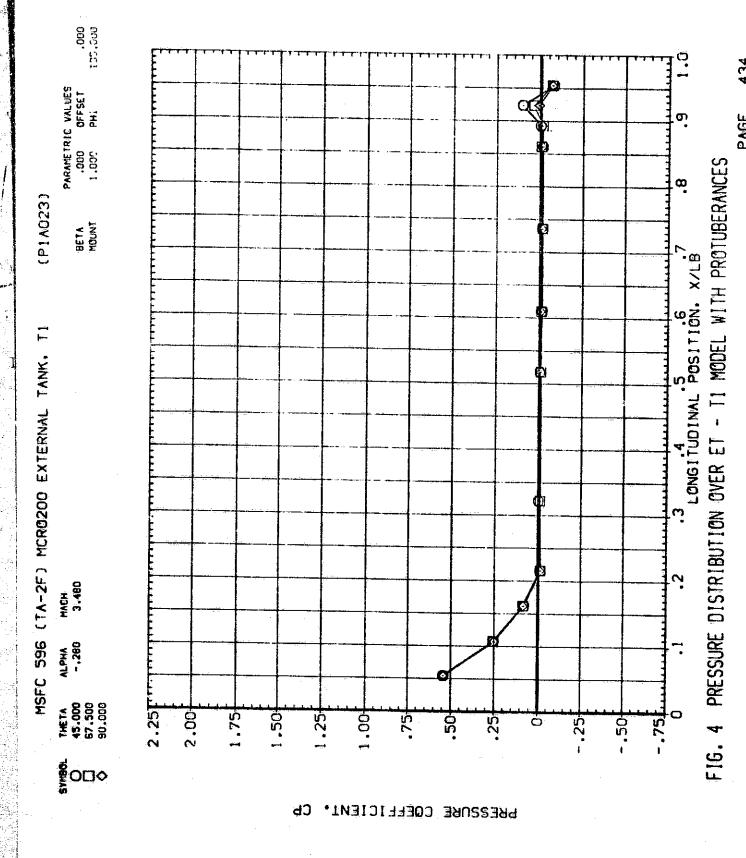
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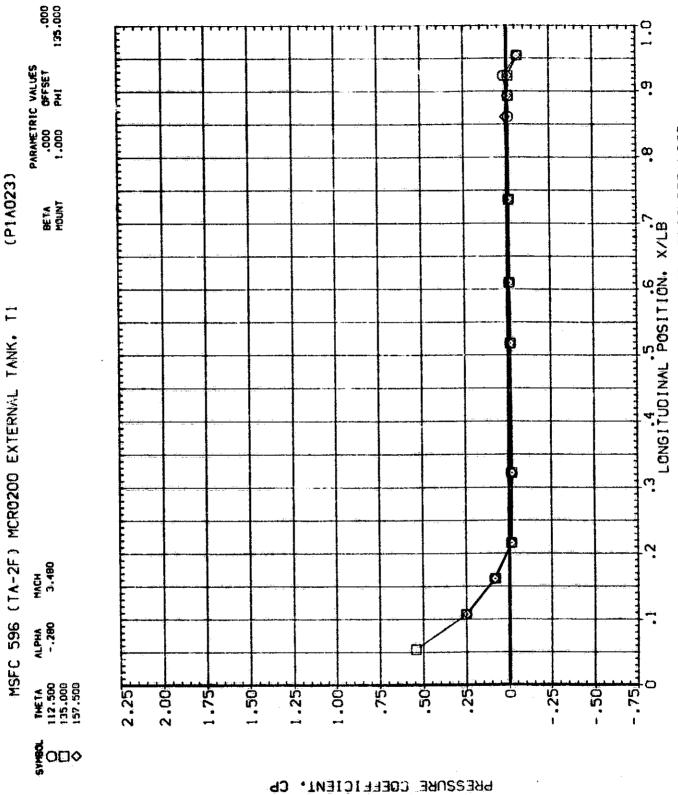


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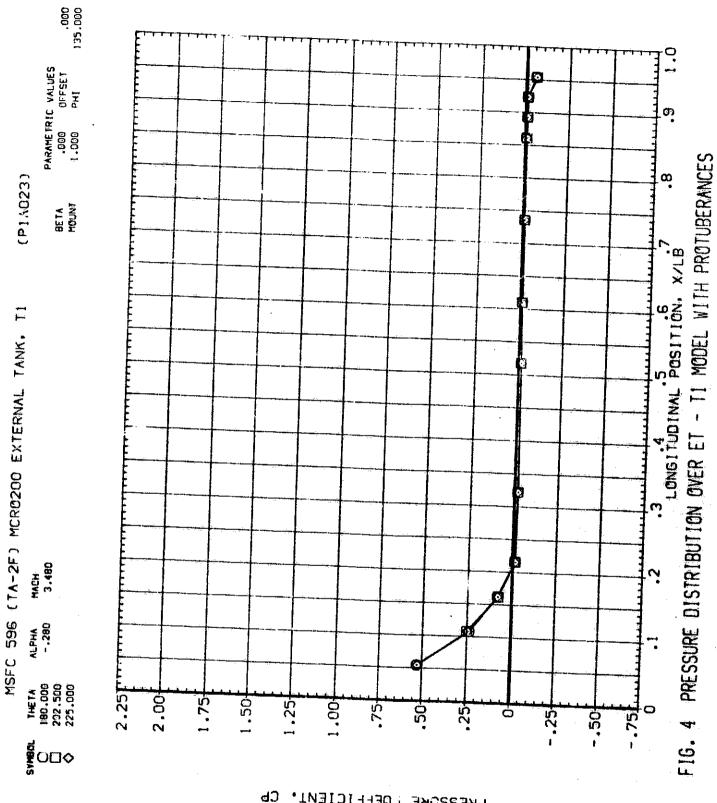


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PRESSURE DISTRIBUTION OVER ET - T1 MODEL WITH PROTUBERANCES FIG. 4



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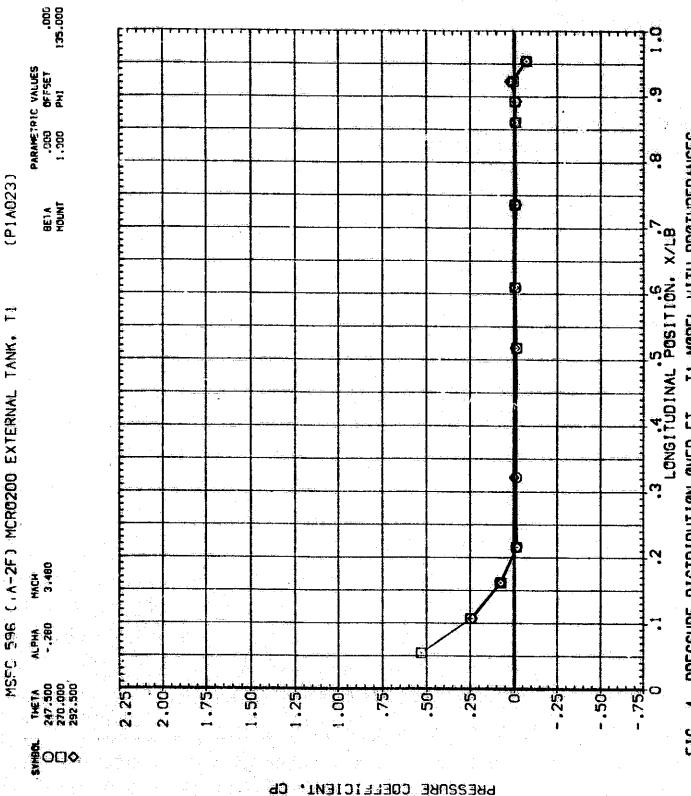


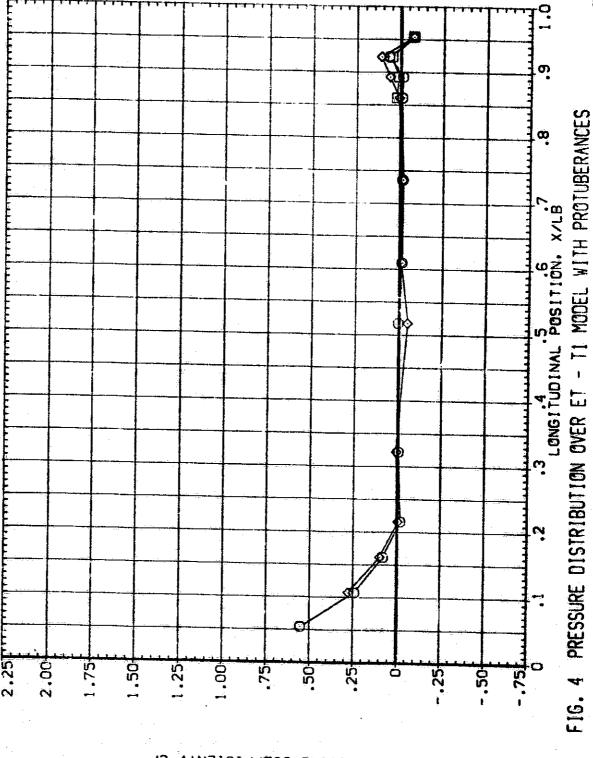
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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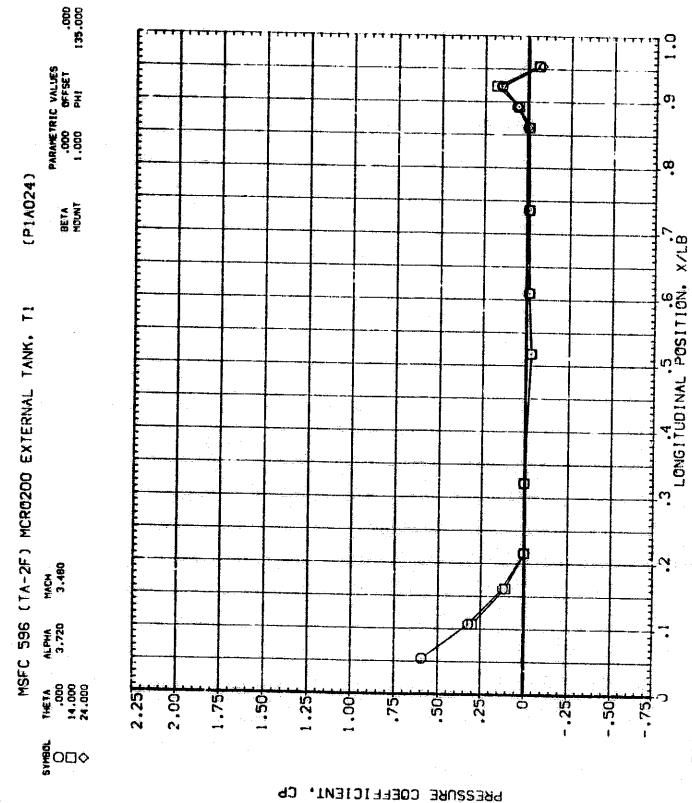
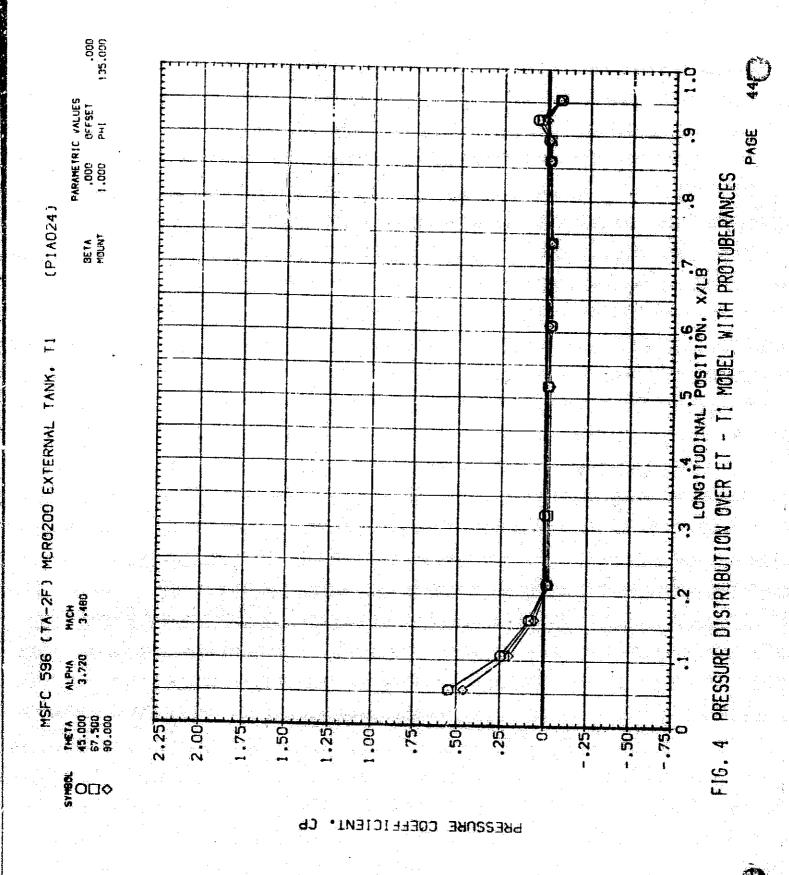
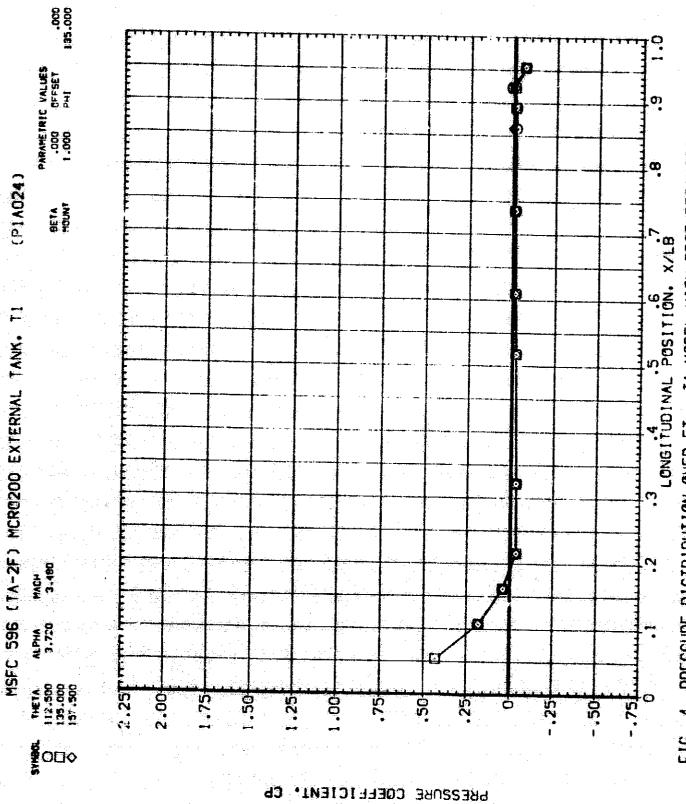
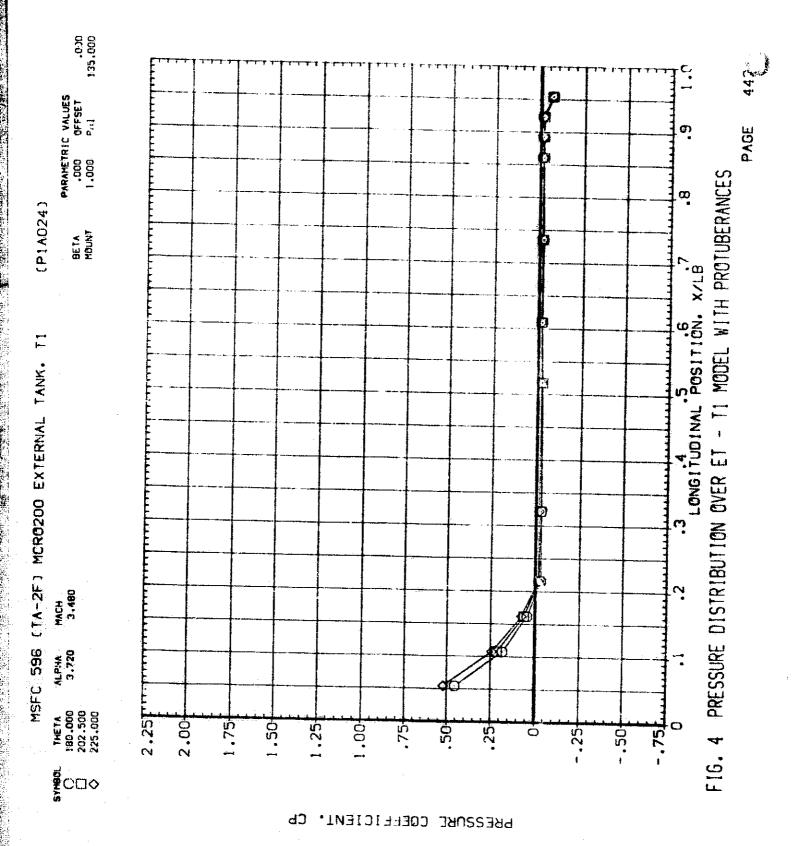


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

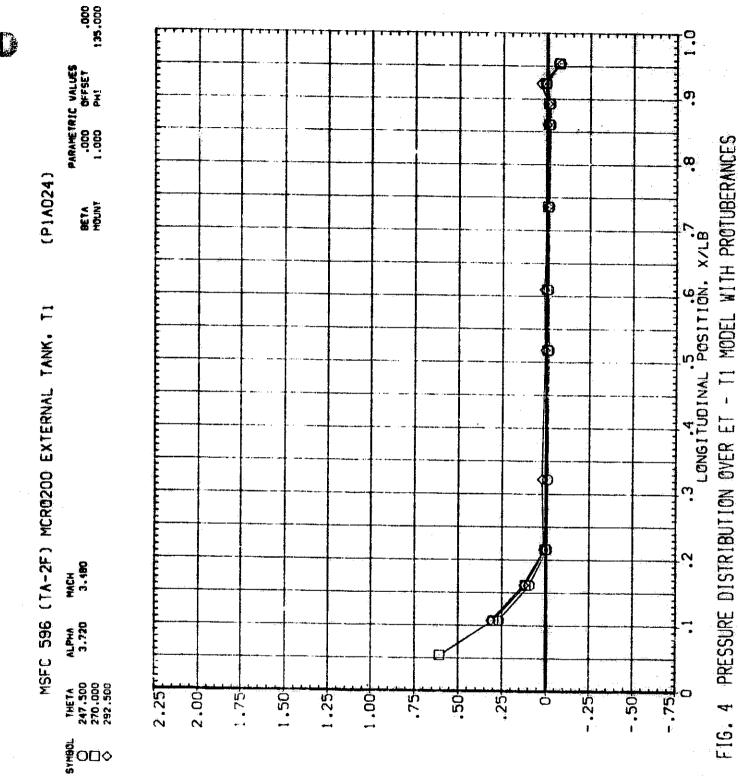




- TI MODEL WITH PROTUBERANCES PRESSURE DISTRIBUTION OVER ET F16. 4

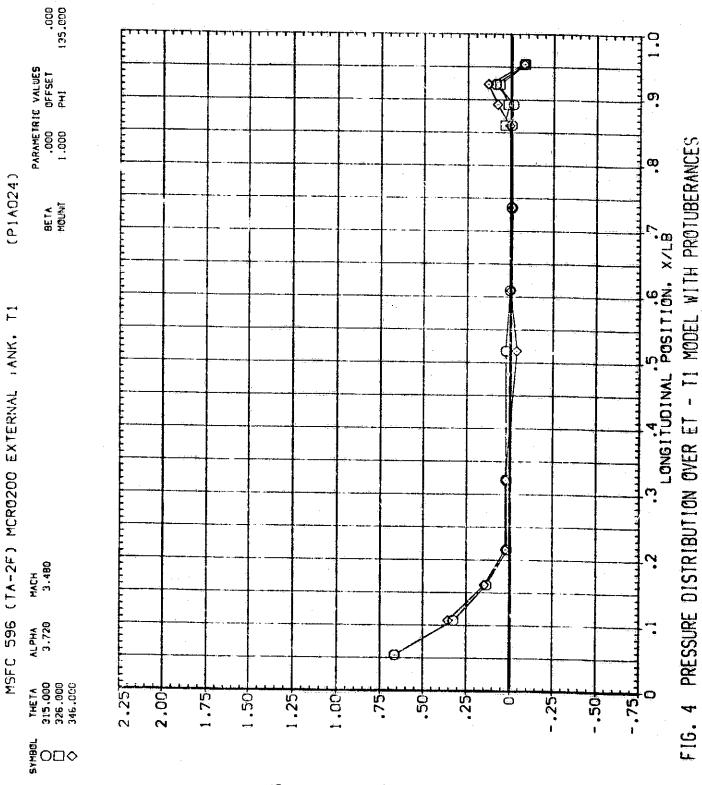






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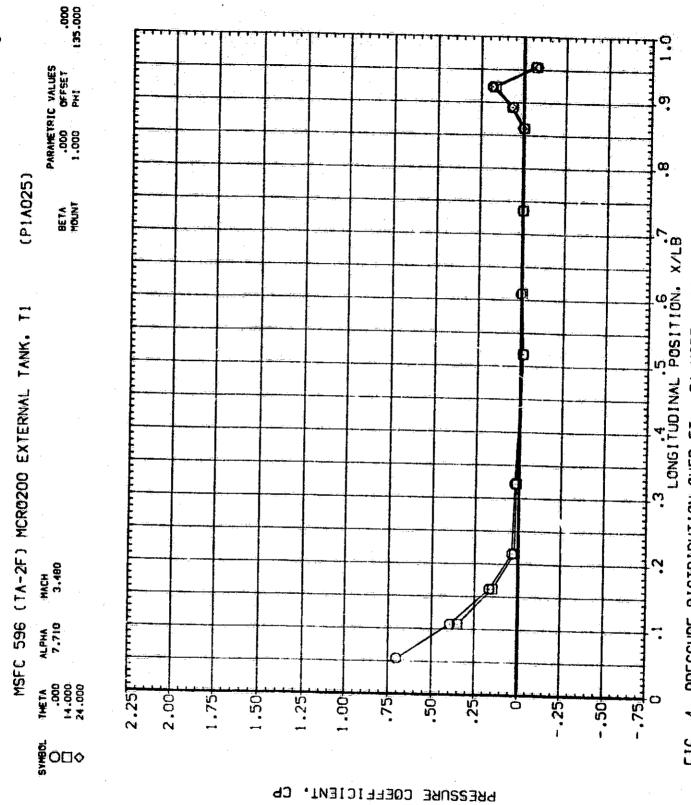


FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

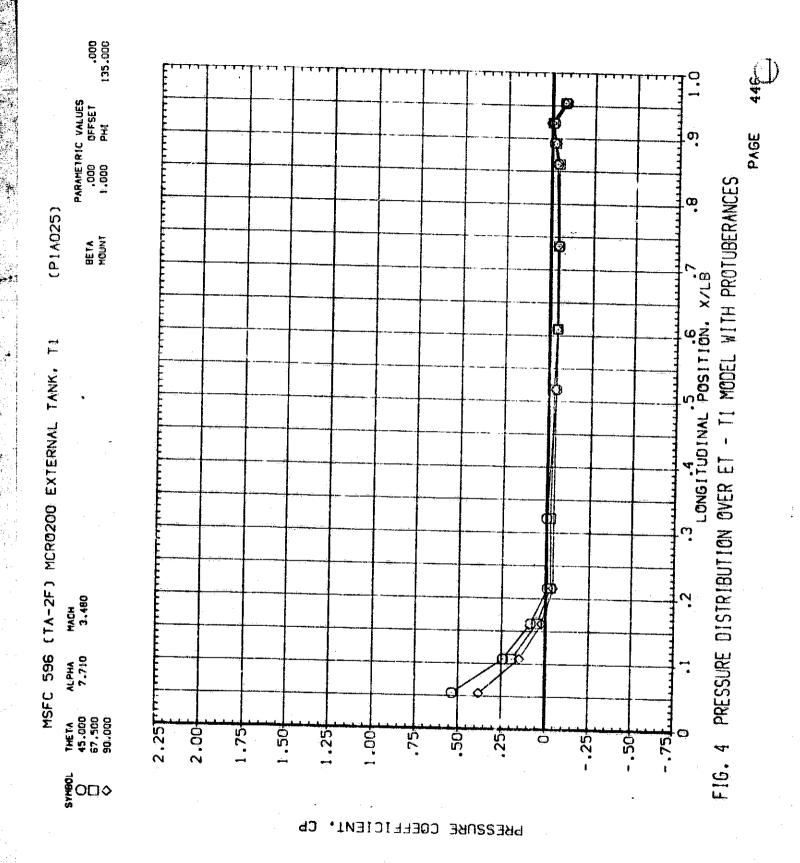
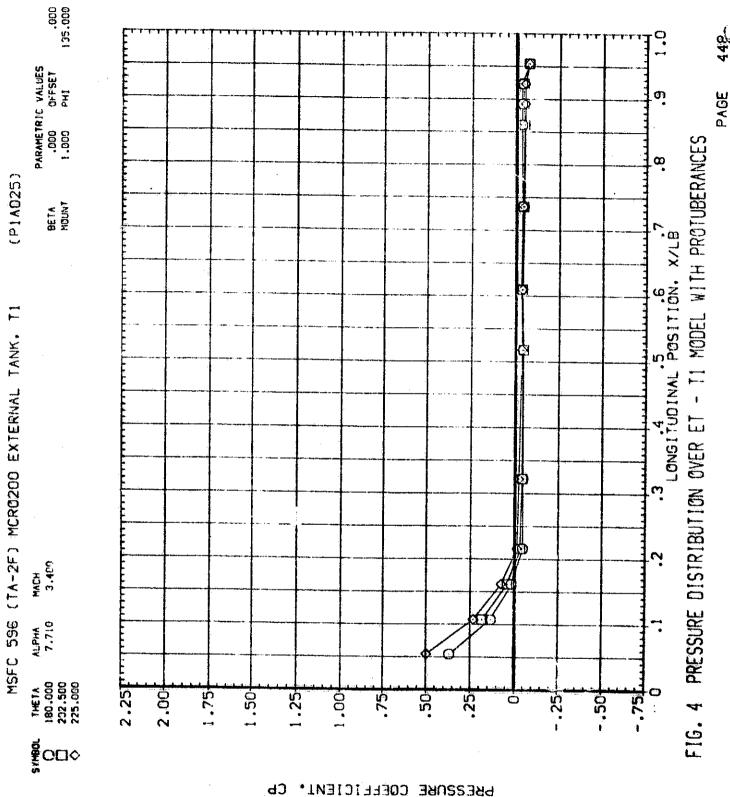
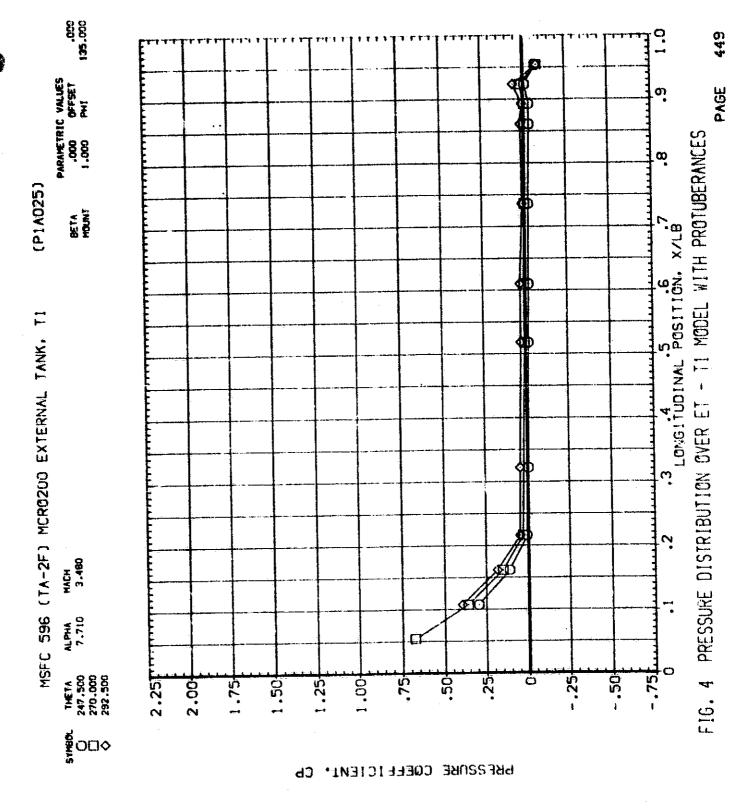


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES





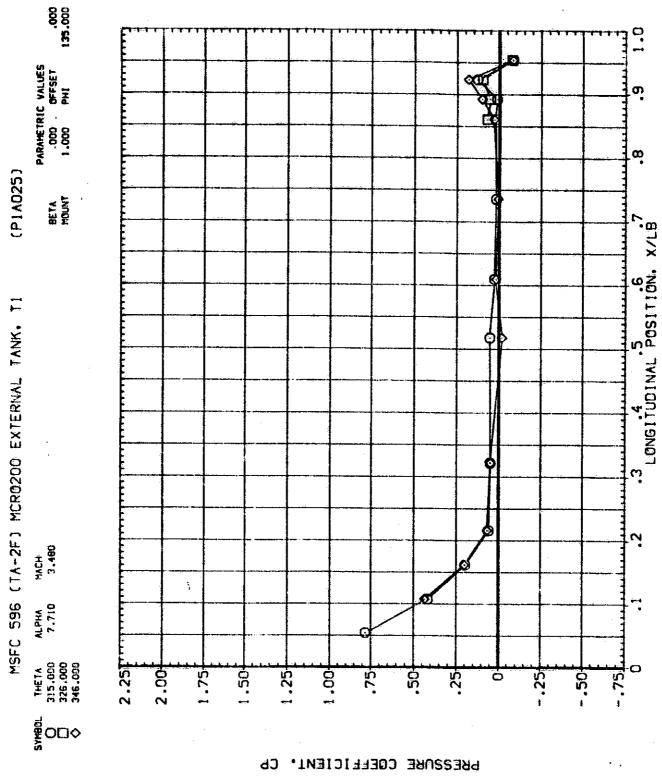
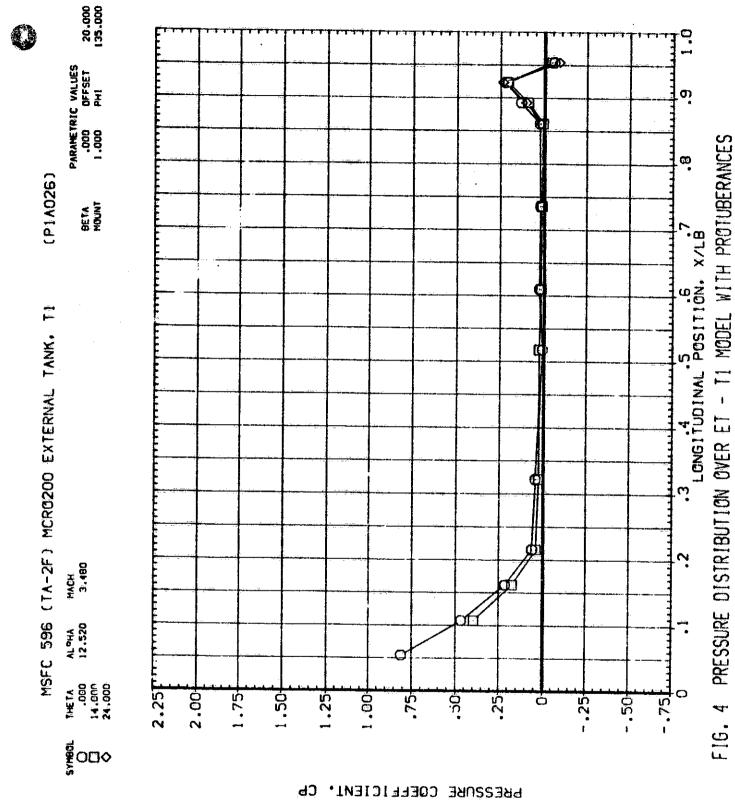


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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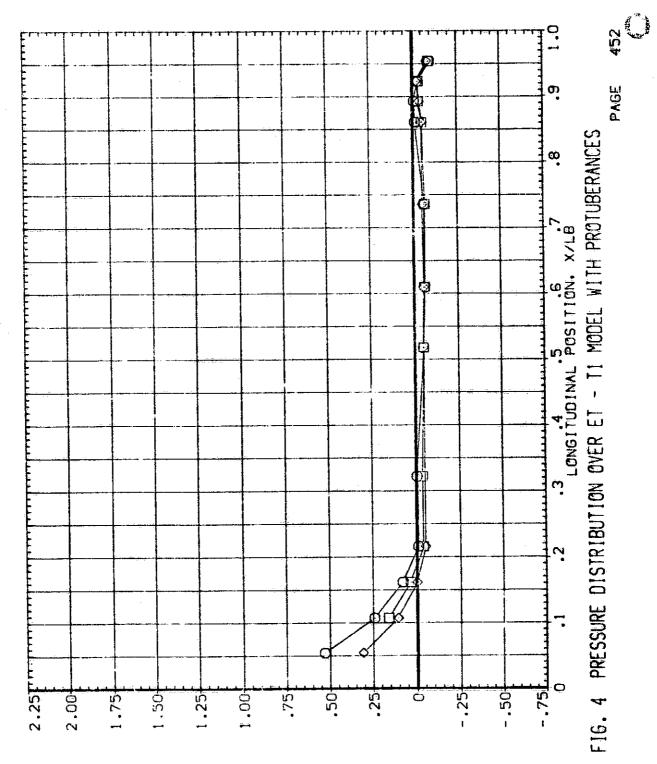
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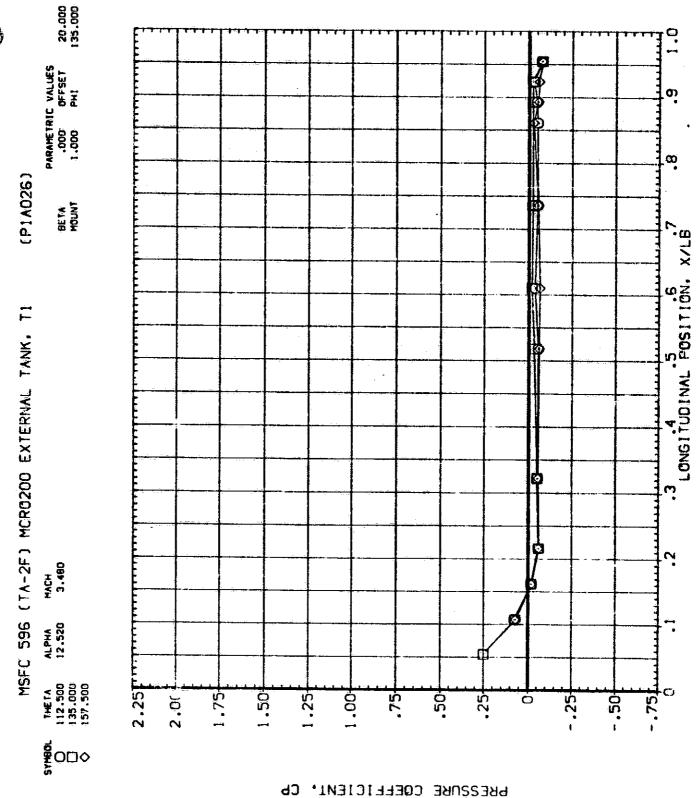
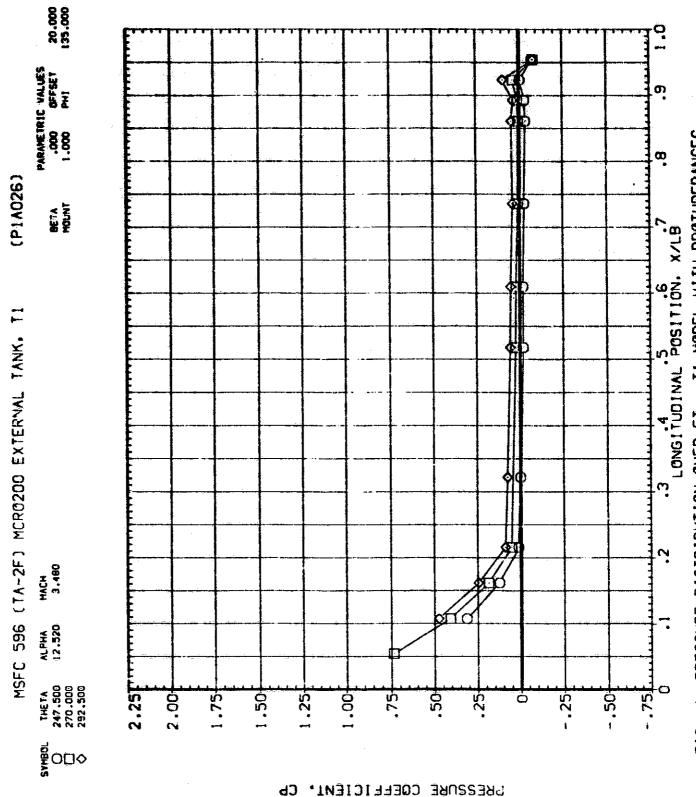


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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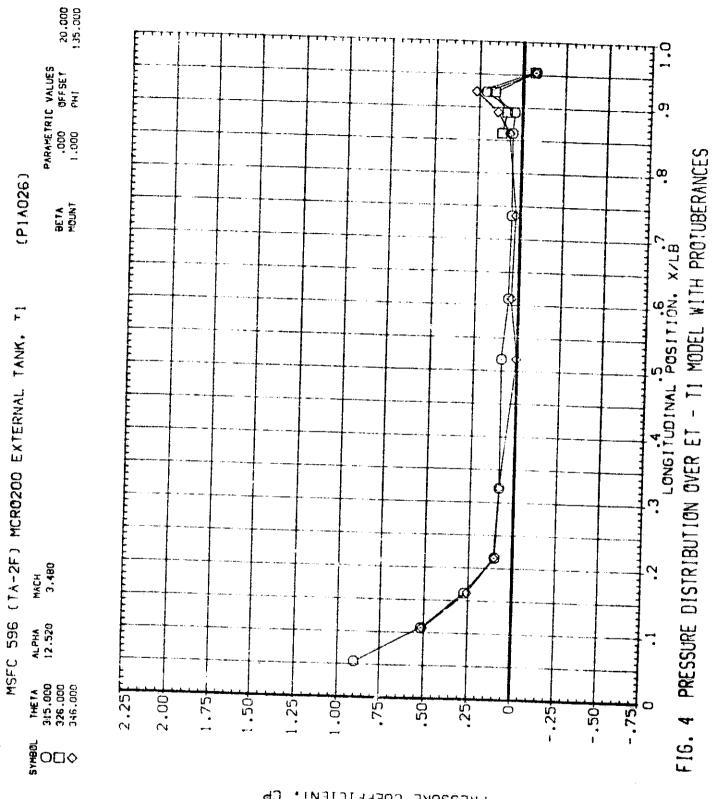


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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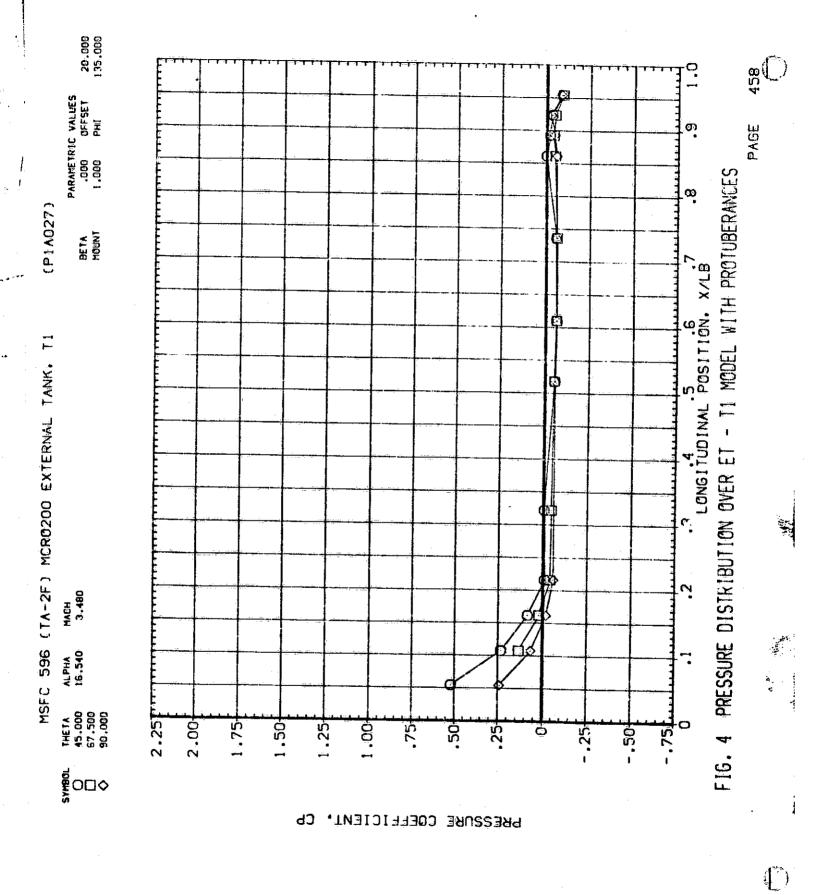
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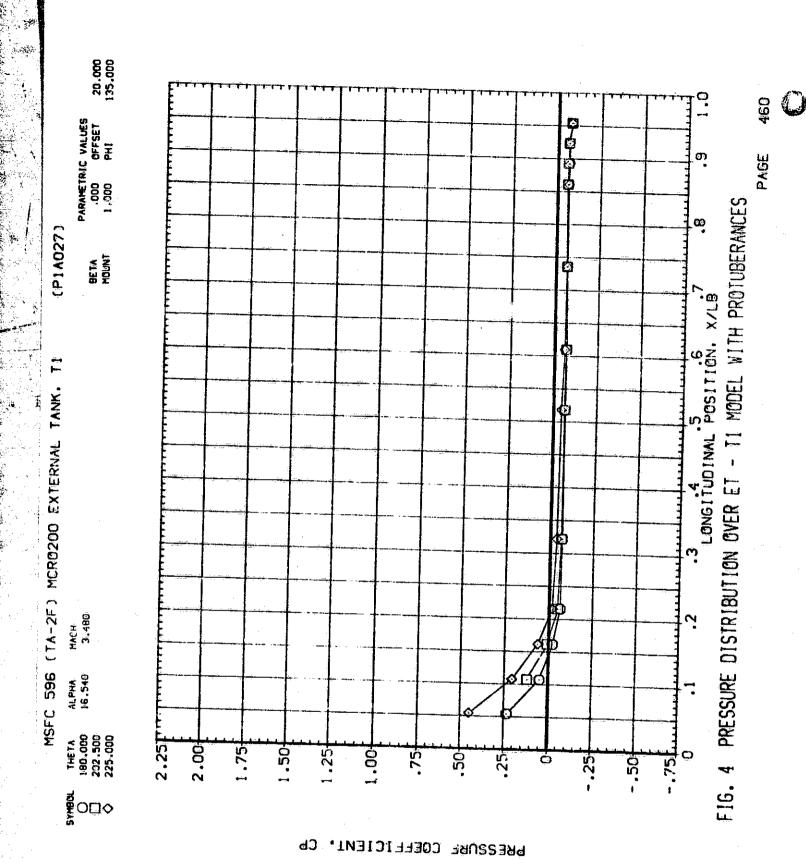
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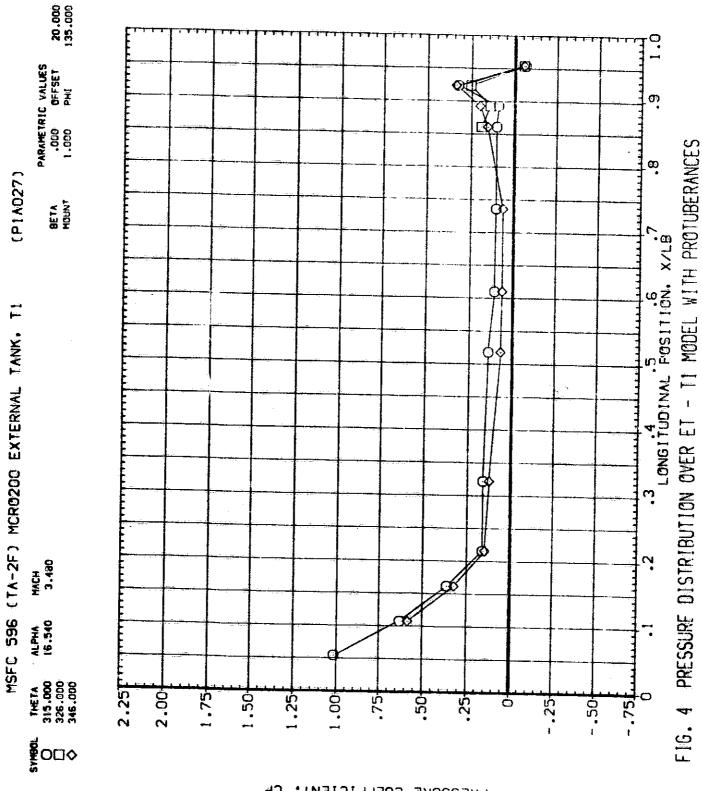
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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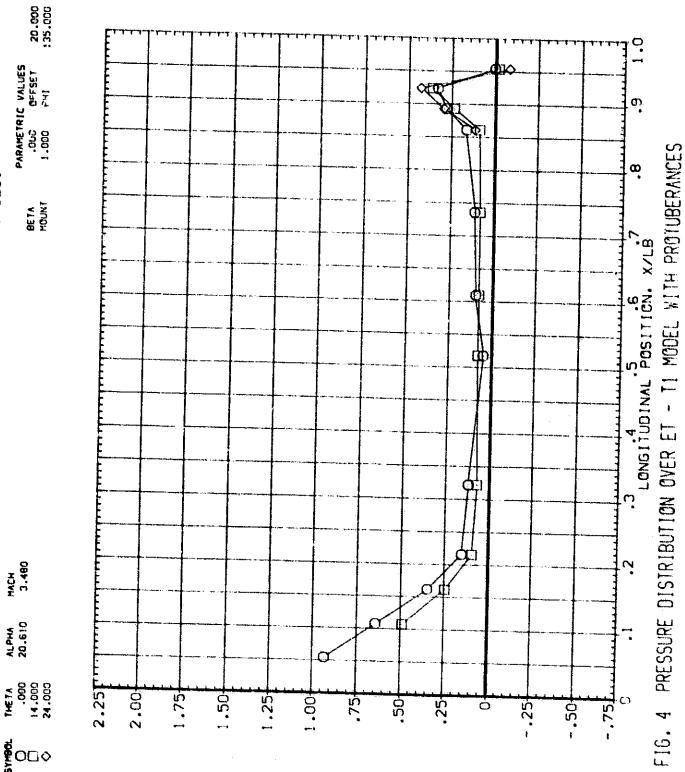


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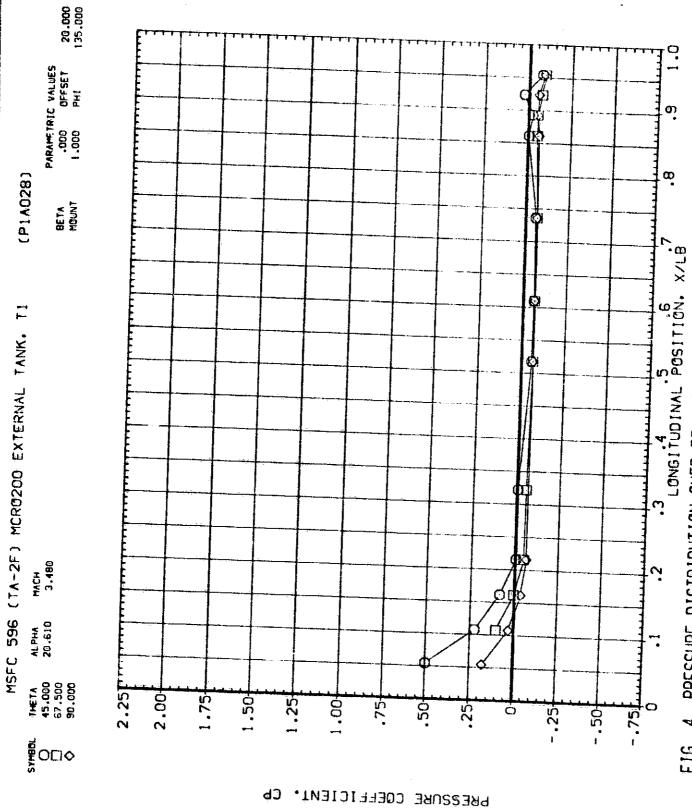
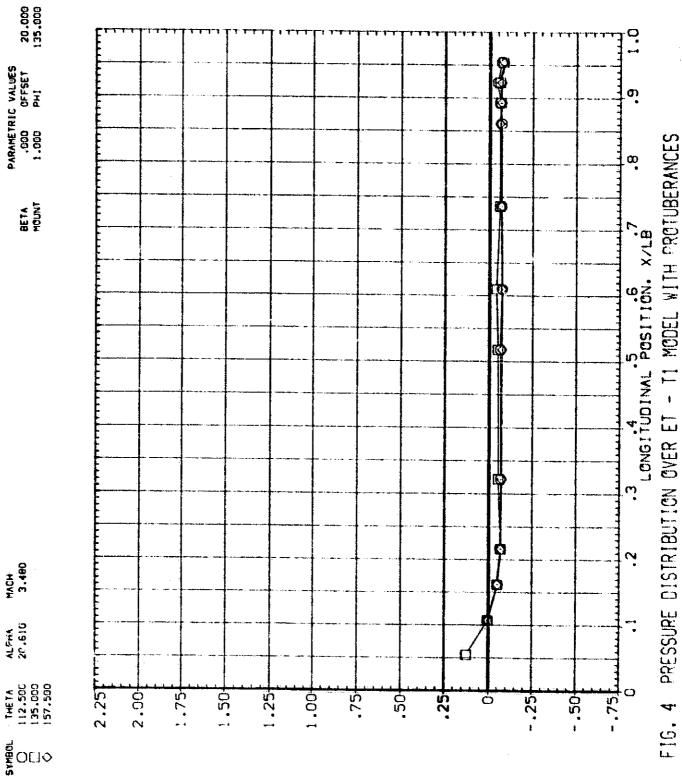


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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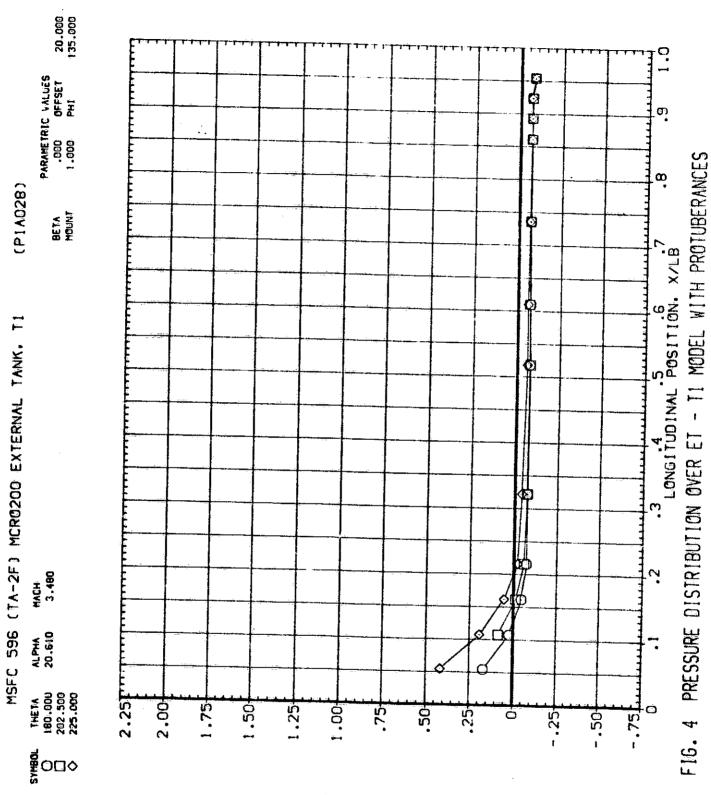


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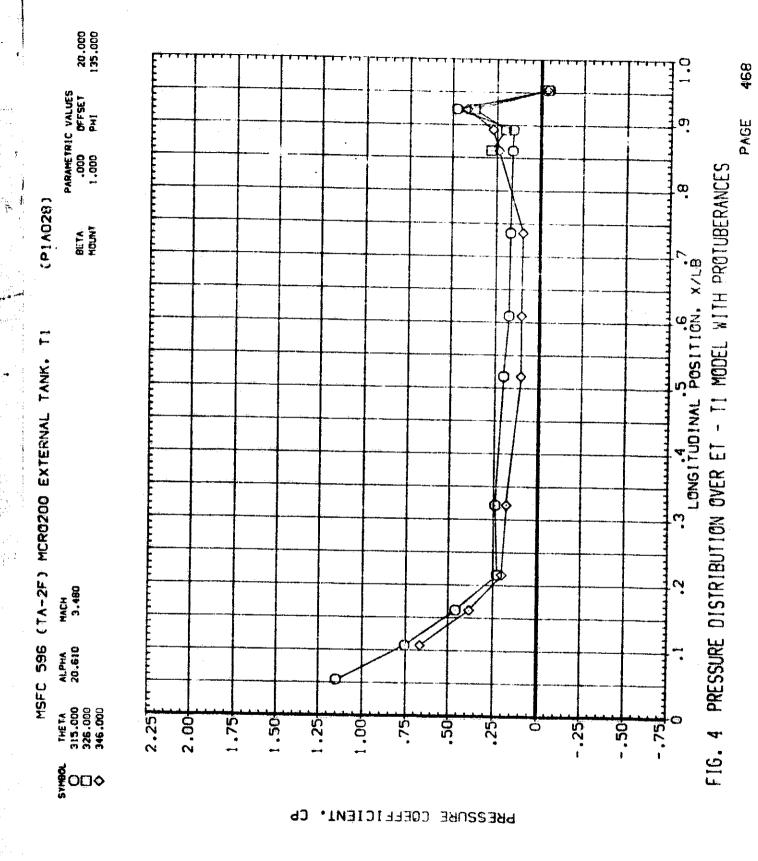


FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES -.75 -.25 -.50

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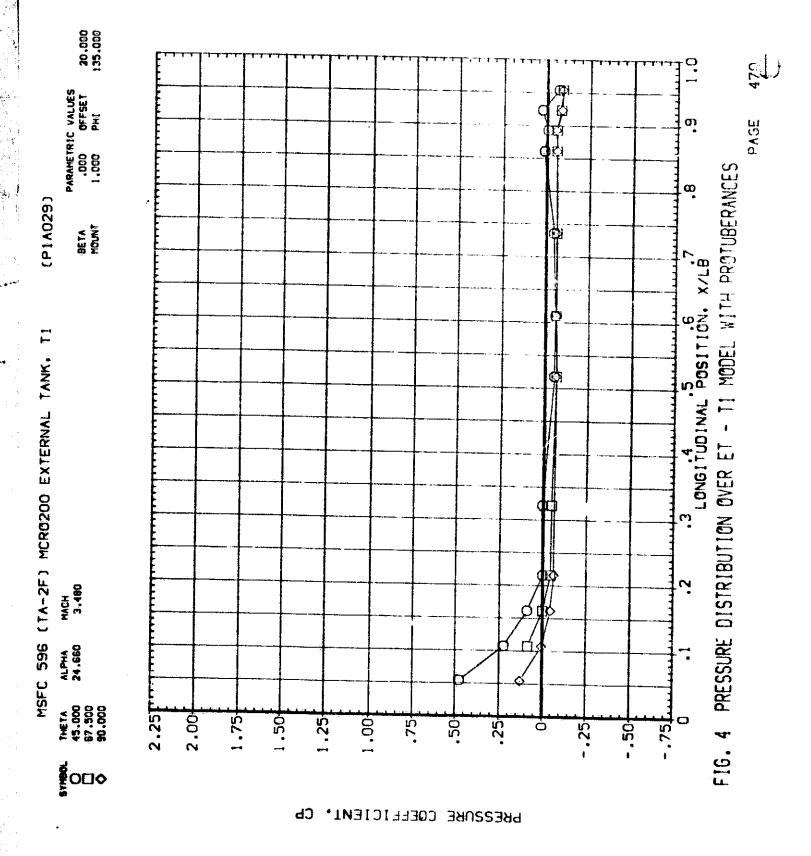
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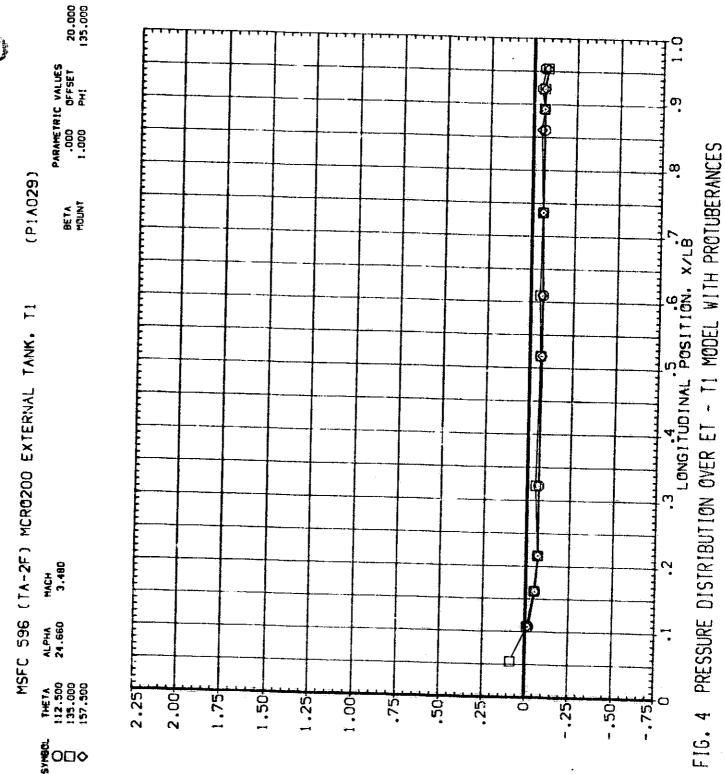
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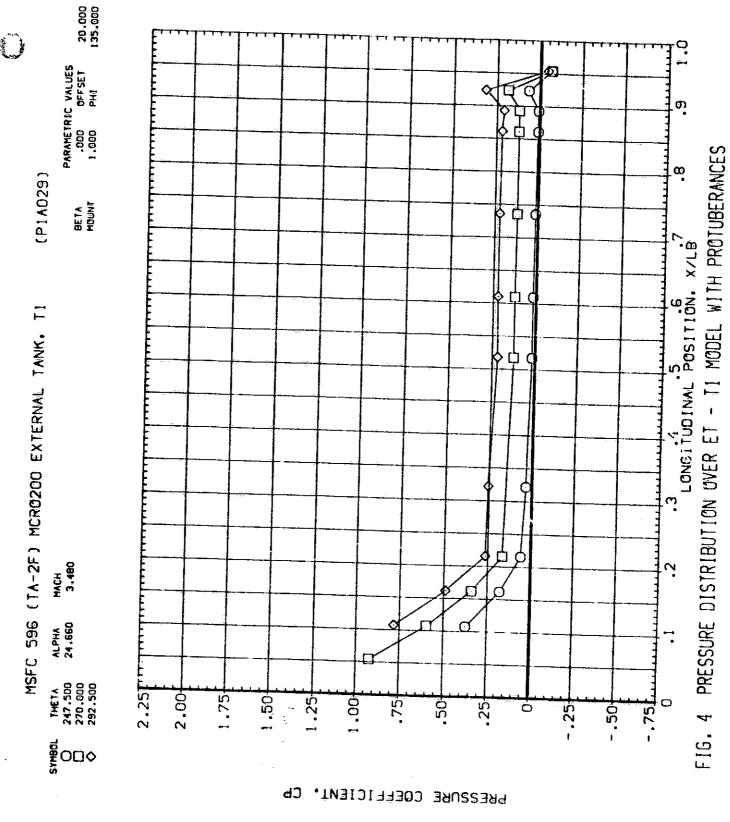
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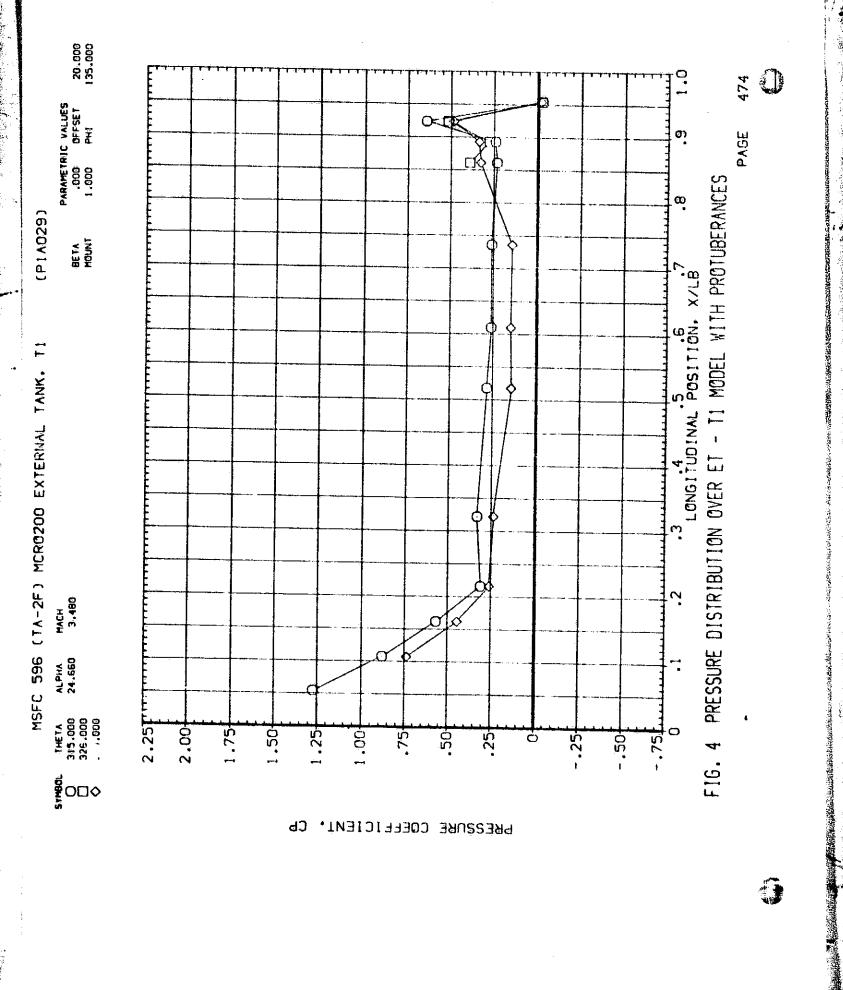
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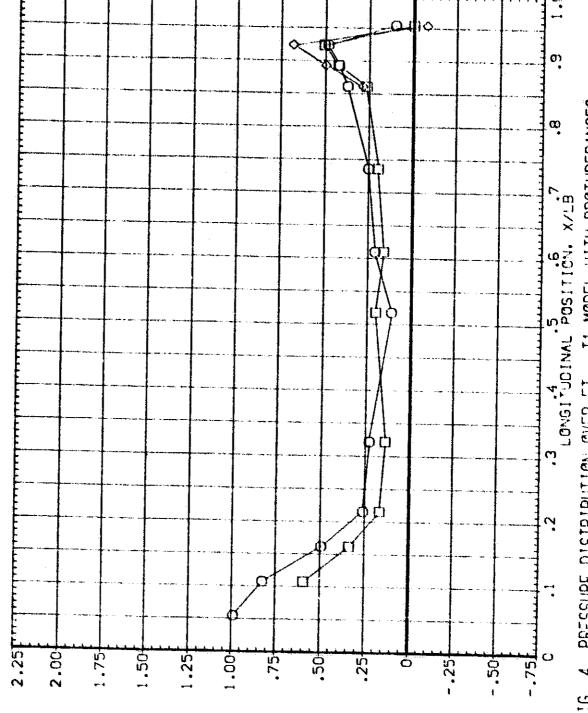
FIG. 4 PRESSURE DISTRIBUTION OVER





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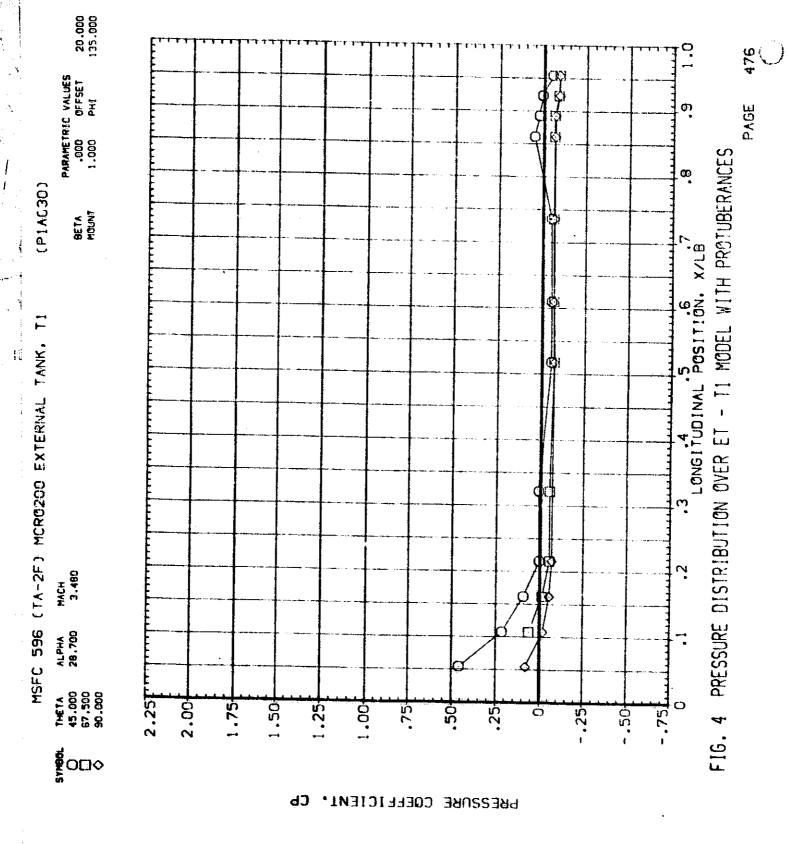
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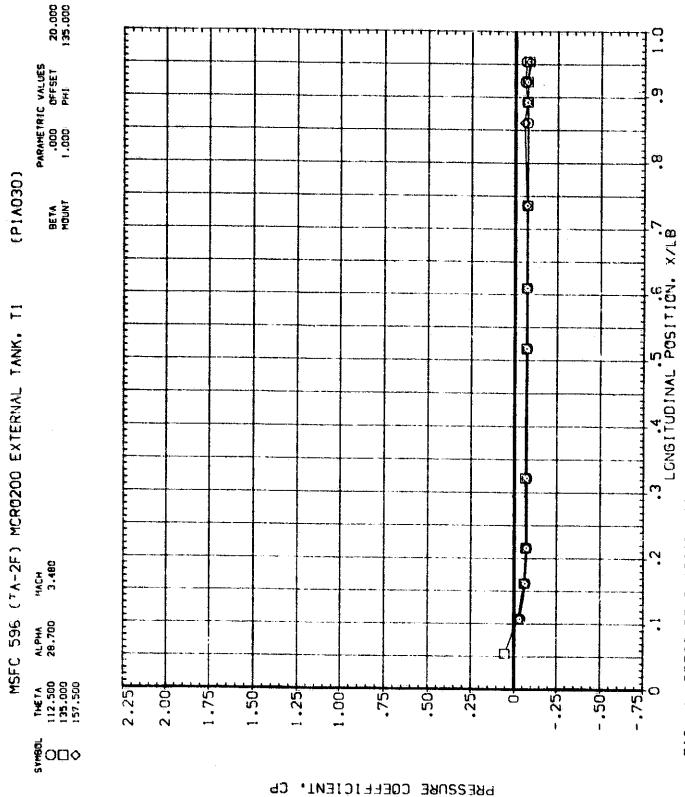


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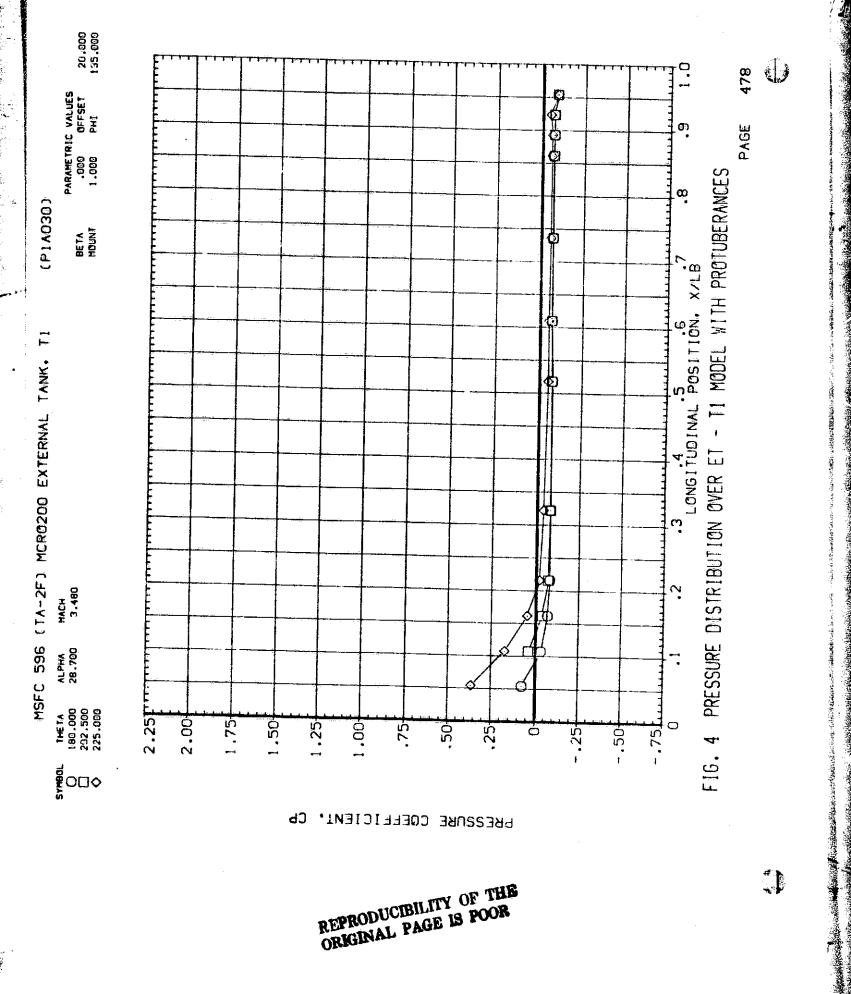
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

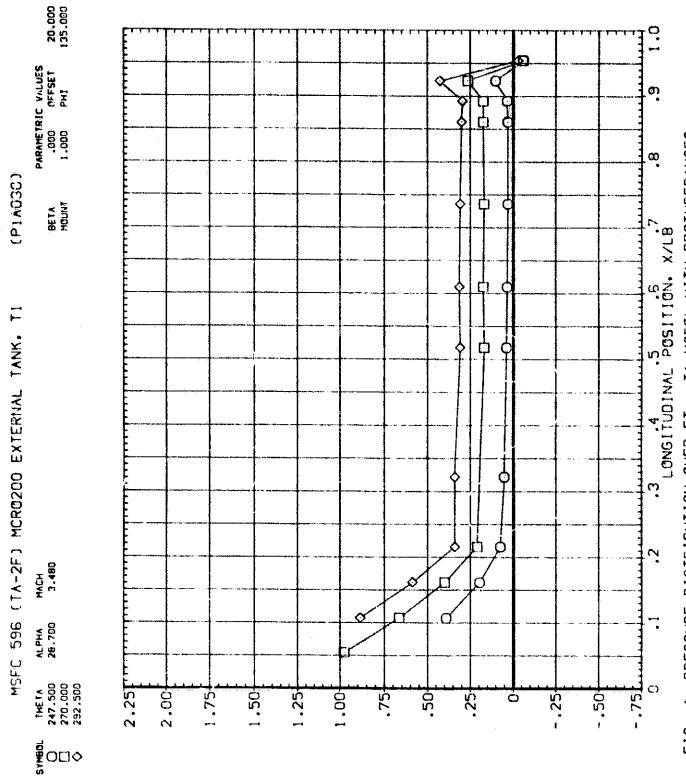




- TI MODEL WITH PROTUBERANCES PRESSURE DISTRIBUTION OVER ET F16, 4



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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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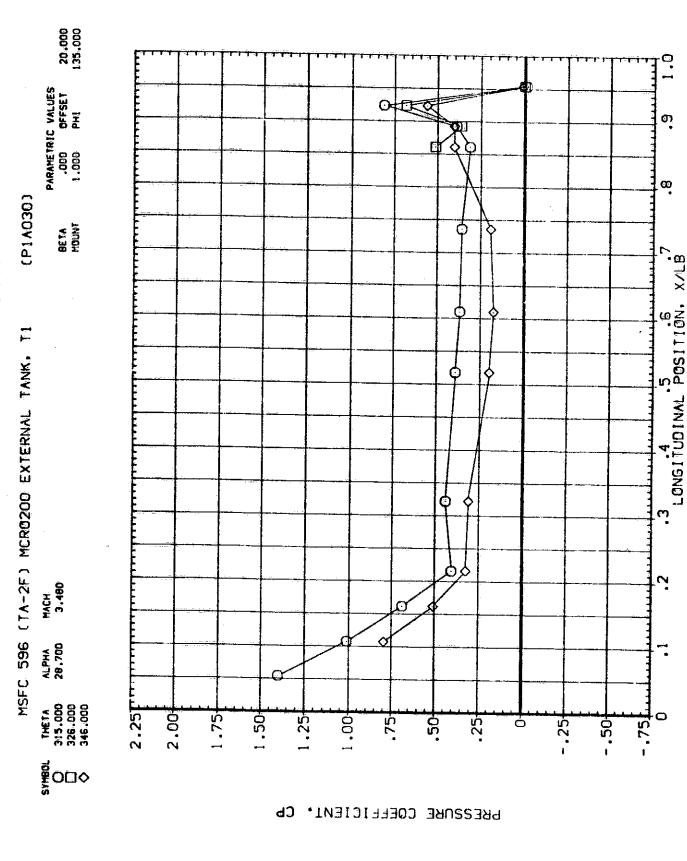
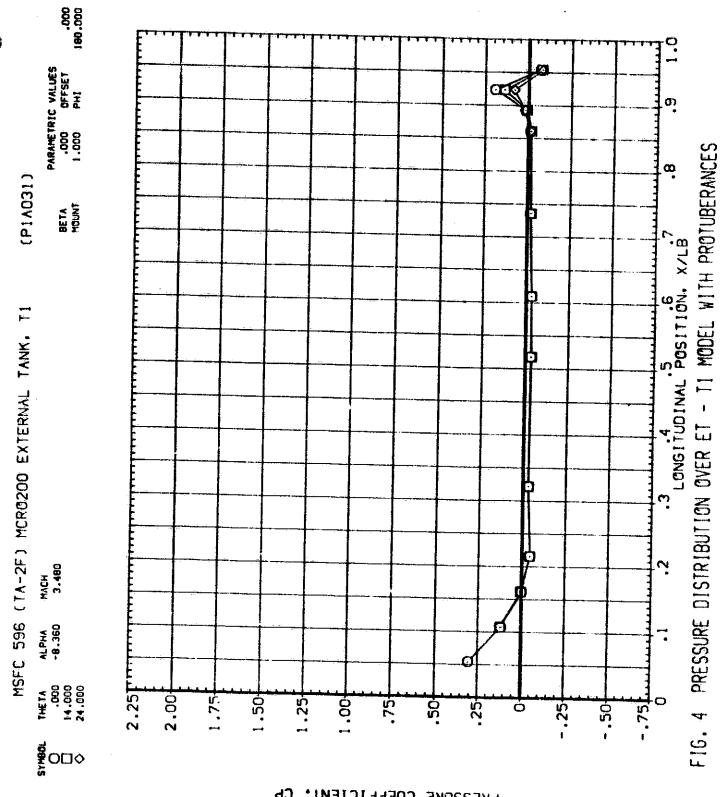


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

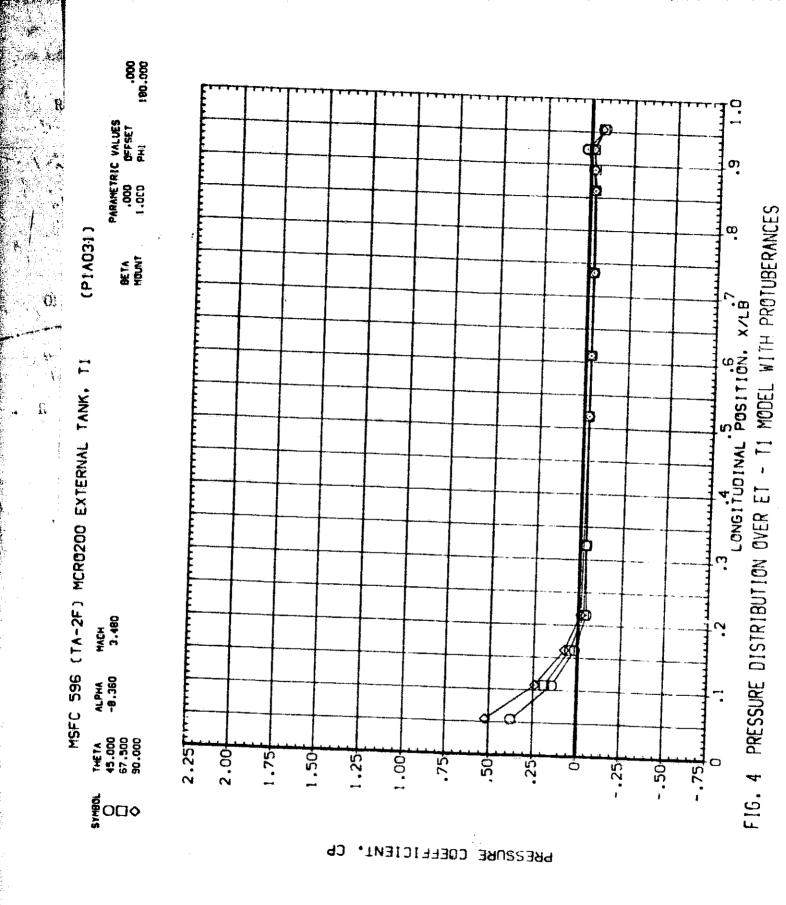
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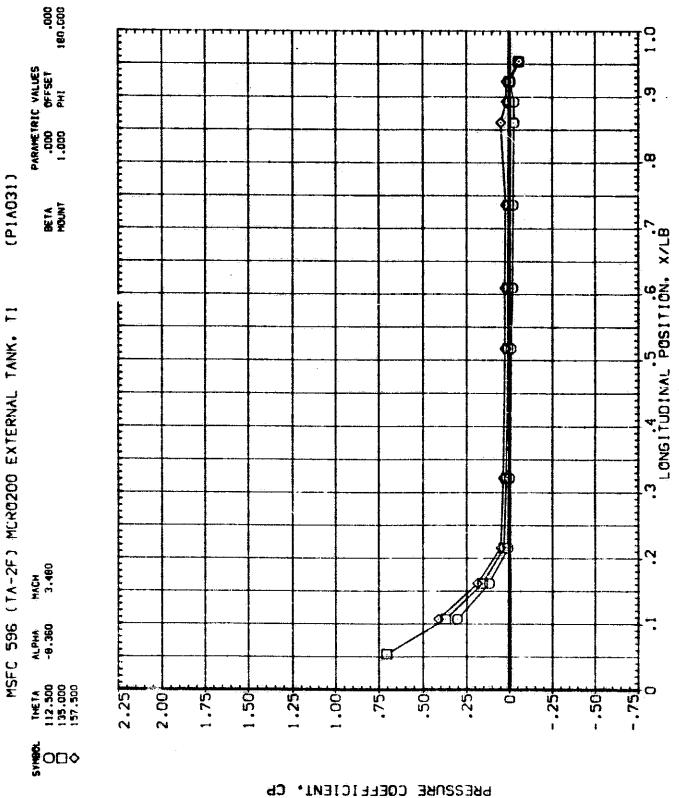


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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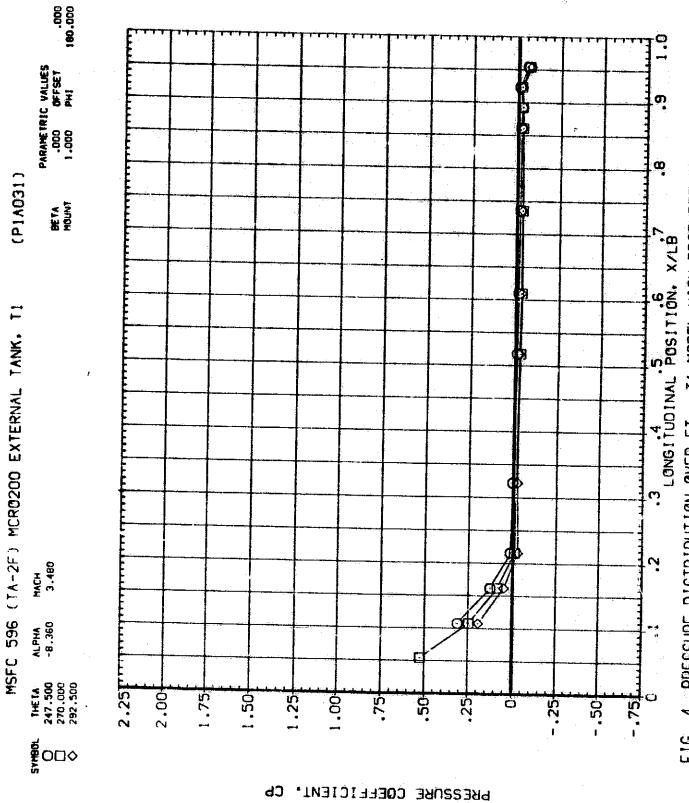
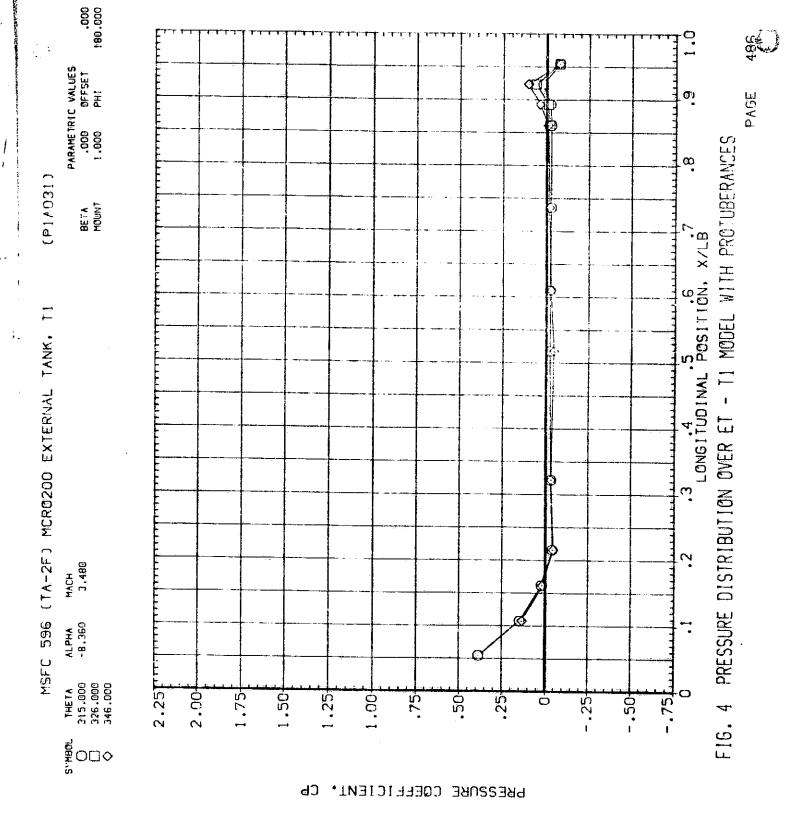
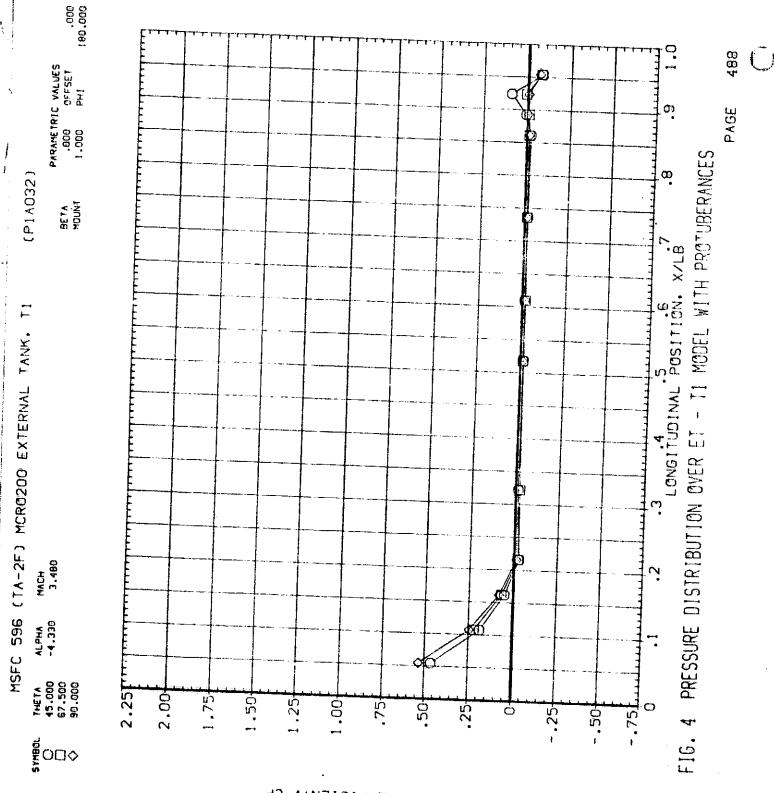


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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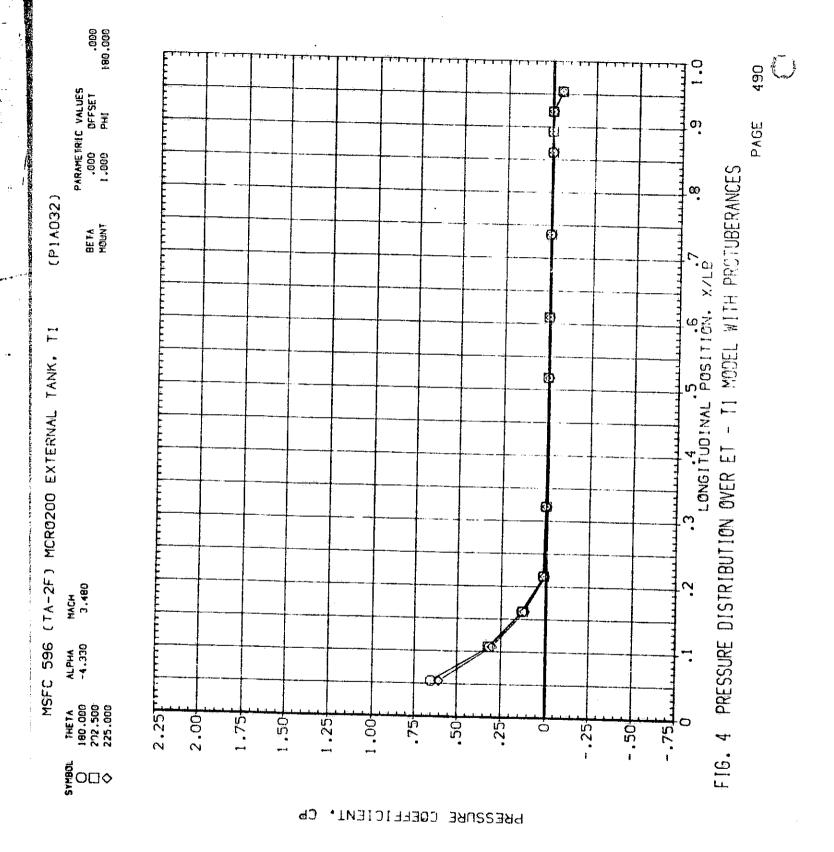
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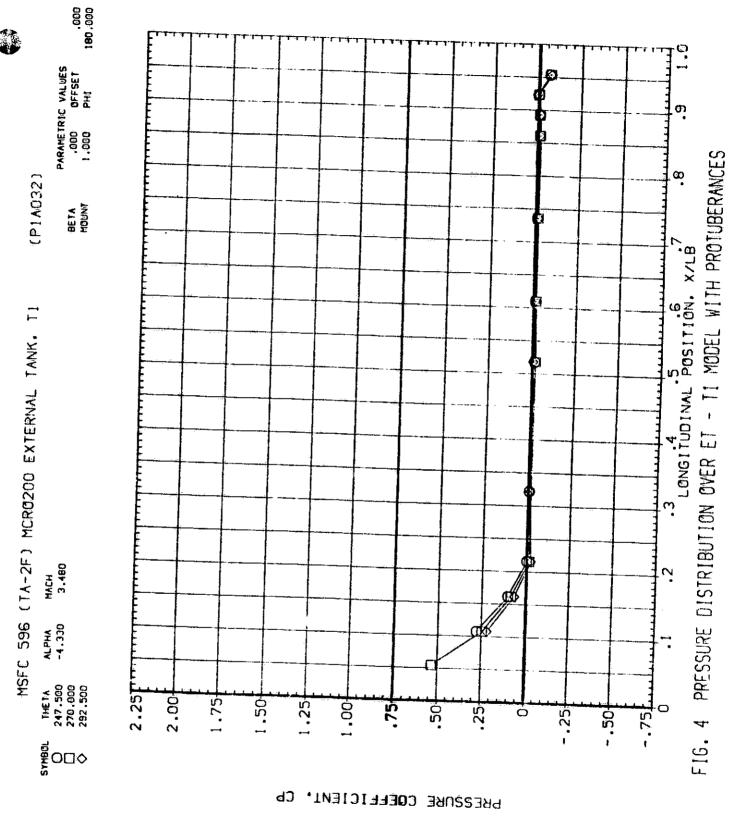
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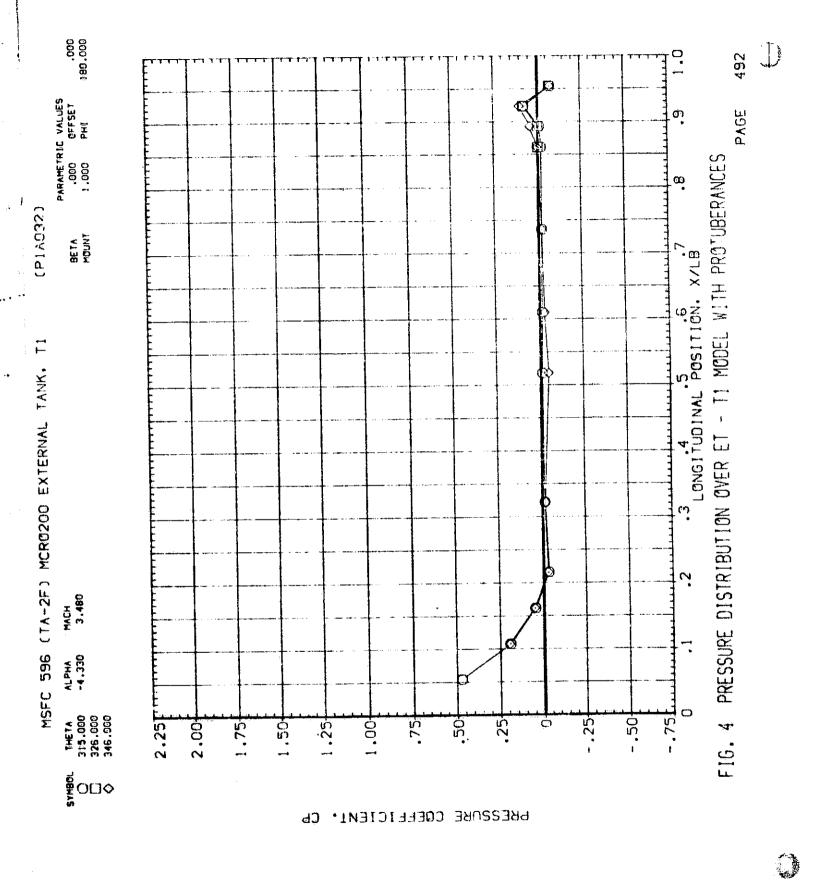
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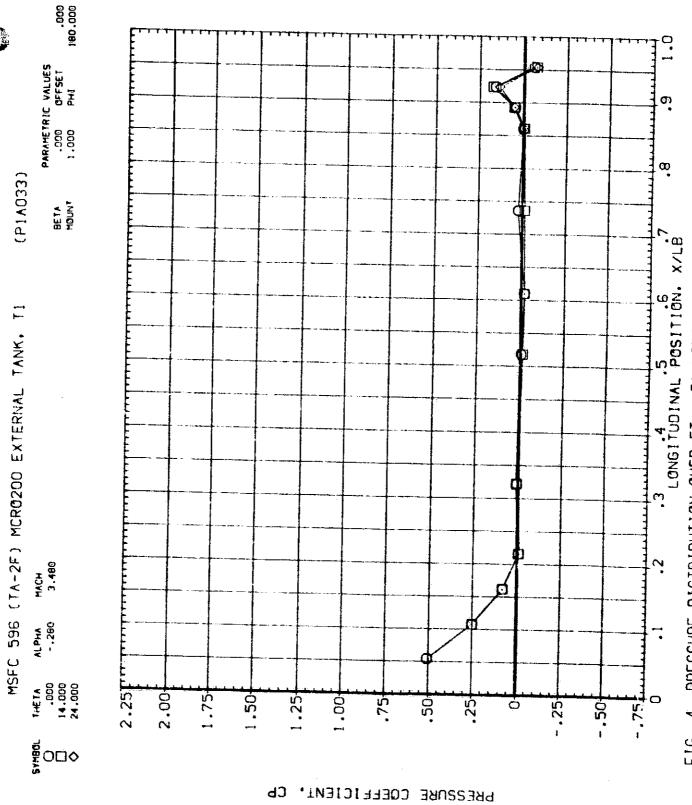
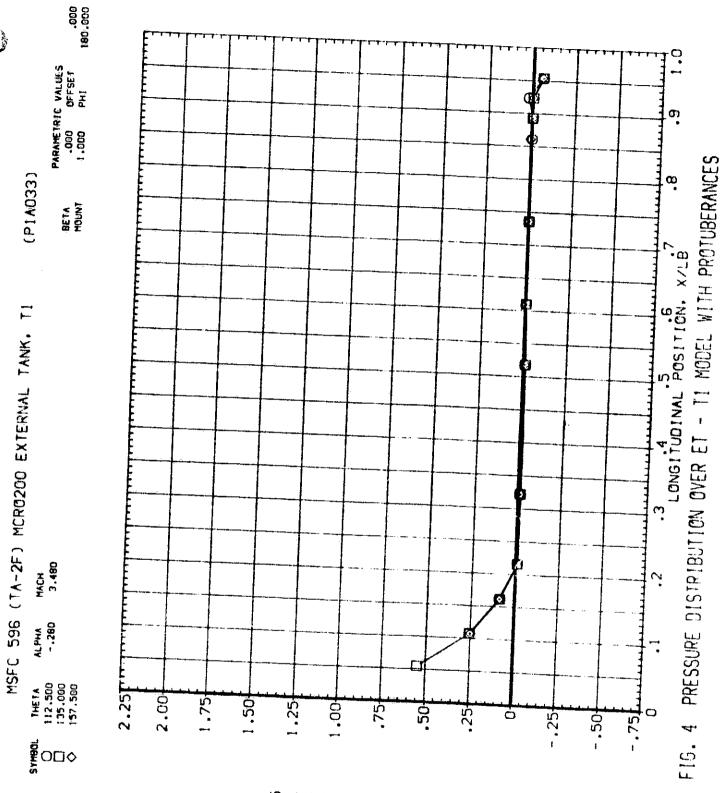


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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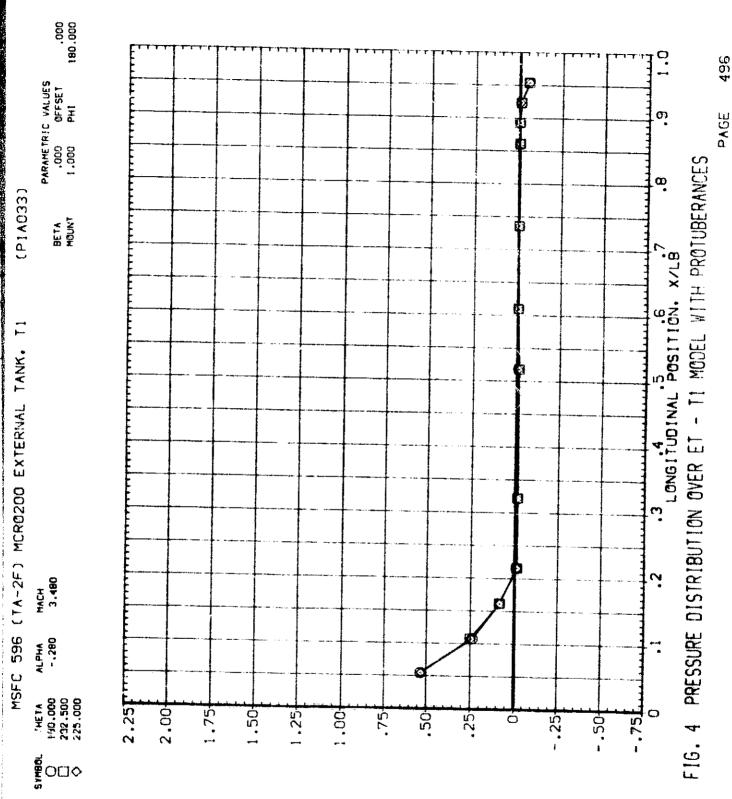
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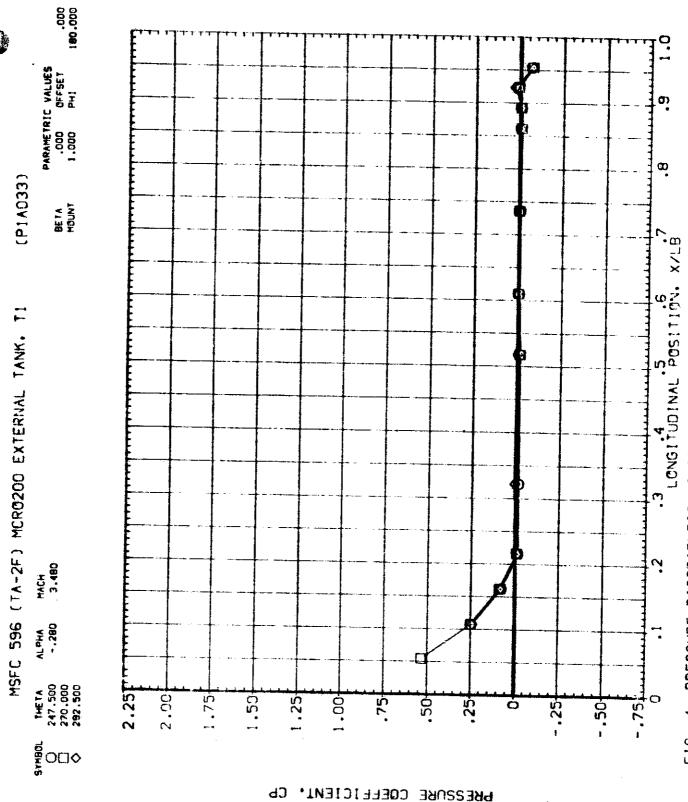


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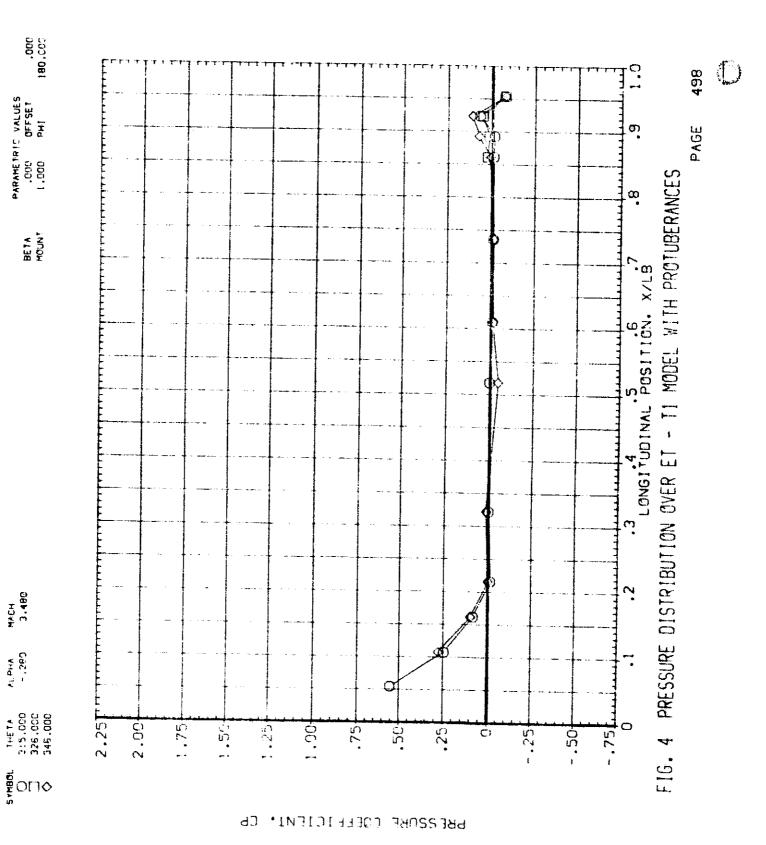
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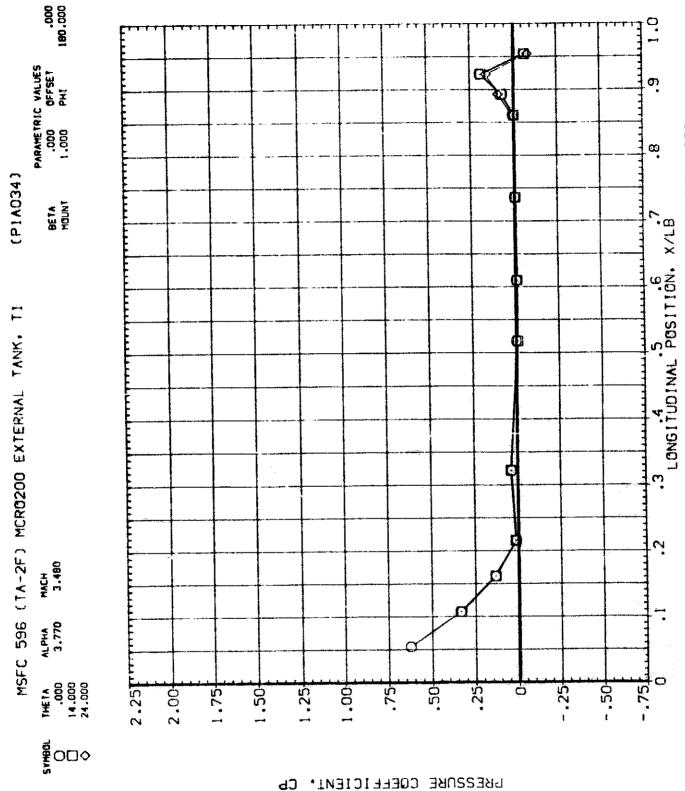
- TI MODEL WITH PROTUBERANCES 딥 FIG. 4 PRESSURE DISTRIBUTION OVER



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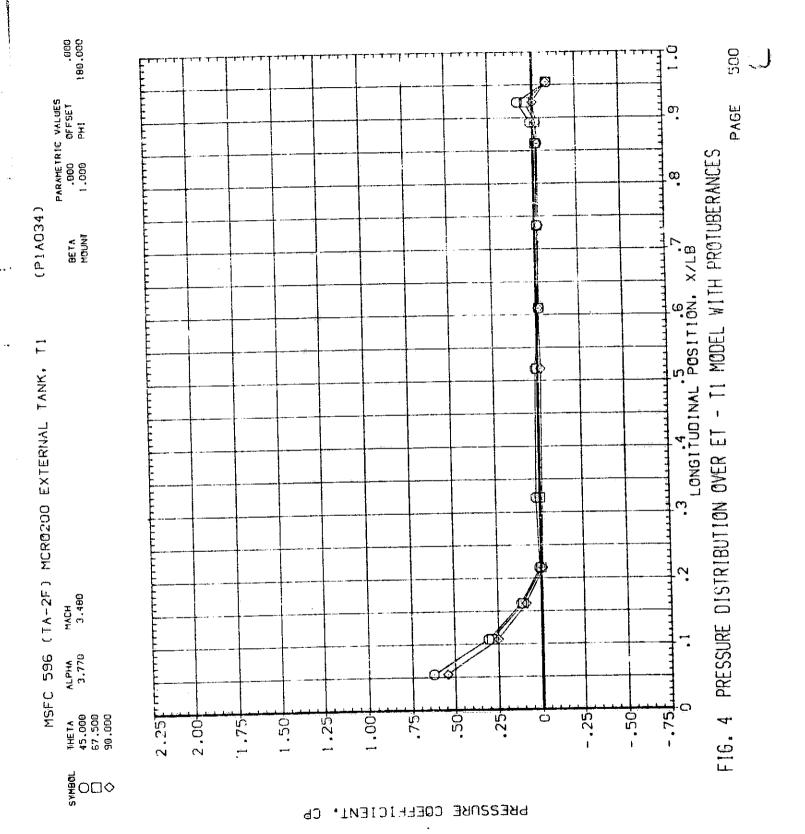


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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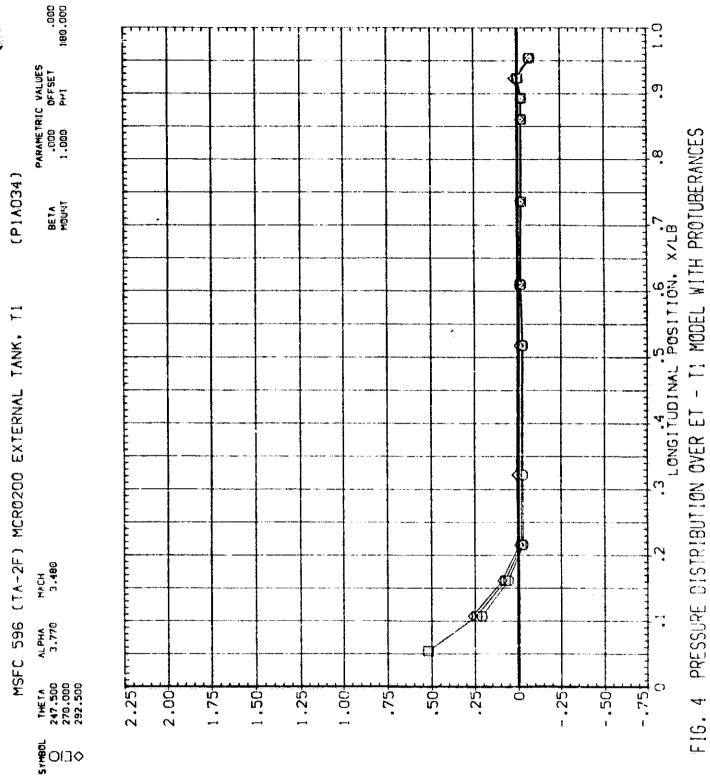
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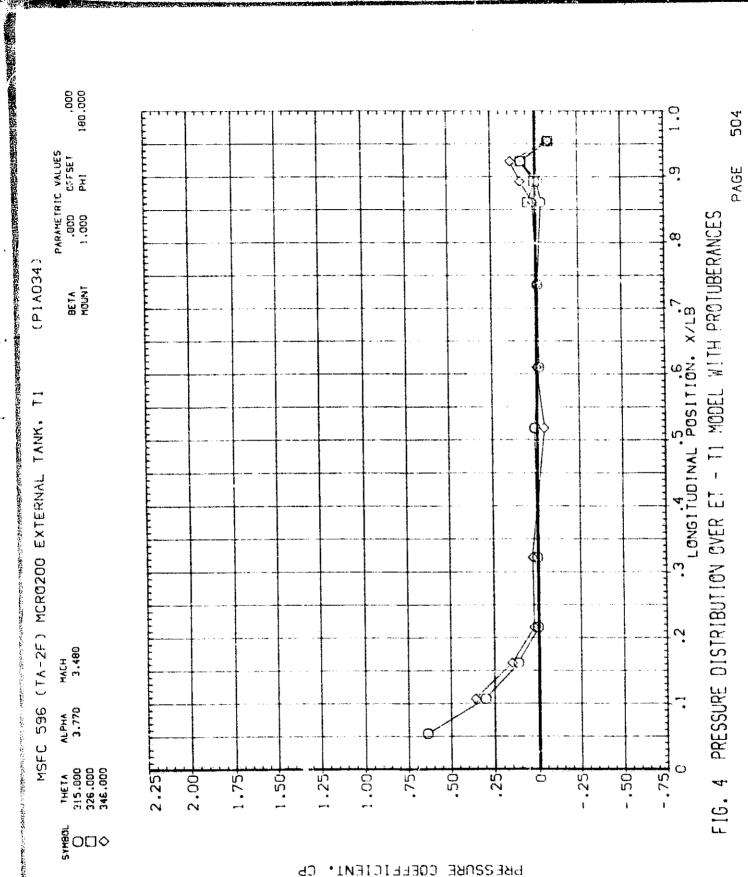
FIG. 4 PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES

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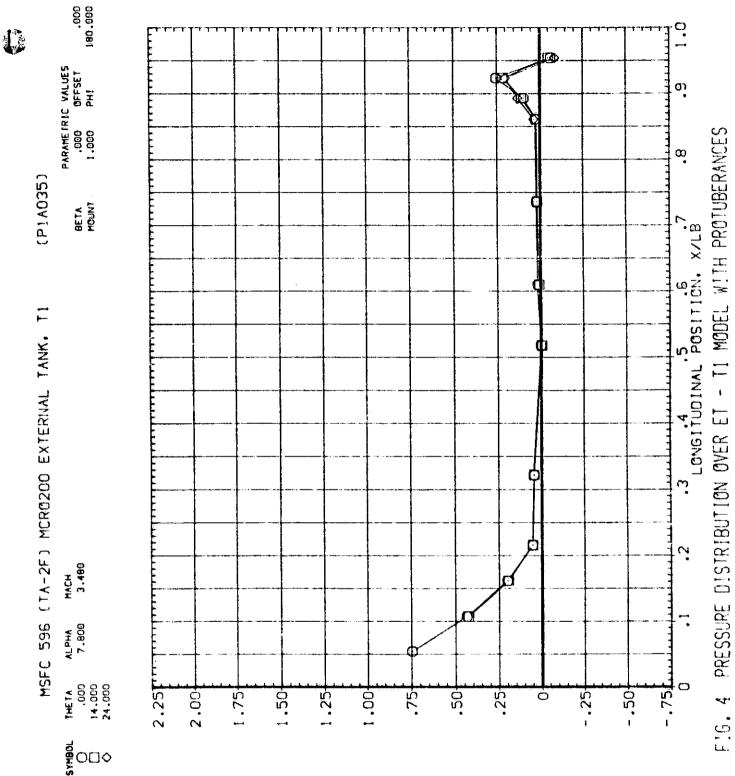


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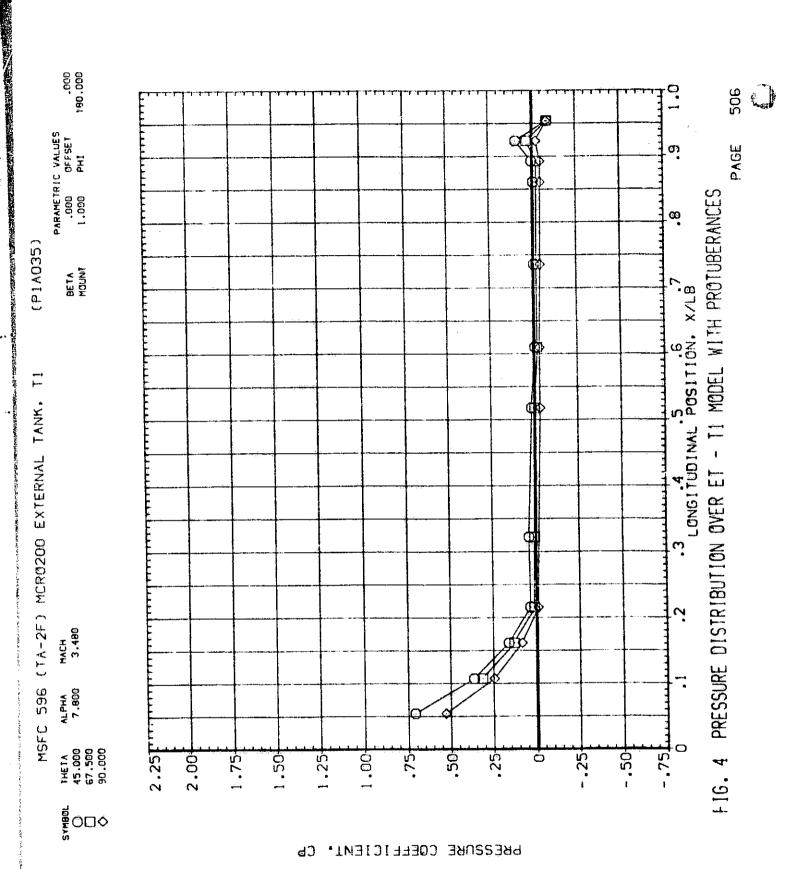
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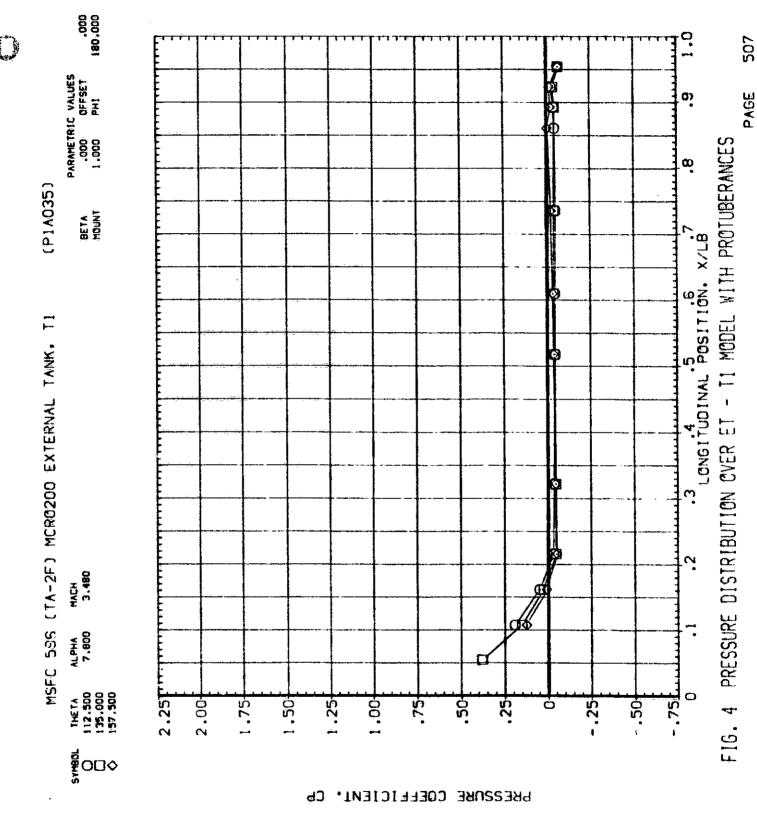
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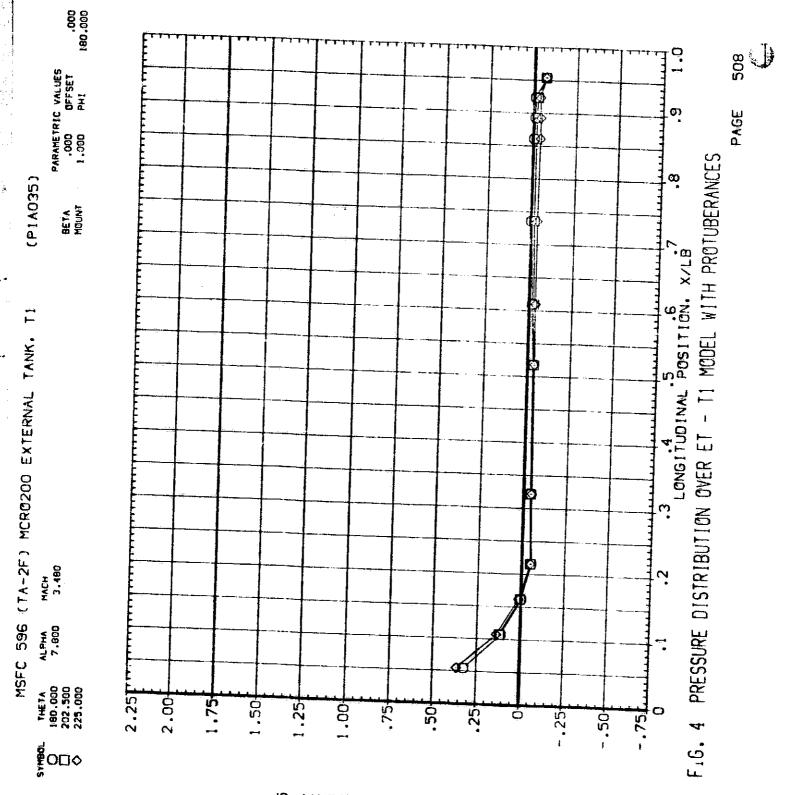


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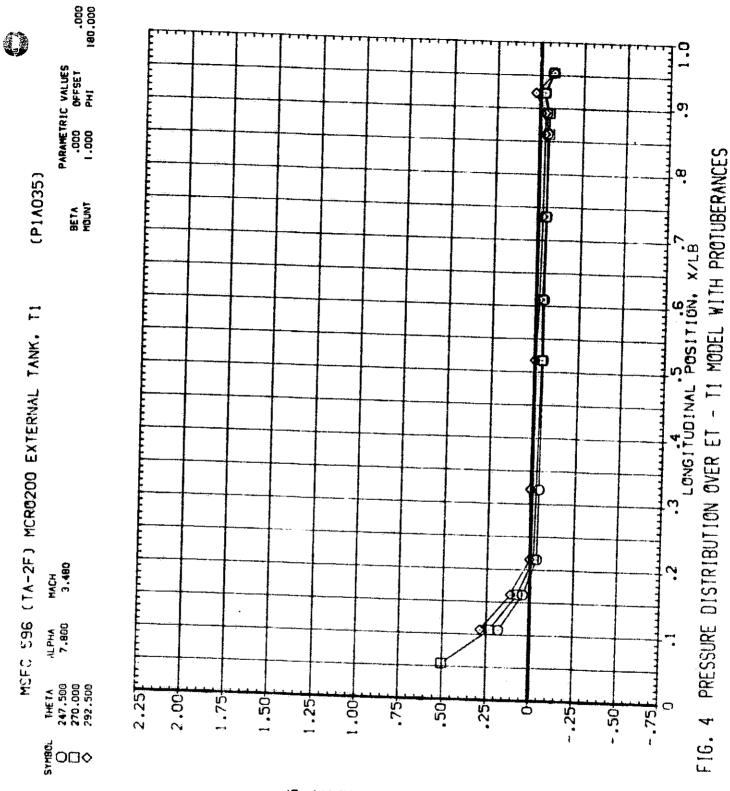
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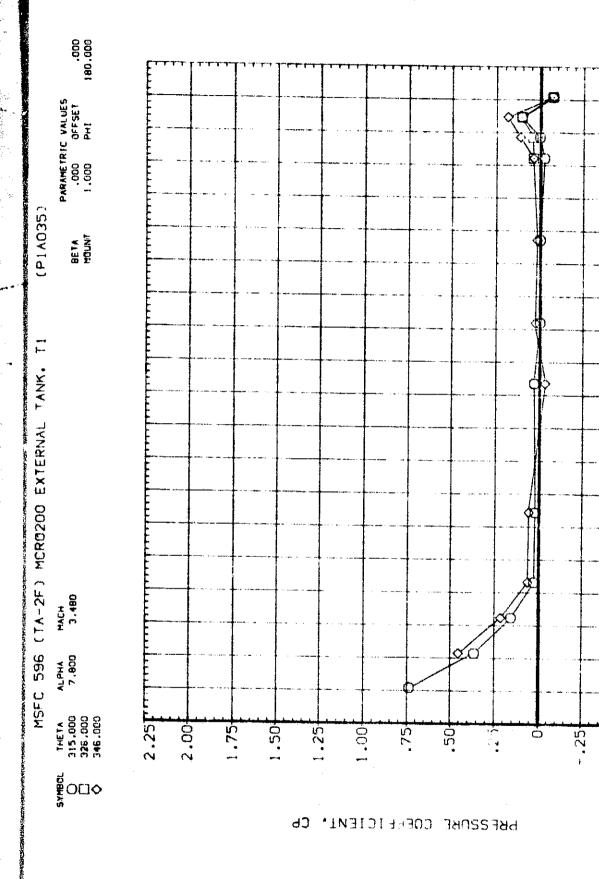
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PAGE - TI MODEL WITH PROTUBERANCES LONGITUDINAL POSITION. X/LB FIG. 4 PRESSURE DISTRIBUTION OVER ET

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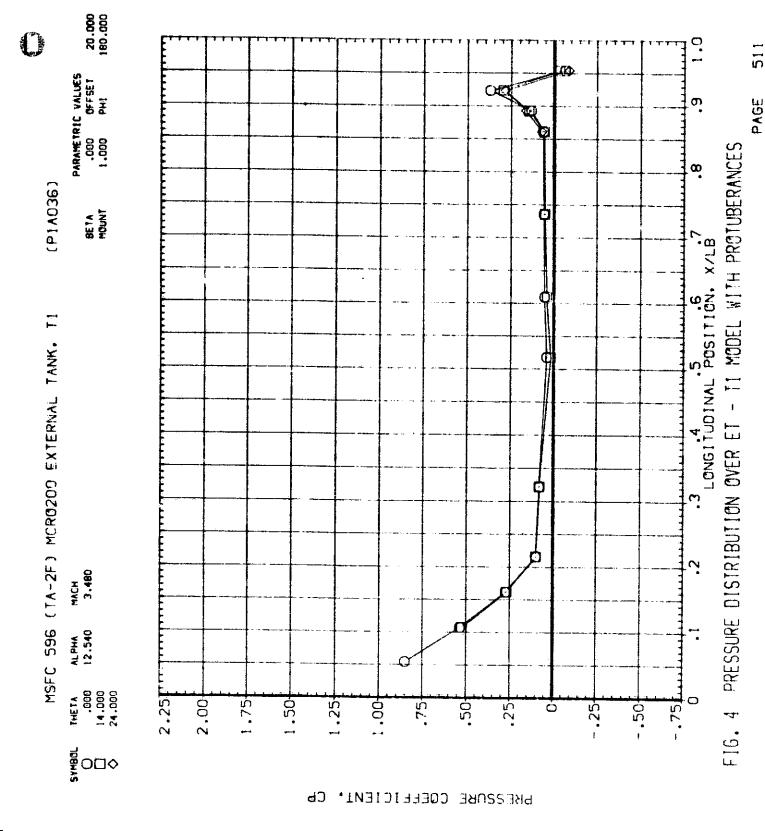
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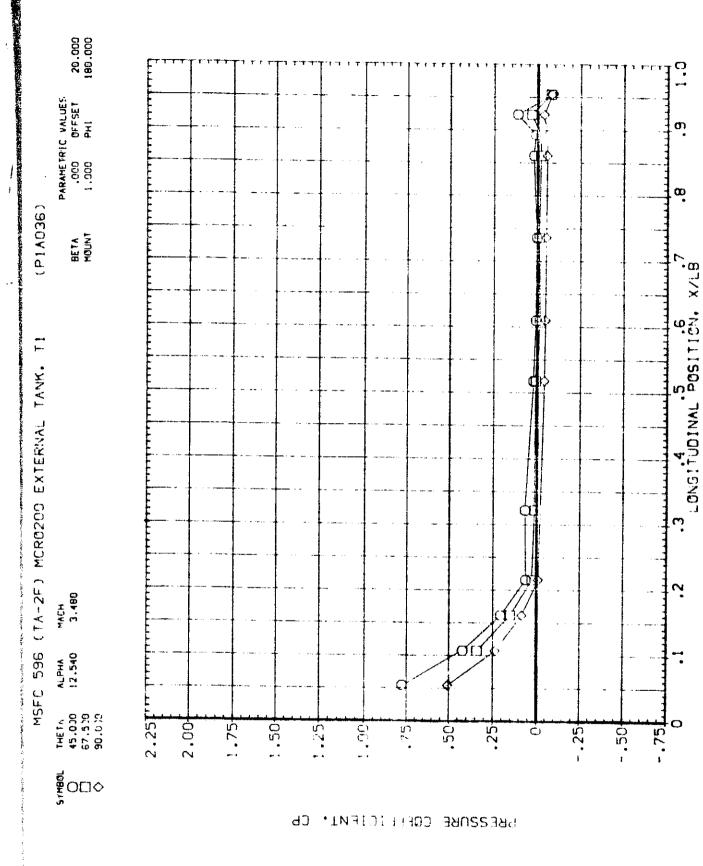
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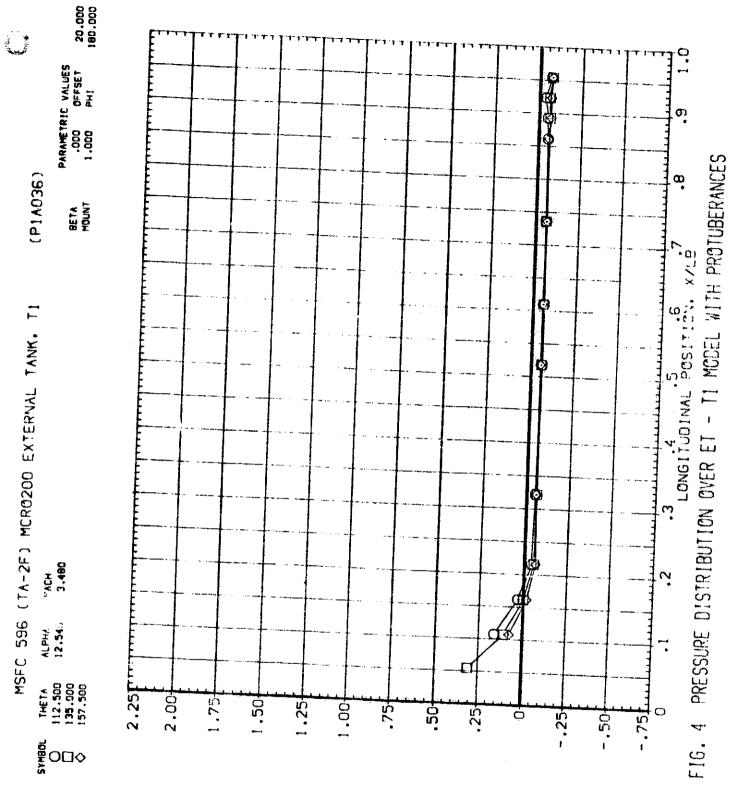
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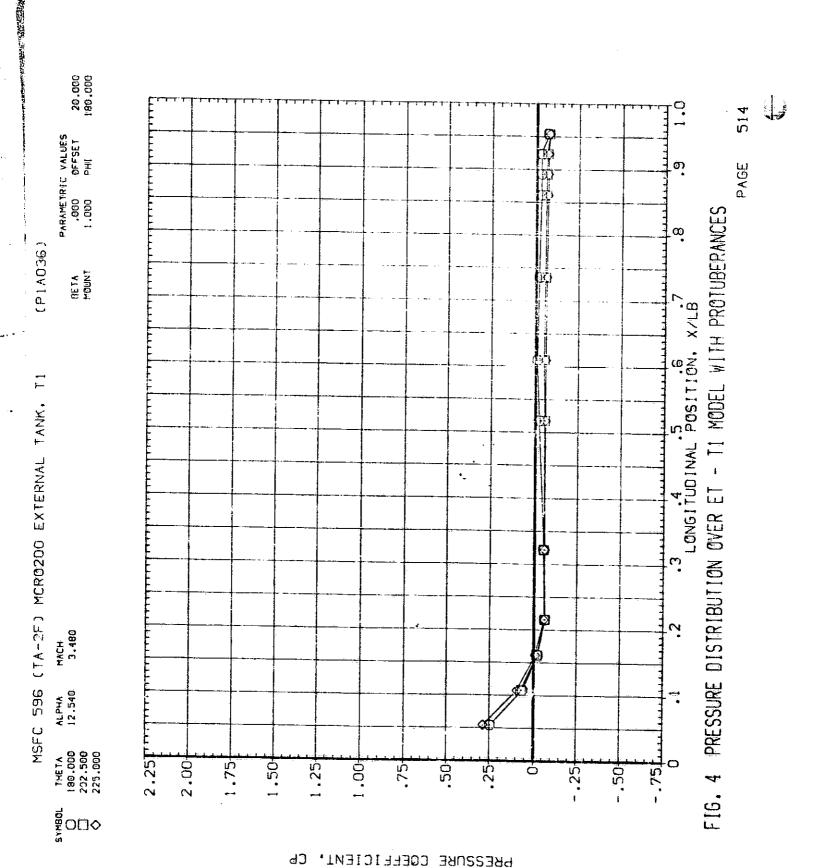


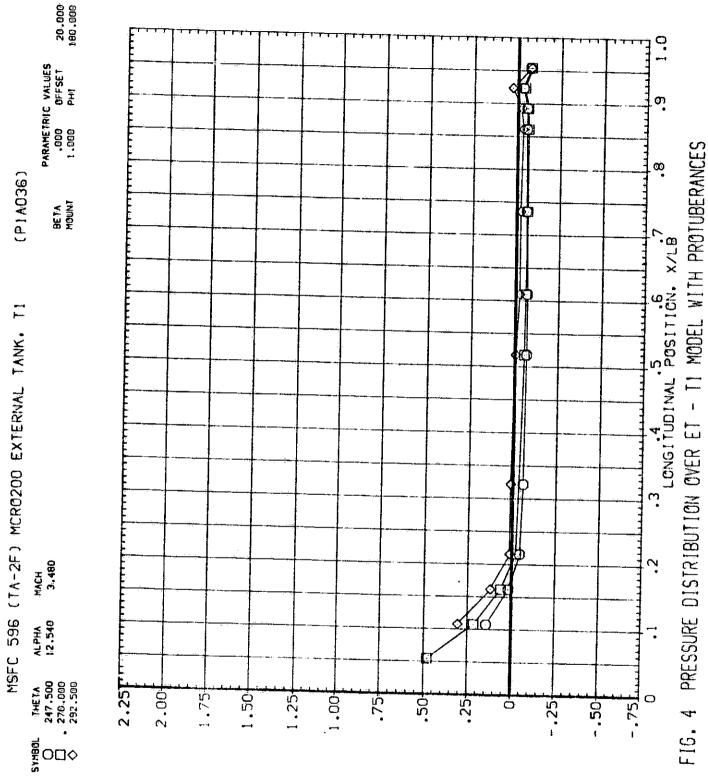
II MODEL WITH PROTUBERANCES ŧ FIG. 4 PRESSURE DISTRIBUTION GVER ET

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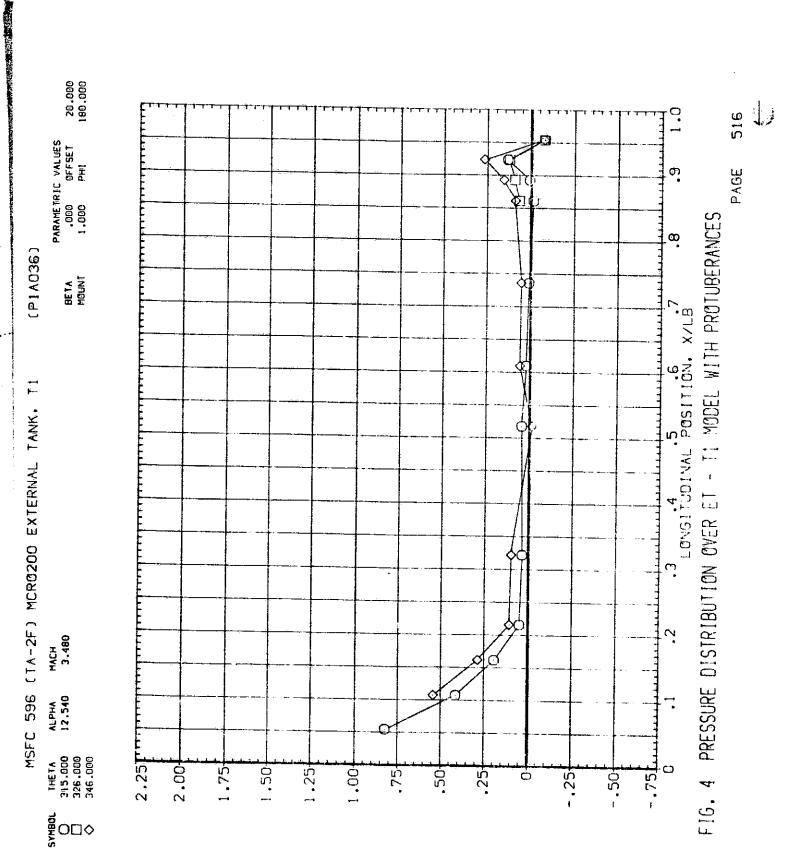


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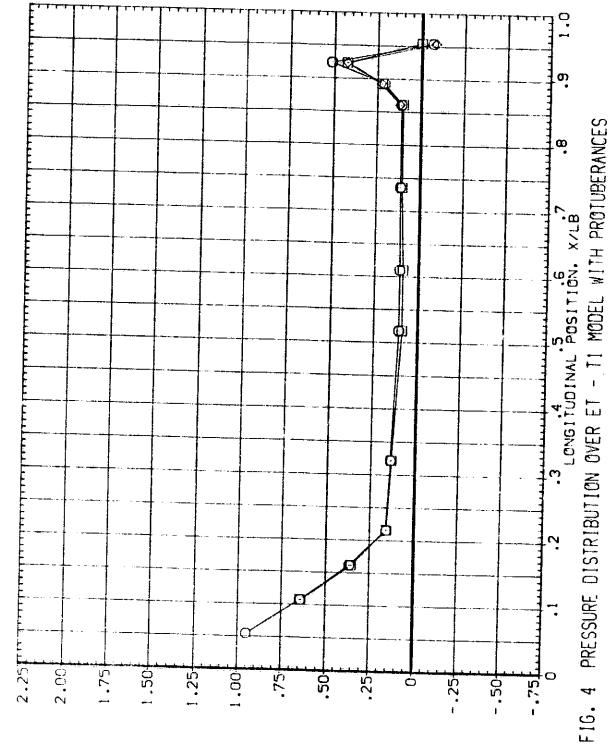


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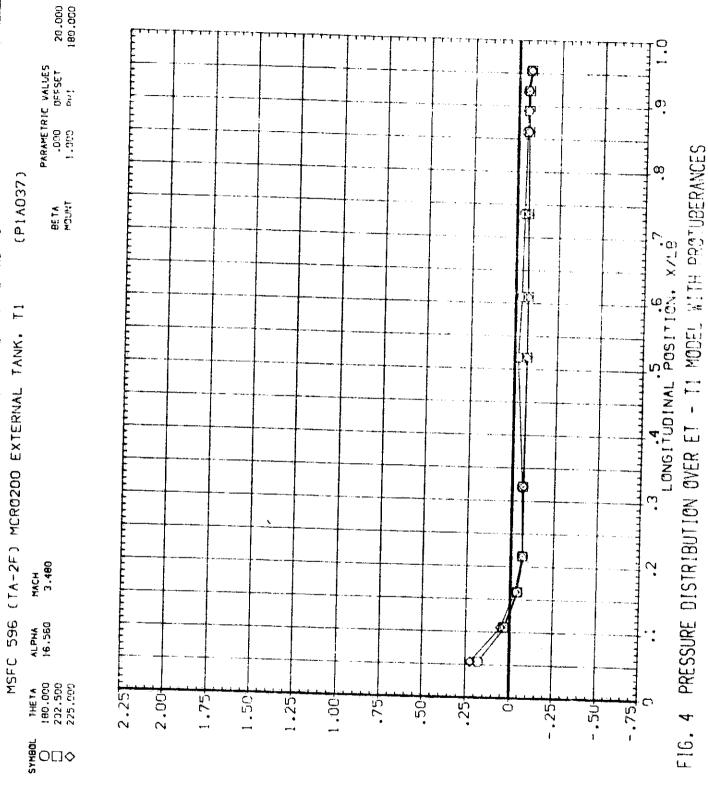
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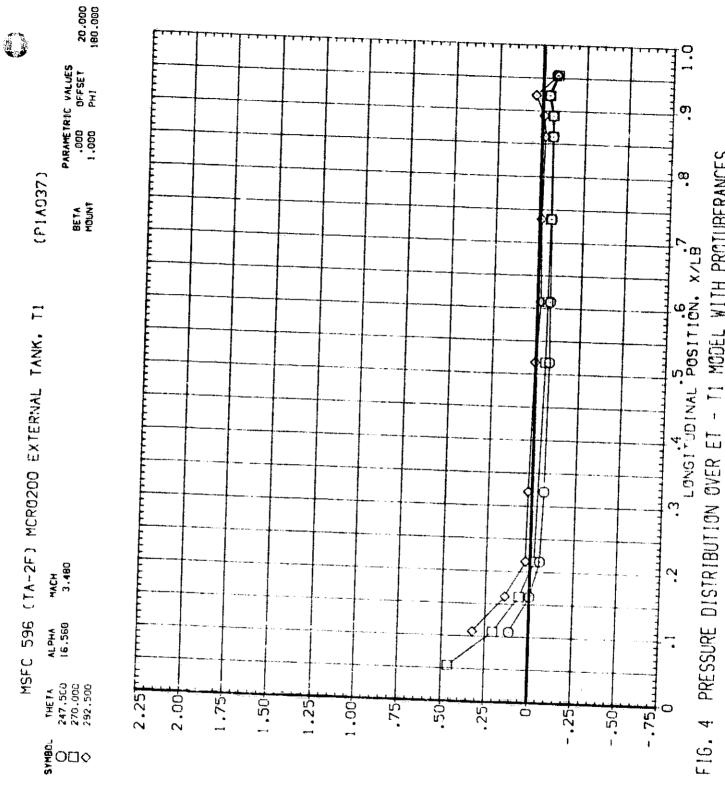


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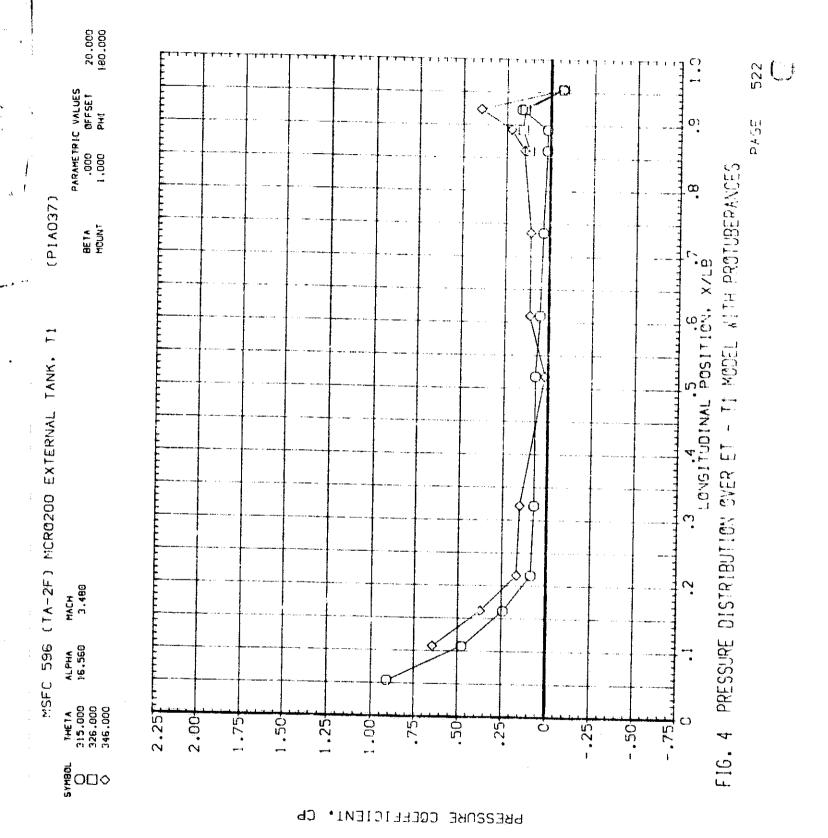
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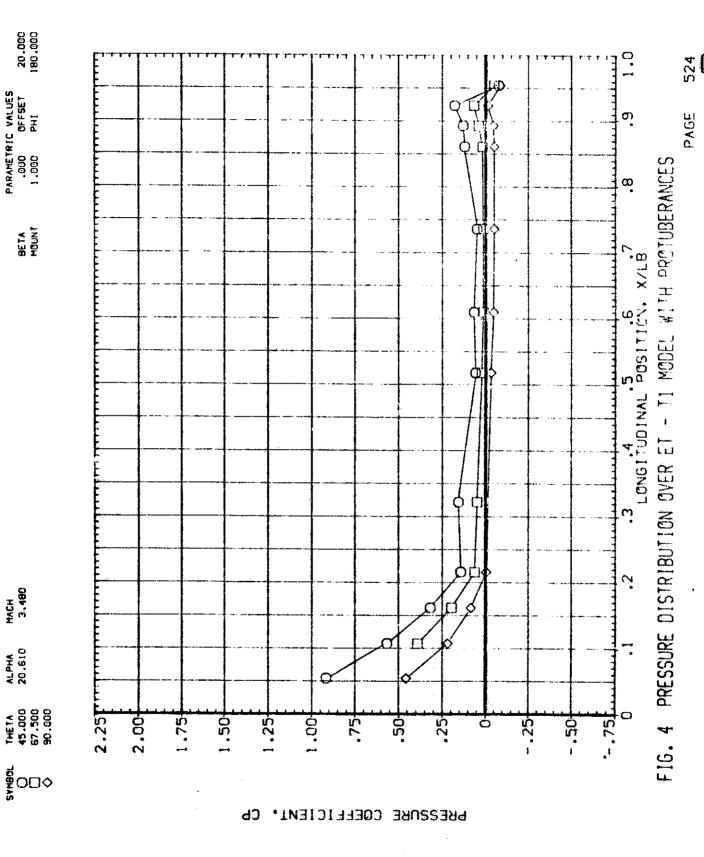
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II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

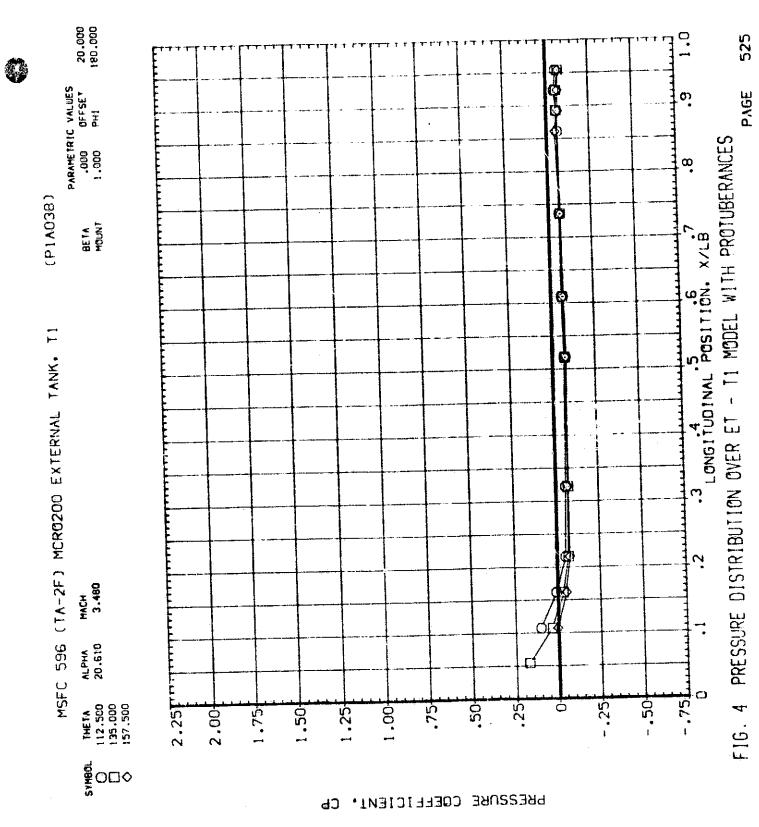


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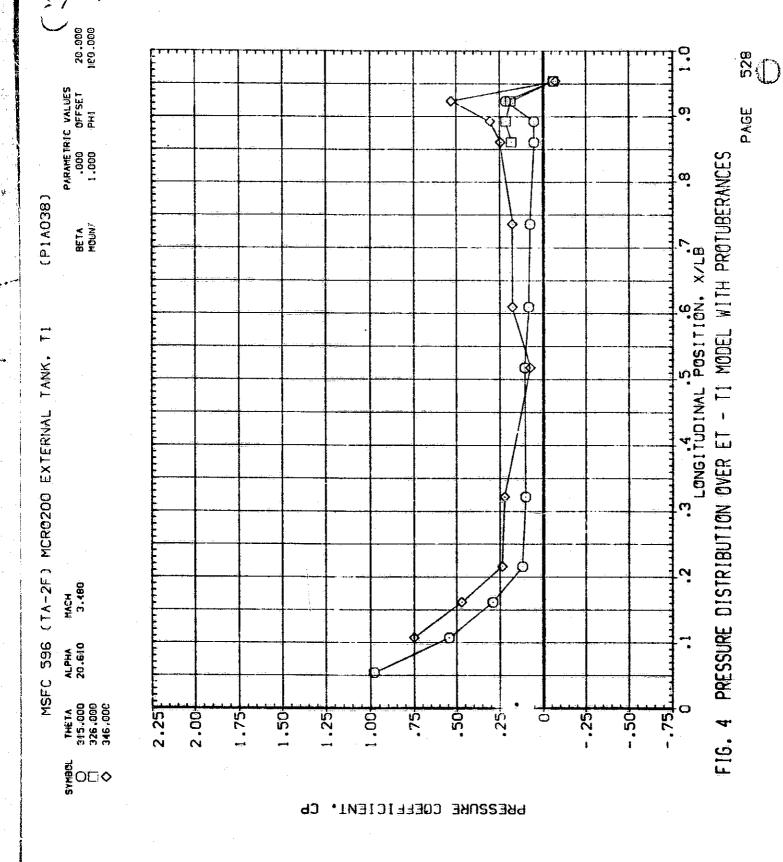
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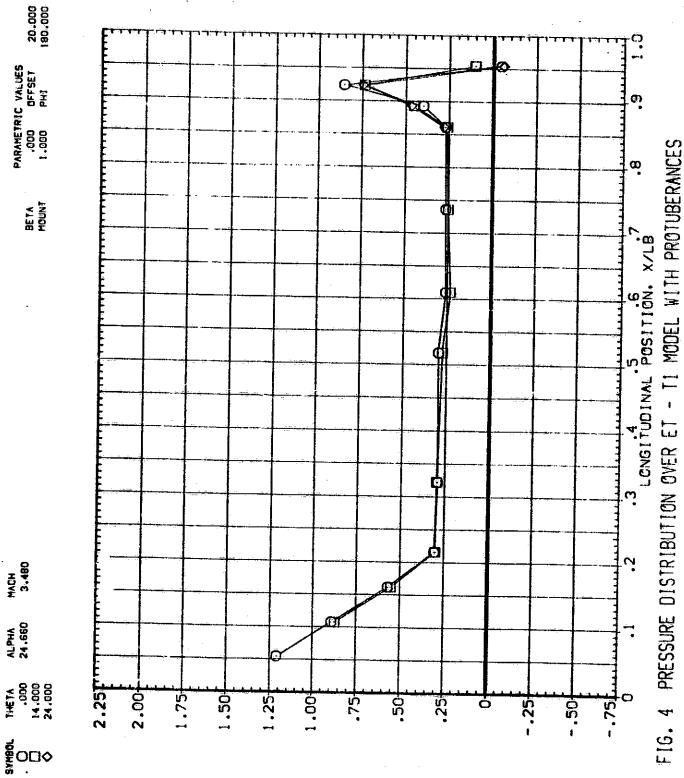
PAGE FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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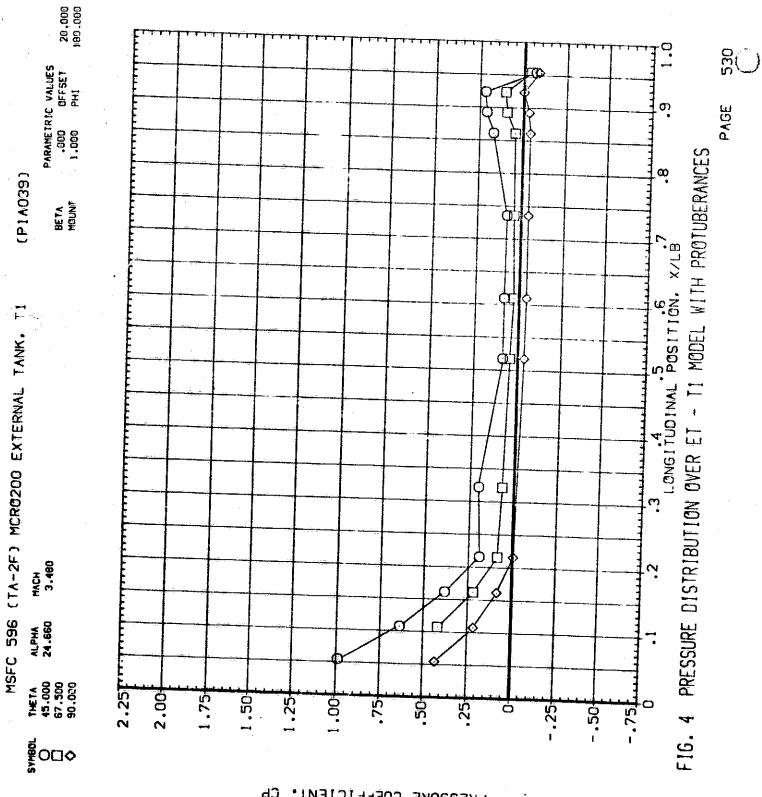
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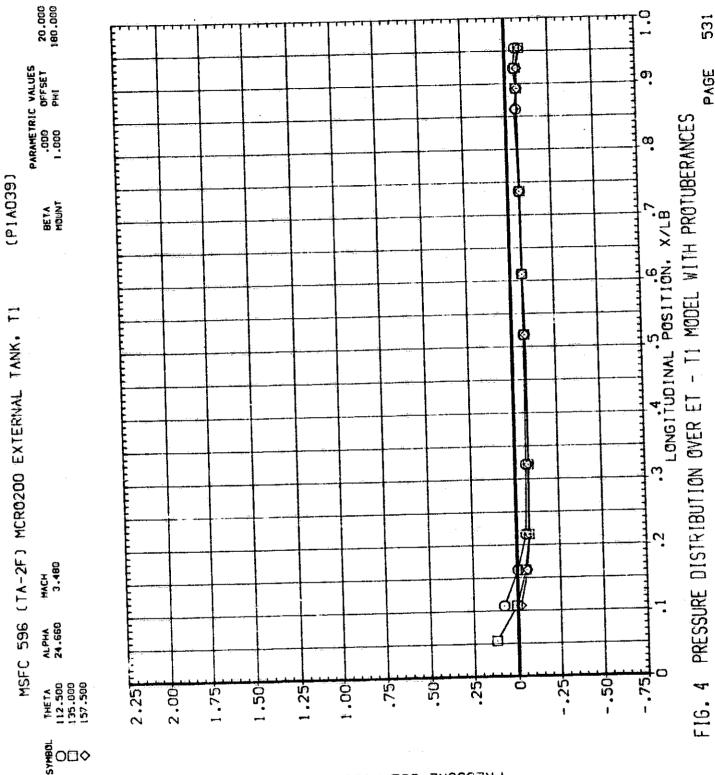


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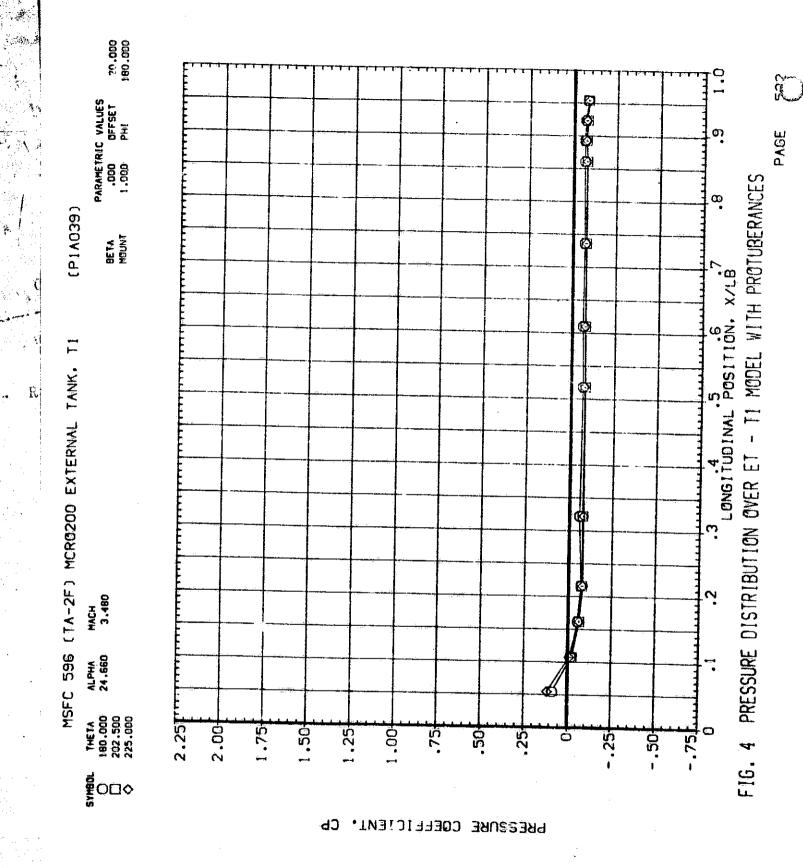
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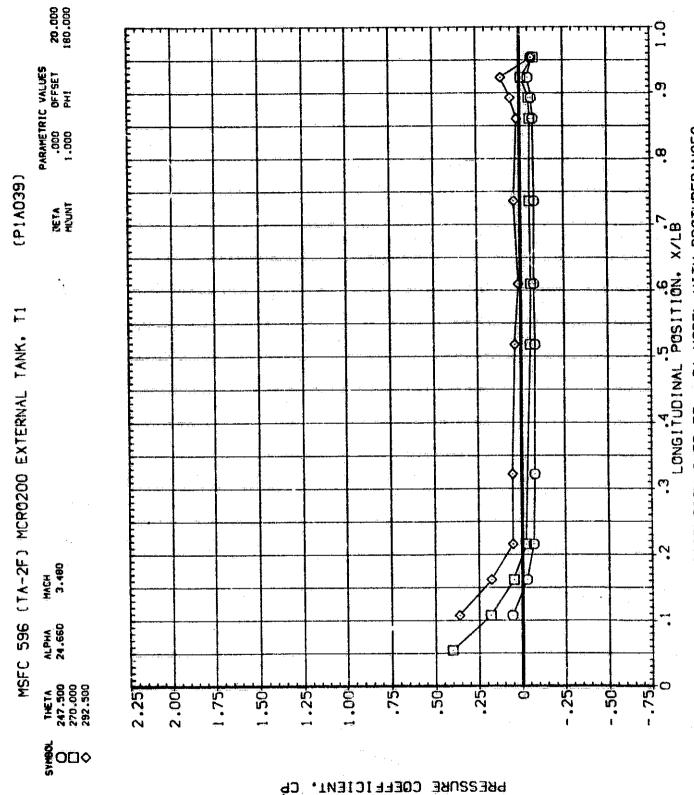
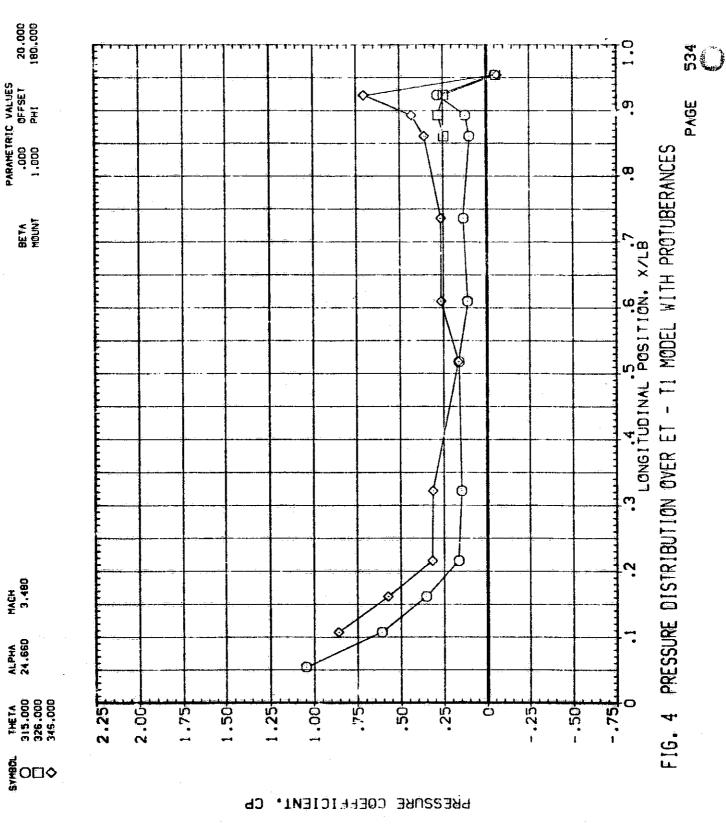


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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FIG. 4 PRESSURE DISTRIBUTION OVER ET

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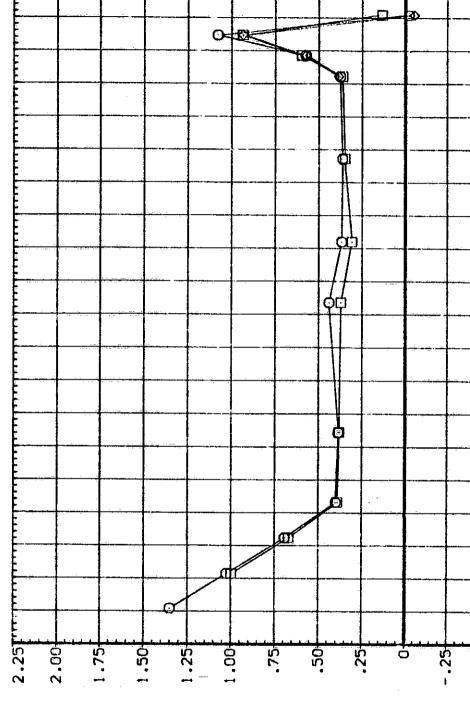
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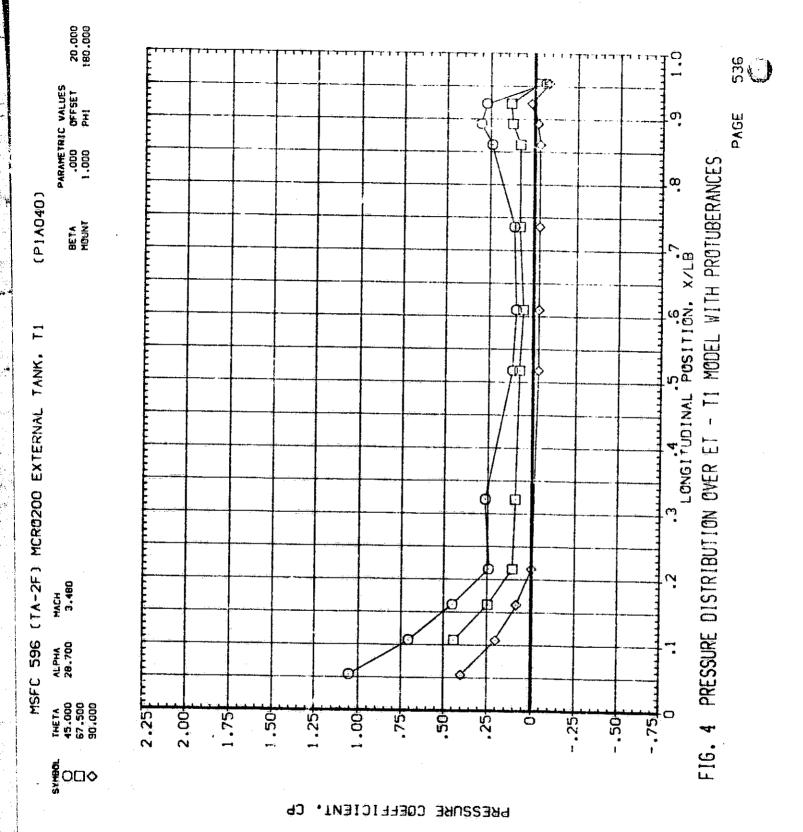
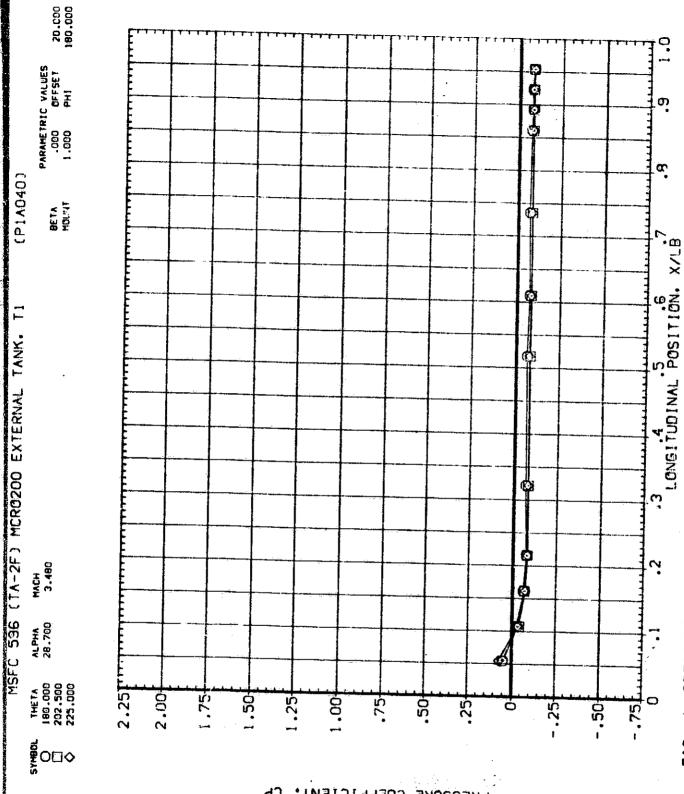


FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES

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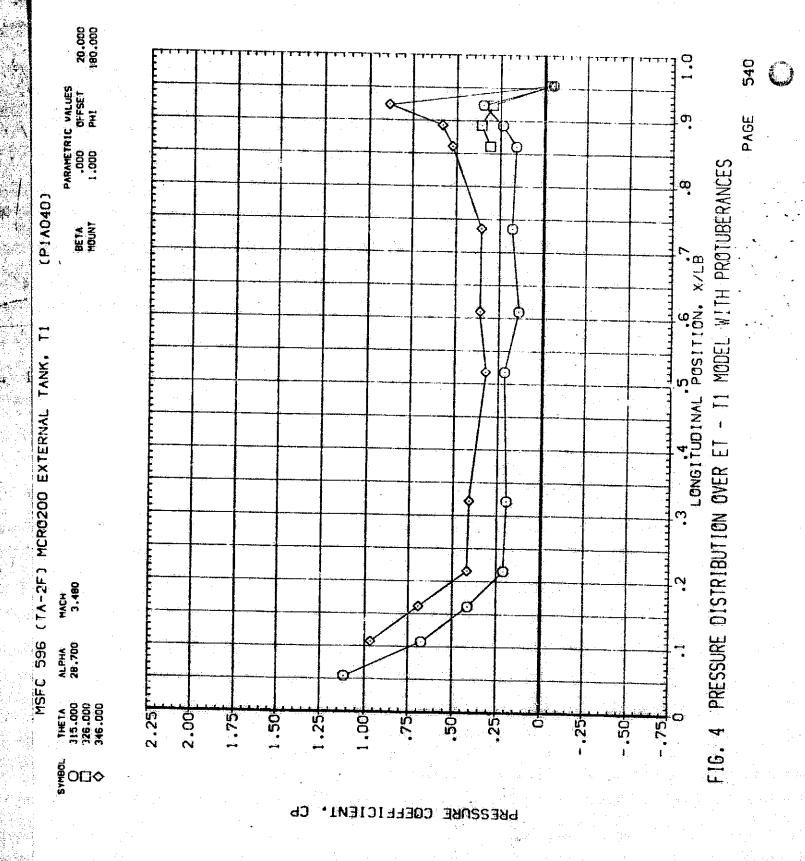
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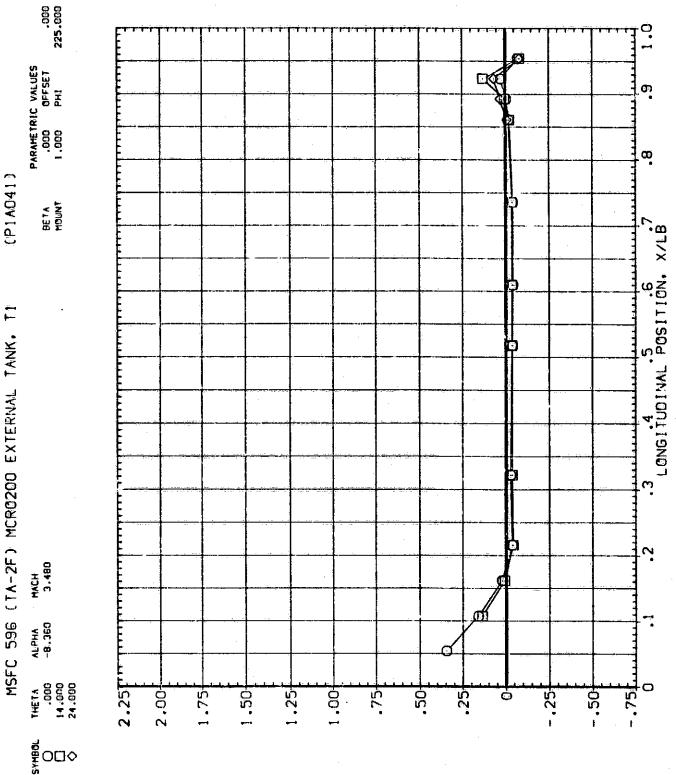
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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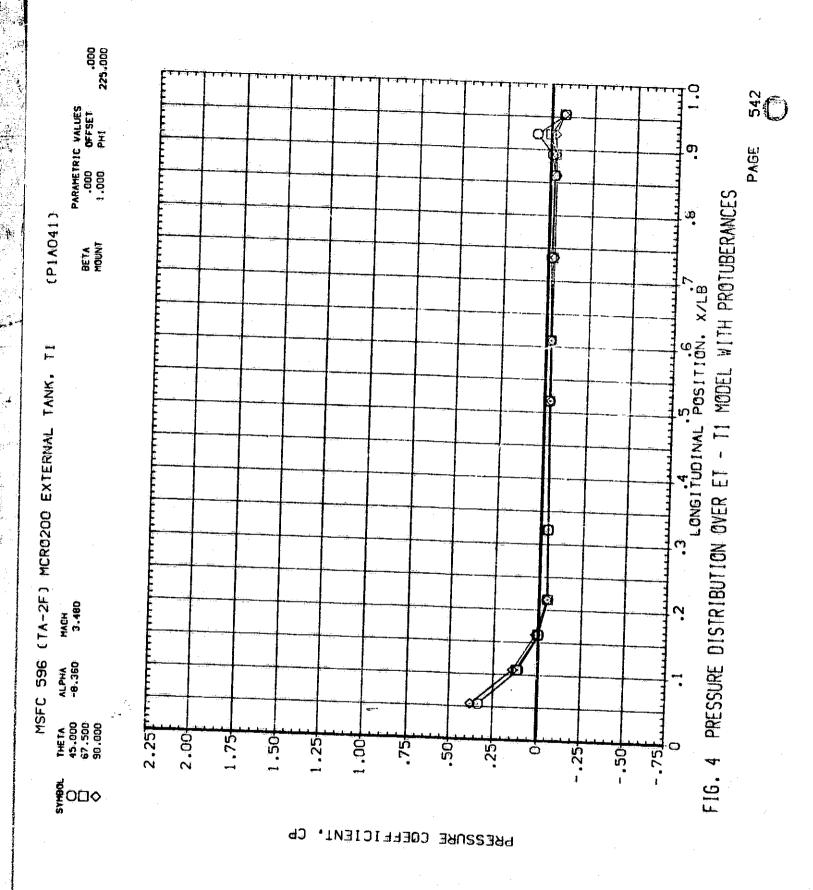
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

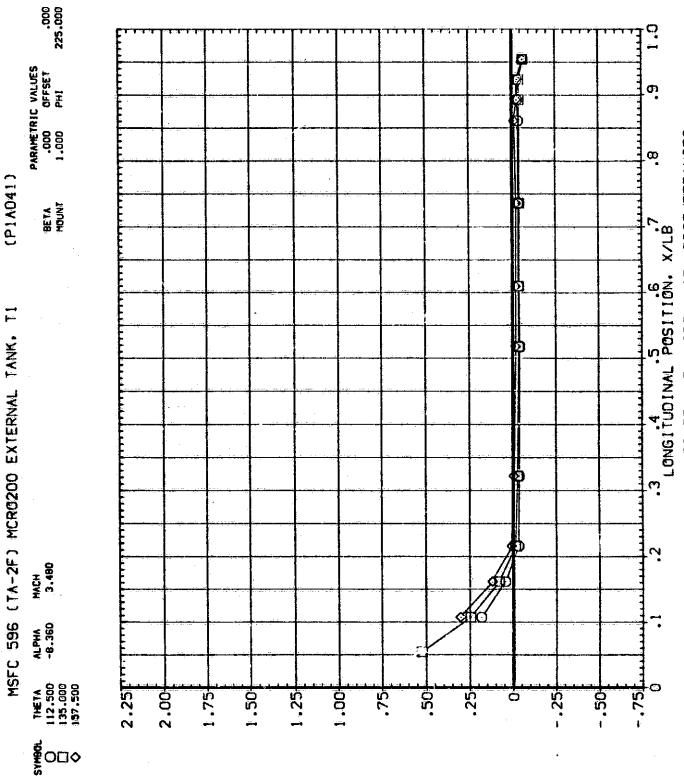




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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES





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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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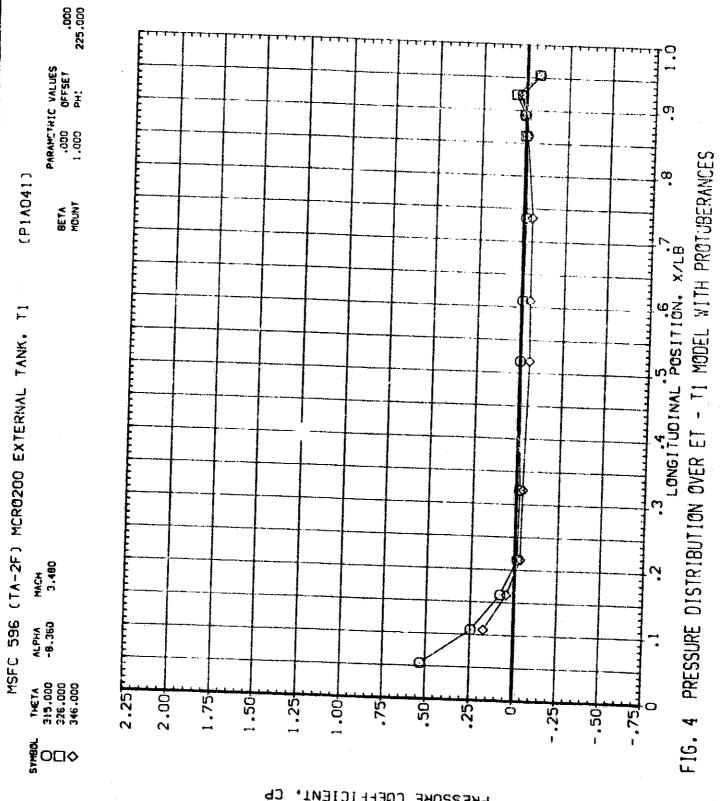
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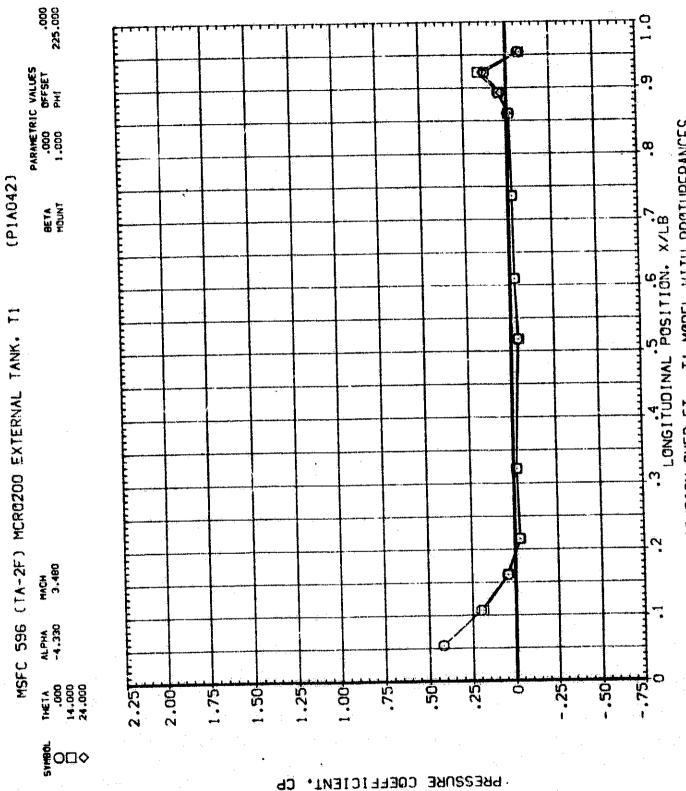
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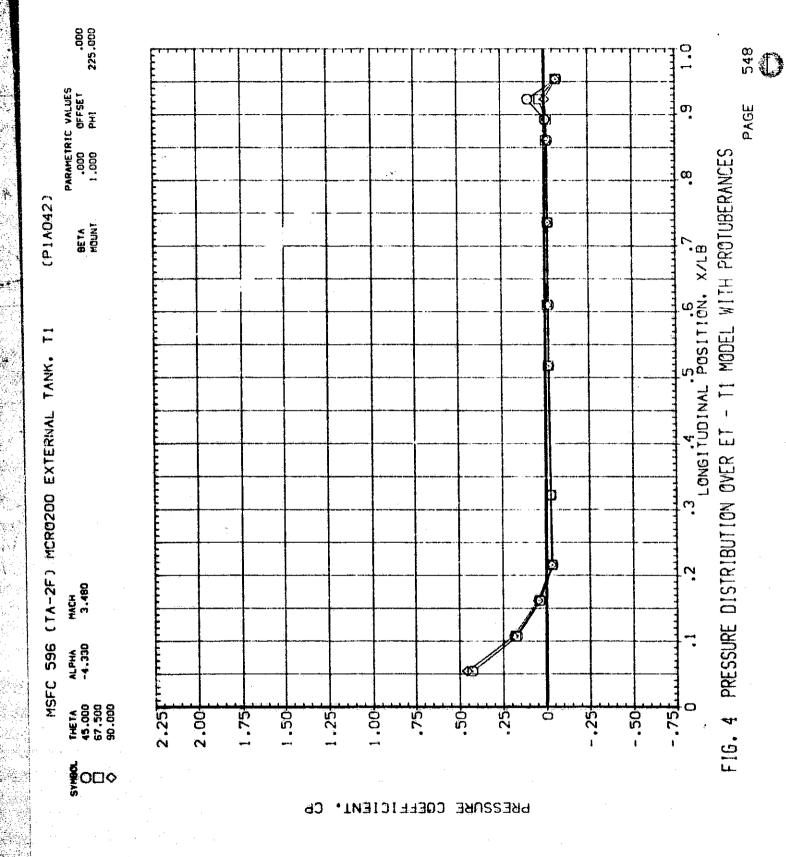
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PAGE FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

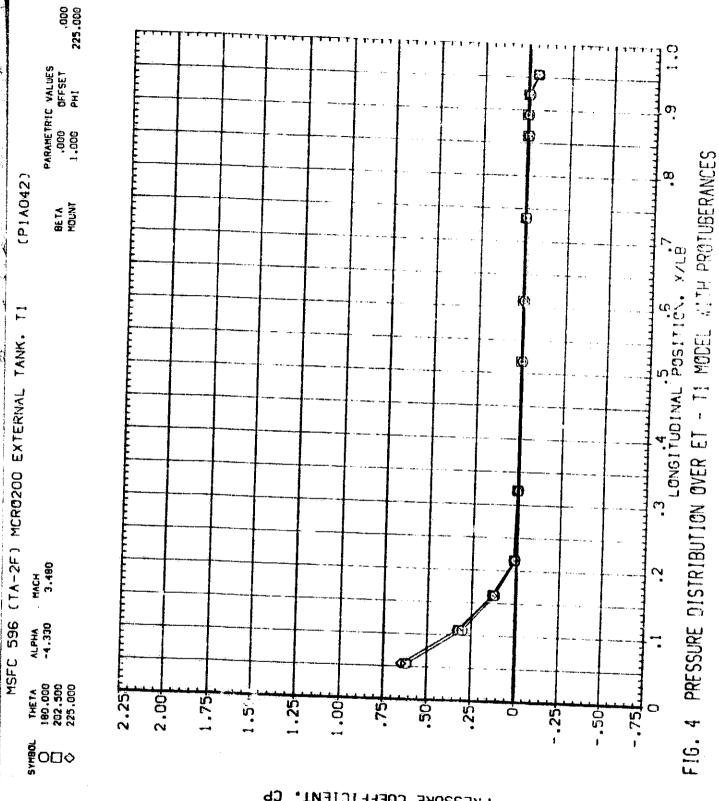


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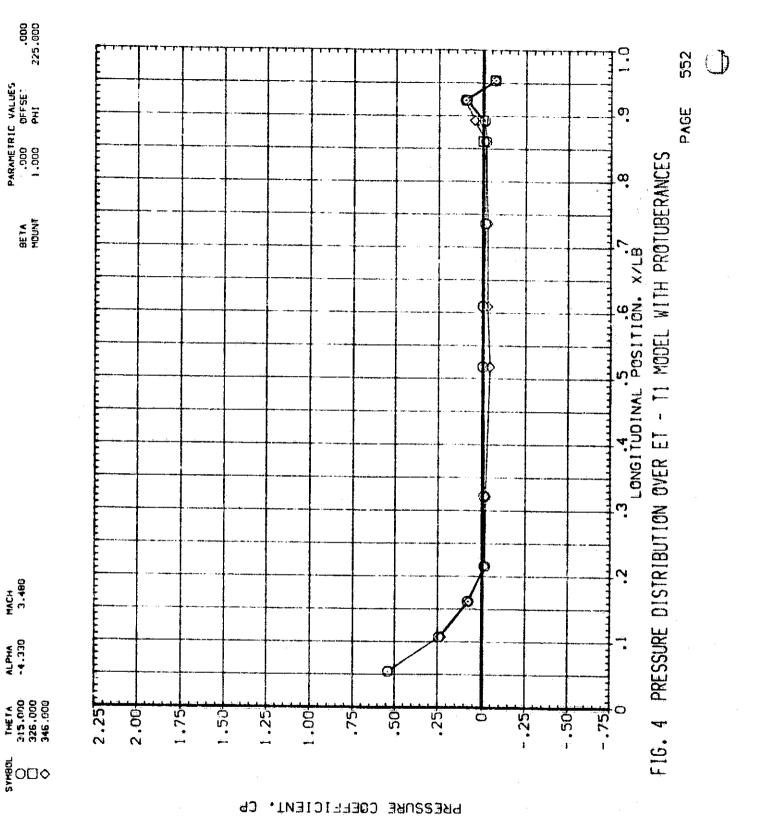
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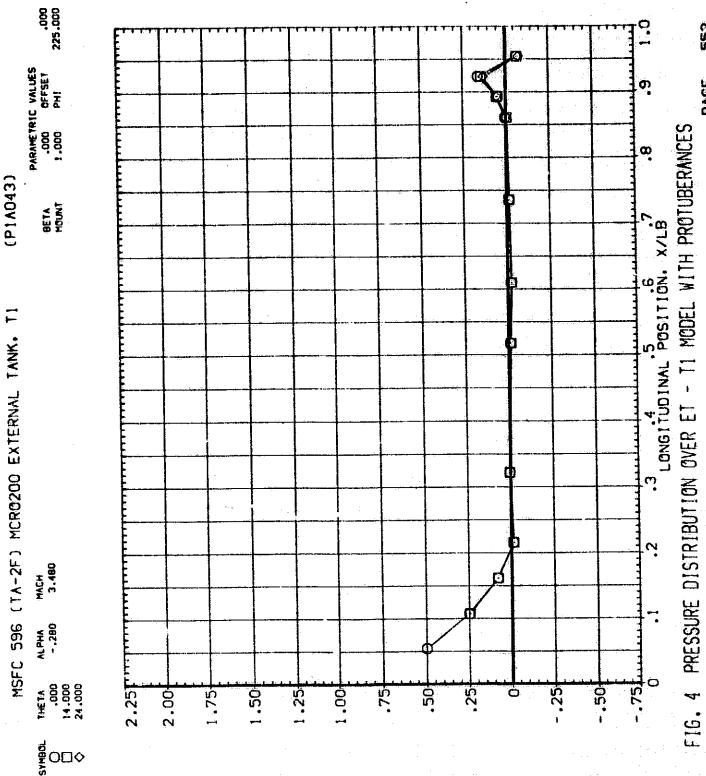


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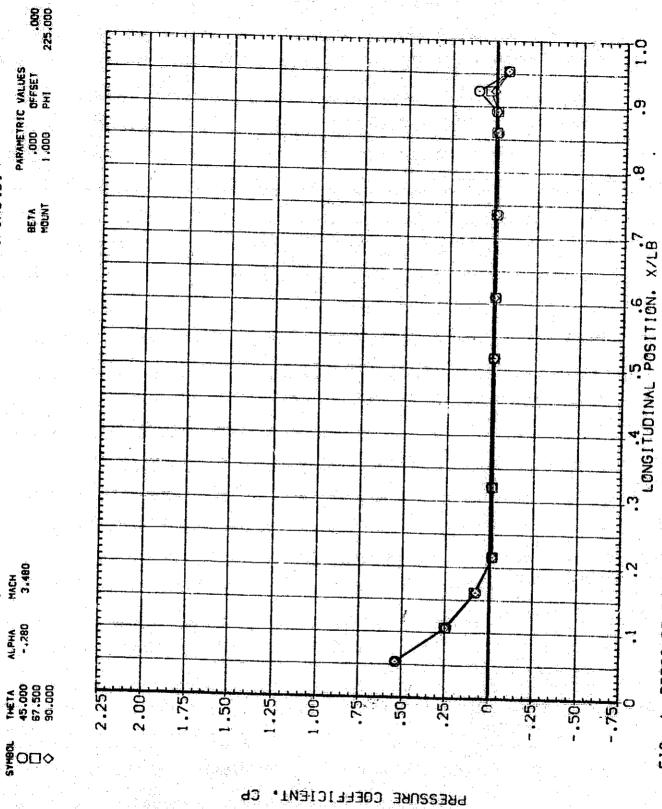
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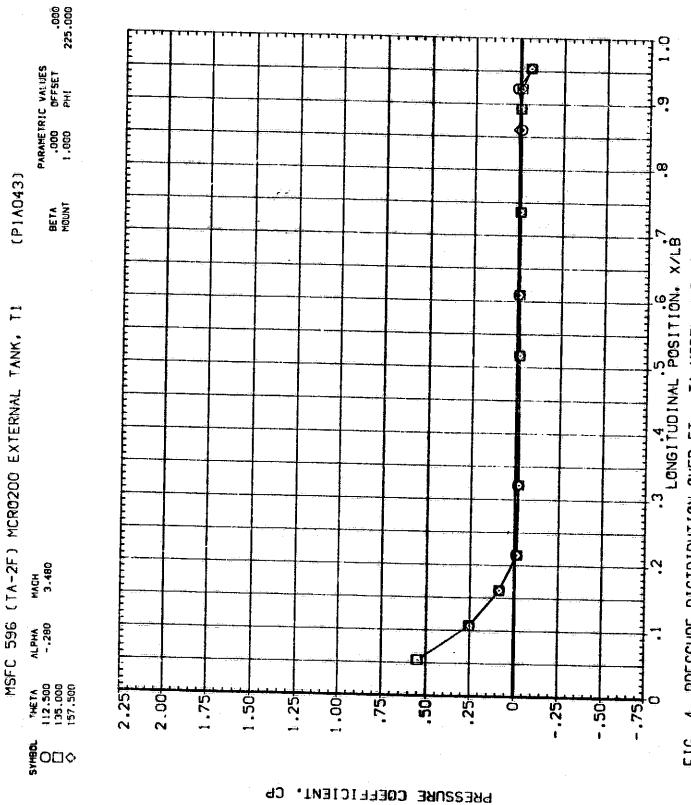


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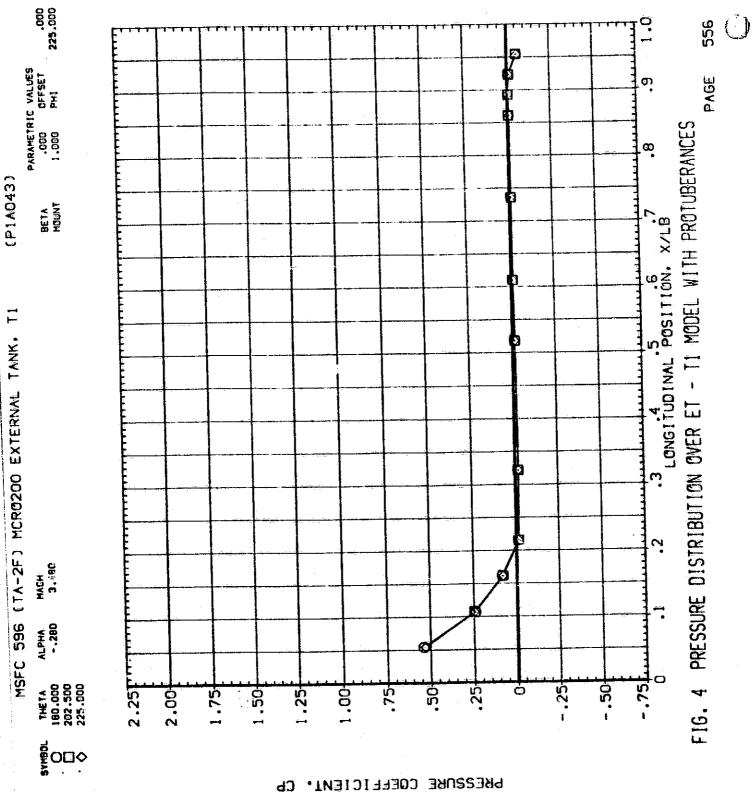
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES





PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES FIG. 4

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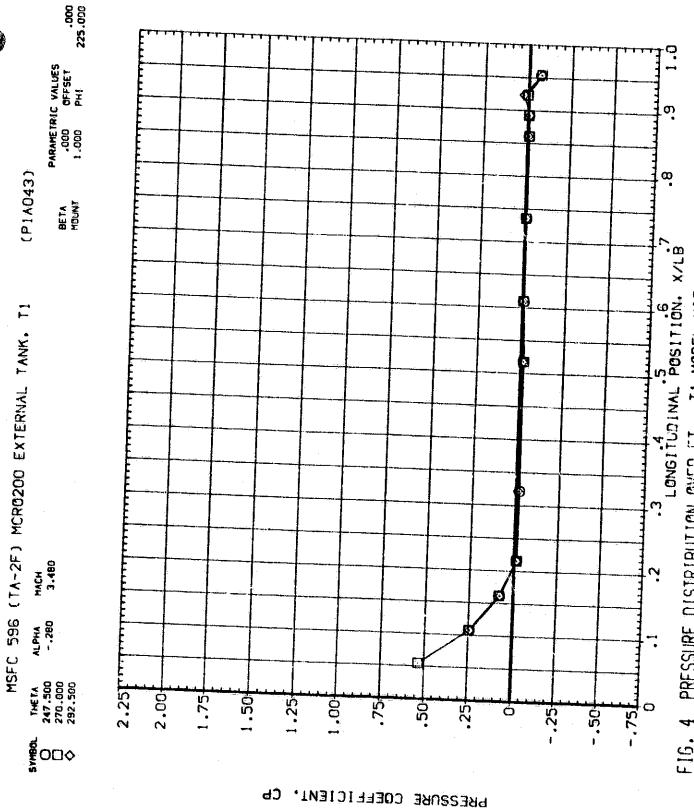
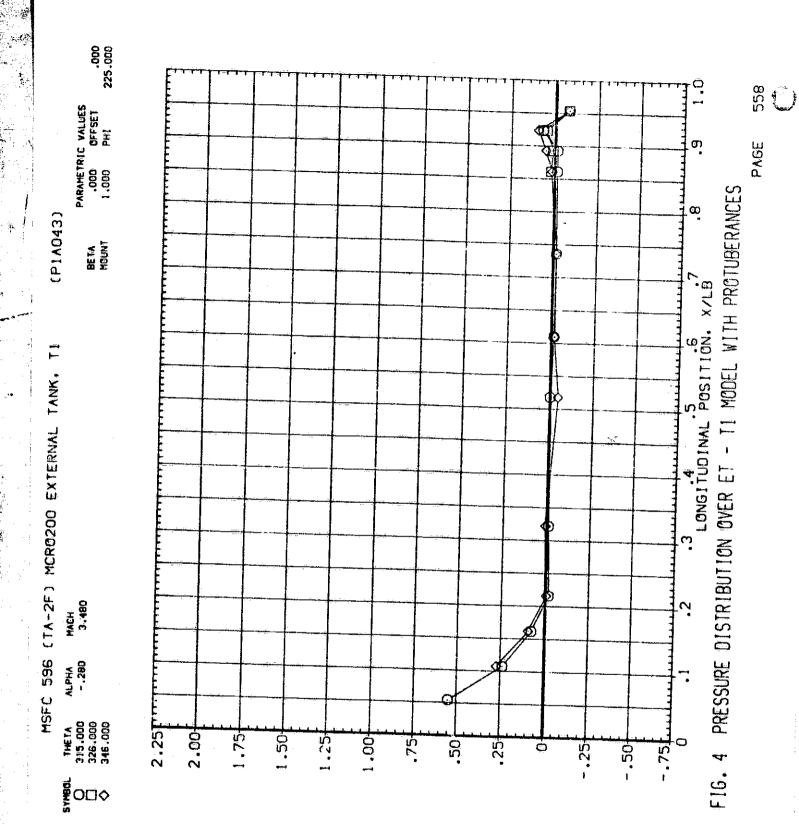


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

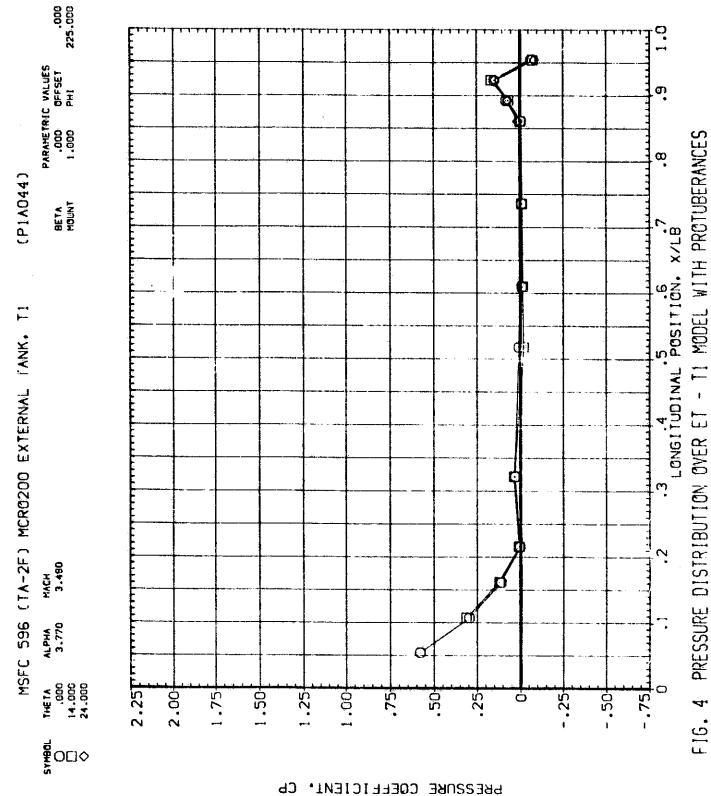
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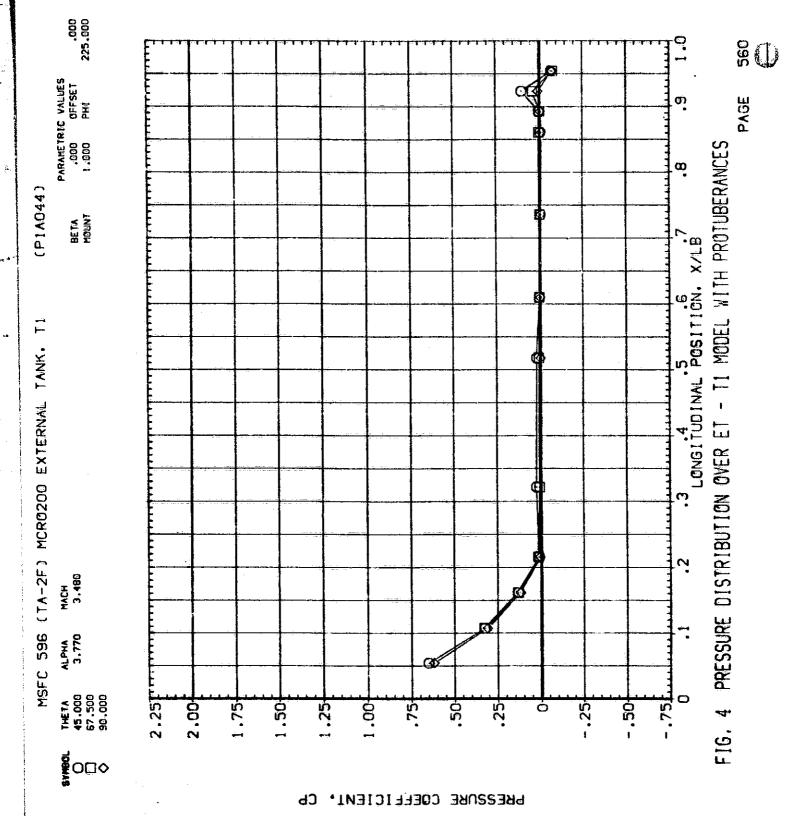
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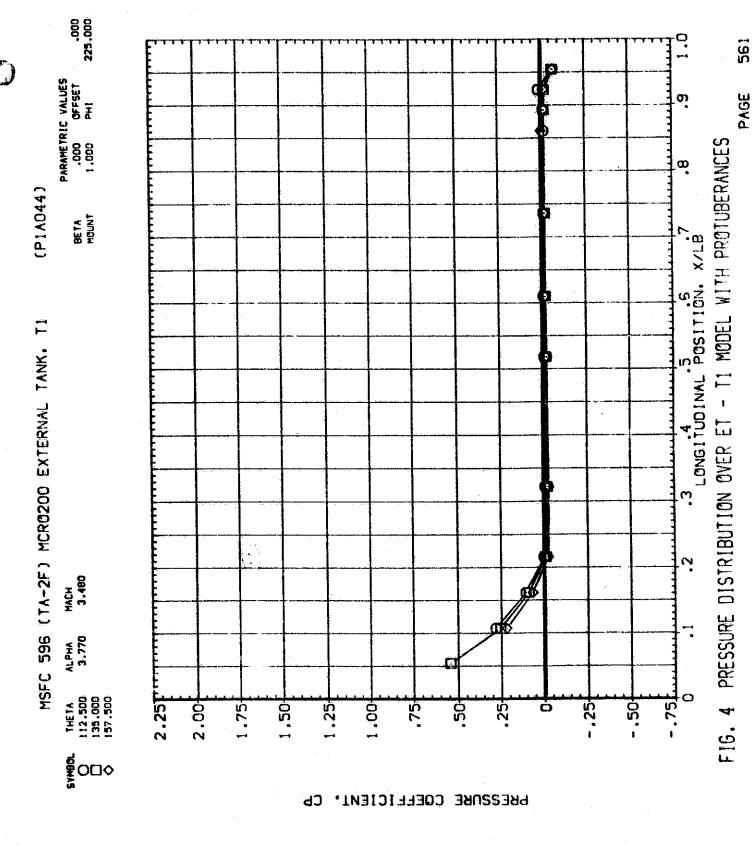
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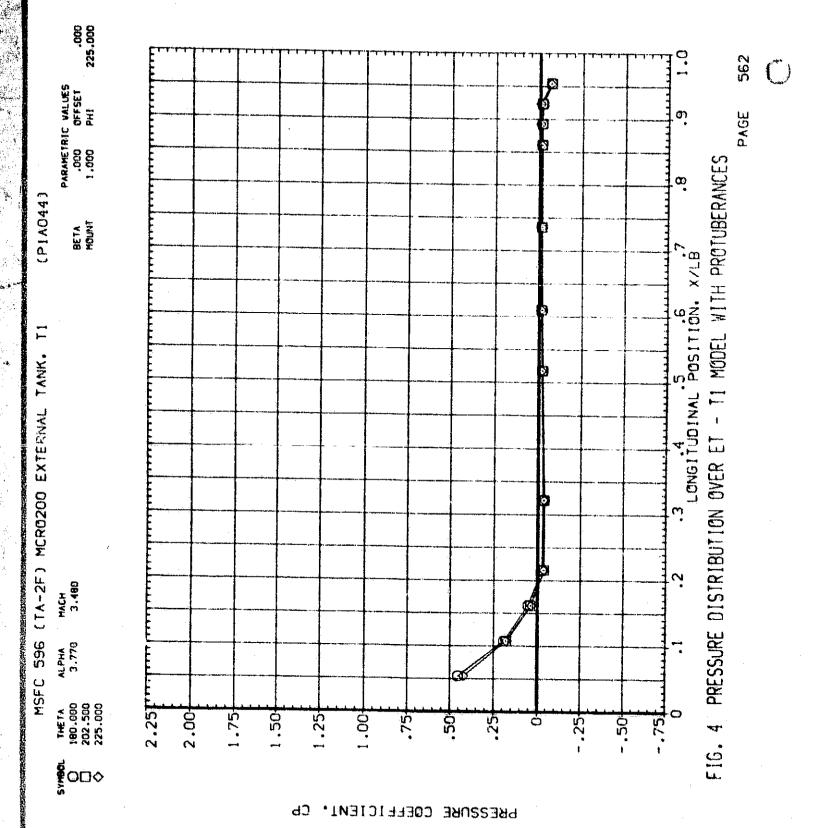


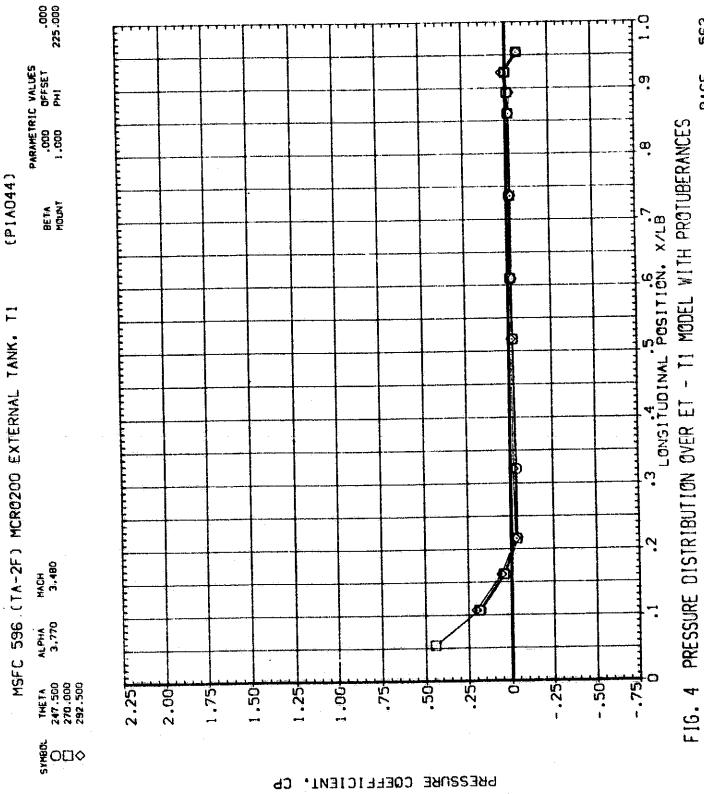






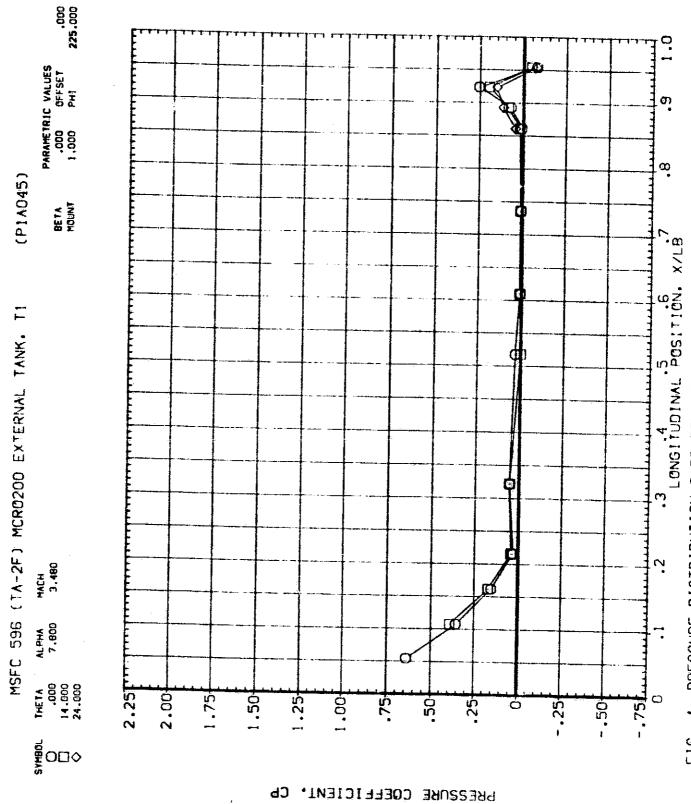




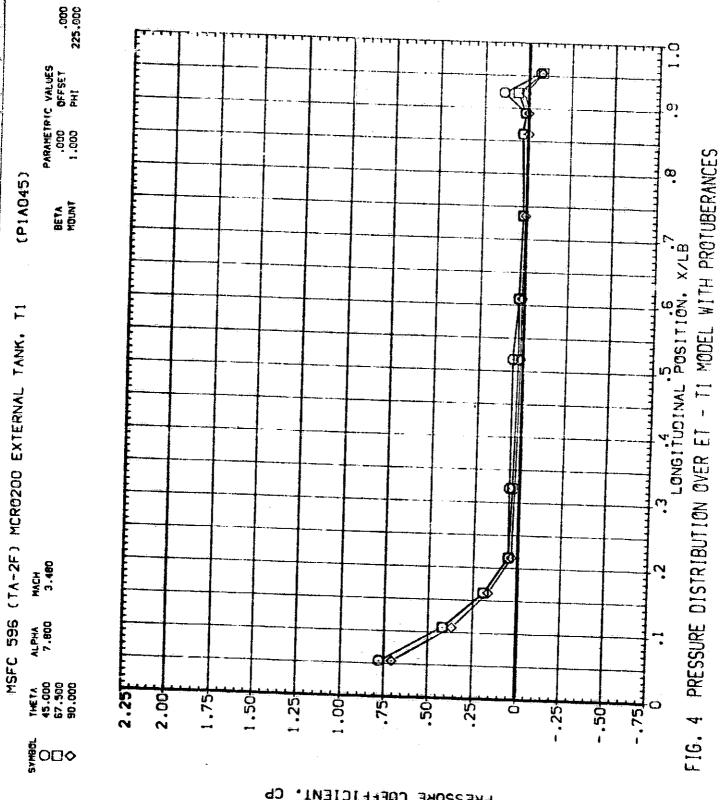


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- TI MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET



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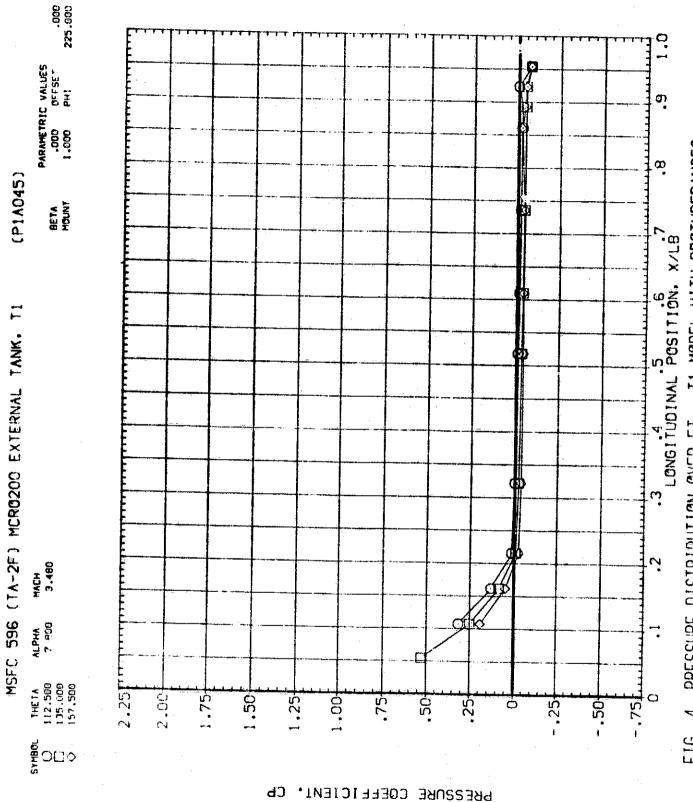
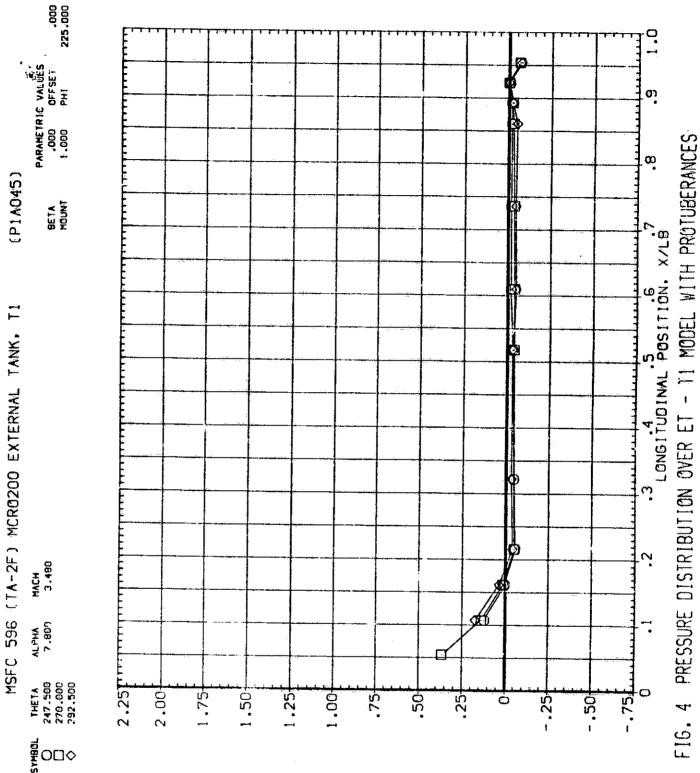


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

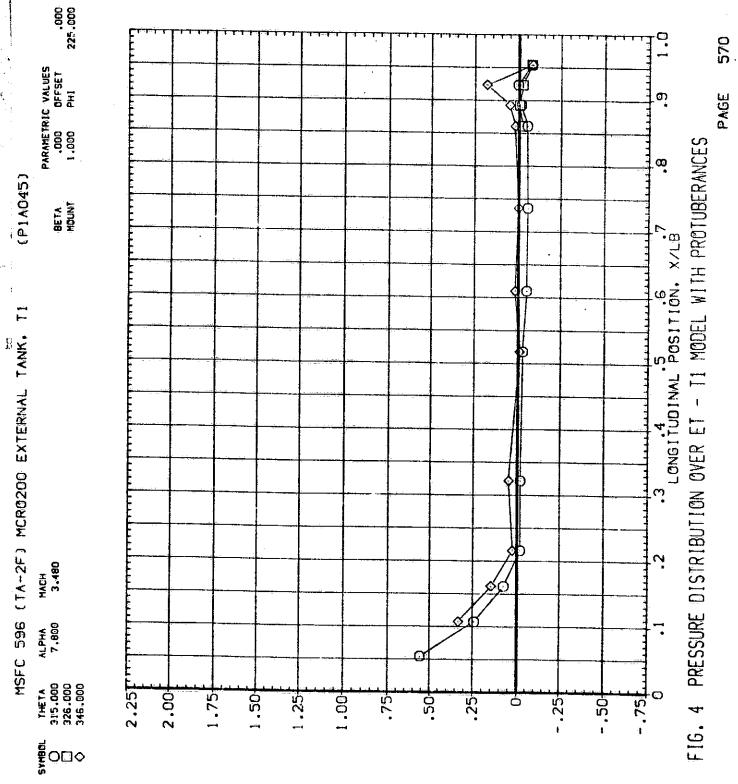
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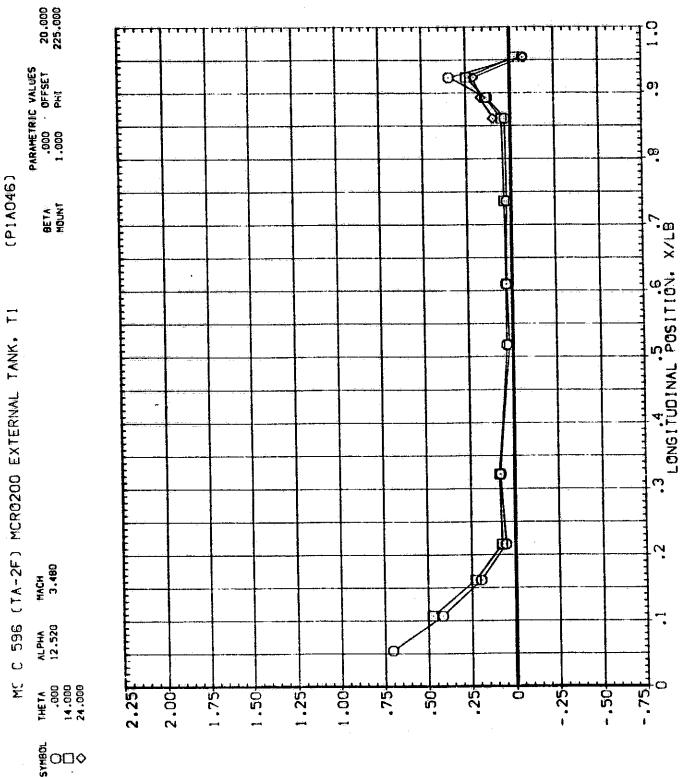
FIG. 4 PRESSURE DISTRIBUTION OVER ET - II MODEL WITH PROTUBERANCES



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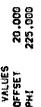
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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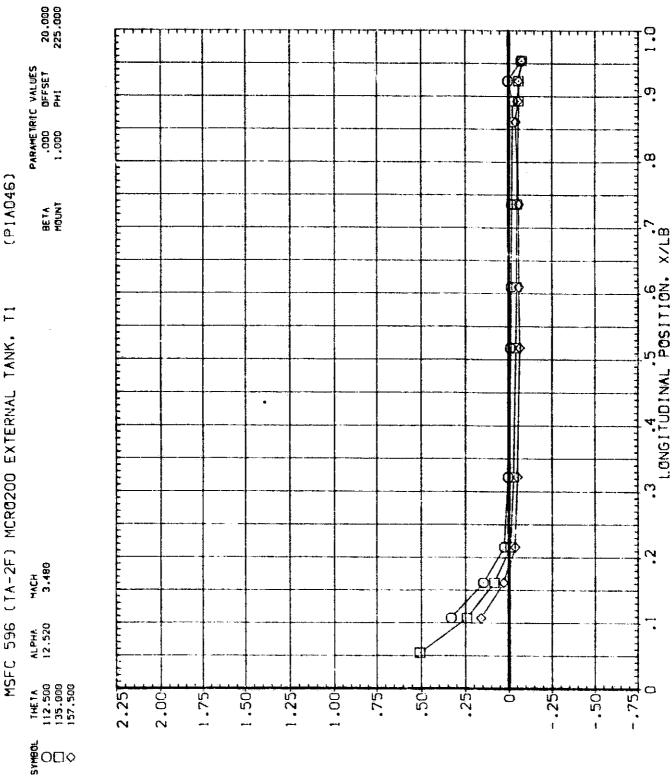
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES LONGITUDINAL POSITION, X/LB

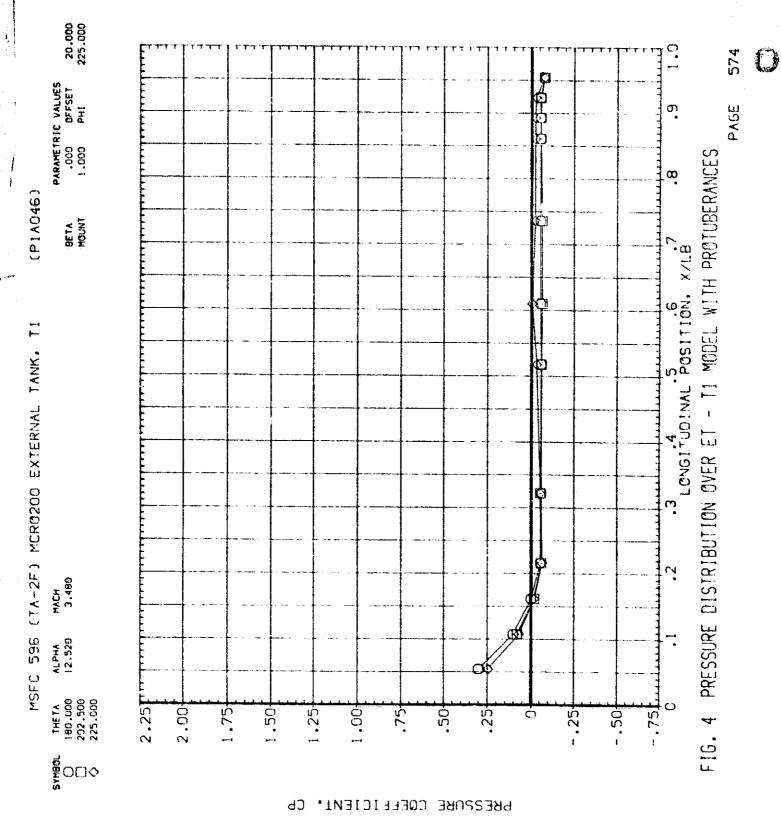
572 PAGE

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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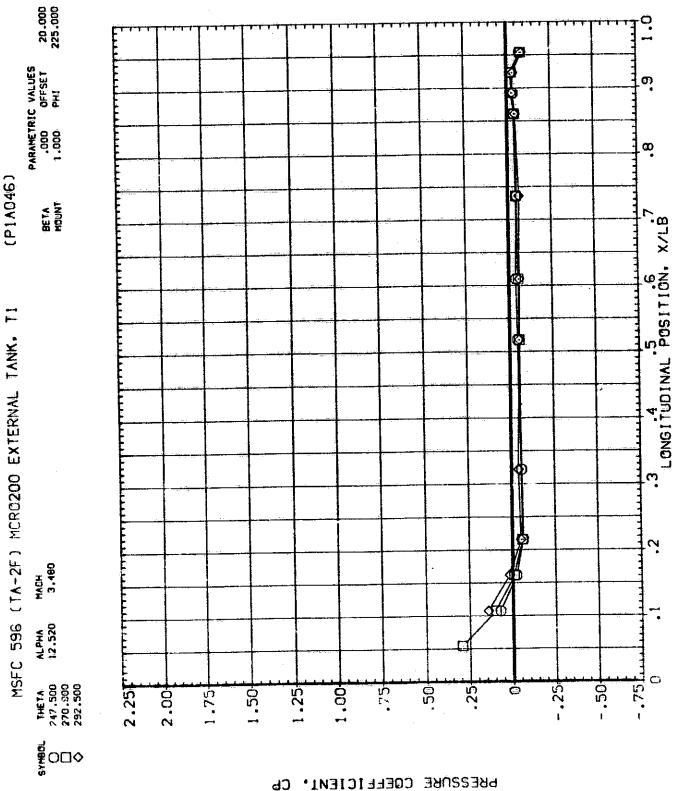
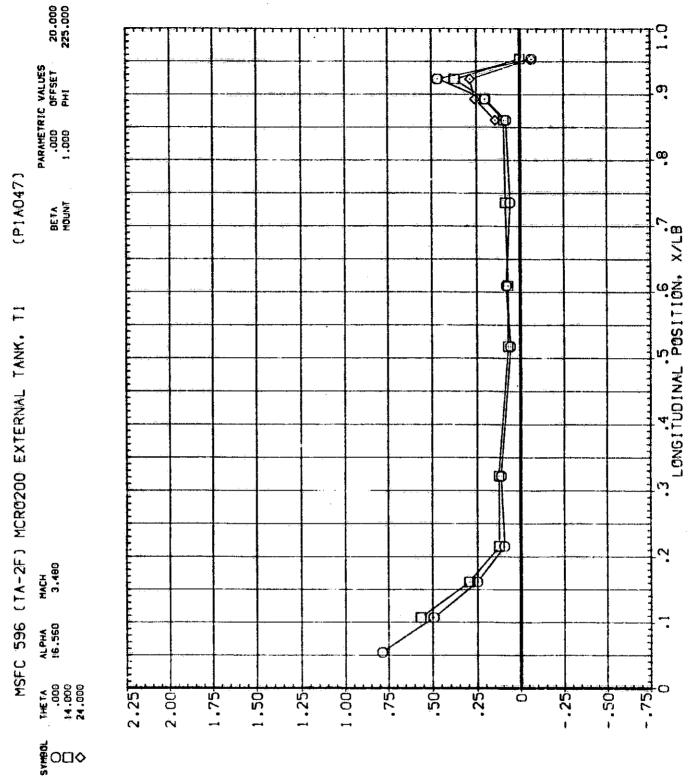


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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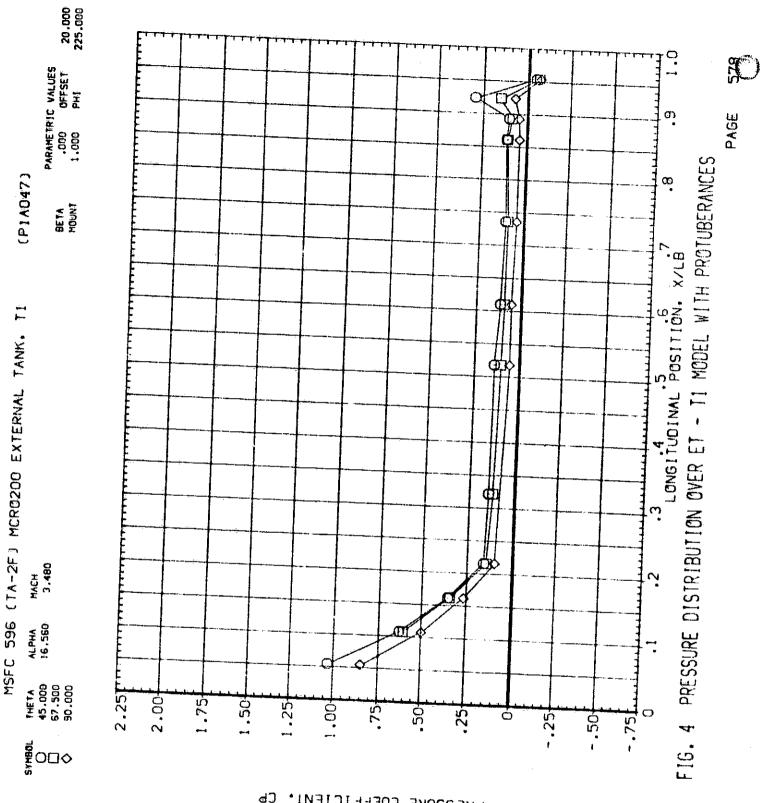
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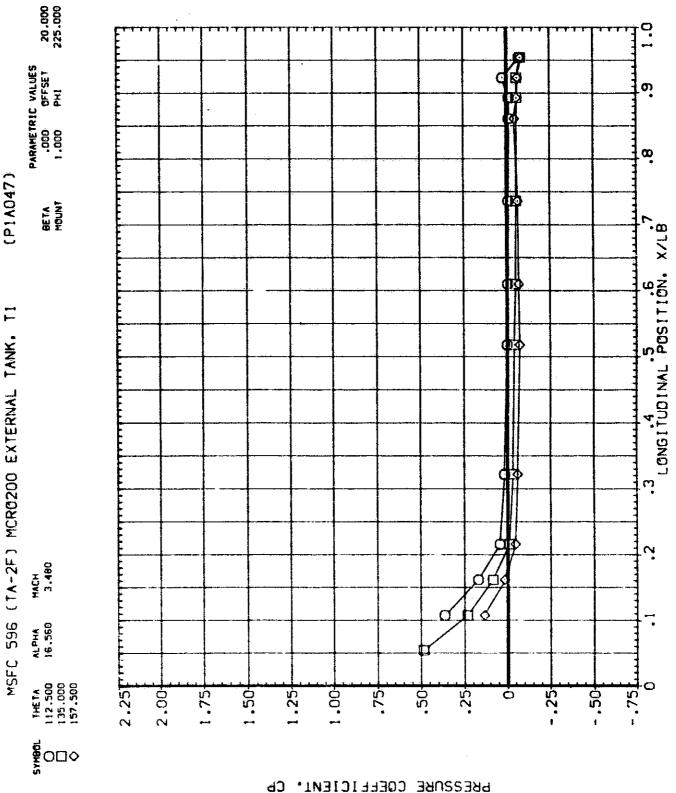
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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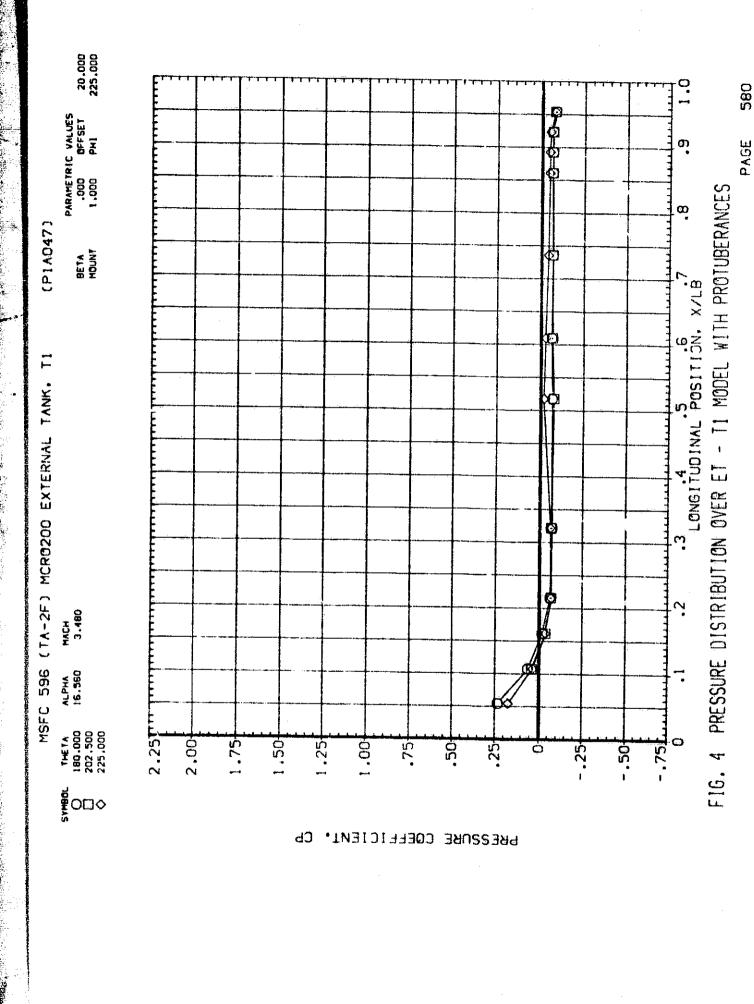


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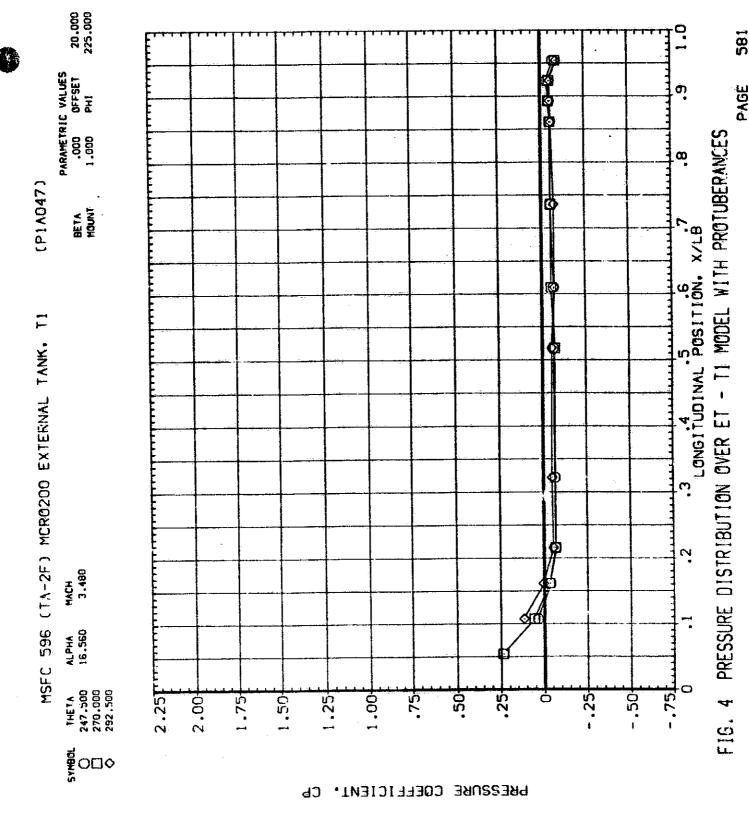
PRESSURE COEFFICIENT.



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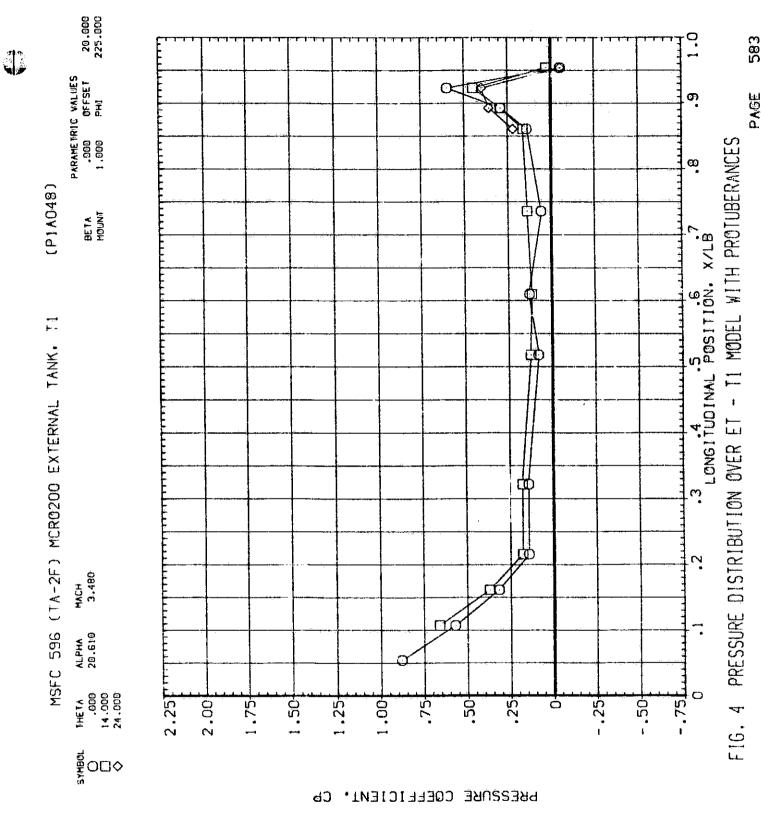
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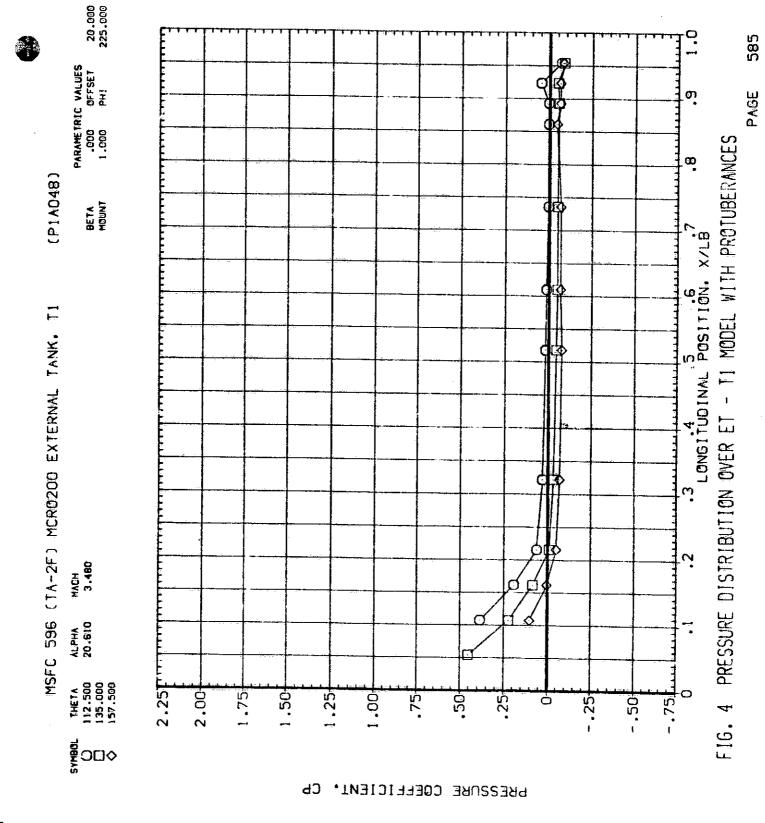
FIG. 4 PRESSURE DISTRIBUTION OVER

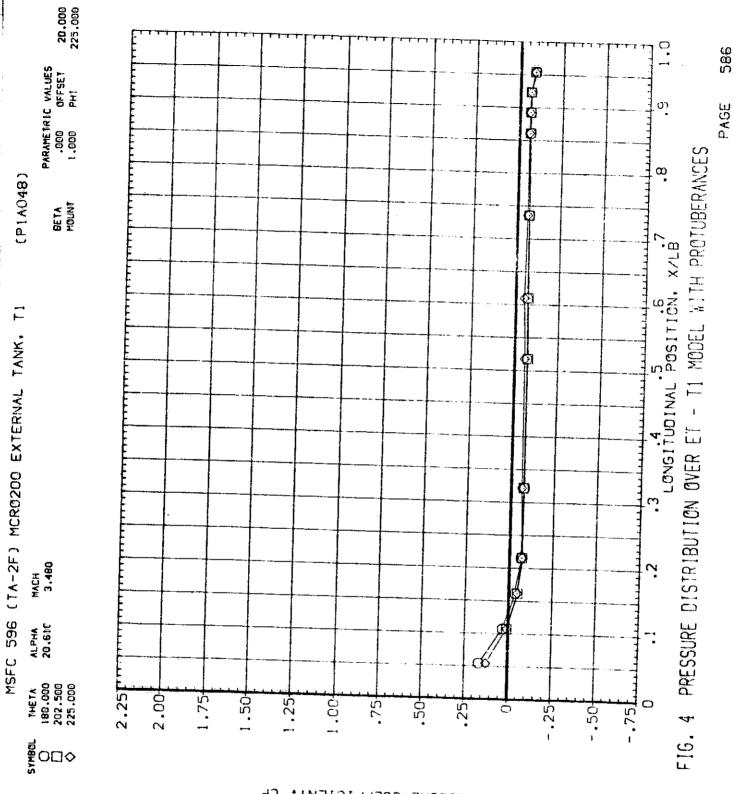
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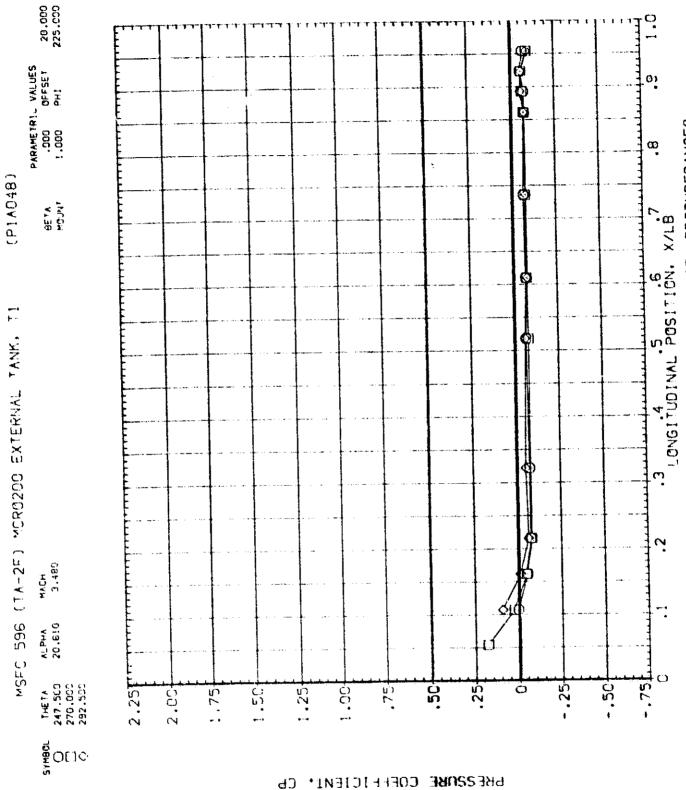


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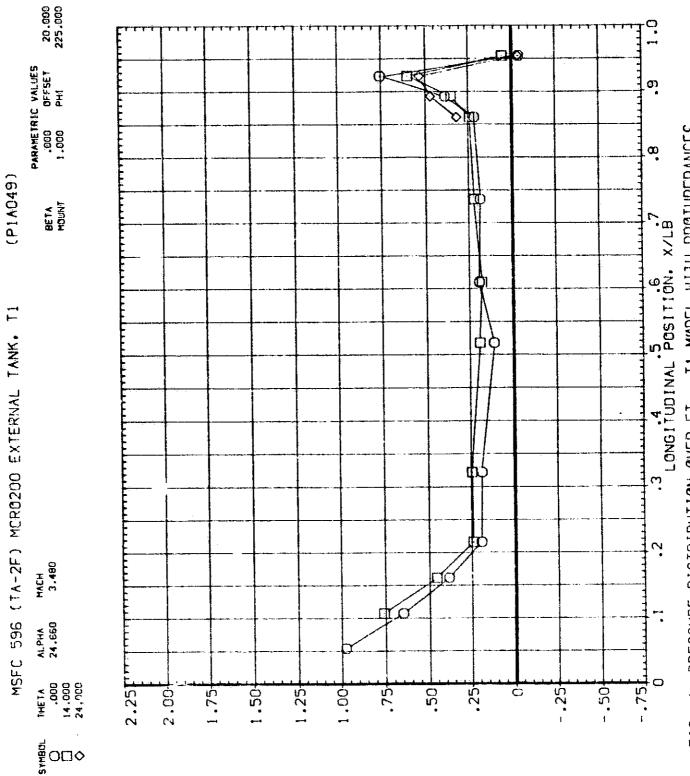
- TI MODEL WITH PROTUBERANCES PRESSURE DISTRIBUTION OVER ET F.G. 4

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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1



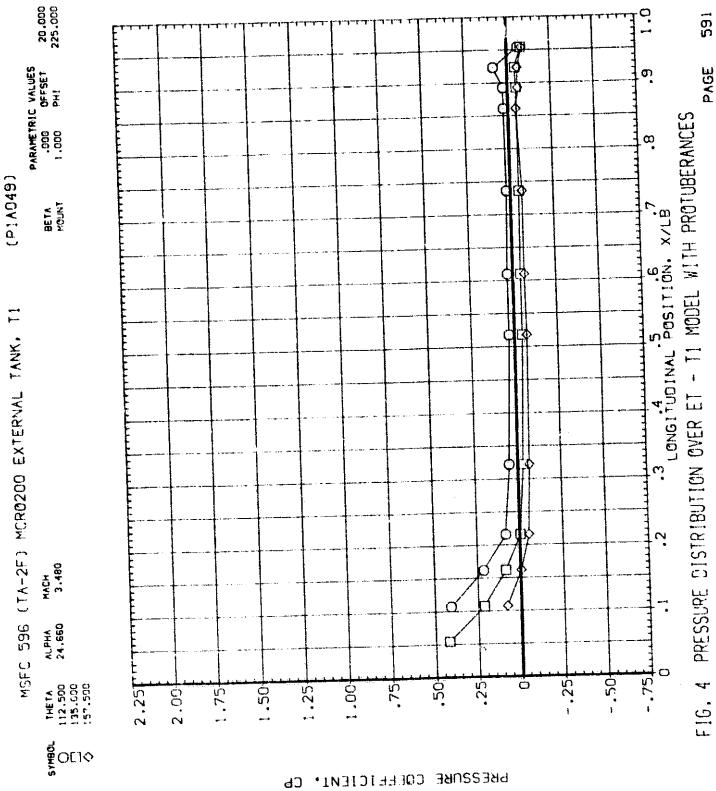
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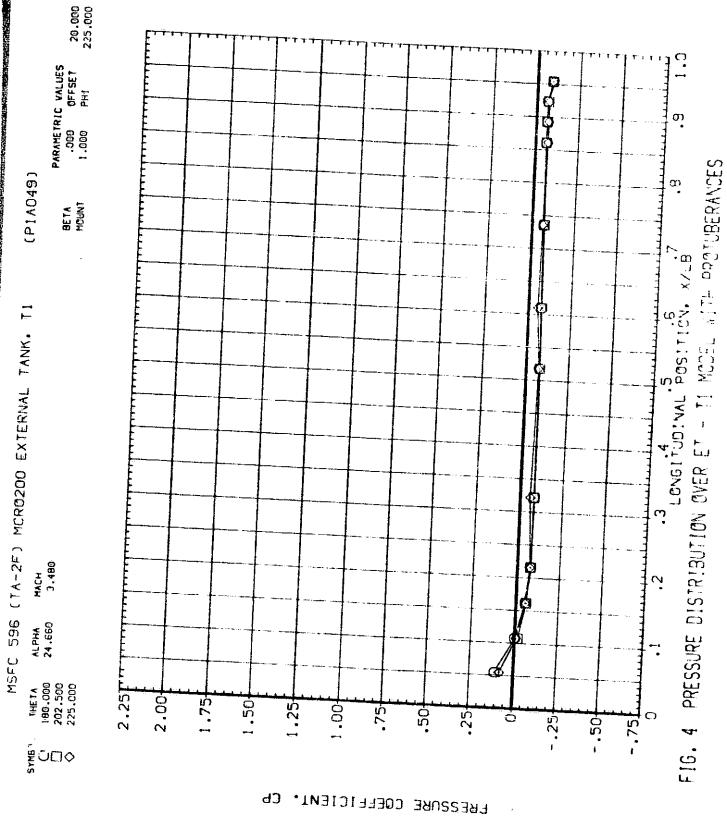
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PAGE - TI MODEL WITH PROTUBERANCES <u>__</u> PRESSURE DISTRIBUTION OVER F16.4

- 11 MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET

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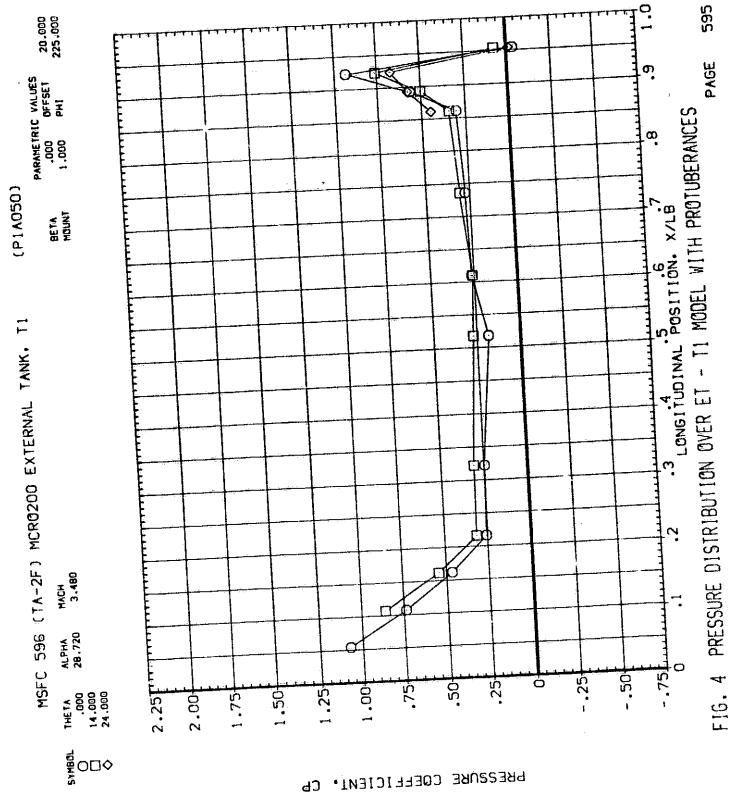


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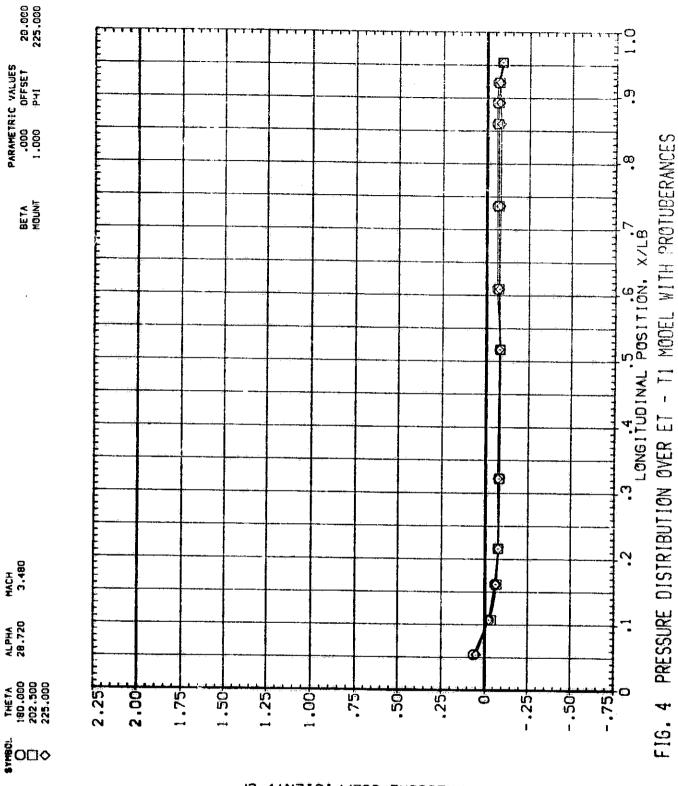
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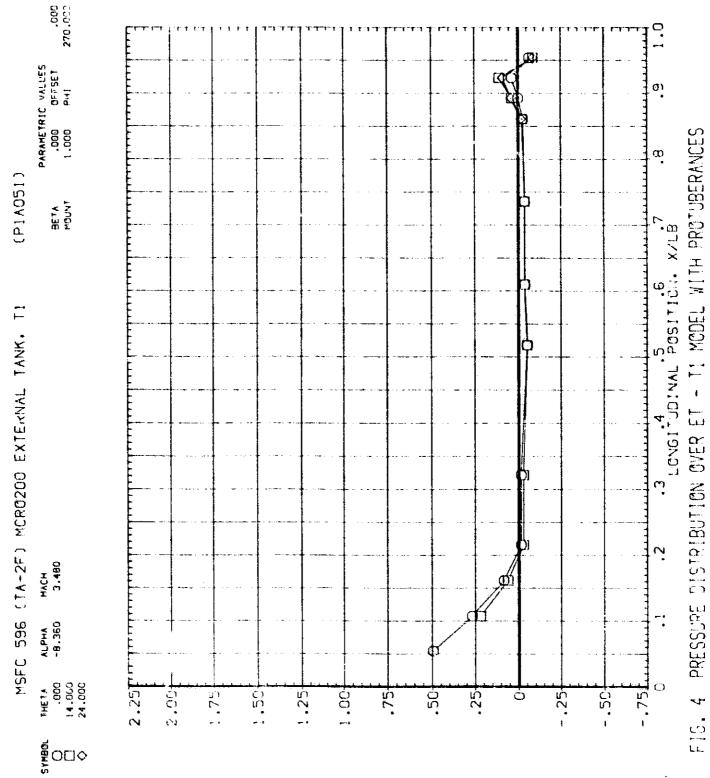
FIG. 4 PRESSURE DISTRIBUTION OVER

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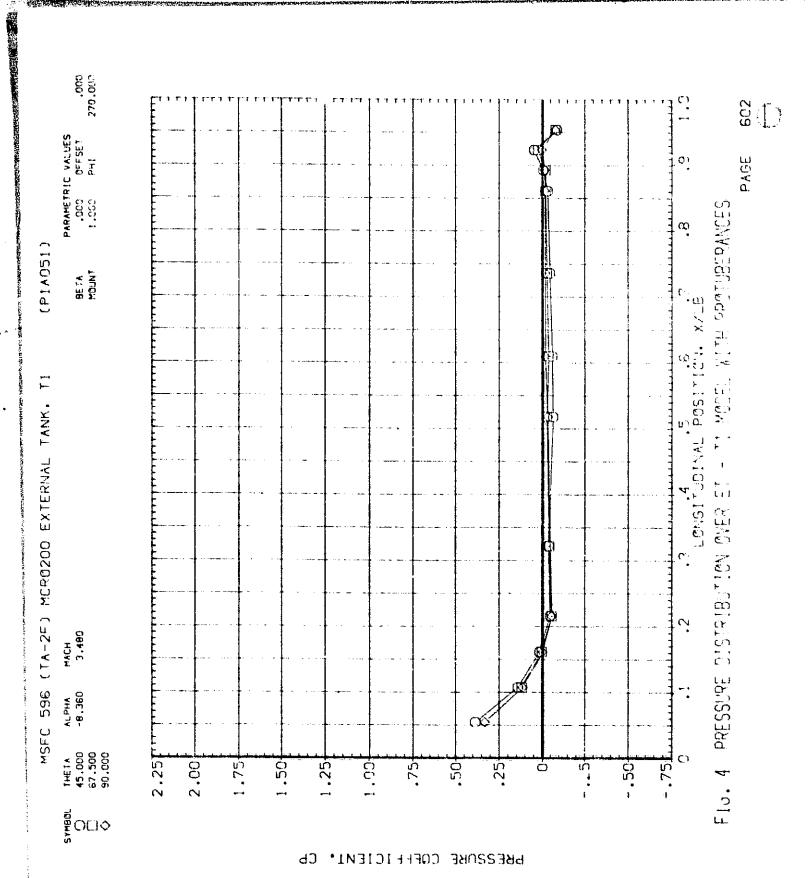


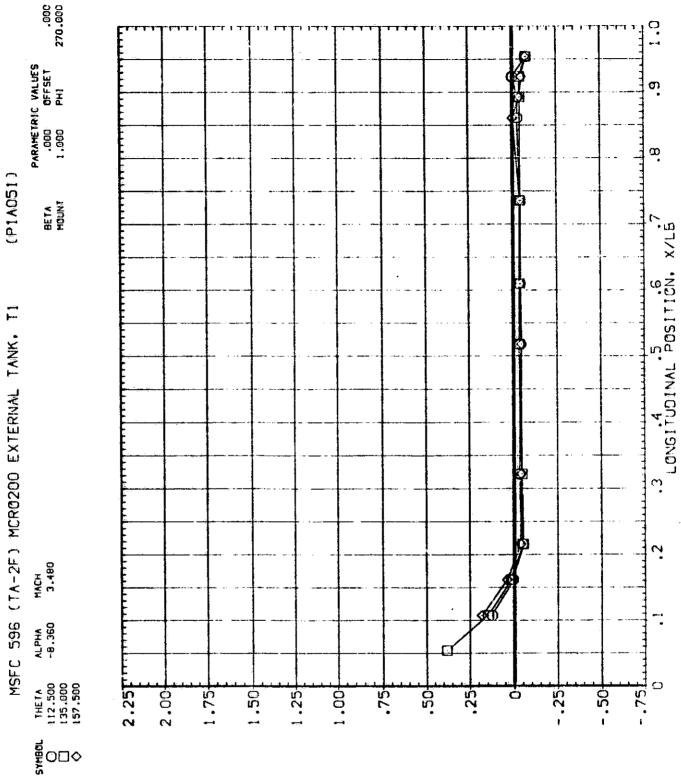
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PRESSURE COEFFICIENT, CP

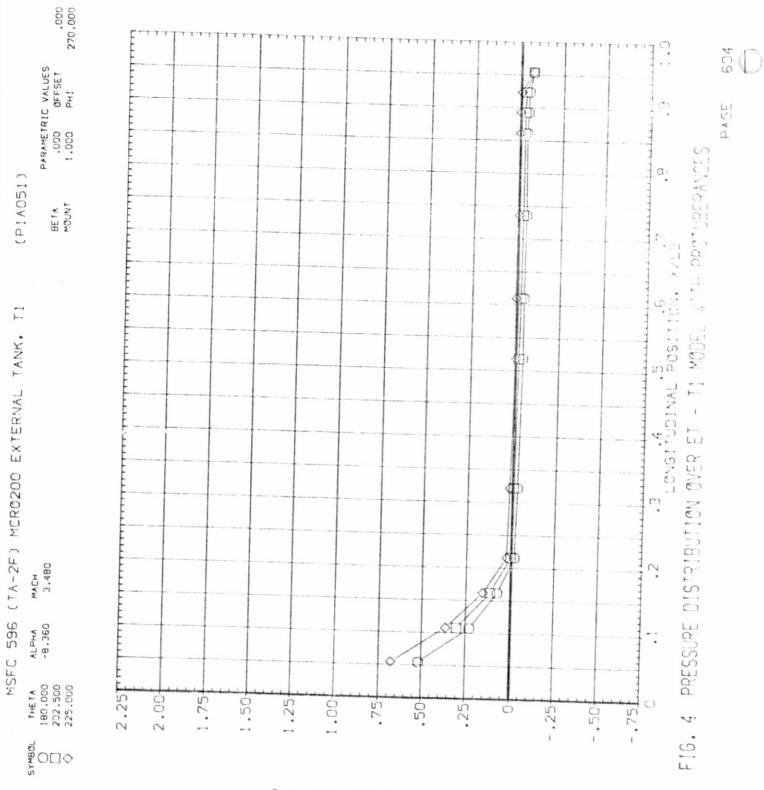






PRESSURE COEFFICIENT,

II MODEL WITH PROTUBERANCES FIG. 4 PRESSURE DISTRIBUTION OVER ET -



PRESSURE COEFFICIENT, CP

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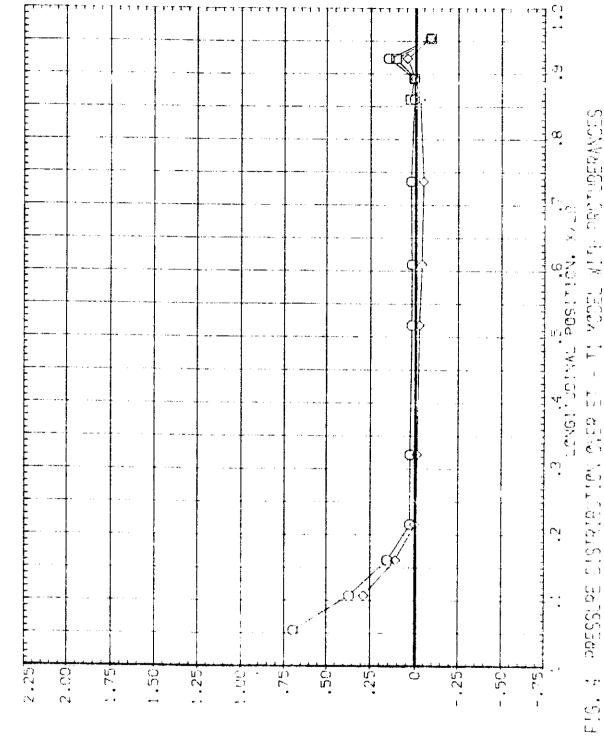
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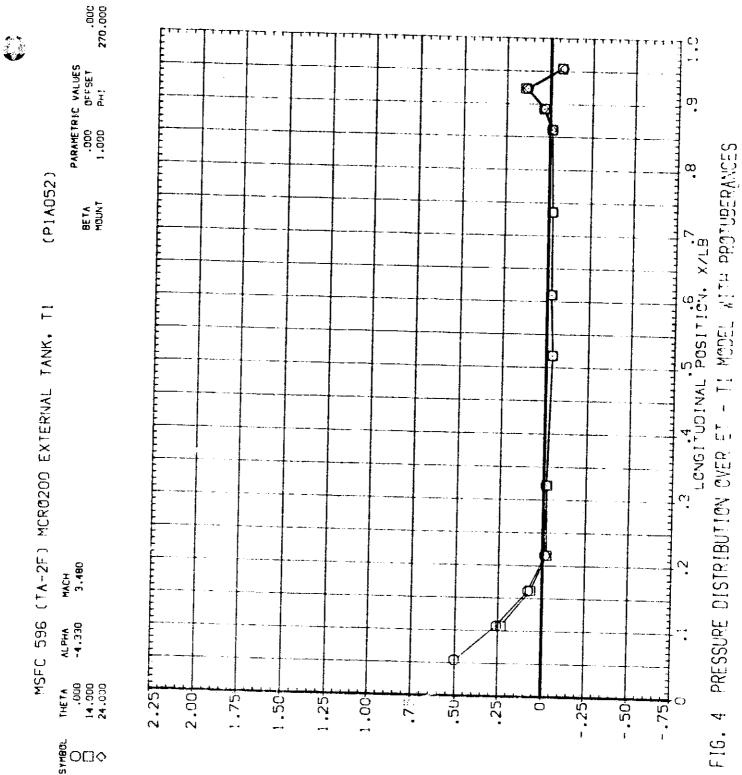
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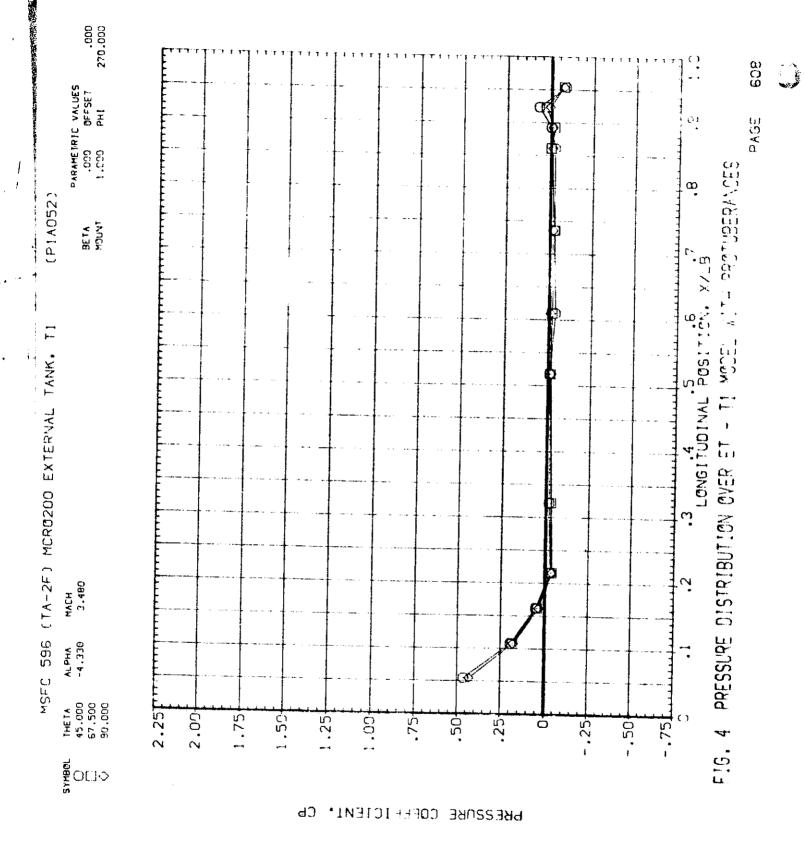
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- TI MODEL WITH PROTUBERANCES

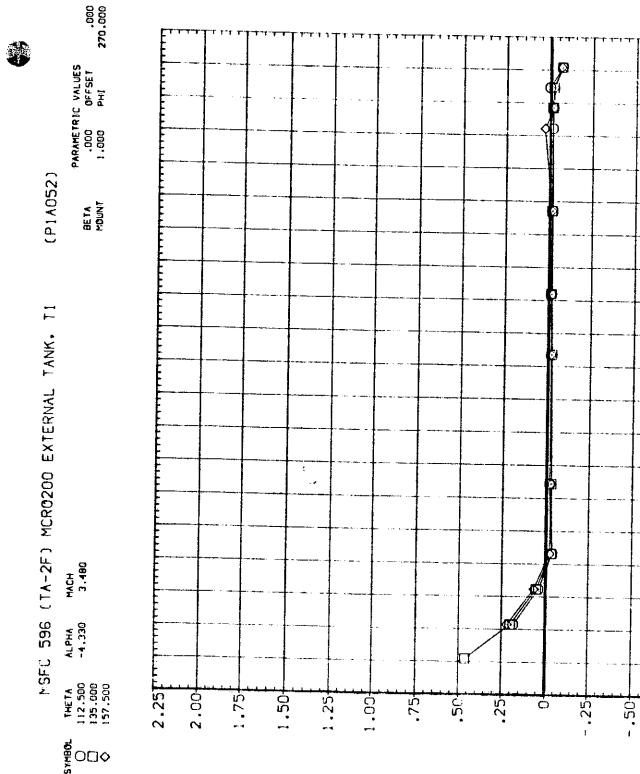
FIG. 4 PRESSURE DISTRIBUTION OVER ET

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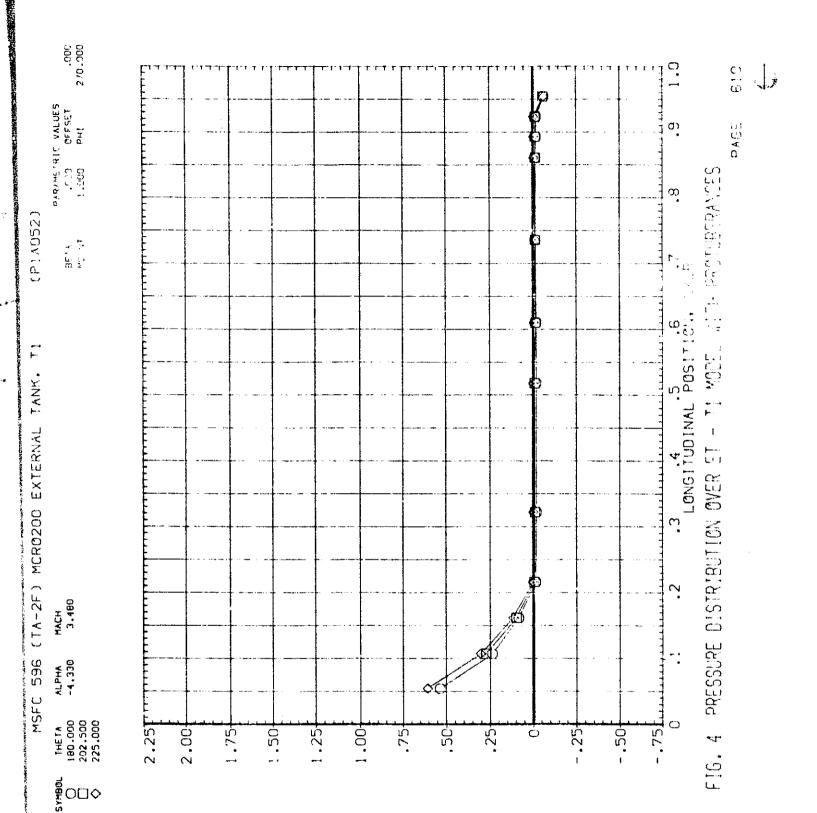
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LONGITUDINAL POSITION. X/LB

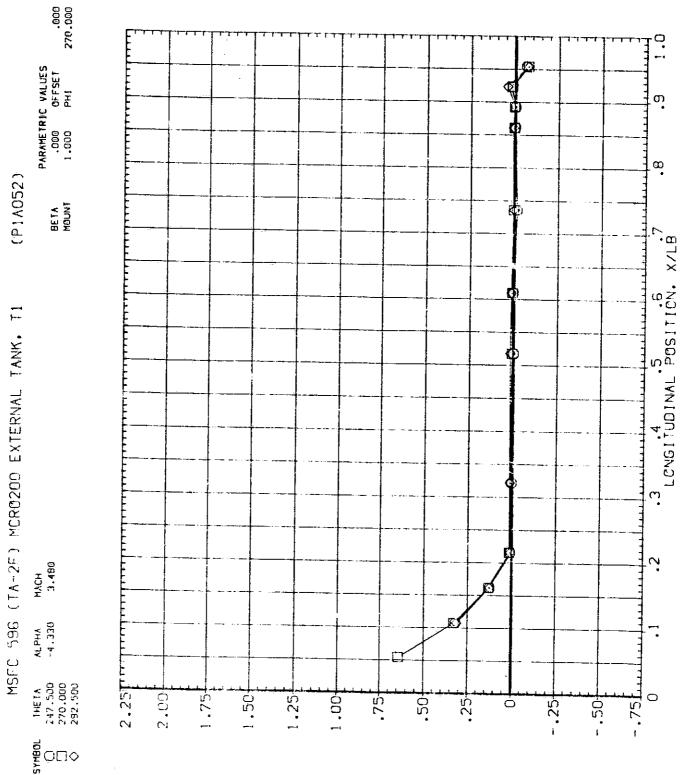


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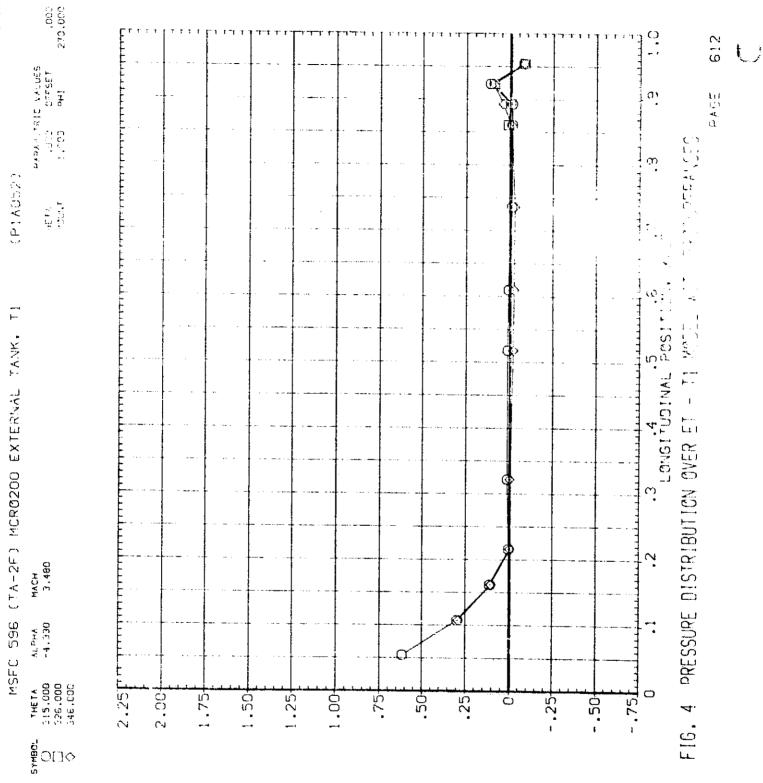


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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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PRESSURE COEFFICIENT, CP

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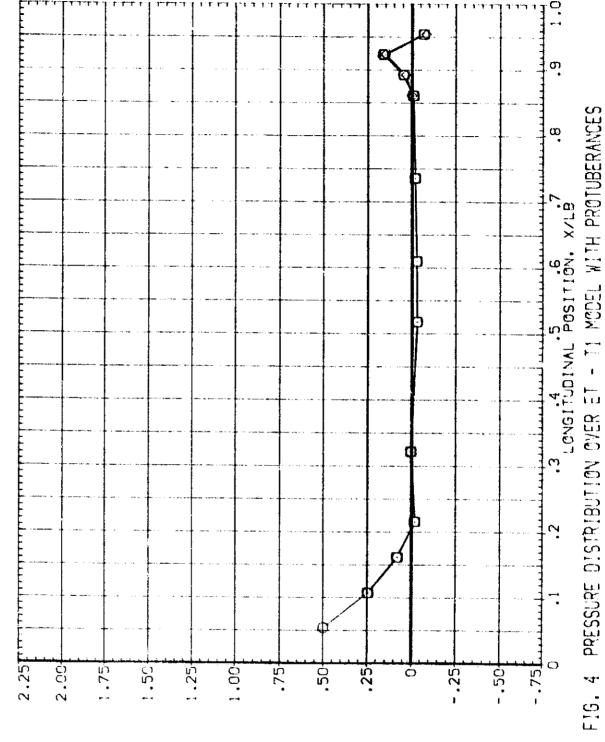
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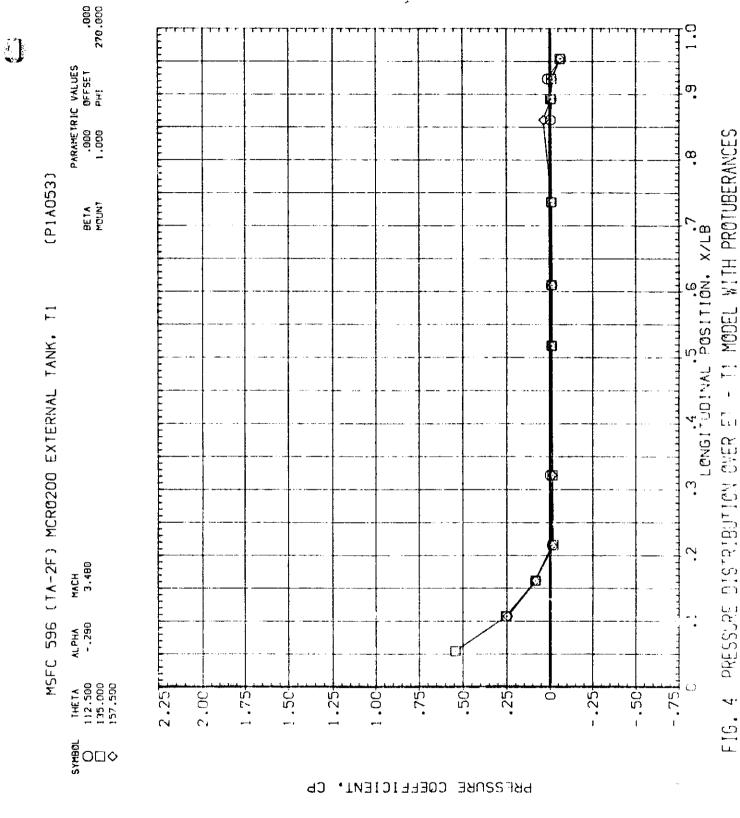
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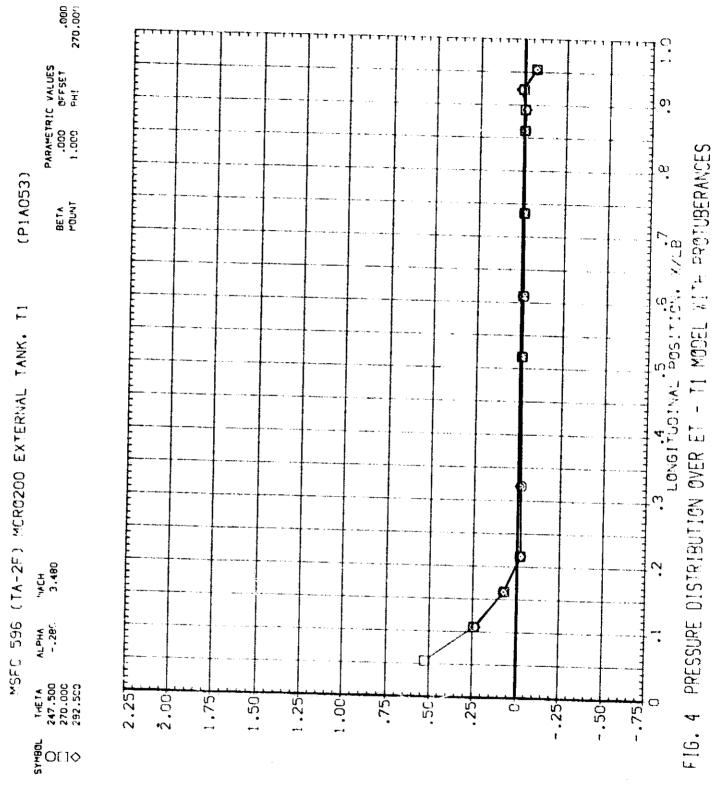
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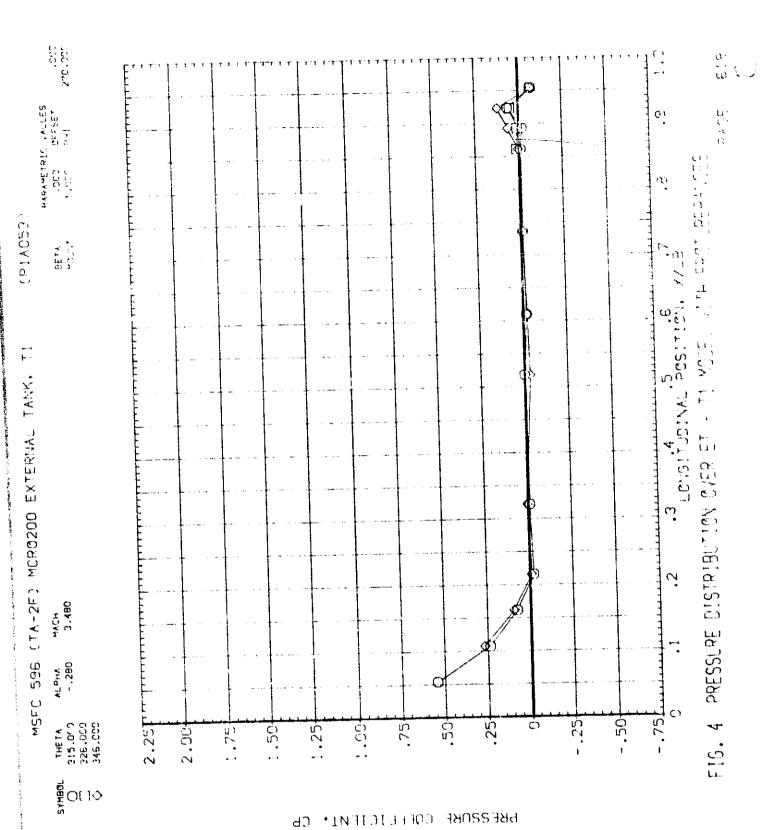
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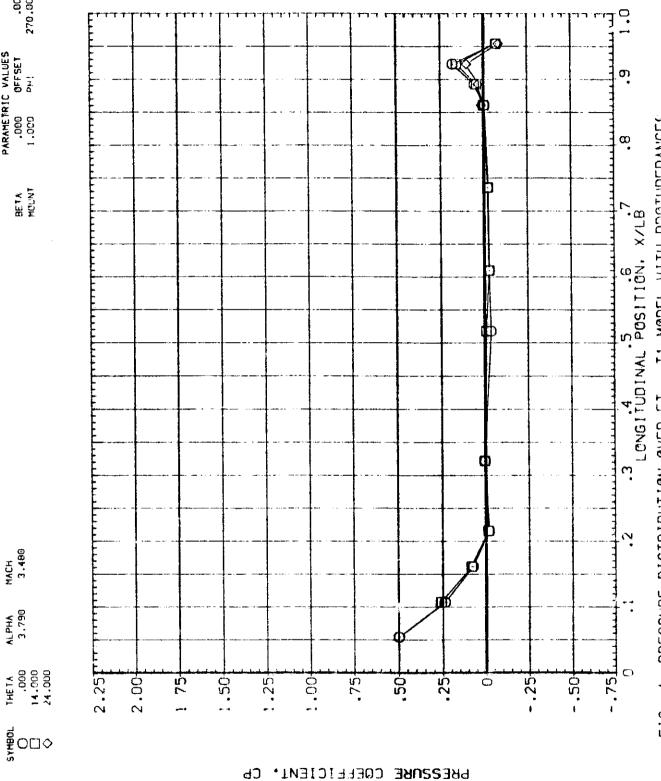
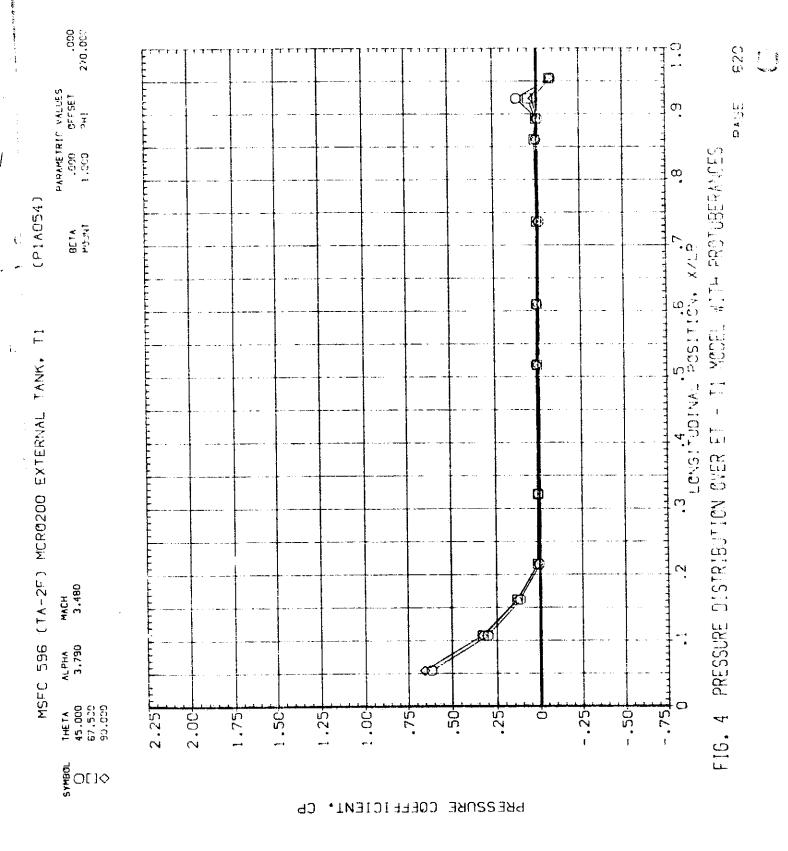
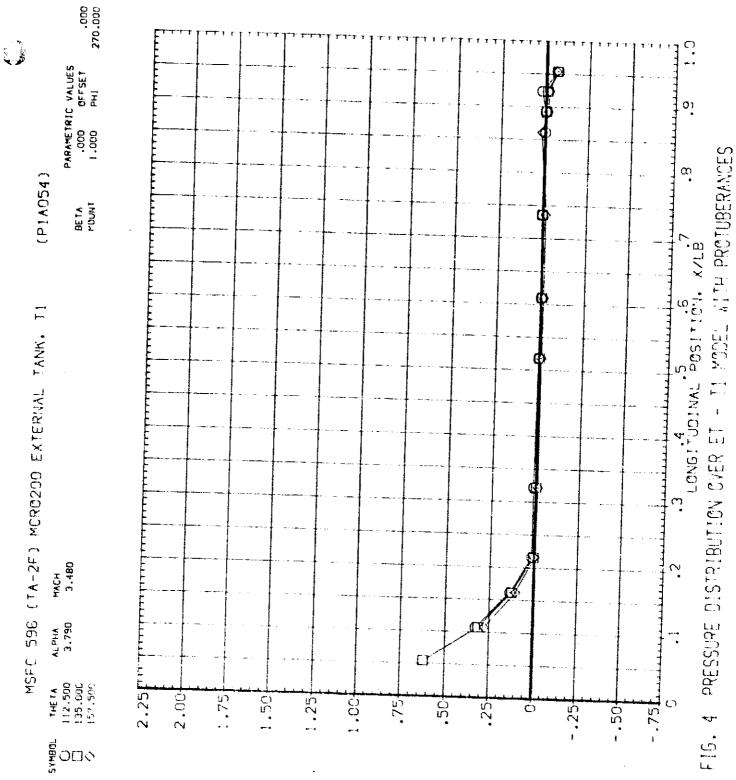


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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PRESSURE COEFFICIENT, CP

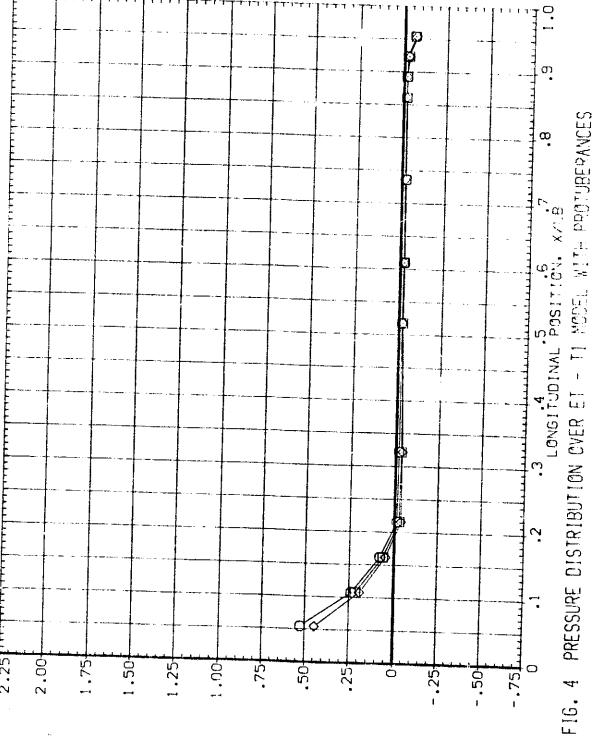
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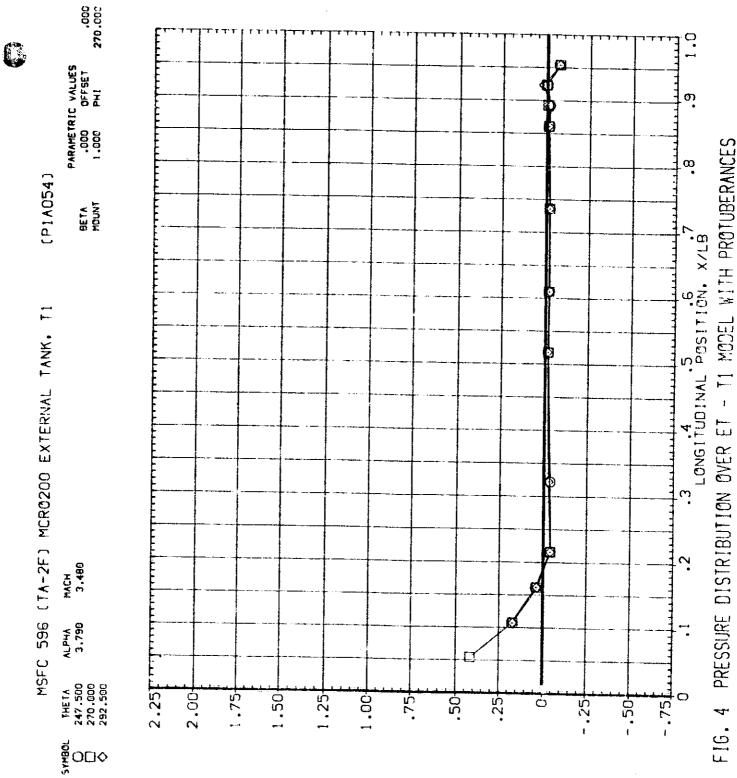
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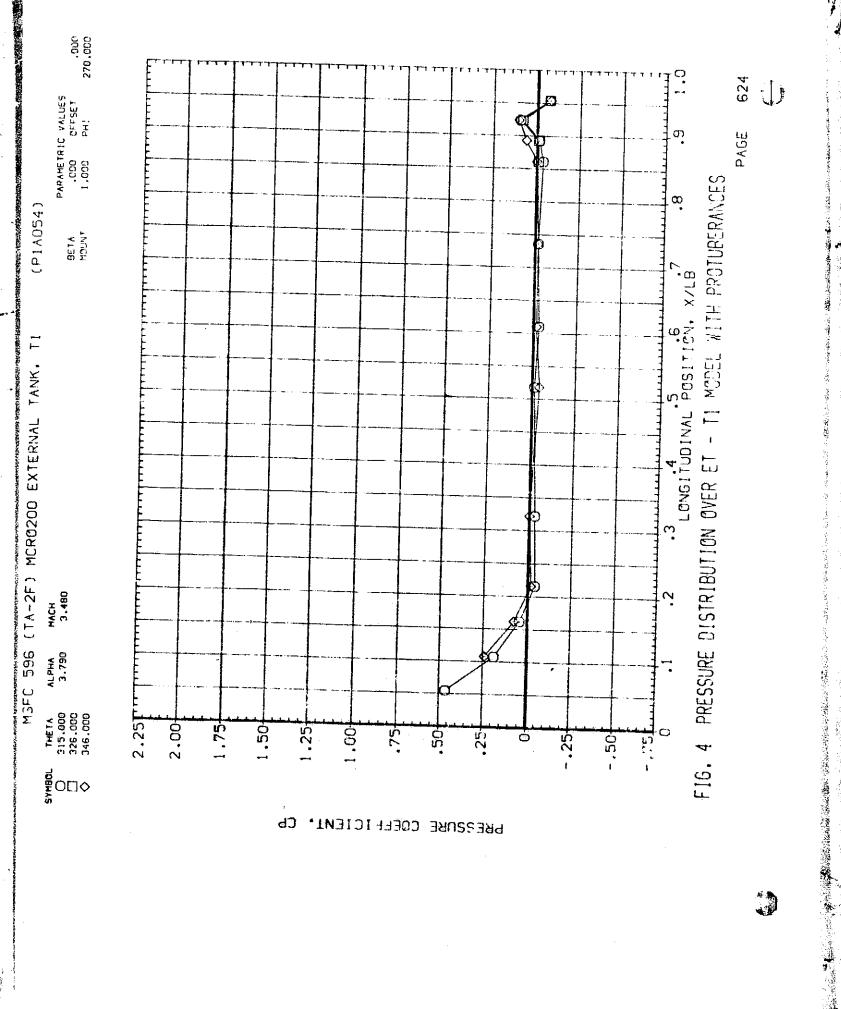
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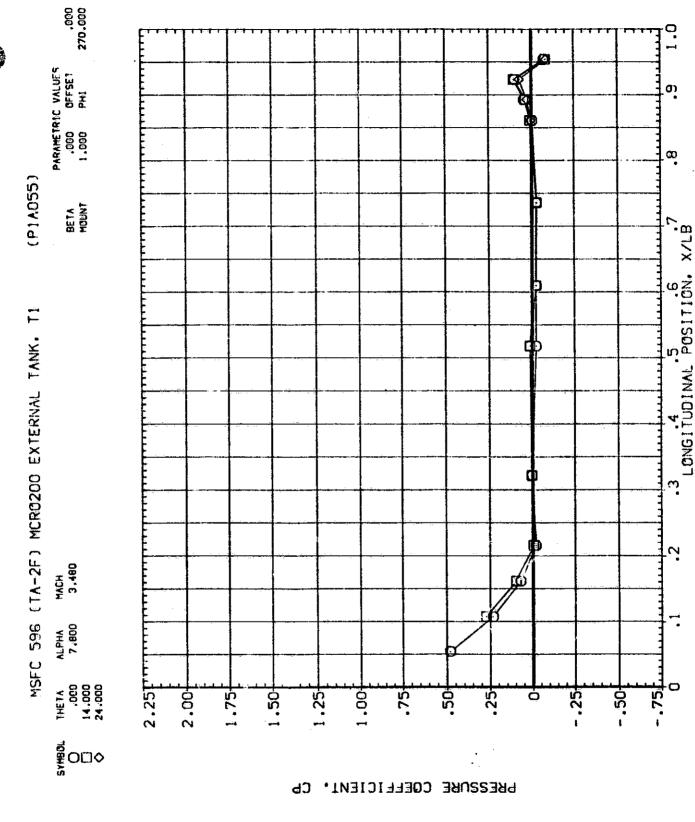


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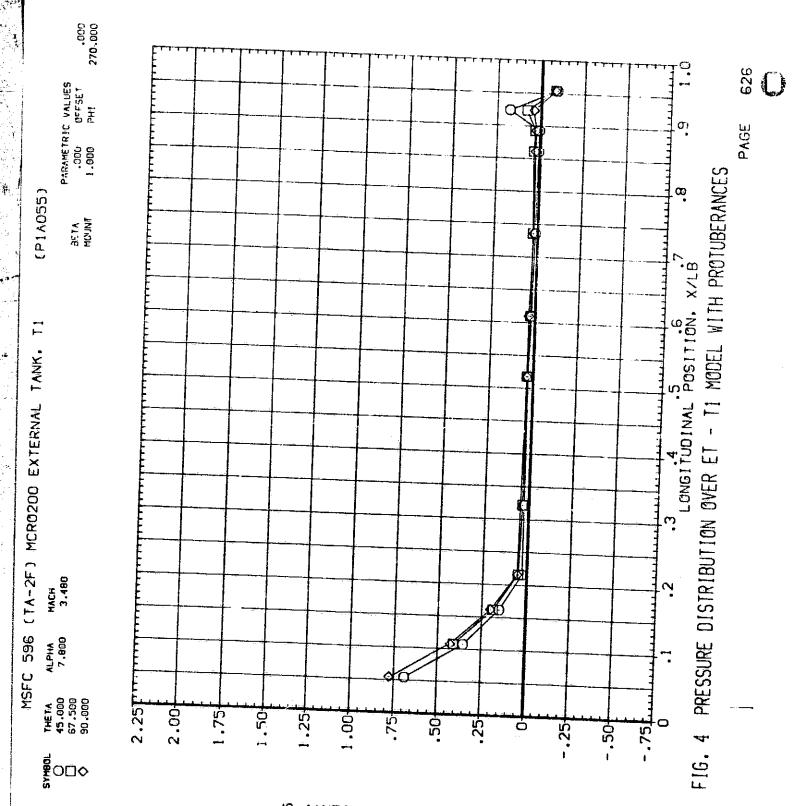
PRESSURE COEFFICIENT,



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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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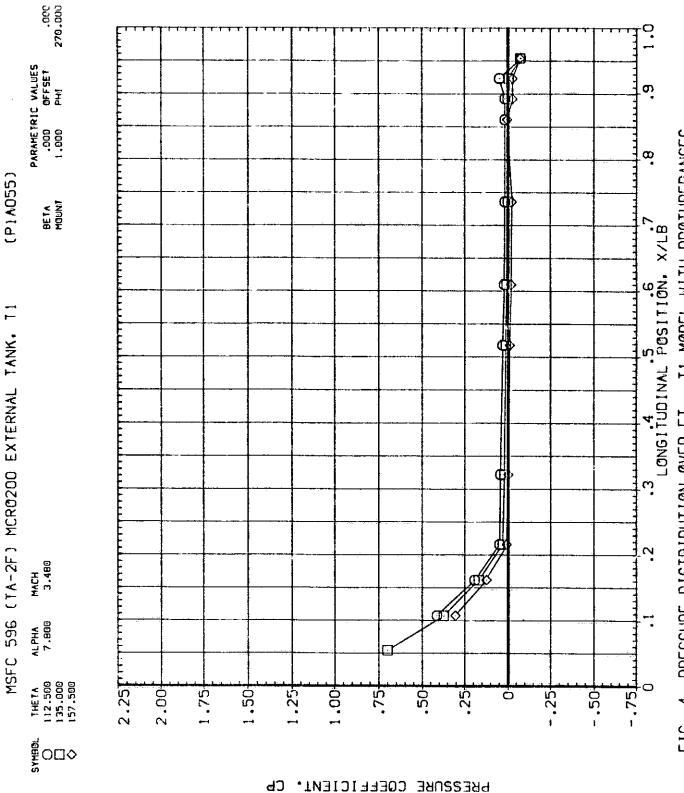
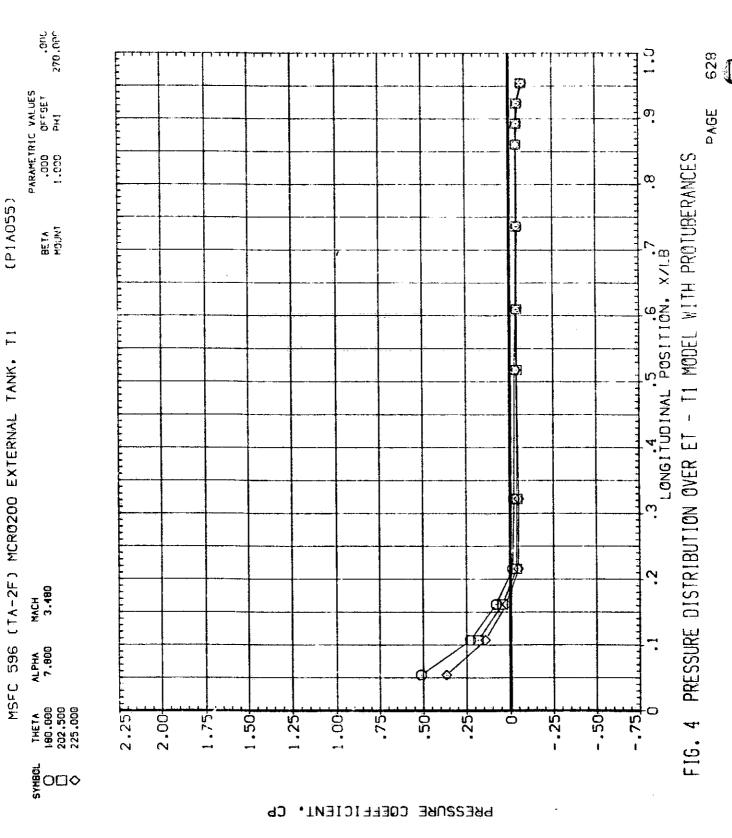
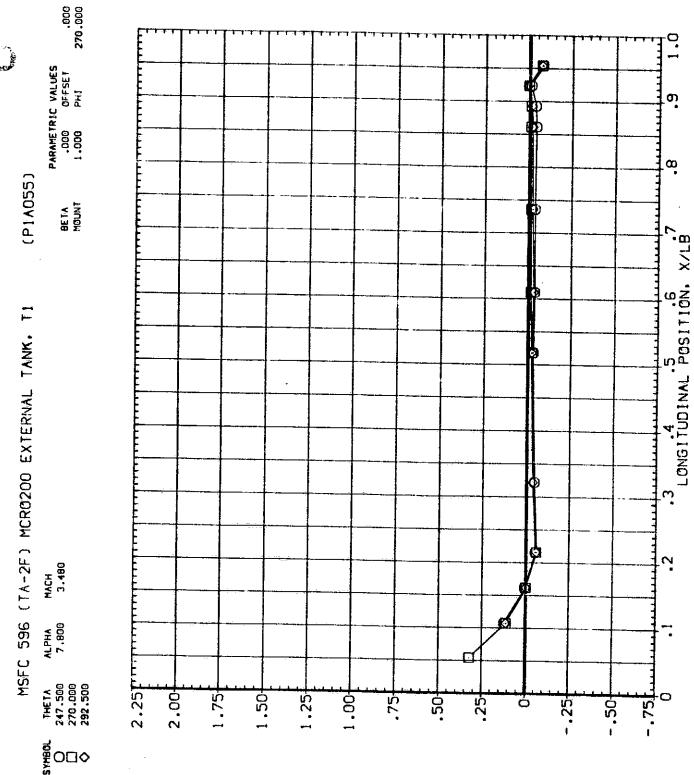


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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PRESSURE COEFFICIENT.

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

PRESSURE COEFFICIENT, CP

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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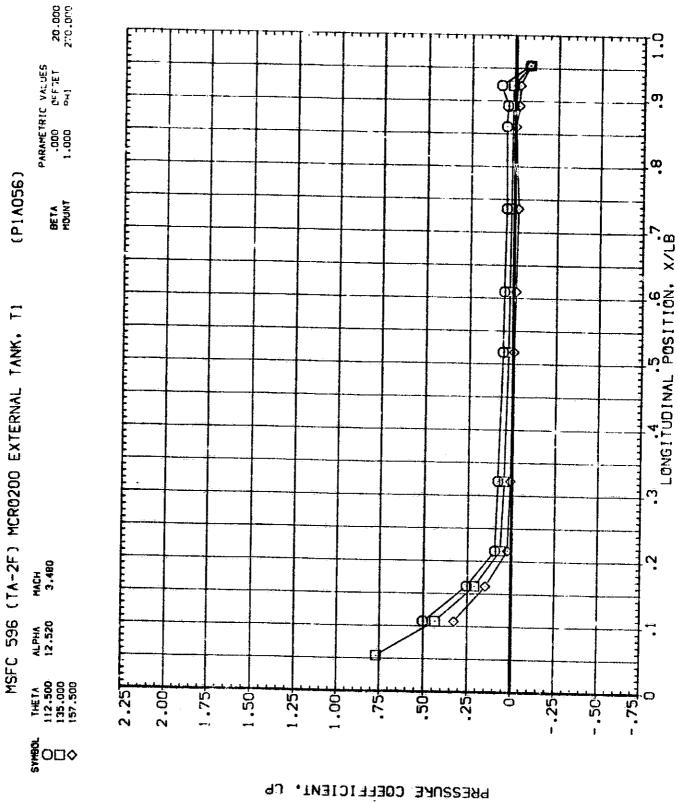
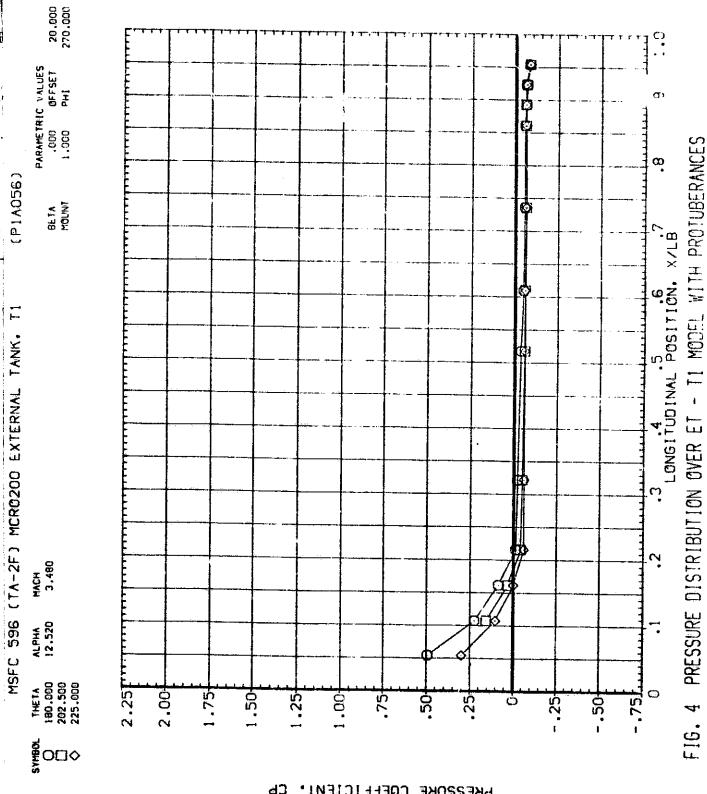


FIG. 4 PRESSURE DISTRIBUTION OVER ET - 11 MODEL WITH PROTUBERANCES

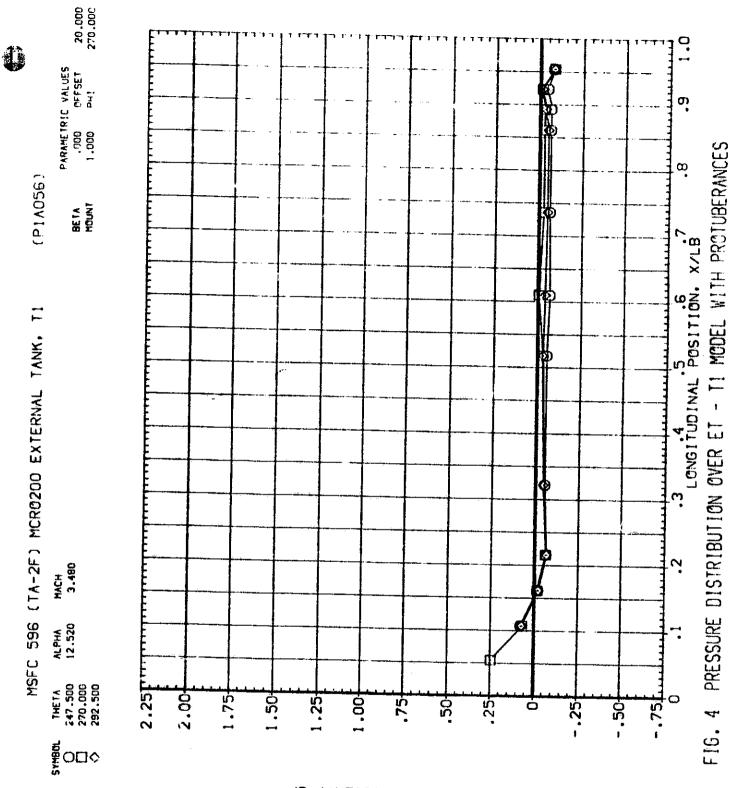
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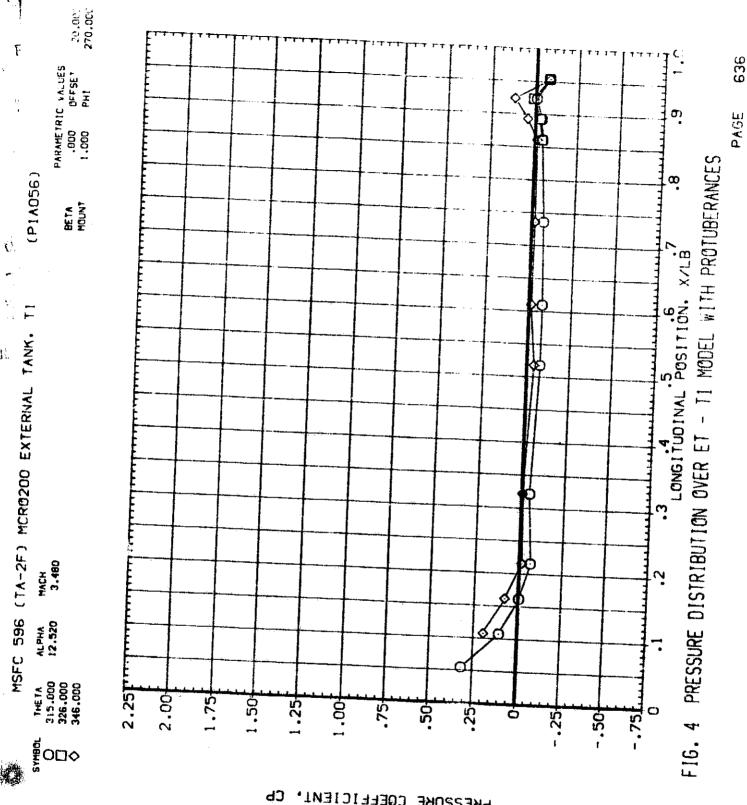
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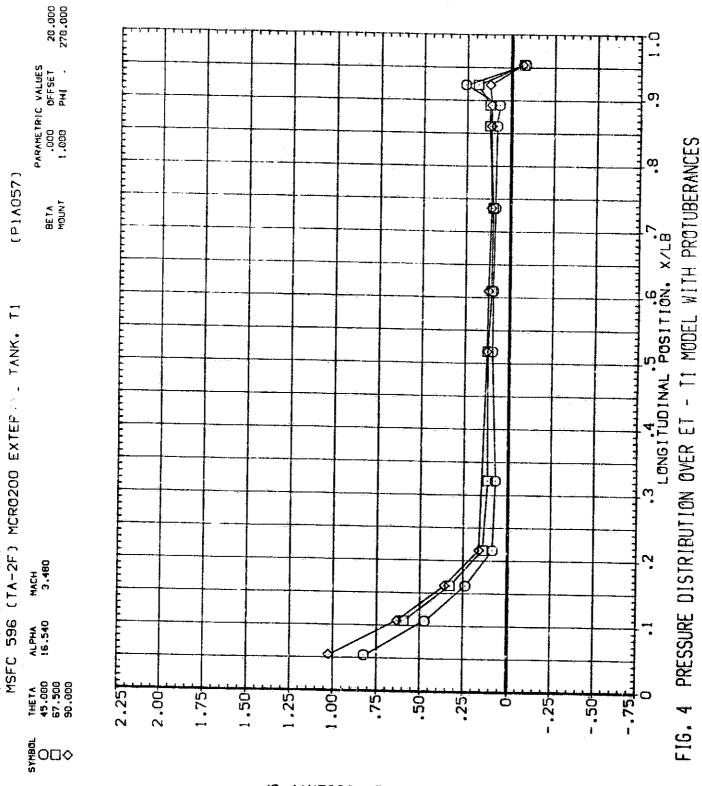
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PRESSURE COEFFICIENT, CP



PRESSURE COEFFICIENT,

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



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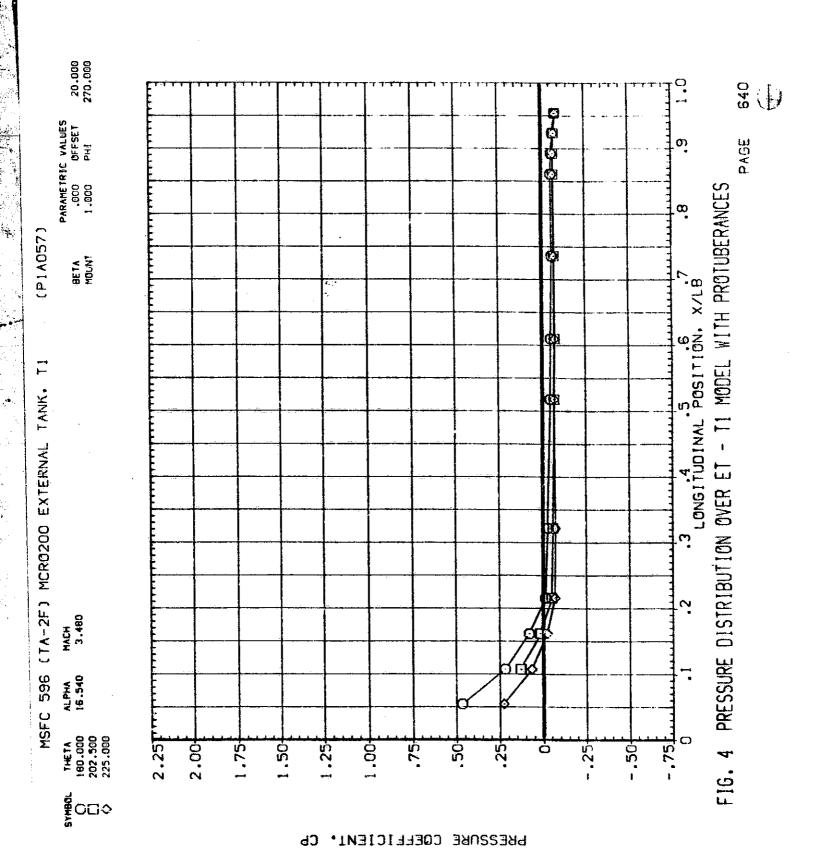
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LONGITUDINAL POSITION, X/LB Ф þ -.75 50 .25 Ó -.25 -.50

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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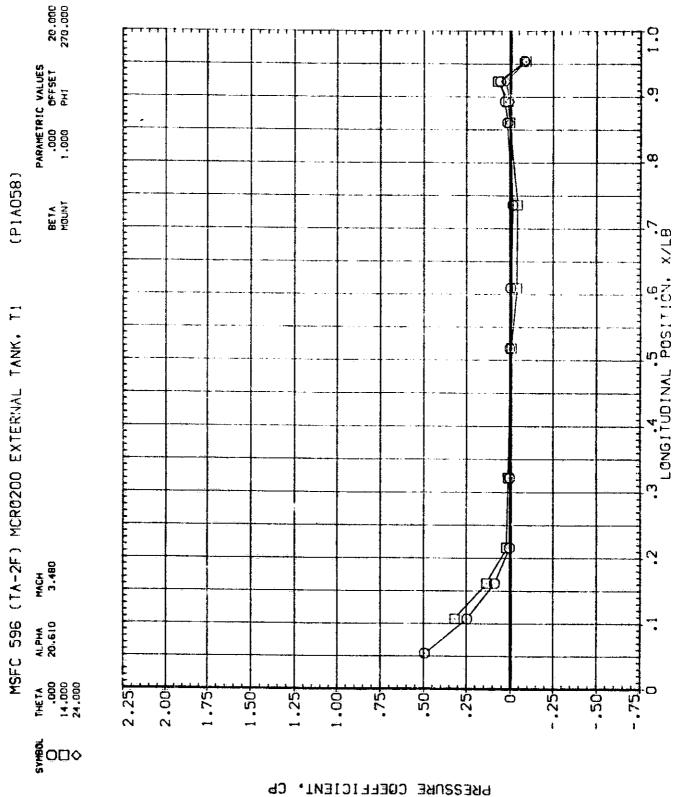
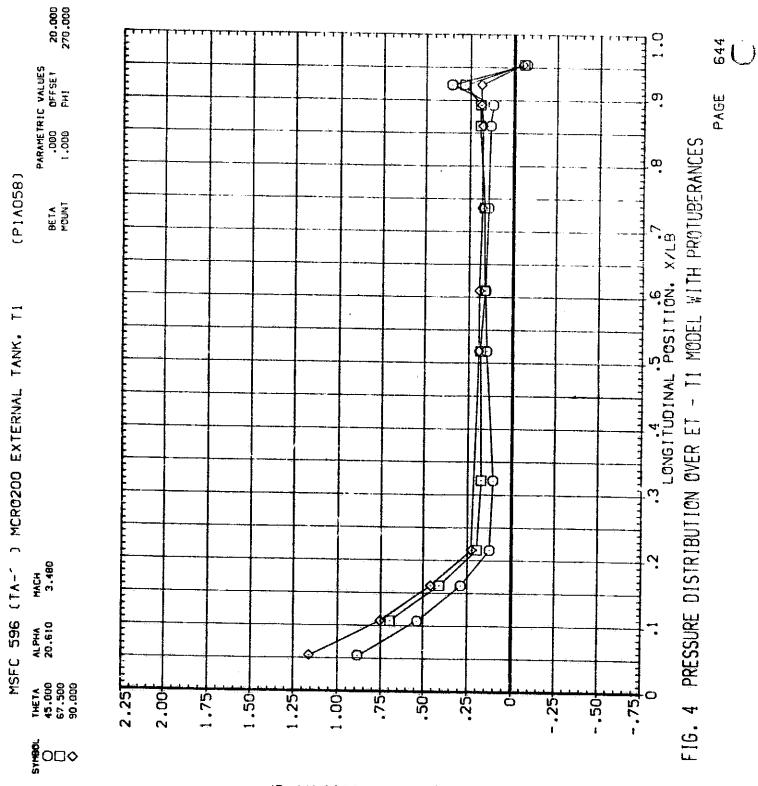


FIG. 4 PRESSURF DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

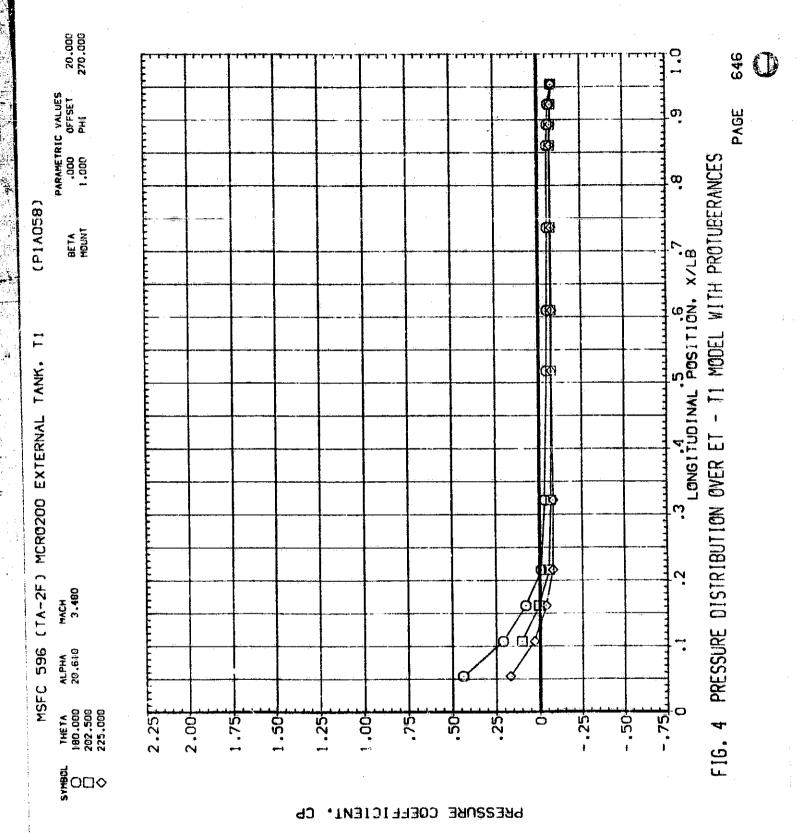
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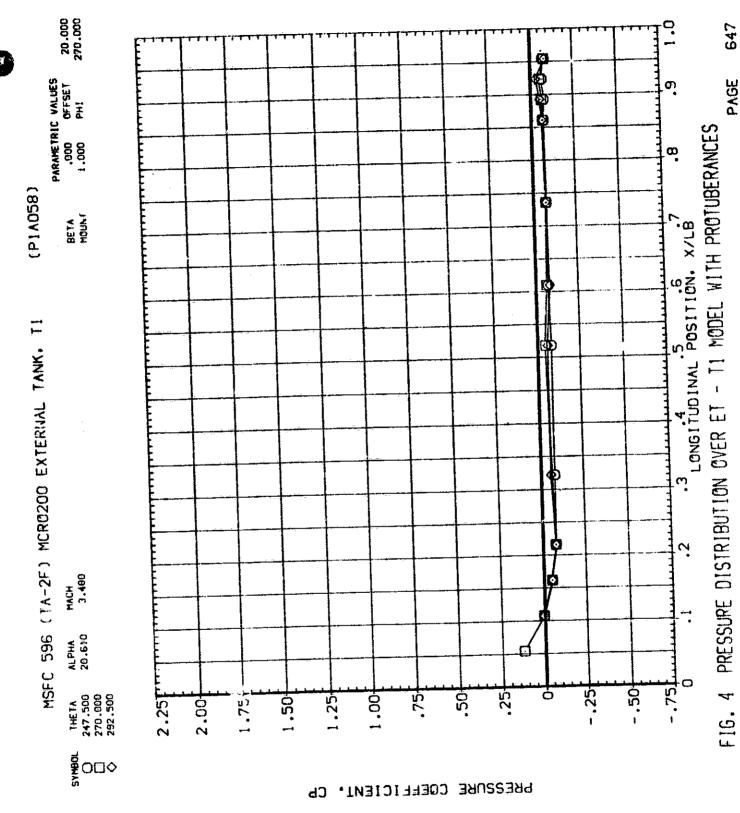
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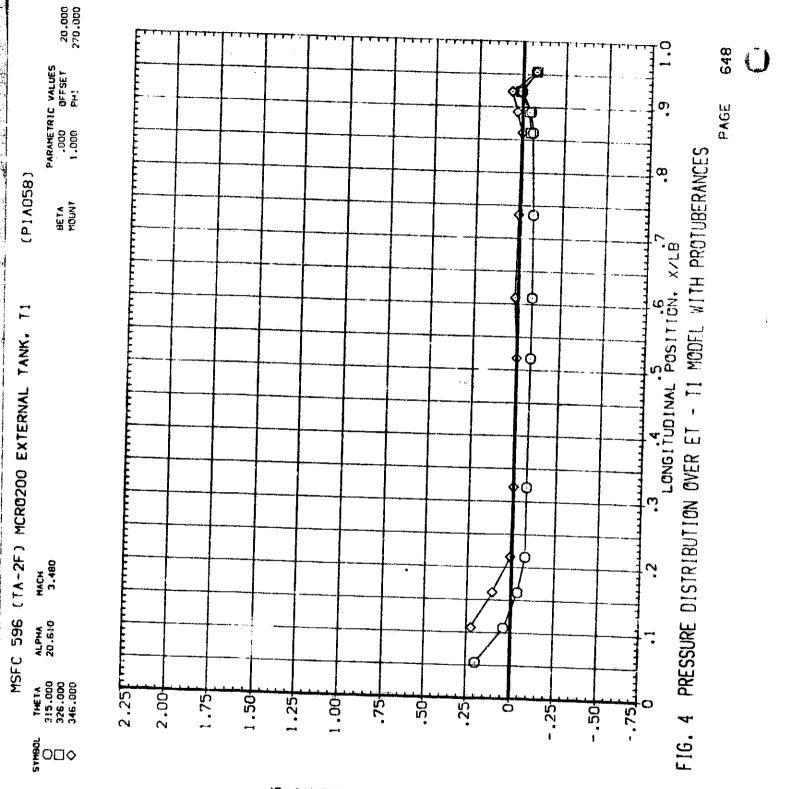
PRESSURE COEFFICIENT, CP

FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

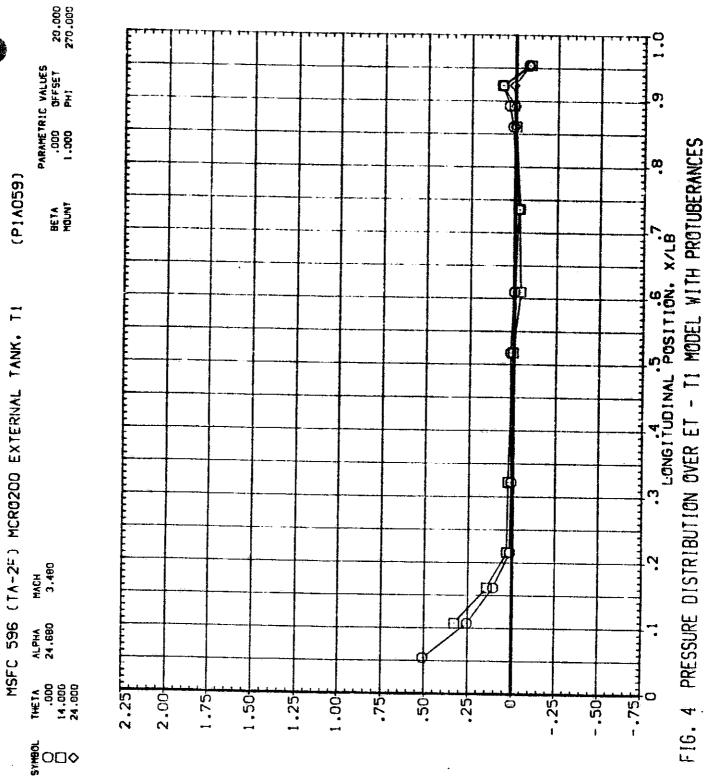


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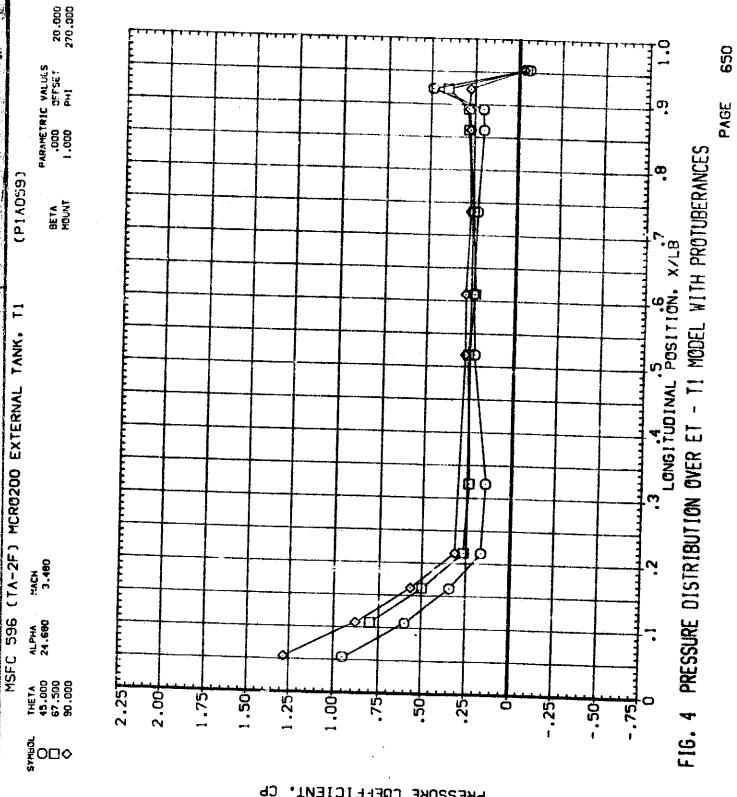




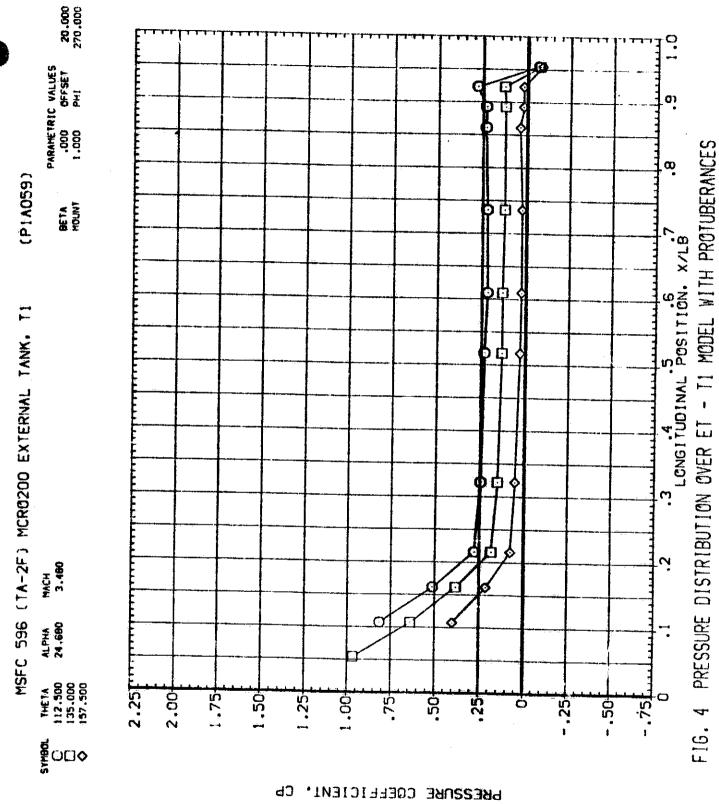
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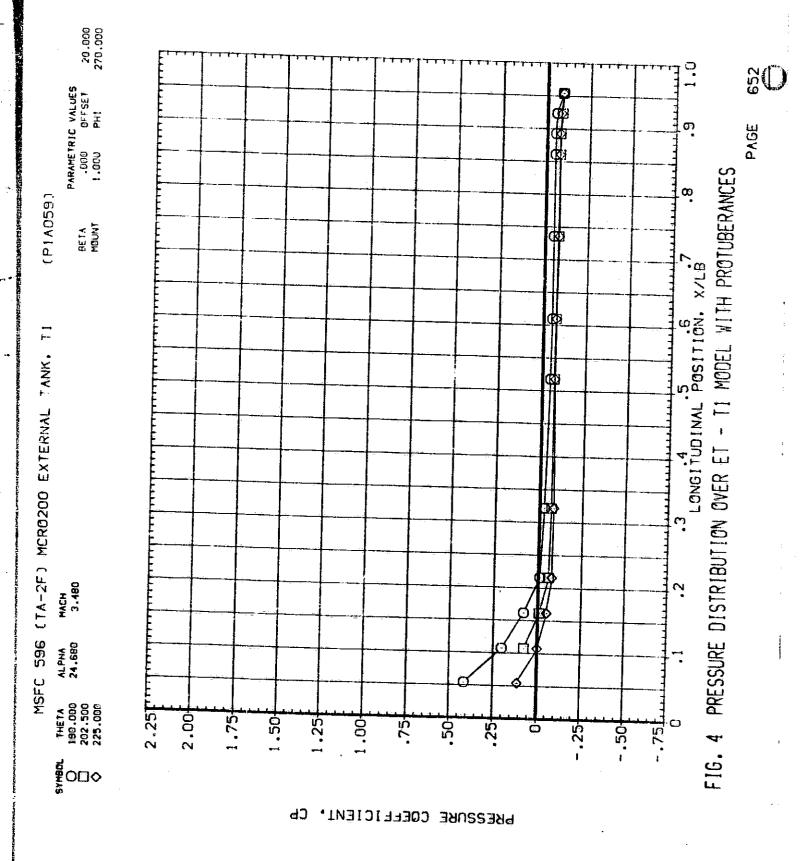


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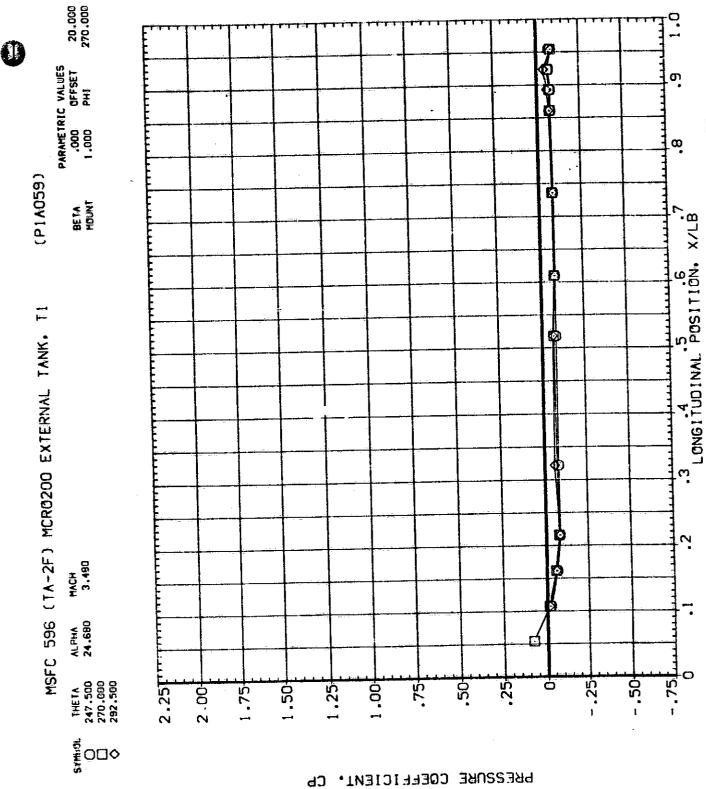
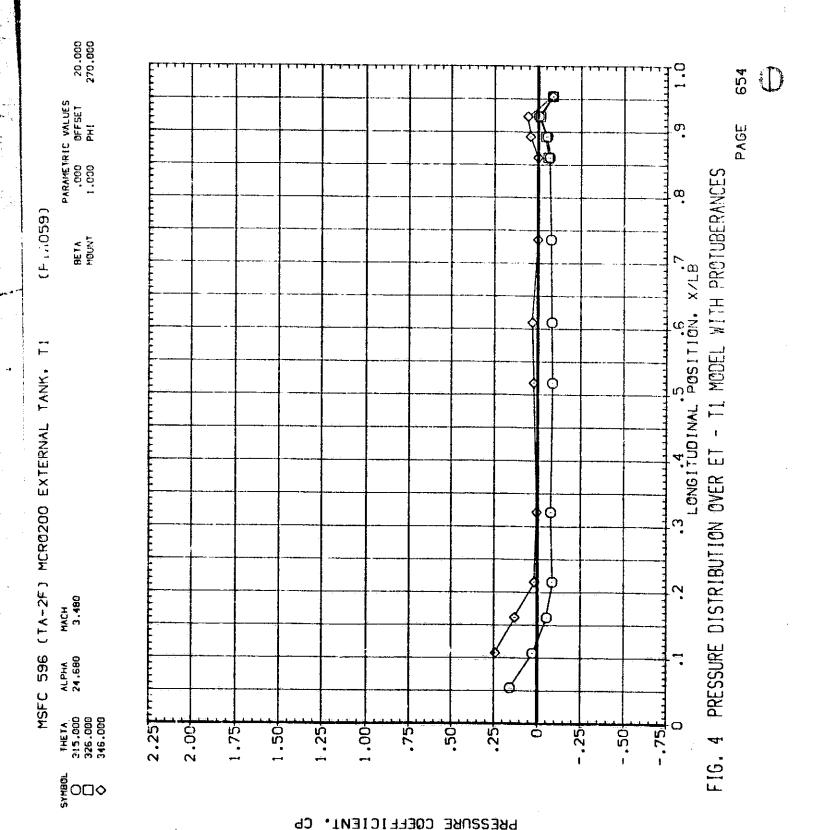
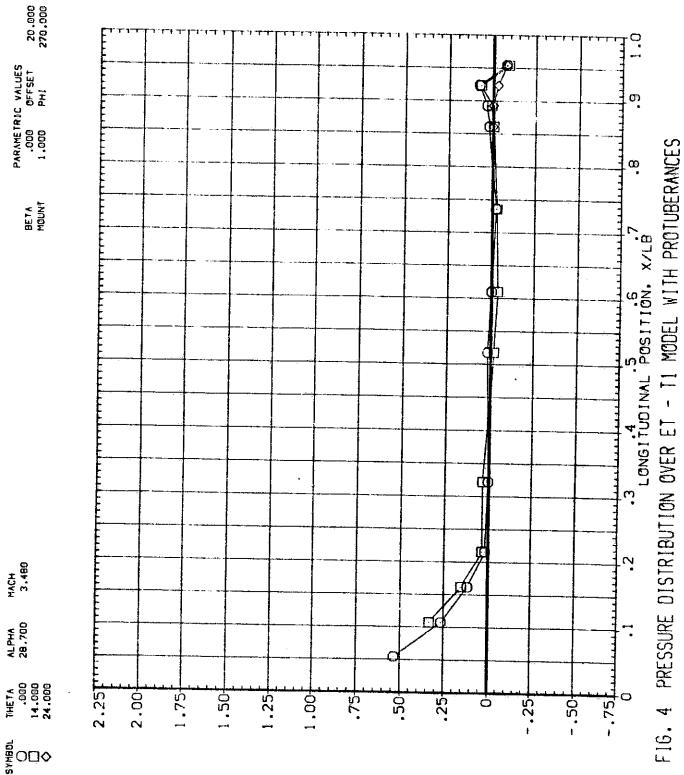


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES





PRESSURE COEFFICIENT. C



(P1A060)

MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, TI

0:0 o, ω 4 .5 .6 .7 POSITION. X/LB LØNGITUDINAL PRESSURE DISTRIBUTION OVER ф þ Q Ŋ 50 -.50 - 75 FIG. 4 -.25 PRESSURE COEFFICIENT, CP

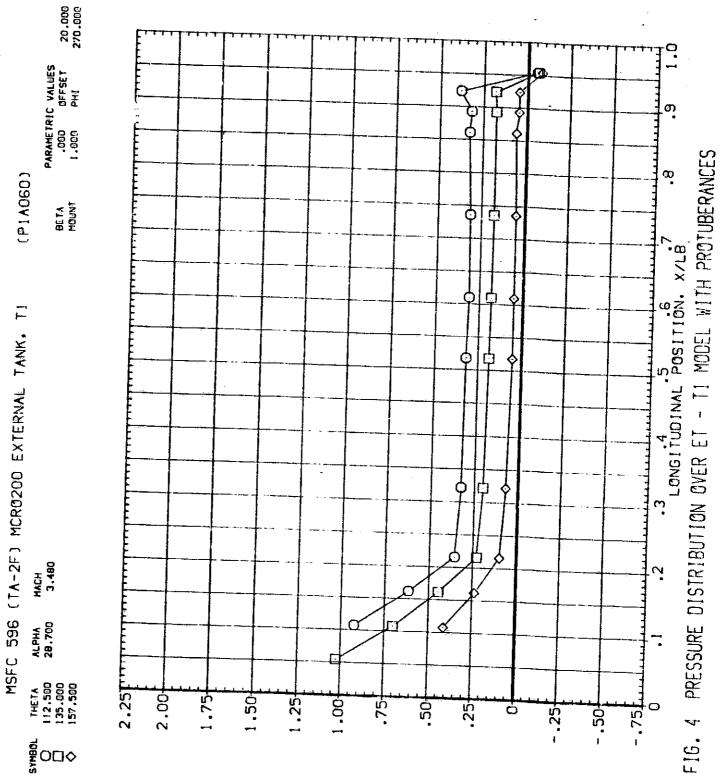
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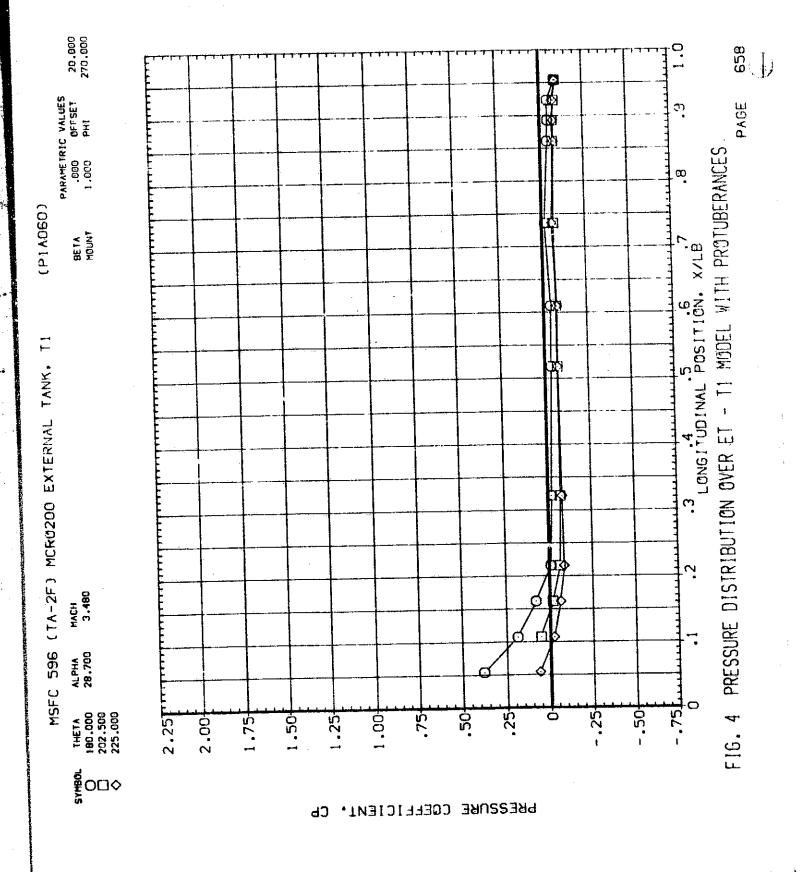
II MODEL WITH PROTUBERANCES

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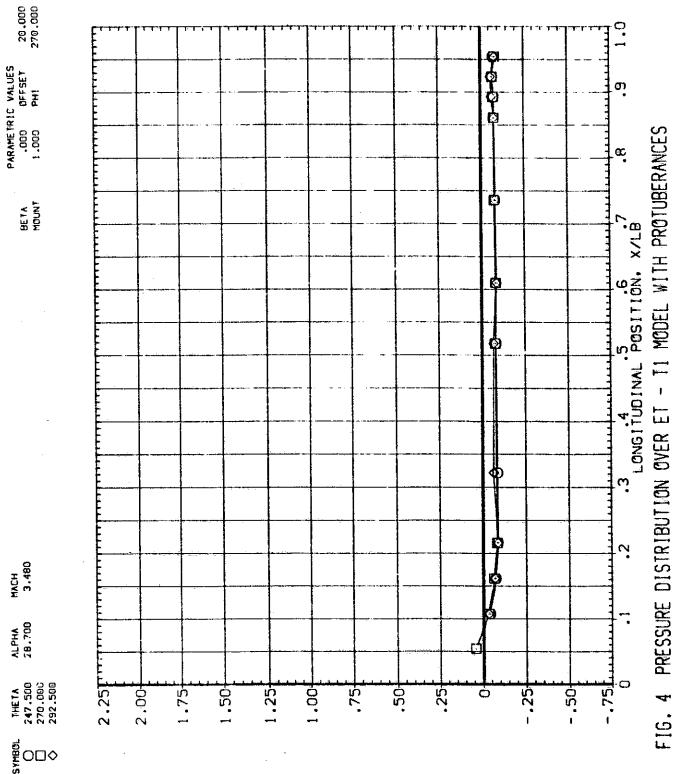


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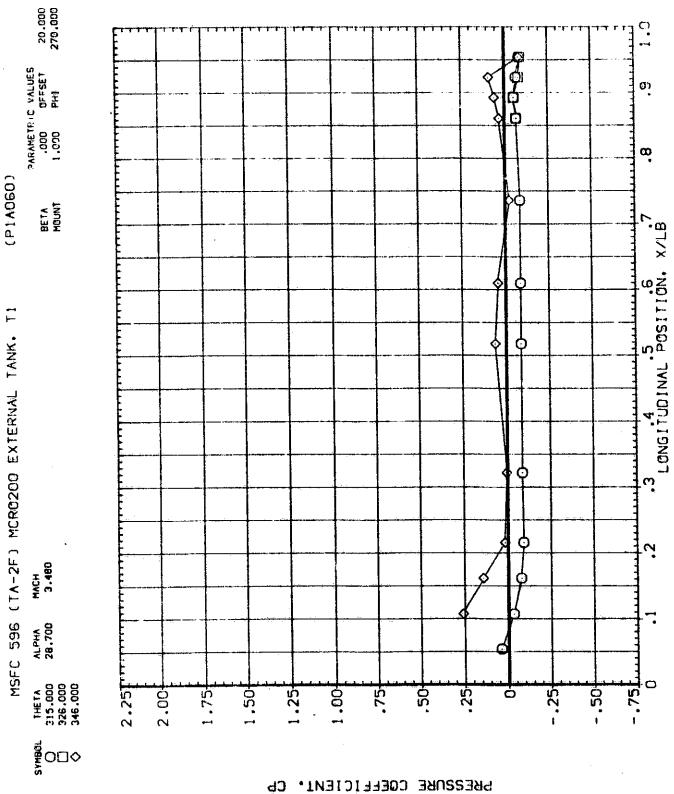
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MSEC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1



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PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES F16. 4

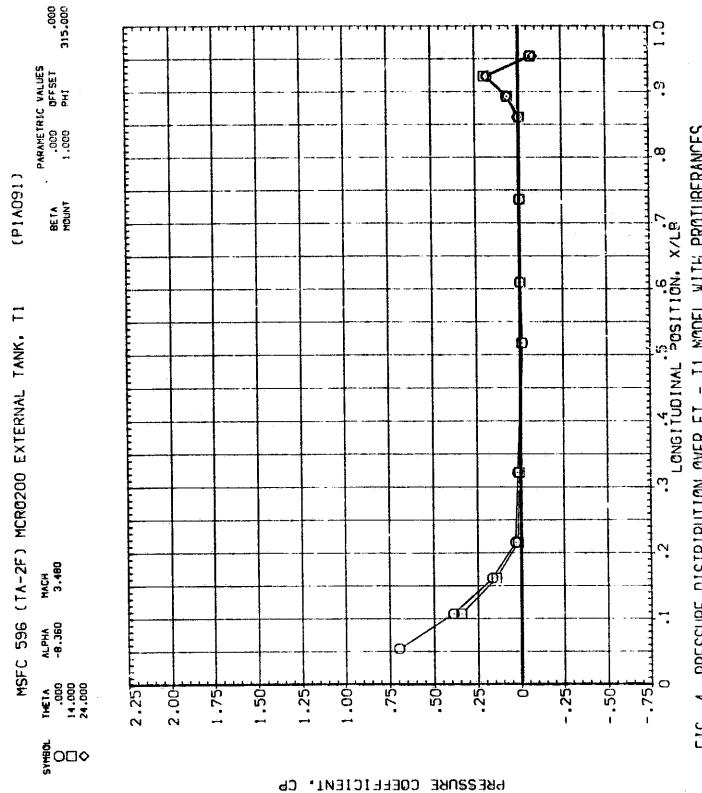
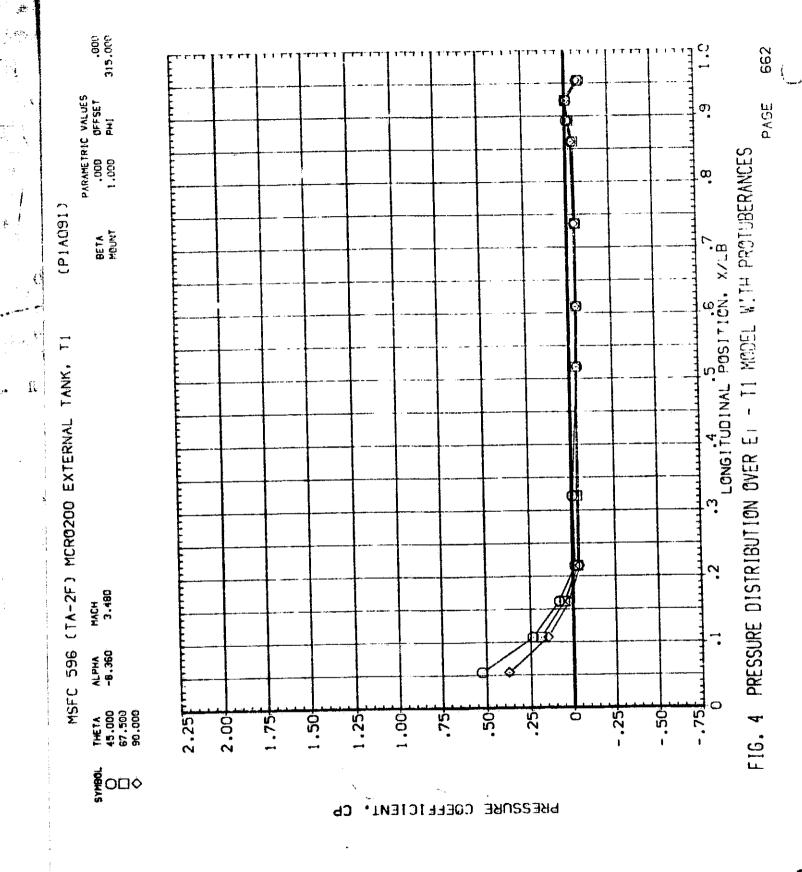


FIG. 4 PRESSURE DISTRIPUTION OVER ET - TI MODEL WITH PROTUBERANCES



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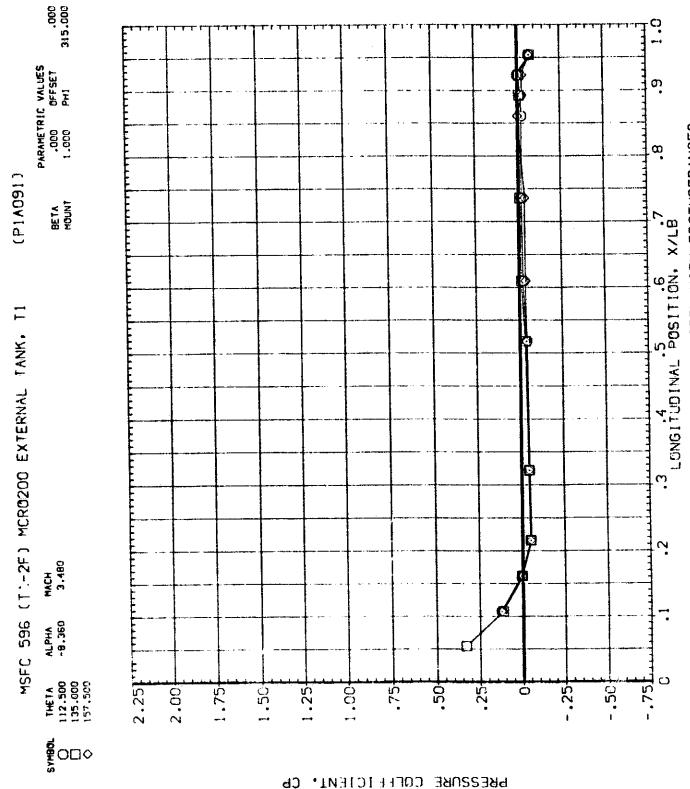
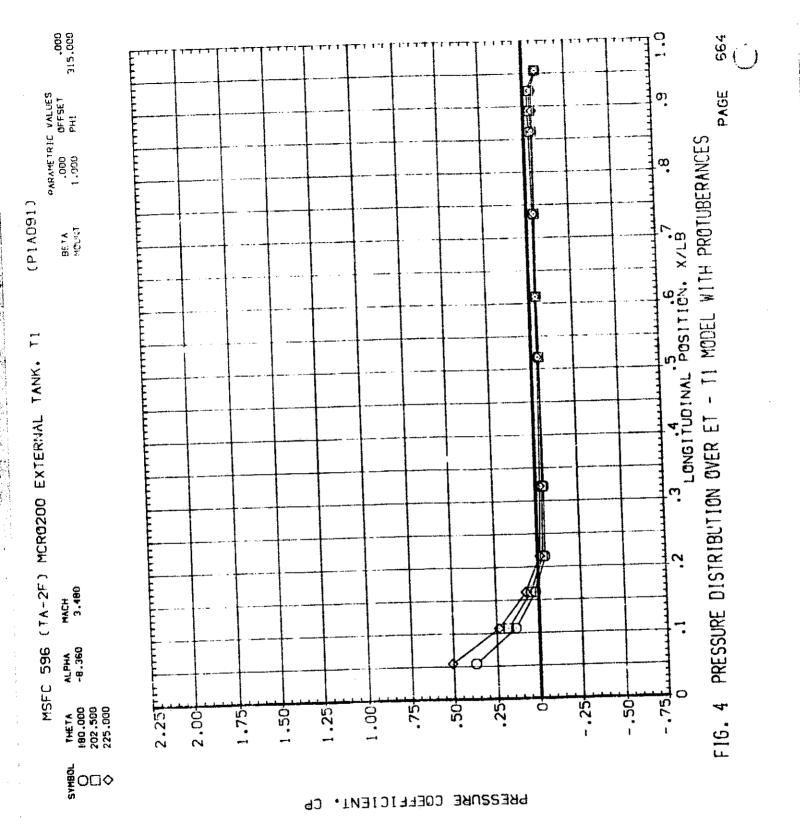


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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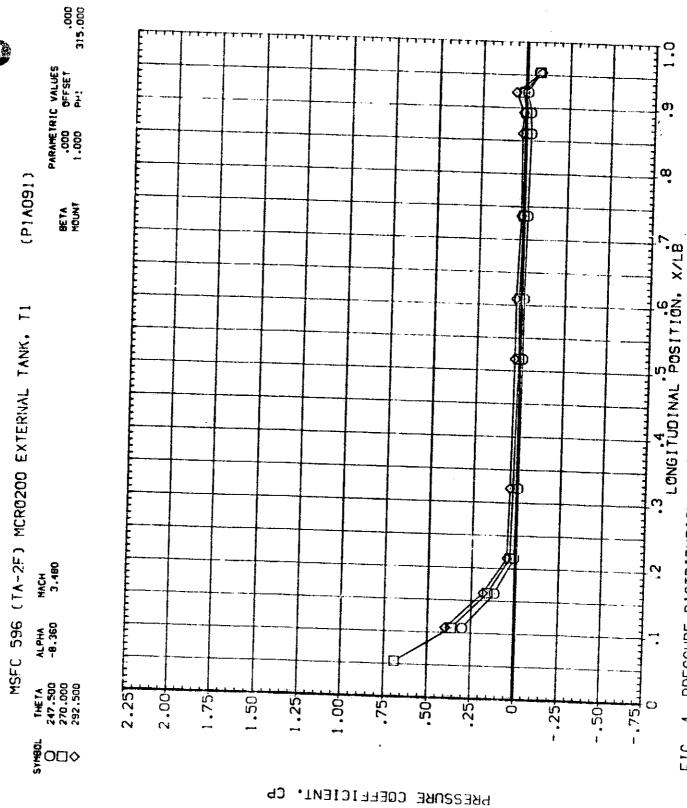
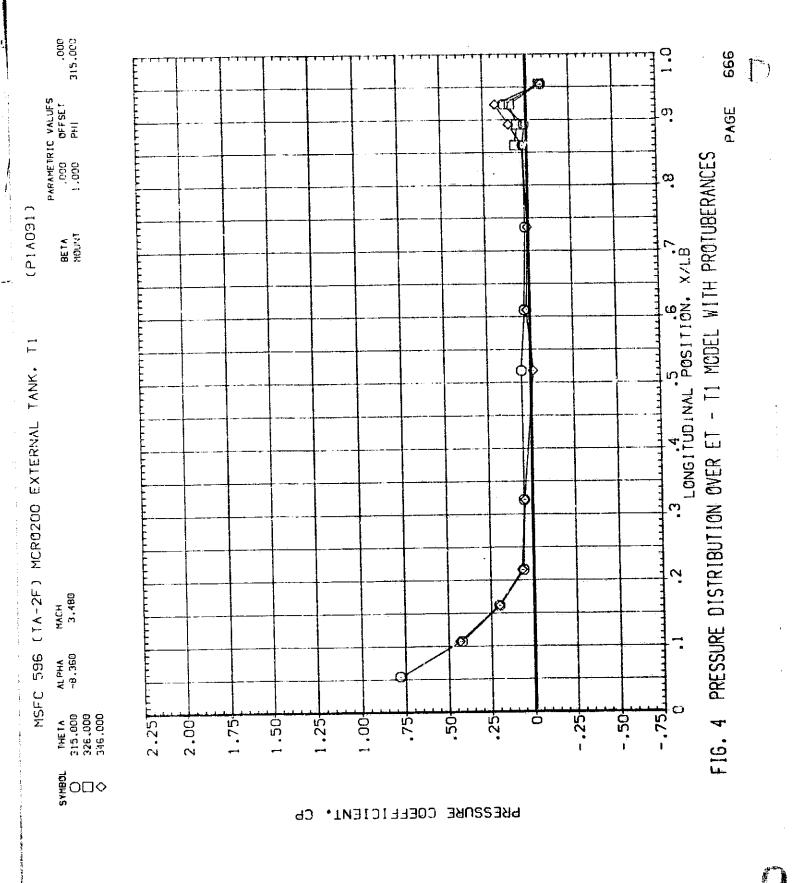
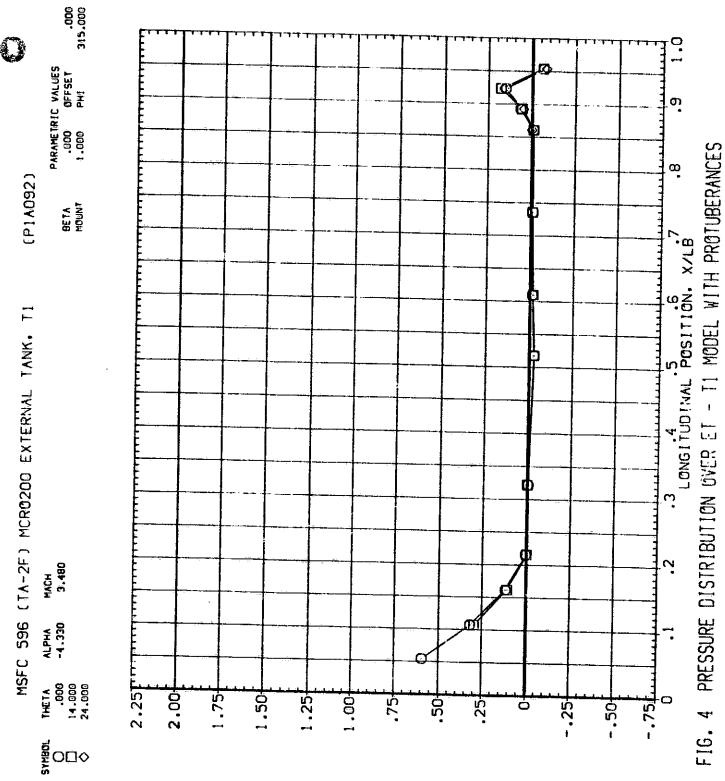


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES





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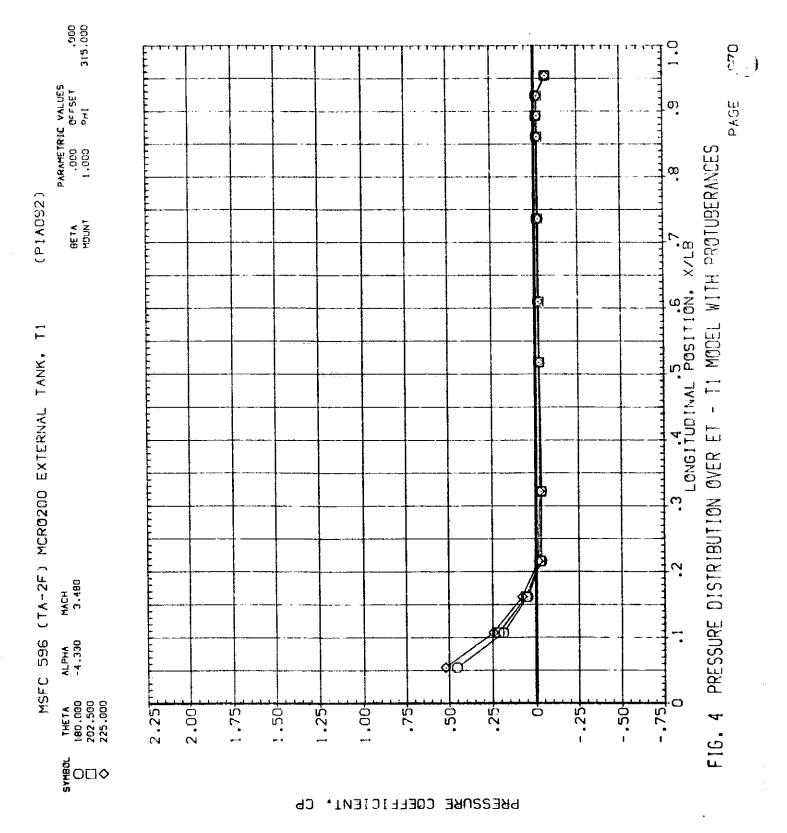
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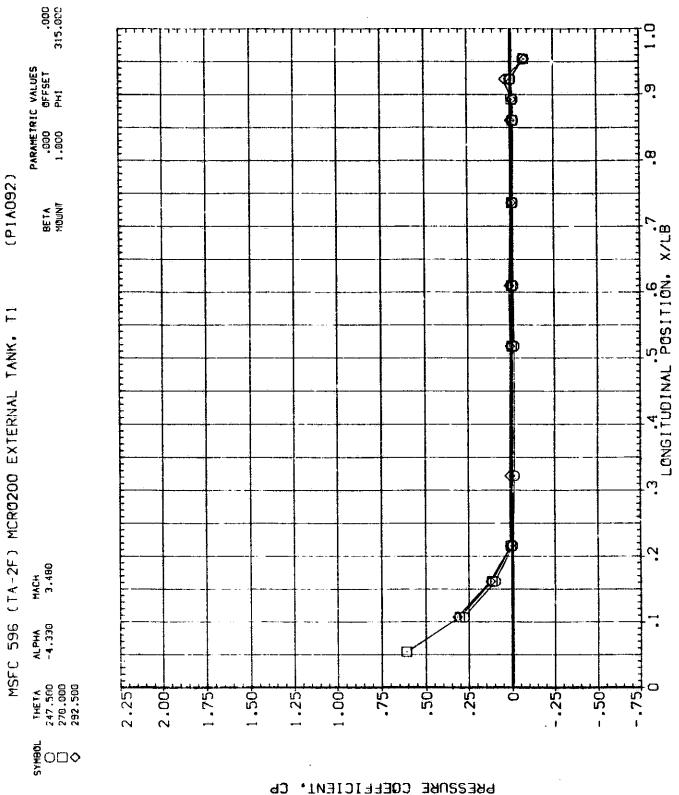
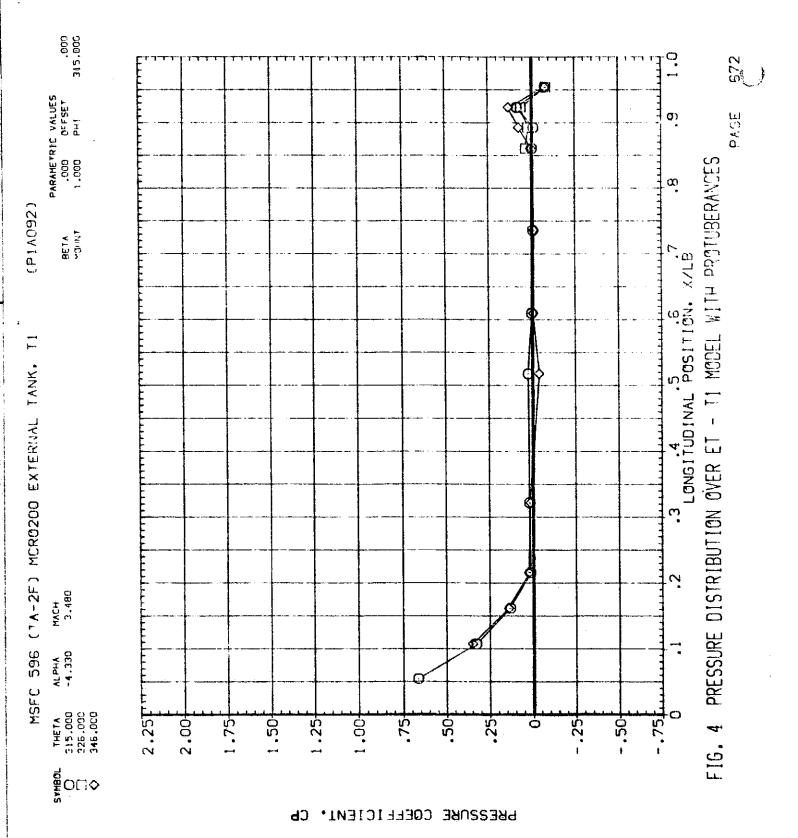
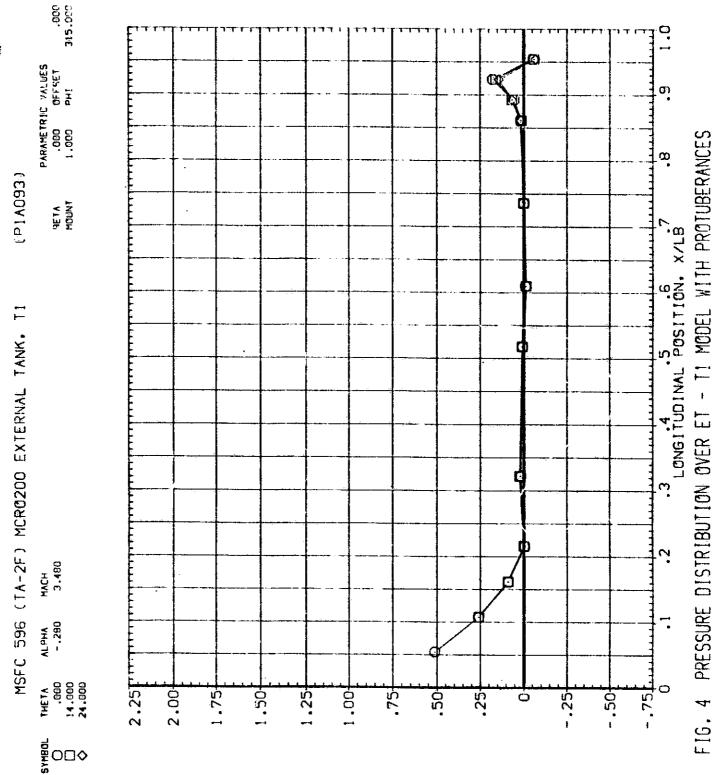


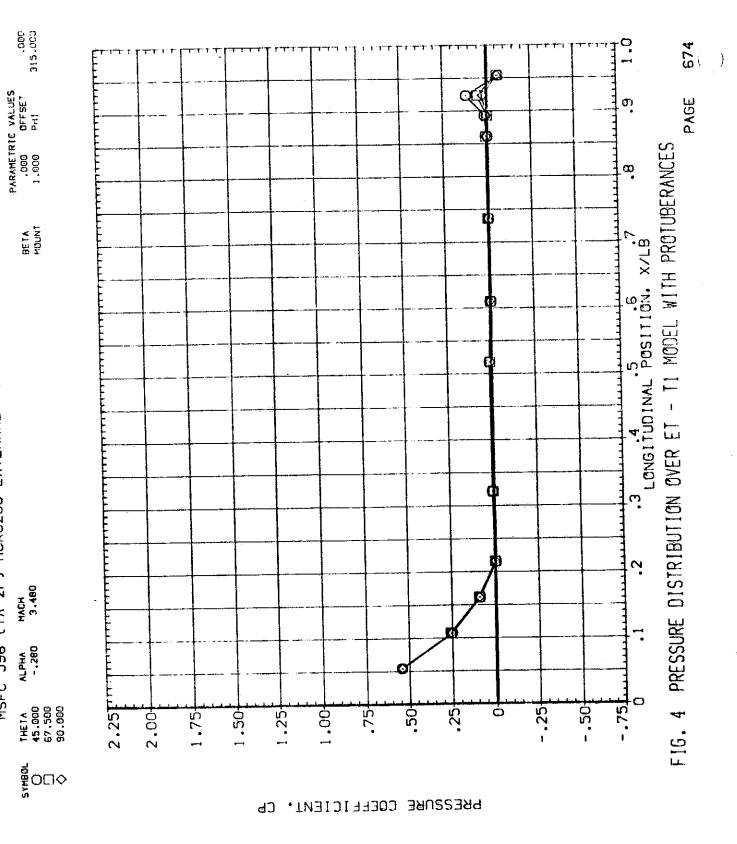
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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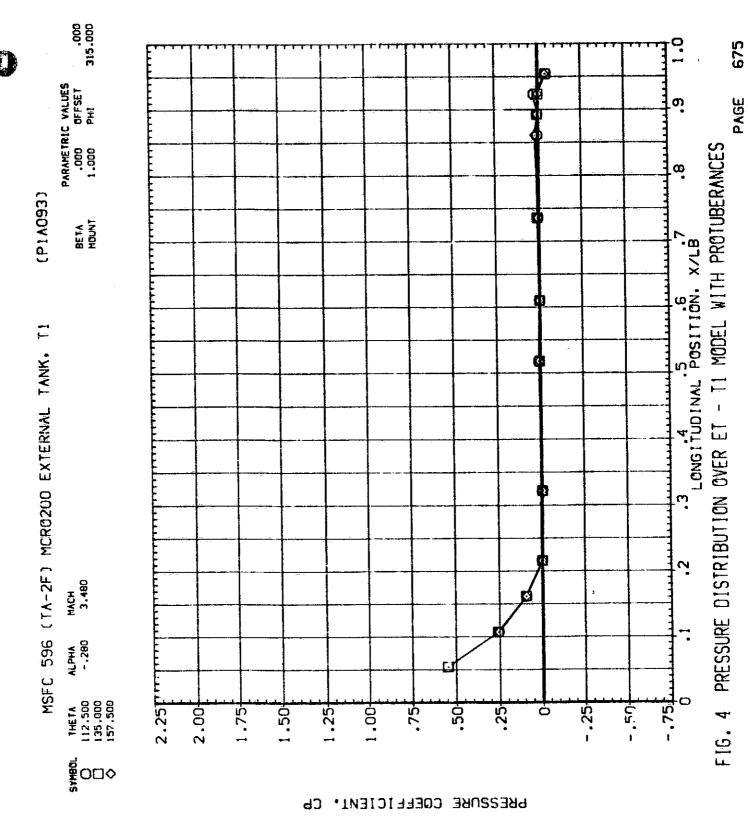


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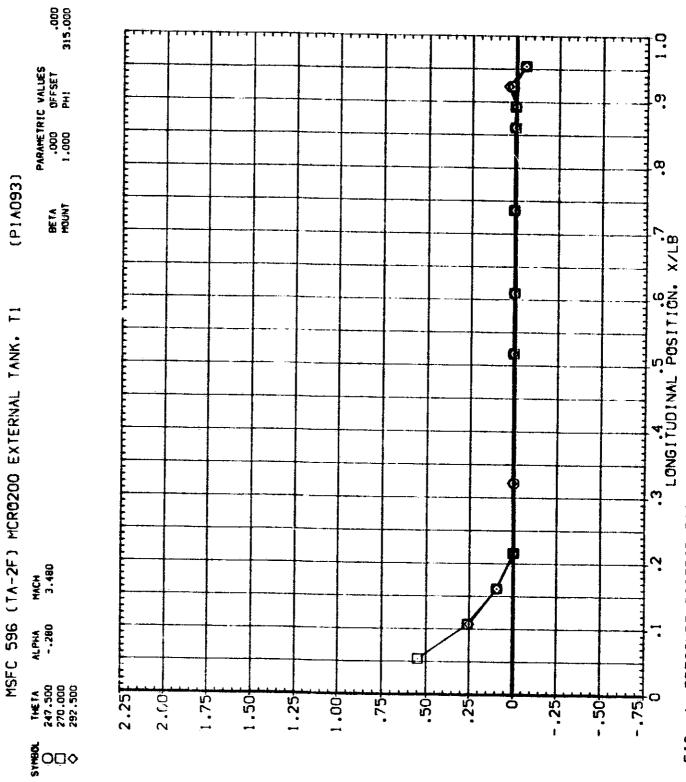
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PAGE 676



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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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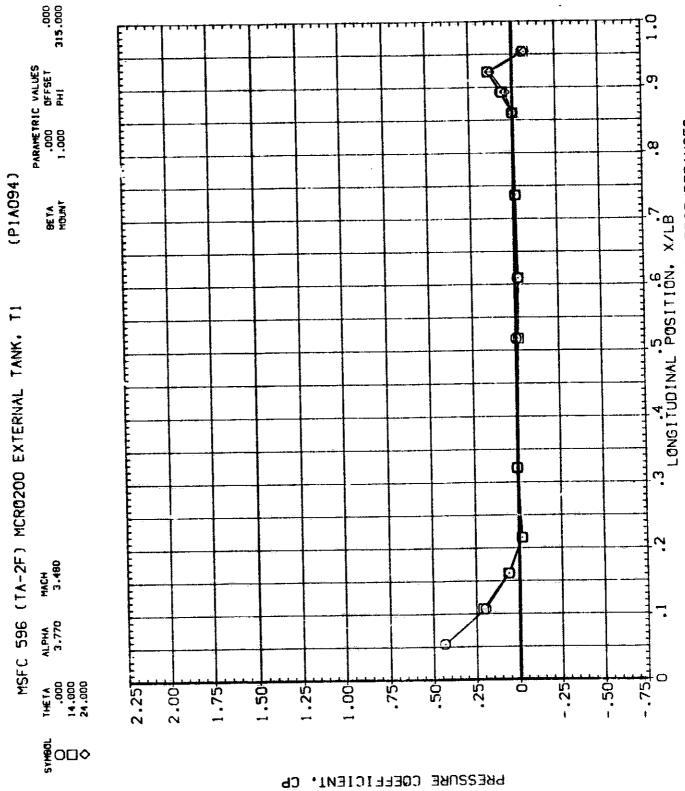
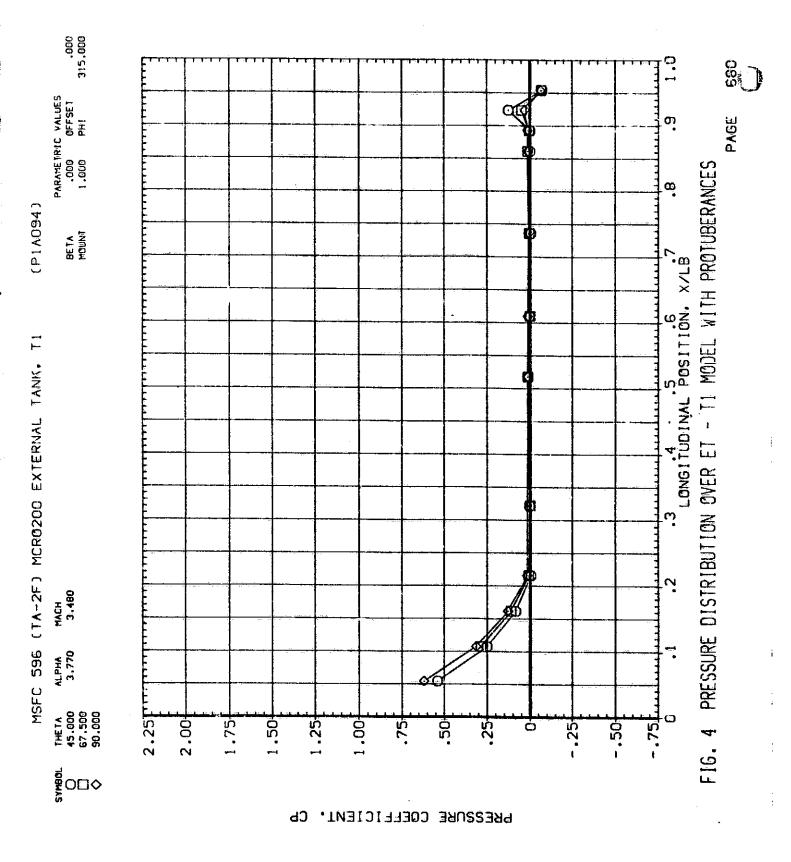
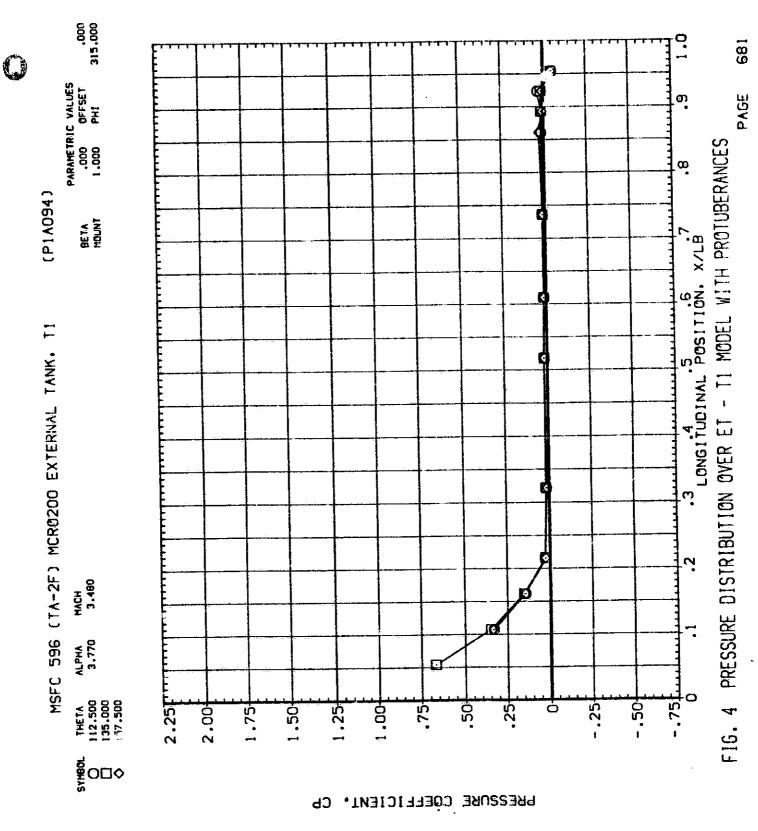
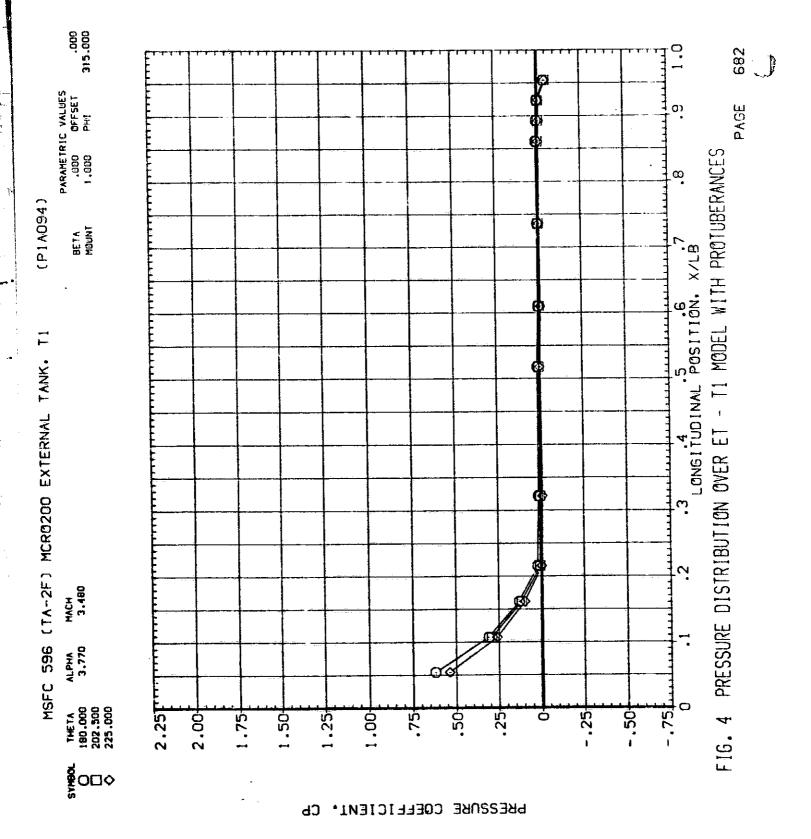


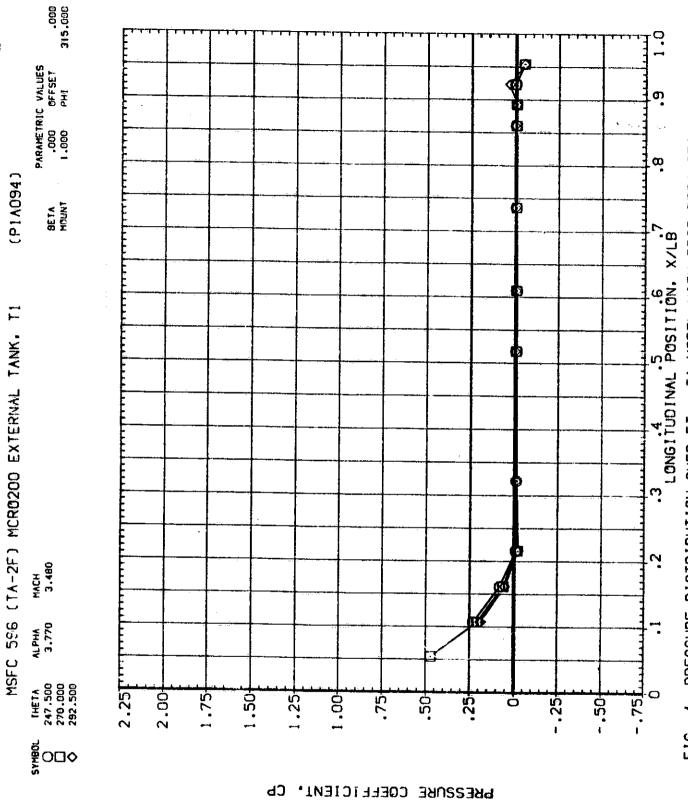
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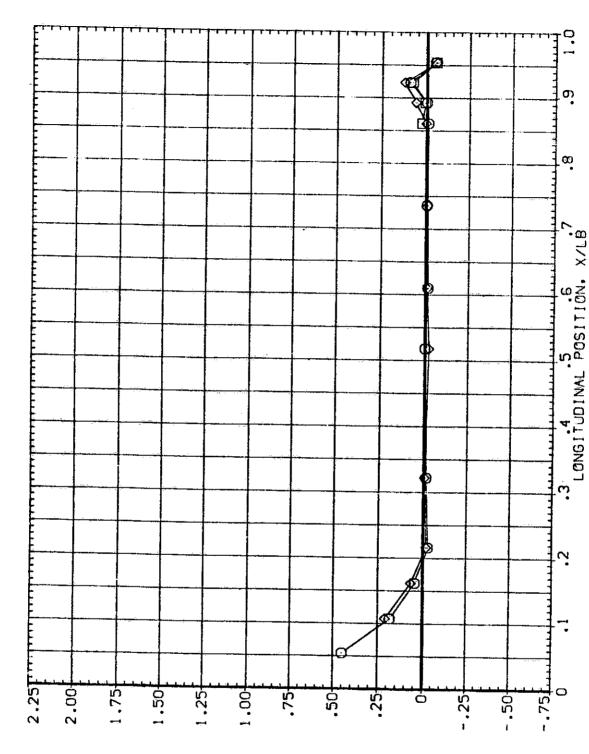
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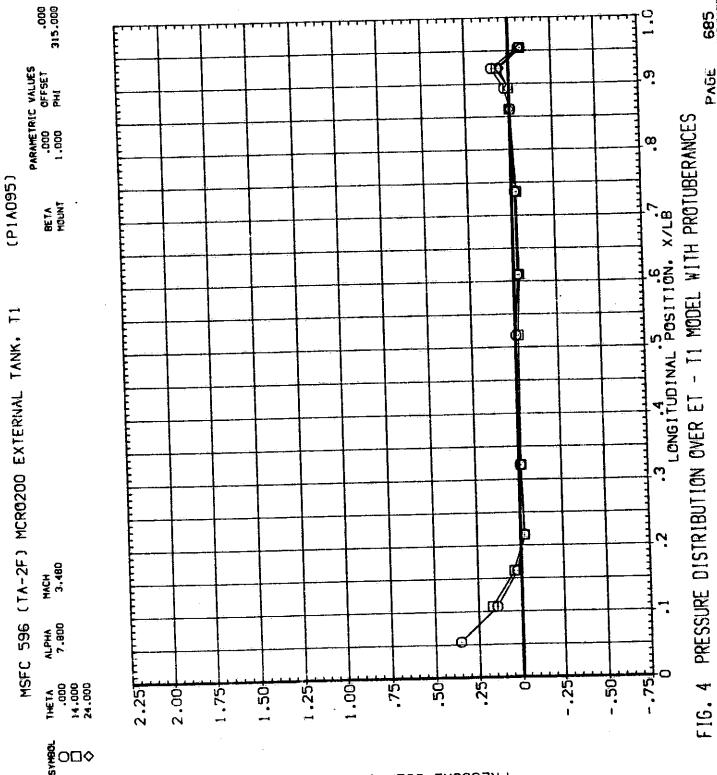
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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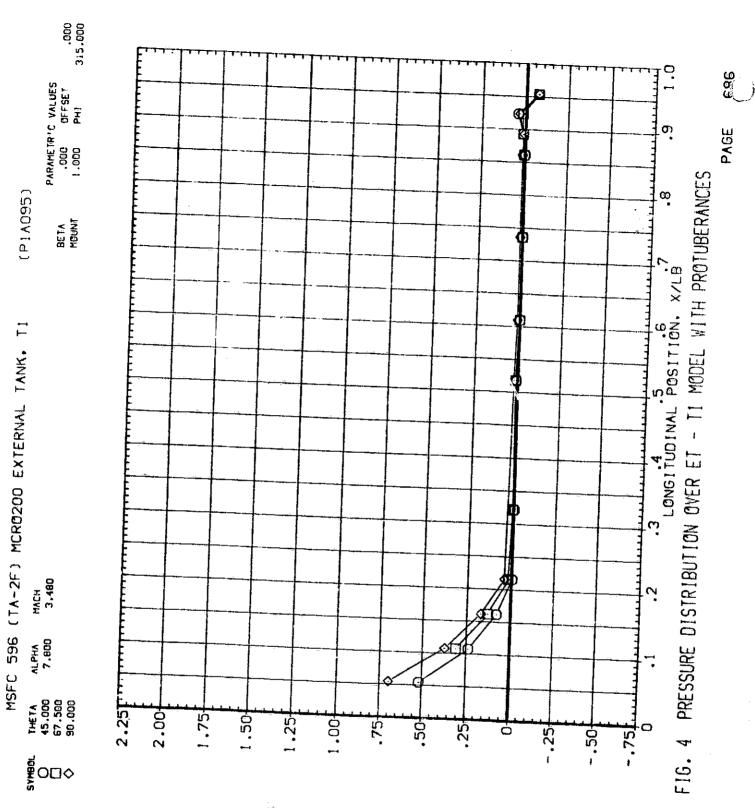


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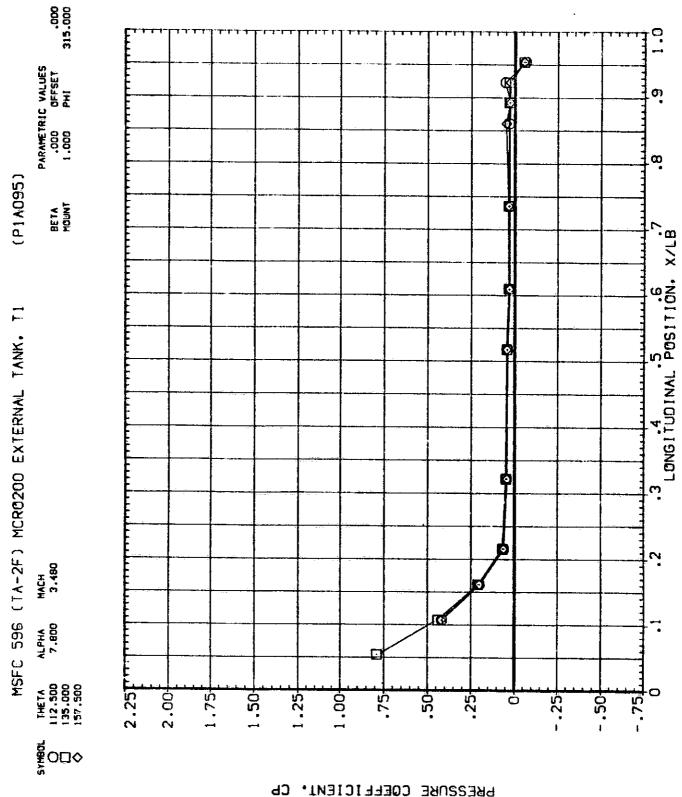
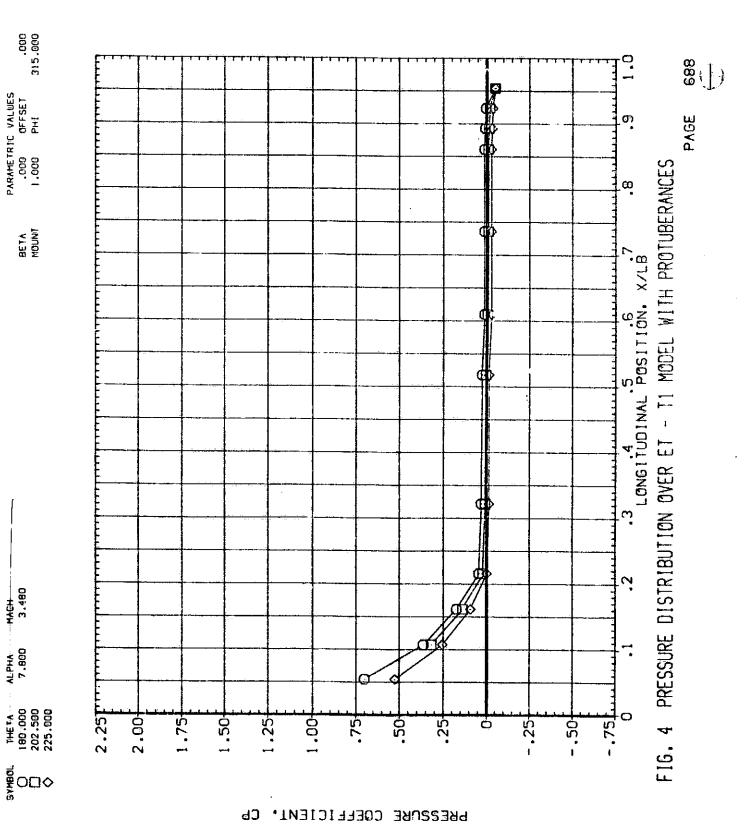


FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

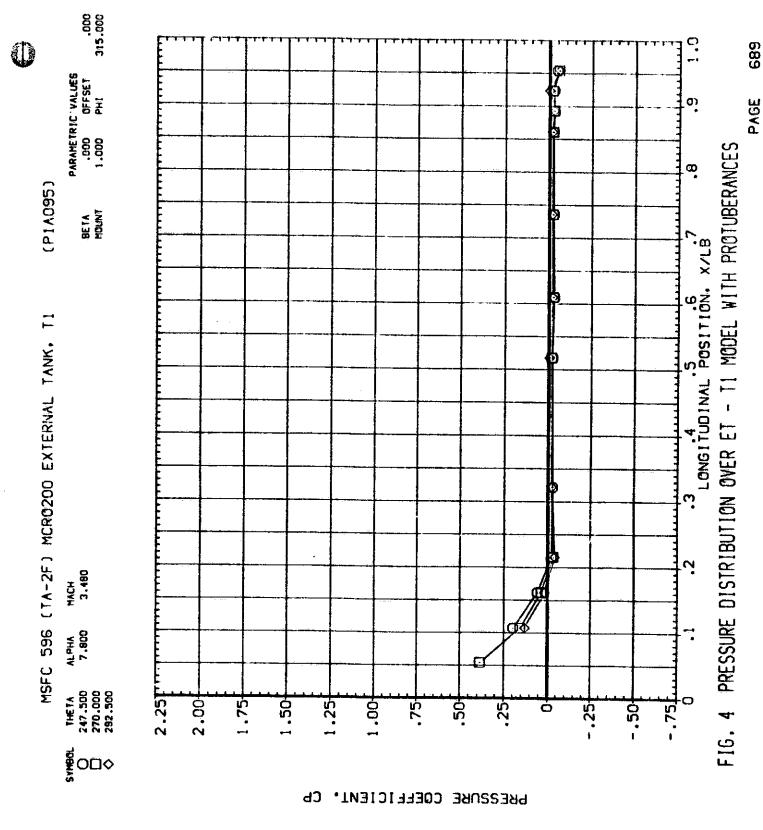


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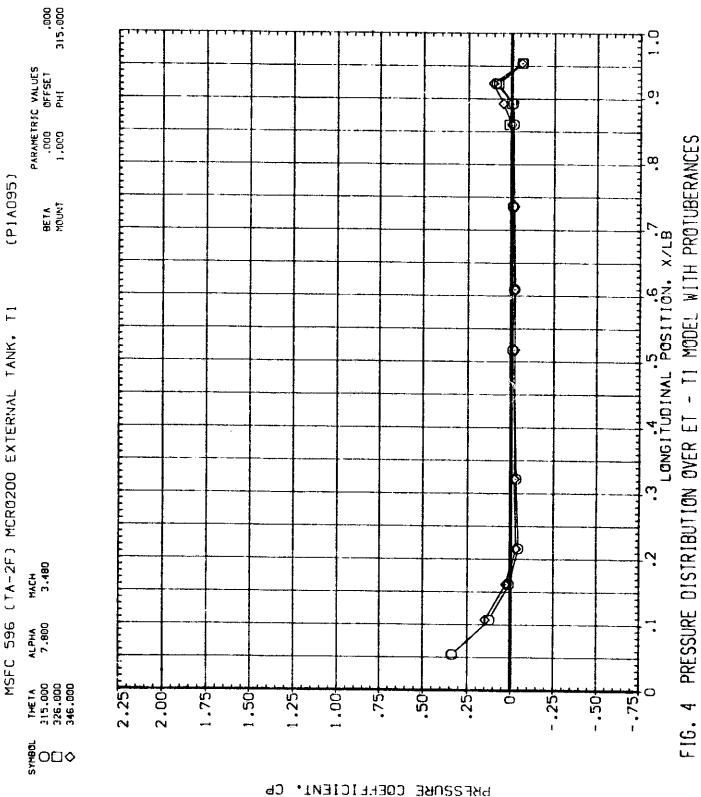
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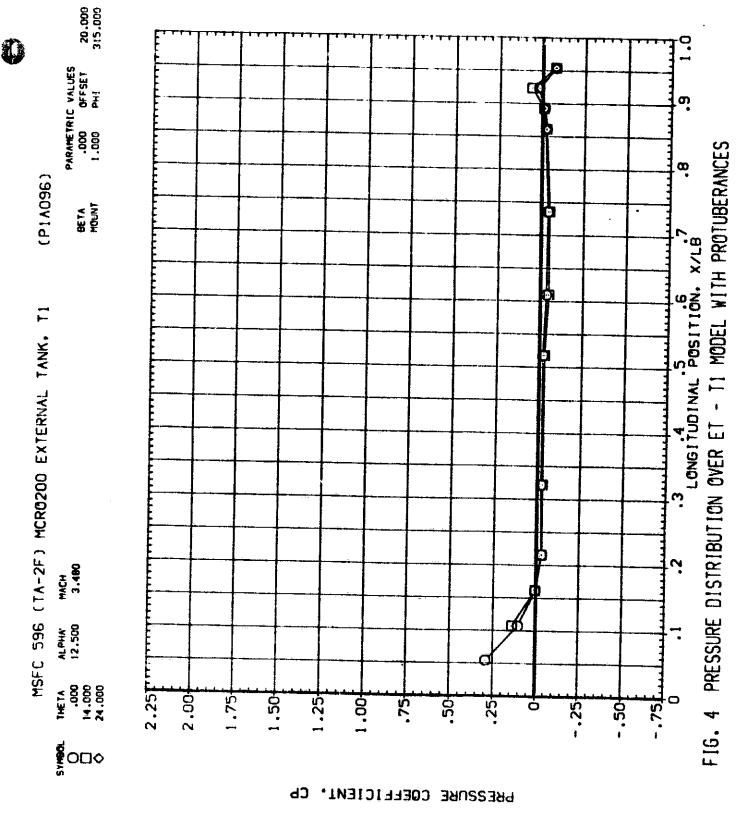
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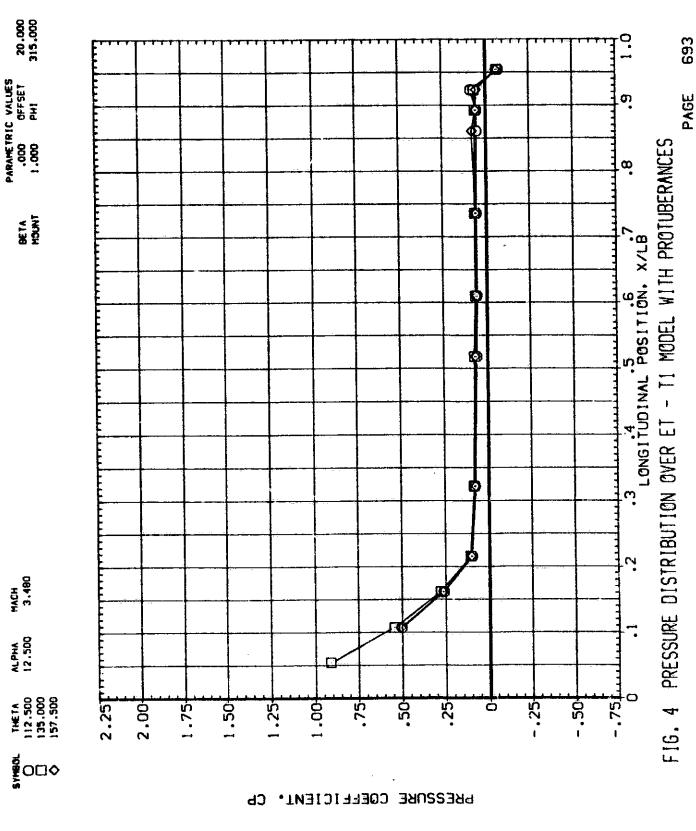


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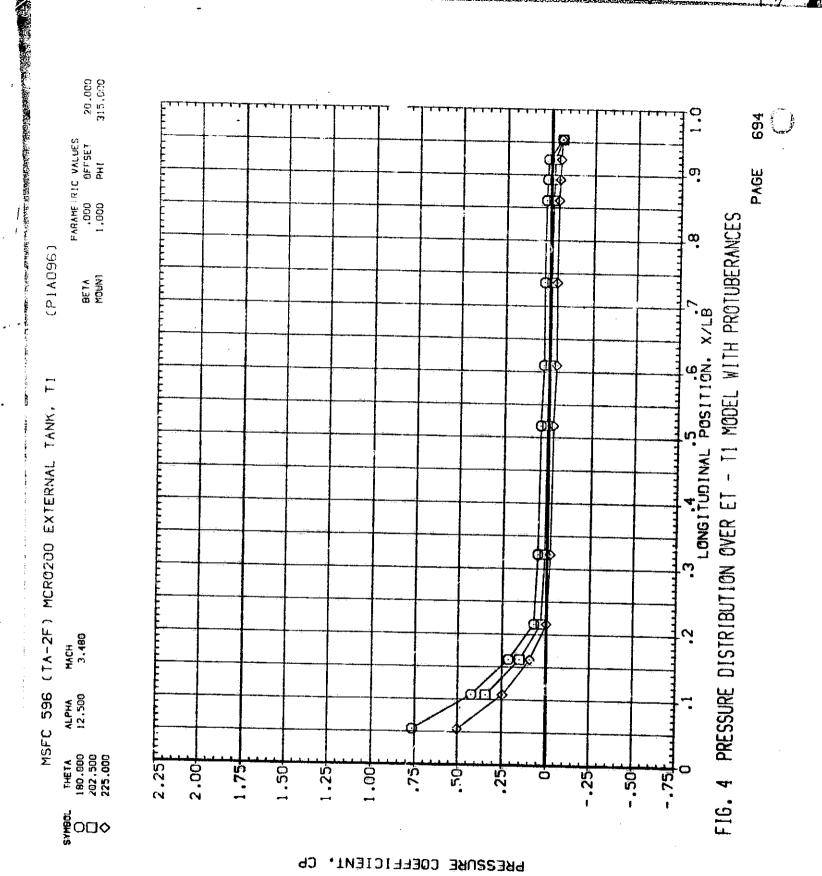
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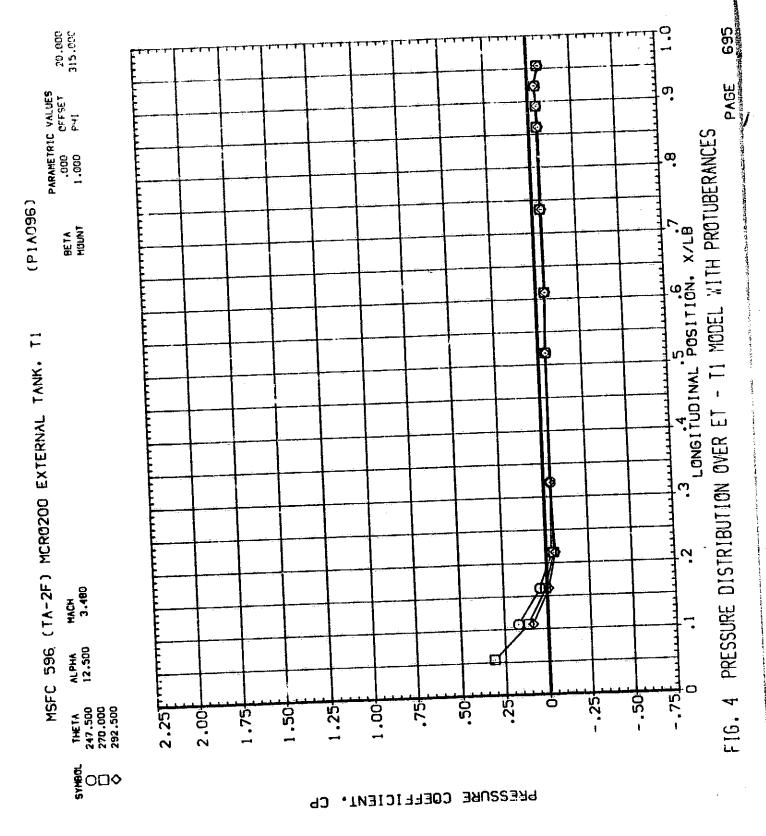


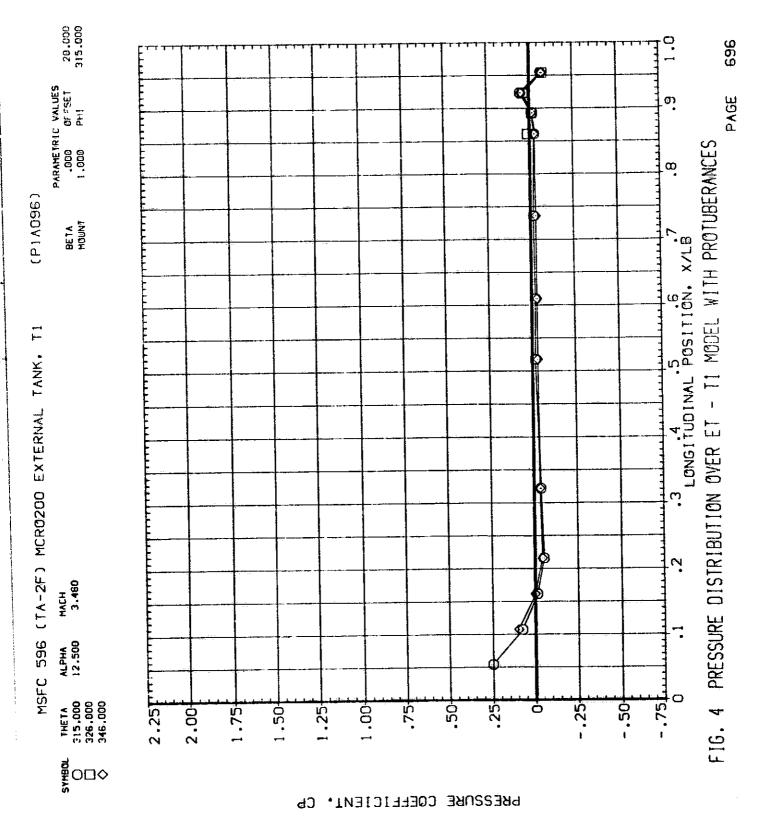
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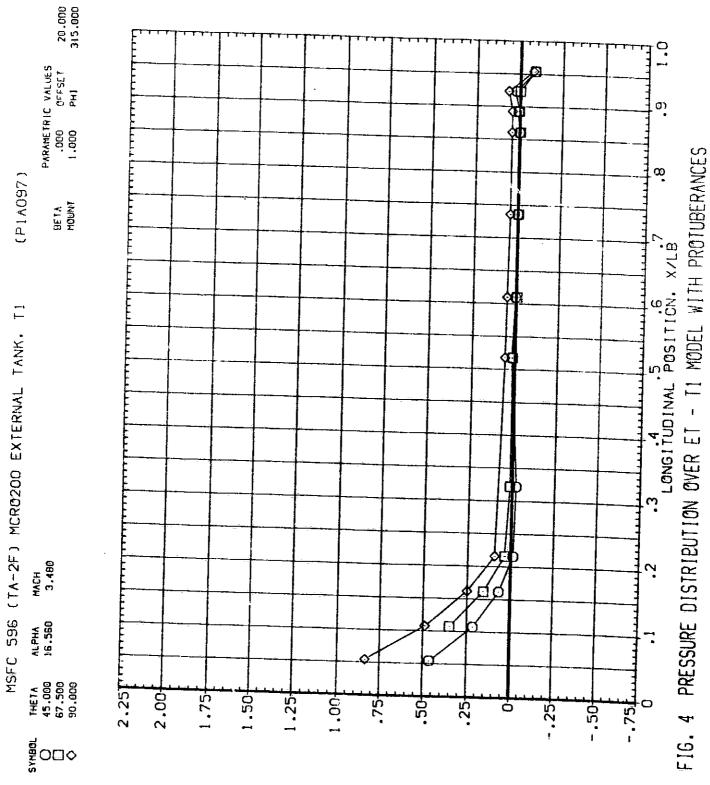
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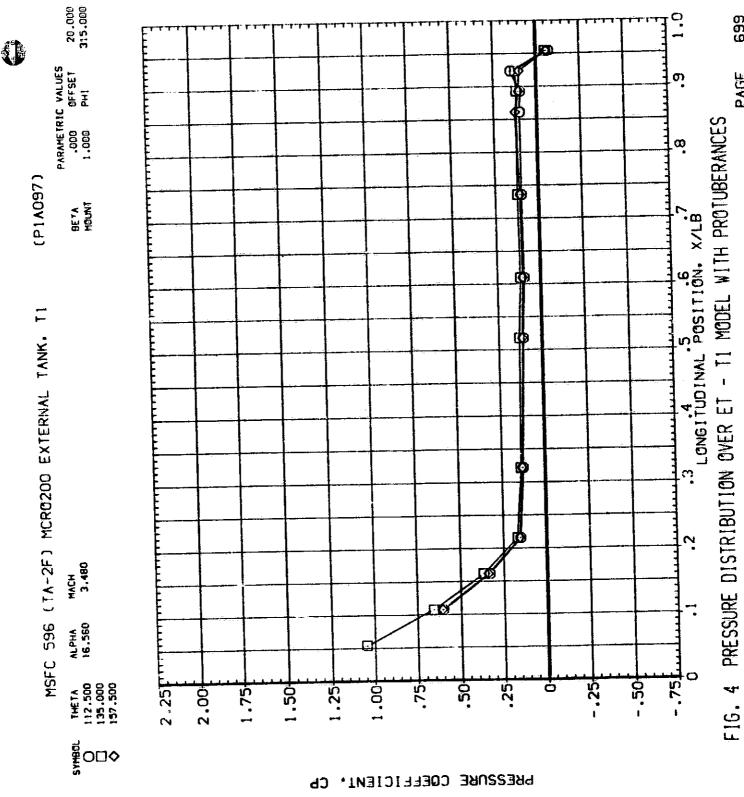
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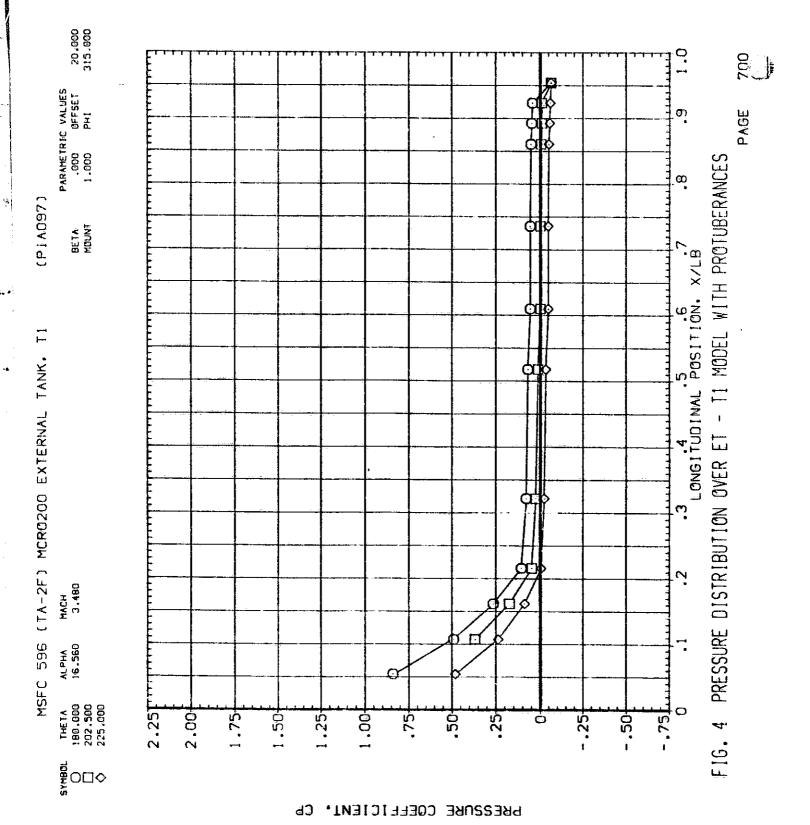
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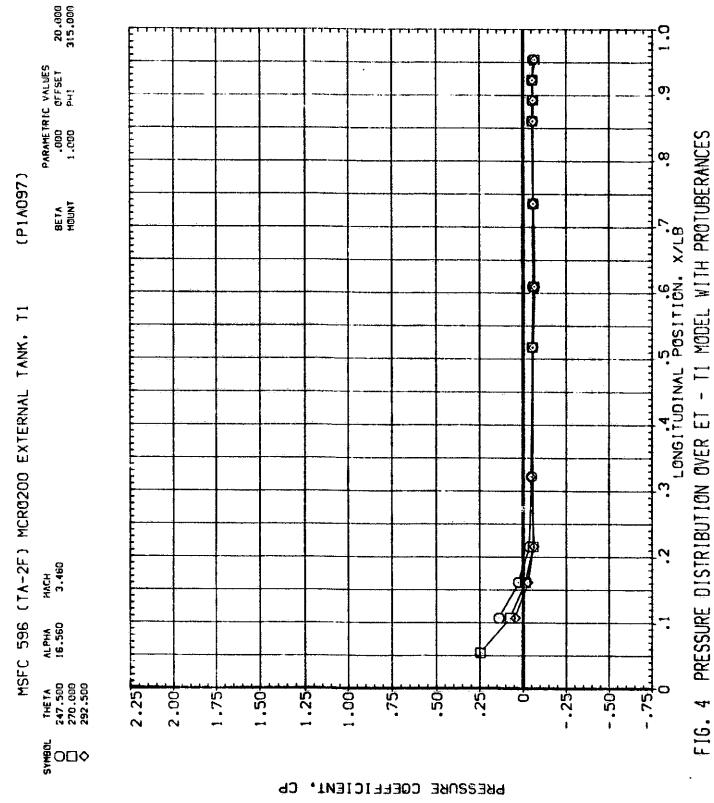
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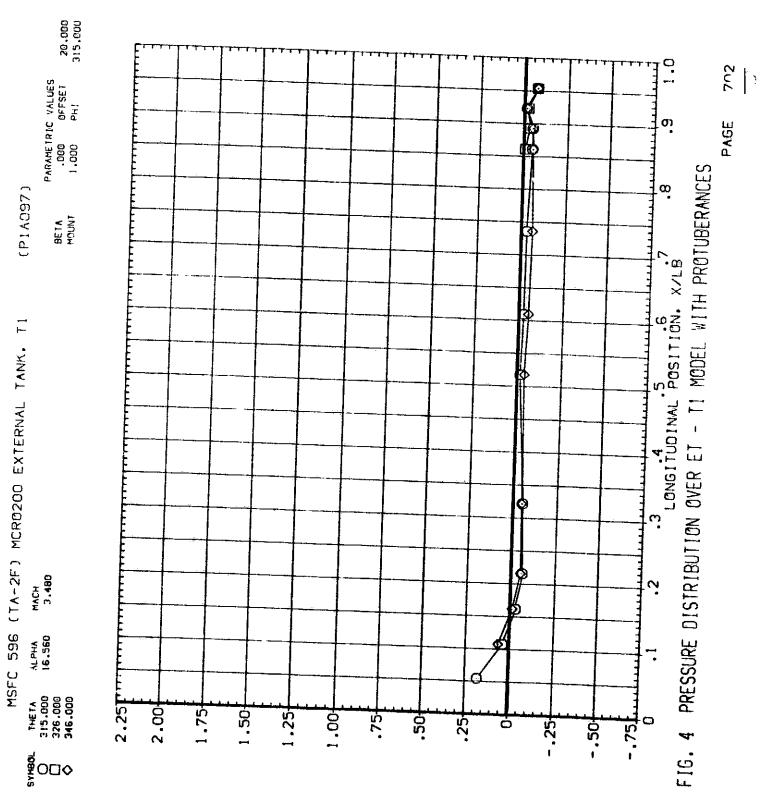


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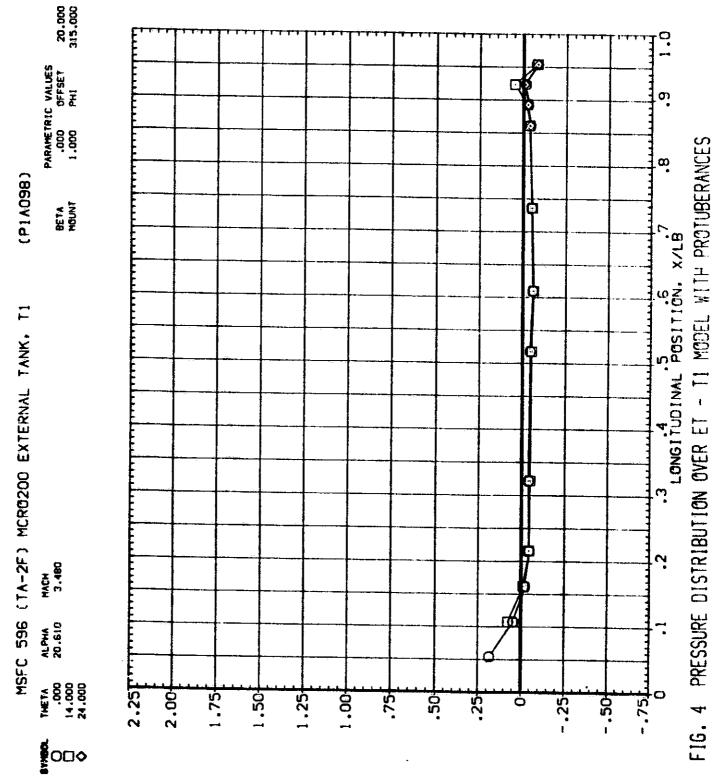


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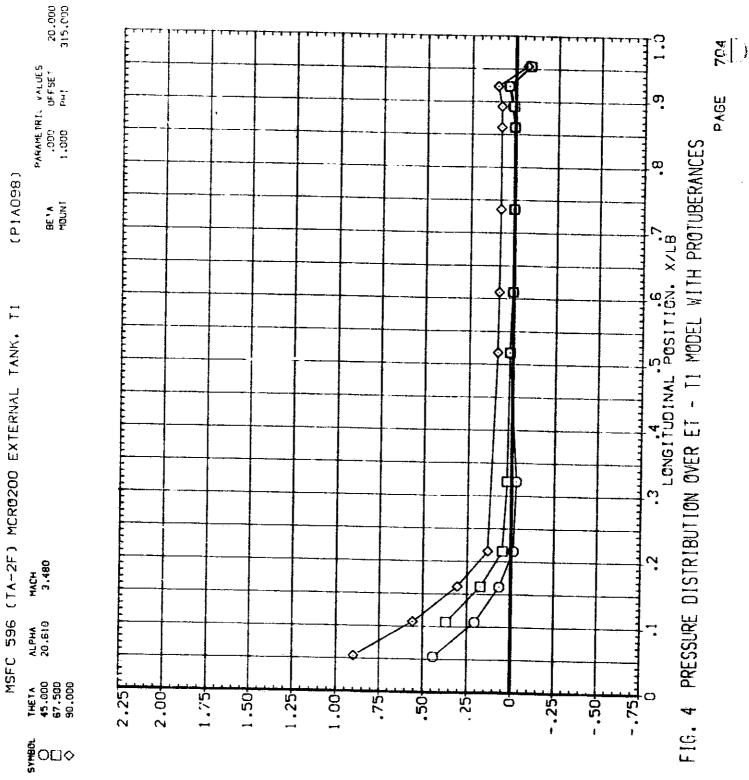




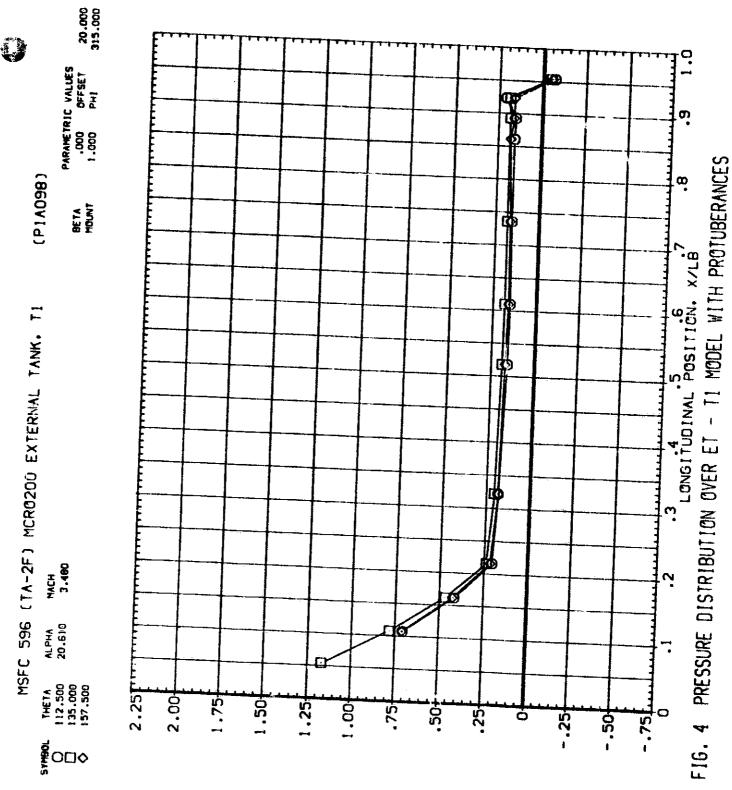
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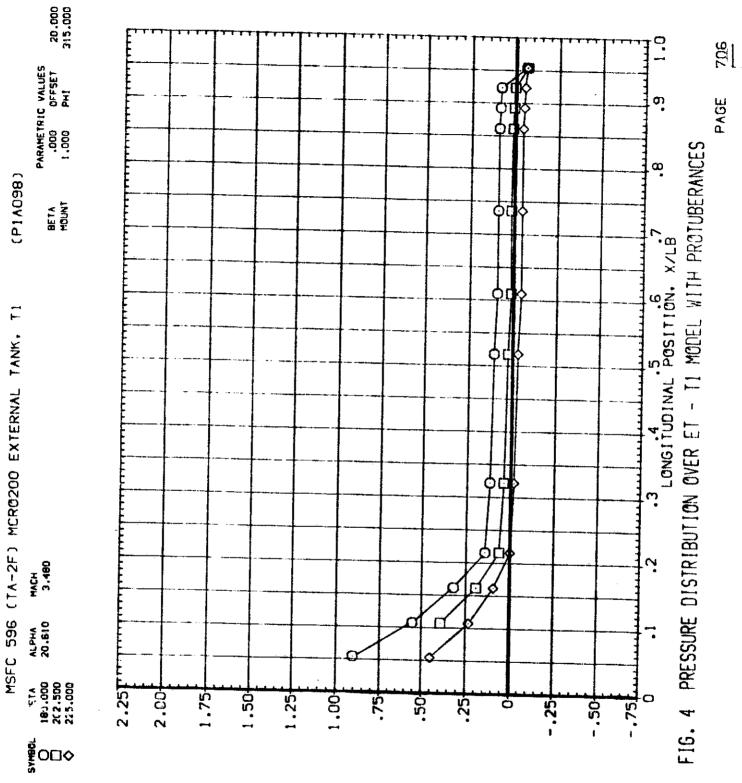
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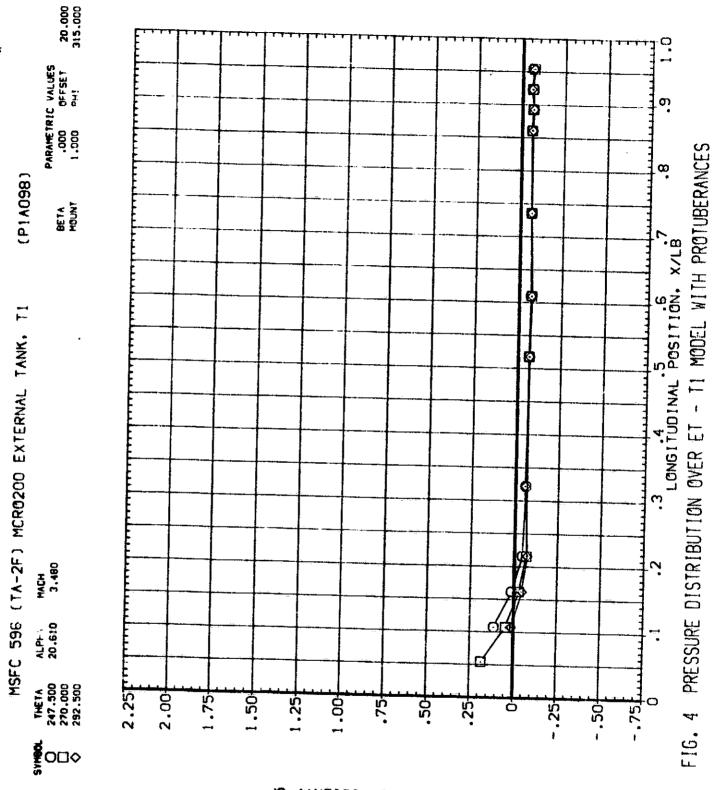
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MSFC 596 (TA-2F) MCR0200 EXTERNAL TANK, T1

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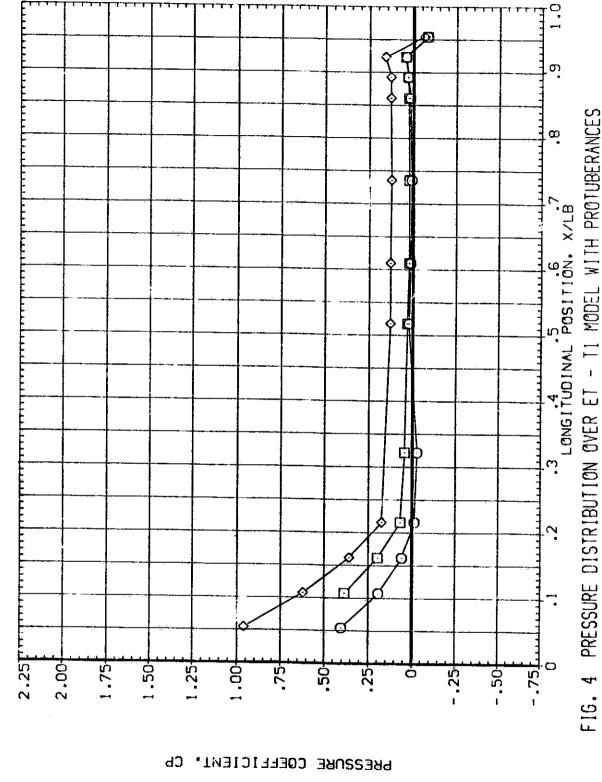
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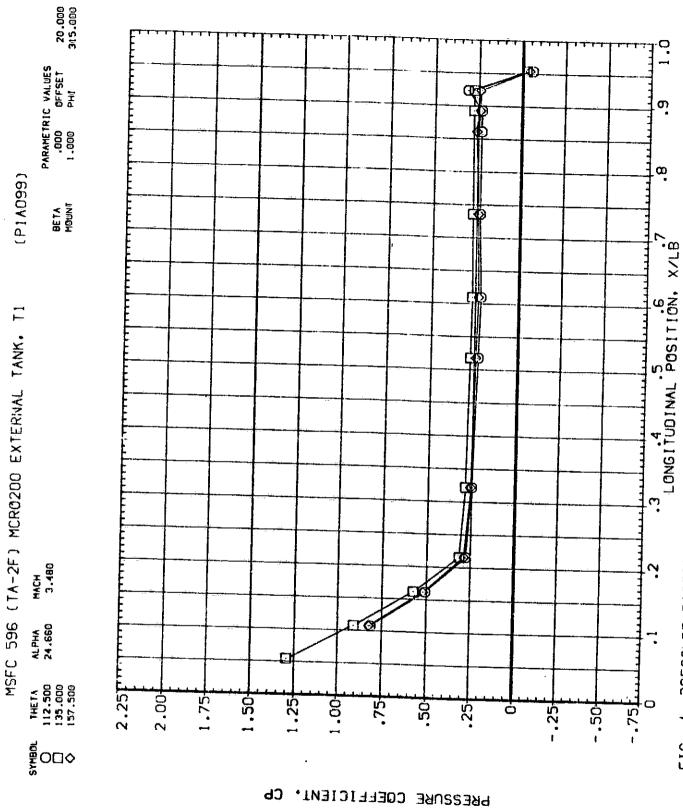
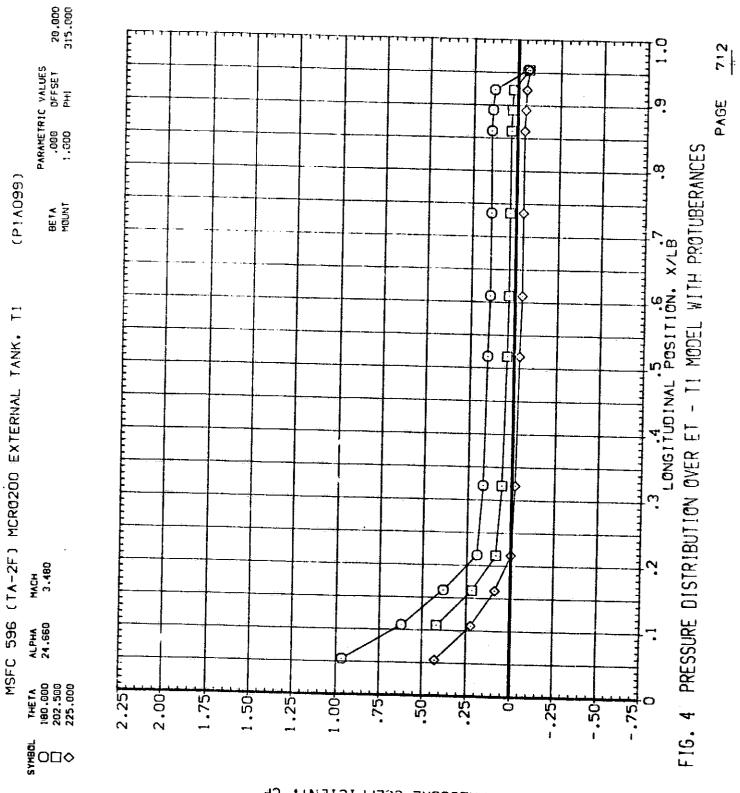


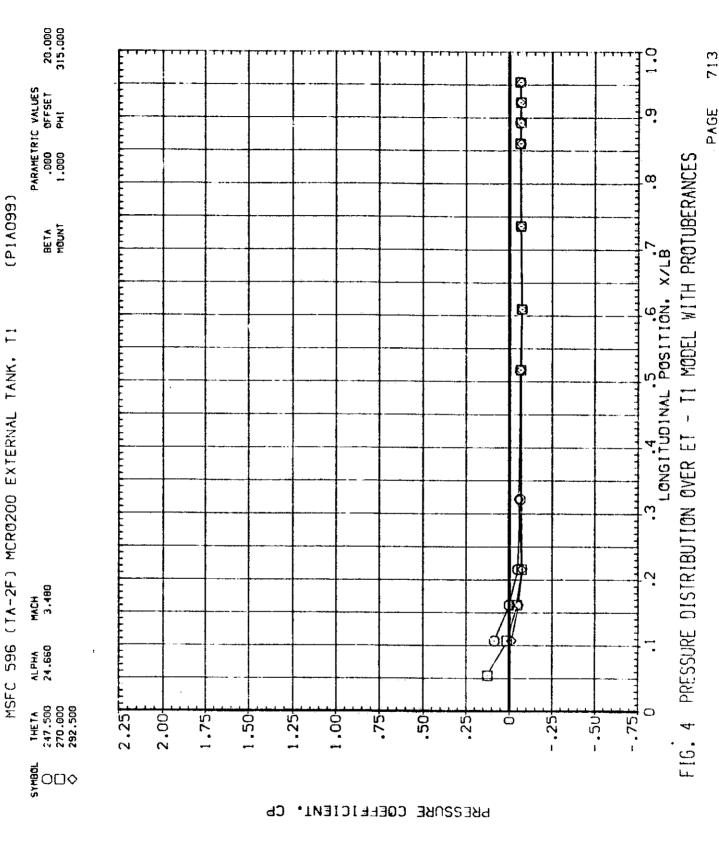
FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES



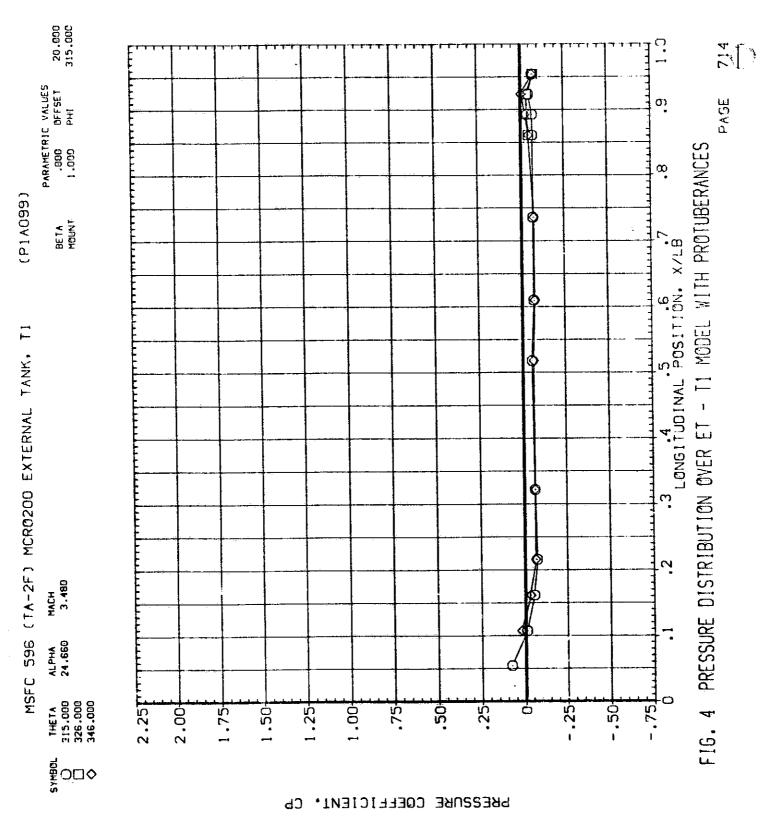
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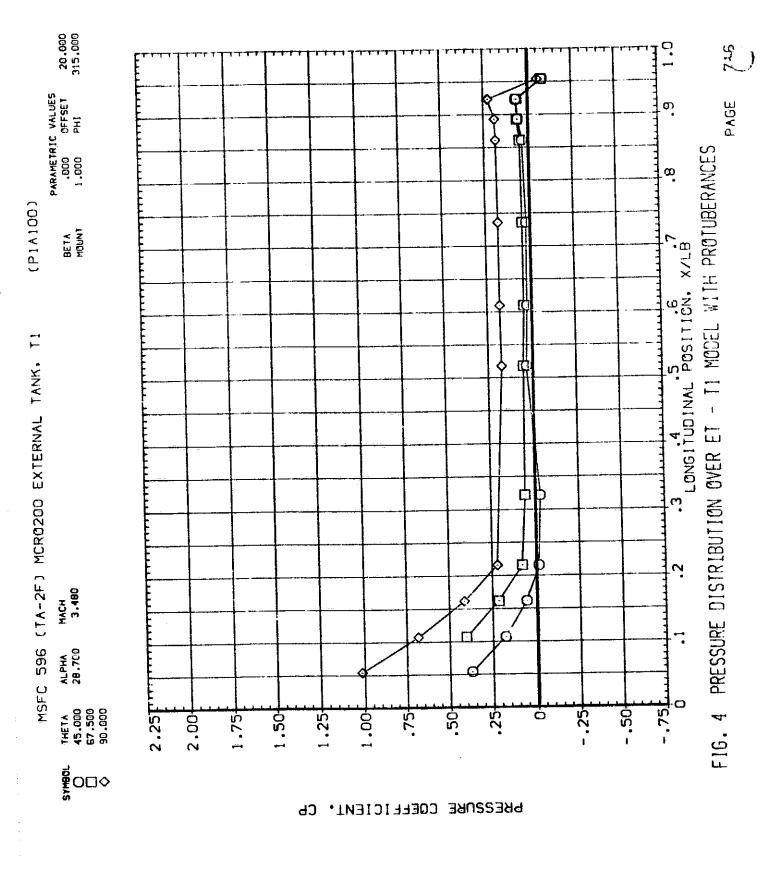


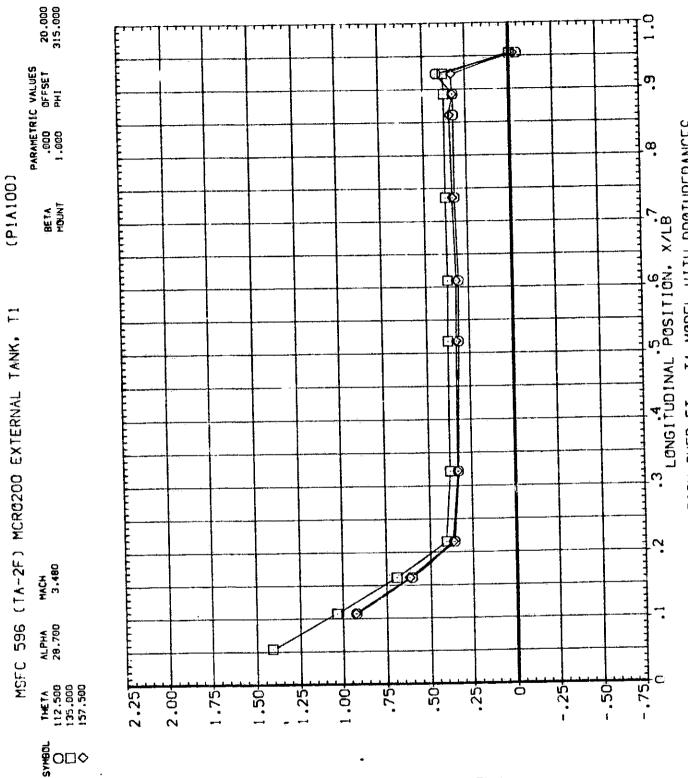
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FIG. 4 PRESSURE DISTRIBUTION OVER ET - TI MODEL WITH PROTUBERANCES

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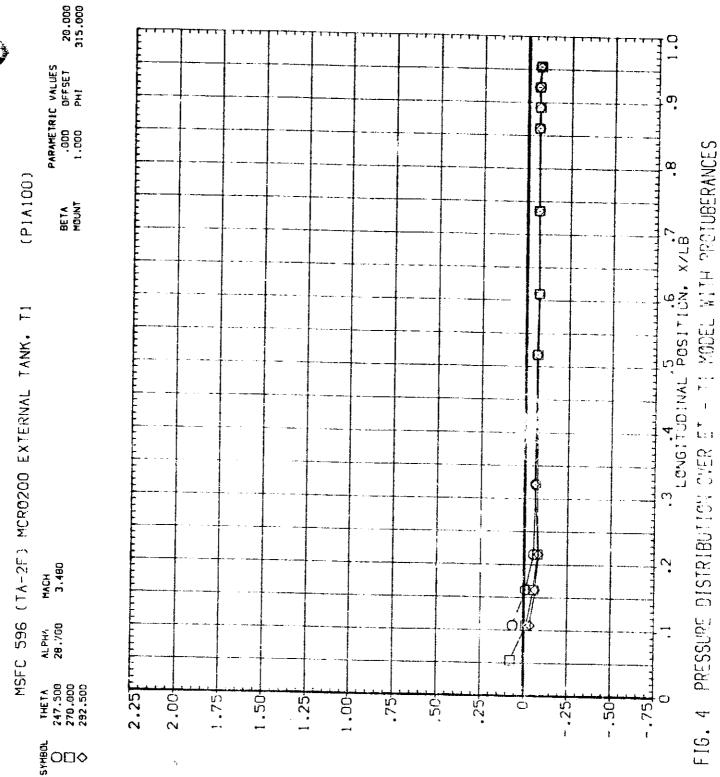
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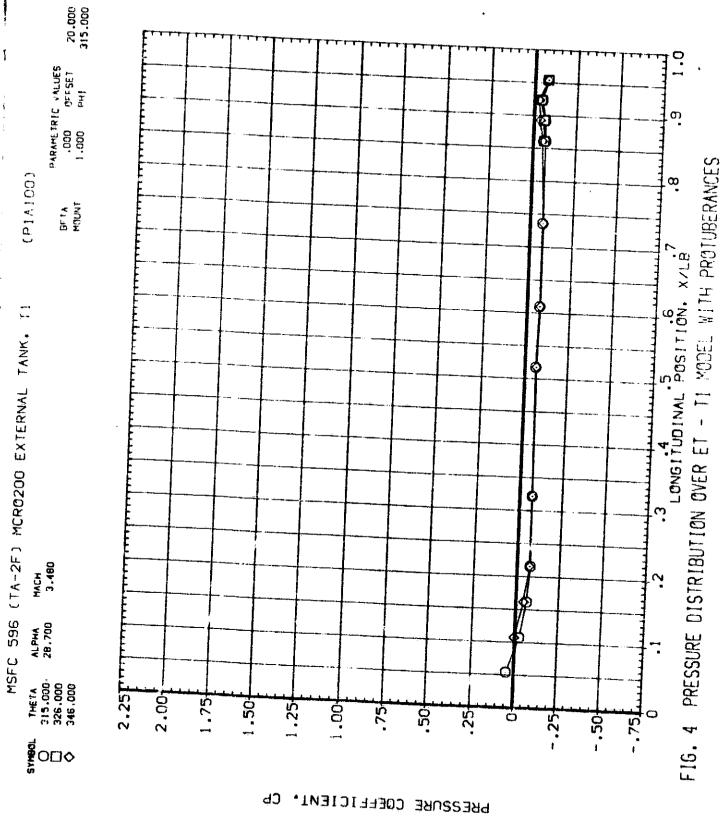
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